



ICAO ENGINE nvPM EMISSIONS DATA SHEET

SUBSONIC ENGINES

ENGINE IDENTIFICATION: GENx-1B64/P2 BYPASS RATIO (-): 9.0
 UNIQUE ID NUMBER: 01P17GE206 PRESSURE RATIO π_{co} (-): 41.4
 COMBUSTOR: TAPS
 ENGINE TYPE: TF RATED OUTPUT F_{oo} (kN): 298.0

REGULATORY DATA

CHARACTERISTIC VALUES:	LTO_{mass}/F_{oo} (mg/kN)	LTO_{num}/F_{oo} (particles/kN)	NVPM MASS CONCENTRATION ($\mu\text{g}/\text{m}^3$)
LTO/ F_{oo} AND MAX nvPM _{mass}	7.9	7.50E+13	161
AS % OF CAEP/10 LIMIT	-	-	4.0
AS % OF CAEP/11 LIMIT (InP)	2.3	1.8	
AS % OF CAEP/11 LIMIT (NT)	3.7	2.7	

MEASURED DATA

MODE	POWER SETTING (% F_{oo})	TIME minutes	FUEL FLOW kg/s	EMISSIONS INDICES*		NVPM MASS CONCENTRATION PEAK nvPM _{mass} ($\mu\text{g}/\text{m}^3$)
				EI_{mass} (mg/kg)	EI_{num} (particles/kg)	
TAKE-OFF	100	0.7	2.280	1.7	7.63E+10	
CLIMB OUT	85	2.2	1.869	1.7	7.93E+10	
APPROACH	30	4.0	0.600	3.0	6.59E+13	
IDLE	7	26.0	0.193	2.3	2.18E+13	
LTO TOTAL (kg, mg, number of particles)			787	1701	1.61E+16	-
NUMBER OF ENGINES				1	1	1
NUMBER OF TESTS				3	3	3
AVERAGE LTO/ F_{oo} VALUES (mg/kN, particles/kN)				5.7	5.40E+13	-
MAX EI VALUES (mg/kg, particles/kg) AND MAX MASS CONC. ($\mu\text{g}/\text{m}^3$)				6.9	1.91E+14	125

* Emissions Indices are corrected for thermophoretic loss and fuel hydrogen content

DATA FOR EMISSIONS INVENTORIES (ESTIMATIONS FOR ENGINE EXIT PLANE VALUES)

MODE	POWER SETTING (% F_{oo})	CORRECTED EMISSIONS INDICES	
		EI_{mass_SL} (mg/kg)	EI_{num_SL} (particles/kg)
TAKE-OFF	100	2.3	1.05E+11
CLIMB OUT	85	2.2	1.10E+11
APPROACH	30	4.1	3.28E+14
IDLE	7	2.9	7.55E+13

AMBIENT CONDITIONS

	From	To	FUEL	
BAROMETER (kPa)	97.4	98.2	HEAT OF COMBUSTION (MJ/kg)	43.15
TEMPERATURE (K)	289.9	297.9	HYDROGEN CONTENT (%mass)	13.79
HUMIDITY (kg water/kg dry air)	0.0079	0.0127	AROMATICS CONTENT (%vol)	17.7
			NAPHTHALENE CONTENT (%vol)	0.38
			SULPHUR CONTENT (ppm by mass)	12

MANUFACTURER: General Electric Company
 TEST ORGANIZATION: General Electric Company
 TEST LOCATION: PTO, Ohio
 TEST DATES: 05/09/2019-06/09/2019

REMARKS

1. GE Aviation Report R2018AE129/Rev. 0
2. Engine S/N 598-426
3. EI_{mass_SL} calculated from average EI_{mass} and KSL_{mass}
4. EI_{num_SL} calculated from average EI_{num} and KSL_{num}