

Comment Response Document (CRD) to Notice of Proposed Amendment (NPA) 2009-02b

for a draft Agency Opinion on a Commission Regulation establishing the Implementing Rules for air operations of Community operators

and

a draft Decision of the Executive Director of the European Aviation Safety Agency on Acceptable Means of Compliance and Guidance Material related to the Implementing Rules for air operations of Community operators

'Part-NCC and Part-NCO'

CRD c.1 – Comment Response Summary Table (CRST) Part-NCC

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Part-OPS	Commentators have been coded as: IS: industry sector BA: business aviation GA: general aviation H: helicopter IA: industry association	
	MS: Member State INDIV: individual	
Subpart A - General operating and fight rules - IR Section I –General Requirements		
OPS.GEN.015 Pilot-in-command responsibilities and authority	Pilot-in-command (PIC) responsibilities should be aligned with the Basic Regulation (BR) and Annex 6.	Accepted. In accordance with the new drafting principles, the Implementing Rules (IRs) make a reference to the Essential Requirements (ER) of the Basic Regulation, where such requirements are addressed in more detail in the IRs.
ALL AIRCRAFT		
(a) The pilot-in-command shall be responsible for:	1/ Concerns were expressed regarding the use of the term pilot-in-command rather than commander as used in EU-	1/ 'Pilot-in-command' is used for non-commercial operations, and also aligns with ICAO Annex 6.

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		 OPS. 2/ 2 IS (GA): request definition of who is PIC for training and check flights – instructor and/or trainee pilot? Clarification requested as to whether both can log the time as flight time/instruction time; 3/ INDIV: request to define and use consistently PIC throughout the NPA. 	 Following stakeholder feedback, the PIC is termed "commander" for commercial air transport (CAT) operations. 2/ This issue is within the scope of Part-FCL but beyond the scope of Part-NCC. 3/ Part-NCC and Part-NCO only use the term 'pilot-in-command'. A definition is provided in Annex I.
(1)	the initiation, continuation, termination or diversion of a flight, when involved in non-commercial operations with other than complex motor-powered aircraft;	MS, IS (BA, GA): request to address this rule to all operations.	Accepted for Part-NCC. Text aligned with Annex 6 Part II Section 3, which states that the PIC is responsible for operational control. ORO.GEN describes the corresponding responsibilities of the operator.
(2)	compliance with all operational procedures and checklists;	MS: Need to address PIC authority to issue instructions to crew and ensure procedures are complied with and reference operations manual.	Accepted. Text modified accordingly.
(3)	not commencing a flight unless he/she has confirmed that all operational limitations referred to in paragraph 2.a.3. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), can be complied with;	 1/ The PIC should be "satisfied that" but should not have to "confirm that". 2/ 2 MS: request to remove reference to ER and to repeat the text in this IR; 	This rule implements ER 2.a.3. 1/ Accepted. Changed to align with BR and ICAO Annex 6. 2/ Reference kept but the items

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		3/ IS (GA): request to review the list and reduce the requirements;	of the ER have been added to the IR. 3/ Not accepted. List has been aligned with Annex 6 Part II.
(4)	not commencing nor continuing a flight beyond the nearest suitable aerodrome or landing site when flight crew members' capacity to perform functions is significantly reduced from causes such as fatigue, sickness or lack of oxygen;	 1/ A definition of a "suitable aerodrome" is required. "Adequate" is understood but not "suitable". 2/ IS (BA): Need to allow scope for the PIC to select an aerodrome that is suitable to the situation, such as one that has appropriate medical facilities in the case of illness. 3/ INDIV: The term "landing site" is not used elsewhere, so change it to "operating site" which is used. 	 1/ A definition for 'weather- permissible aerodrome' is added to Annex I, in addition to that for 'adequate aerodrome' that was published in CRD OPS I and Opinion 04/2011. 2/ The "suitable", now "weather- permissible" aerodrome does not take this into account. However, the operator is free to add such a criteria for the PIC to check before commencing a flight. 3/ Accepted. Changed accordingly.
(5)	admission to the cockpit or, in the case of balloons, the pilot compartment;		

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	(6)	deciding, in accordance with the Configuration Deviation List (CDL) or Minimum Equipment List (MEL), as applicable, whether or not to accept an aircraft for flight with unserviceable equipment; and	MS: align with OR.OPS.010.GEN and replace "unserviceable equipment" with "inoperative or missing item(s)"	Accepted. Text has been transposed in the new rule of (a)(3) and (4).
	(7)	recording utilisation data, and all known or suspected defects in the aircraft at the termination of the flight, in the aircraft log book or journey log book for the aircraft.		Text amended to allow for recording of information after a series of flights. This is the usual practice in some helicopter operations where each flight may be only a few minutes. A GM to the notion of series of flight has been added.
(b)	dise	e pilot-in-command has the authority to refuse carriage of or embark any person or cargo that may represent a potential ard to the safety of the aircraft or its occupants.		
(c)	app cone	e pilot-in-command shall, as soon as possible, report to the propriate Air Traffic Services (ATS) unit any hazardous flight ditions encountered that are likely to affect the safety of other graft.		
(d)	crev the	withstanding the provision of OPS.GEN.015(a)(4), in a multi- w operation the pilot-in-command may continue a flight beyond nearest suitable aerodrome when adequate mitigating cedures are in place.	1/ Clarification requested concerning managing fatigue, continuation of the flight, and in conjunction a clarification of the definition of suitable and adequate aerodrome;	1/ AMC1 and GM1- NCC.GEN.106(d) clarify that the mitigating measures are related to fatigue. In addition, the term 'weather-permissible aerodrome' is used in place of 'suitable

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		2/ MS: request that the mitigating measures only apply to fatigue and only	aerodrome', and is defined in Annex I.
		reporting to the competent authority;3/ MS: request to delete (d) as it is to	2/ AMC1 and GM1- NCC.GEN.106(d) clarify that the mitigating measures are related to fatigue.
			3/ Not accepted. The AMC and GM provide means and further explanations on how to comply with the rule.
BALLOONS	S		Not transposed, beyond the scope of Part-NCC.
• •	pilot-in-command shall in addition to paragraphs (a), (b), (c) (d) be responsible for:		
(1)	the pre-flight briefing of those persons assisting in the inflation and deflation of the envelope;		
(2)	notwithstanding OPS.GEN.130, ensuring that no person is smoking on board or within the direct vicinity of the balloon; and		
(3)	ensuring that persons assisting in the inflation and deflation of the envelope wear appropriate protective clothing.		
OPS.GEN	.020 Crew responsibilities		

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(a)	Crew members shall be responsible for the proper execution of their duties which are related to the safety of the aircraft and its occupants, in accordance with the applicable requirements and, where applicable, the operations manual.		
(b)	Crew members shall be seated at their assigned stations and shall not perform any activities other than those required for the safe operation of the aircraft during critical phases of flight.	MS: Add "and when deemed necessary by the pilot-in-command in the interest of safety" to the end of this provision and re-structure to clarify.	Accepted. Text amended accordingly.
(c)	Flight crew members shall keep their safety belt fastened while at their stations.	 1/ MS: Clarification of the subject "shoulder harness" requested – for some helicopters, the seat harness is not a belt; 2/ Request to add "and when deemed necessary by the PIC in the interest of safety." 	 1/ A safety harness includes shoulder strap(s) and a seat belt which may be used independently (Annex 6 Part II Note to 2.2.4.4.4). 2/ Accepted. It has been added in (b).
(d)	At least one qualified flight crew member shall remain at the controls of the aircraft at all times.	 1/ Clarify that only a "<i>suitably</i> qualified <i>pilot</i>" shall remain at the controls of the aircraft at all times; 2/ IS (H): request to add " at all times when the rotor is turned under power for the purpose of flight"; 	 1/ The dictionary definition of "suitable" is "right or appropriate for a particular person, purpose, or situation". However as NCC.GEN.100(a)(4) clarifies what qualifications are required by crew members, "suitable" has been deleted. 2/ This is already covered by NCC.GEN.125.

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(e)	Crew members who undertake duties from more than one operator and are subject to flight and duty time limitations and rest requirements in accordance with OR.OPS.FTL shall:	INDIV: request clarification for calculation of flight time limitations where commercial pilots combine airline transport pilot licence (ATPL) and private pilot licence (PPL) flying;	This is addressed under crew responsibilities but not under flight and duty time limitations (FTL) requirements.
	 comply with all flight and duty time limitations and rest requirements applicable to their activities; 		
	(2) inform each operator of their activities;		
	 (3) maintain their individual records regarding flight and duty times and rest periods as referred to in OR.OPS.020.FTL; and 		
	(4) upon request, present their records to each operator before commencing a flight duty period.		
(f)	Crew members shall not undertake duties on an aircraft if they know that they are suffering from or are likely to suffer from fatigue or they feel unfit, to the extent that the flight may be endangered.	5	Not accepted. The intention is not to confer a right, but to specify an obligation.
(g)	A crew member shall report to the pilot-in-command:		This is a CAT requirement. It is not required by Annex 6 Part II and has therefore not been transposed in Part-NCC.
	(1) any fault, failure, malfunction or defect, which he/she believes may affect the airworthiness or safe operation of		

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the aircraft, including emergency systems; and		
(2) any incident that was endangering, or could endanger, the safety of the operation.		
OPS.GEN.025 Common language		Rule title amended to better reflect the content.
All crew members shall communicate in a common language.	 1/ MS: Clarify with whom the communication should be in the same language (i.e. all operations staff); 2/ IS (BA): request to amend text to specify that crew members shall use <i>one</i> common language; 3/ Specify that only safety-related crew and safety-related duties are concerned; 4/ MS: request to align with ICAO Annex 1 to specify the level of language knowledge required to radio/telephone (R/T) communications; 5/ MS: request that, for non-commercial operators, crew shall understand the language in which placards are written; 	Text amended to align with ICAO Annex 6 Part II. 1/ Accepted. Added "with each other". 2/ Not accepted. The proposed text is clear enough in this respect. 3/ Not accepted. The rule addresses crew members. Crew members perform safety related duties. 4/ Not accepted. This is within the scope of Part-FCL, FCL.055. 5/ Not accepted. The proposed text is clear enough in this
OPS.GEN.030 Transport of dangerous goods		respect.

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(a)	The transport of dangerous goods by air shall be conducted in accordance with the 2007-2008 Edition of the Technical Instructions for the Safe Transport of Dangerous Goods by Air published by decision of the Council of the International Civil Aviation Organization. (ICAO Doc 9284-AN/905.).	MS: It is important not to refer to a specific edition of the ICAO Technical Instructions since new editions are published every 2 years. Reference to an out-of-date edition would mean States would not be in compliance with Annex 18 and would cause difficulties for both operators and shippers of dangerous goods.	Accepted. Text edited to include: "as last amended" and the reference to the editions has been removed. This has been solved with the new definition in Annex I of 'Technical Instructions' as the latest effective edition and including the Supplement and any addenda.	
(b)	Dangerous goods shall only be transported by an operator approved in accordance with OPS.SPA.DG, except when:	MS: request to provide an exemption to allow transport of DG for avalanche blasting;	Exemptions have to be dealt with in accordance with the Technical Instructions or in accordance with Article 14 of the BR.	
	(1) they are not subject to the Technical Instructions in accordance with Part 1 of those Instructions.			
	(2) required on board the aircraft in accordance with airworthiness and operational requirements;	MS: Not required since the text is shown in Part 1 of the Technical Instructions and is therefore already covered by (b)(1).	Deleted as already covered by (b)(1)	
	(3) required on board the aircraft for specialised purposes;	MS: Not required since the text is shown in Part 1 of the Technical Instructions and is therefore already covered by (b)(1).	Deleted as already covered by (b)(1)	

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	(4)	carried by passengers or crew members in accordance with the Technical Instructions; or	INDIV: Text needs to specifically refer to only those dangerous goods that passengers and crew are permitted to carry.	Reference to Part 8 of the Technical Instructions added to clarify that only specific items may be carried. Reference to baggage added to avoid duplication.
	(5)	in baggage which has been separated from its owner.		Deleted as a consequence of amendments to (b)(4).
(c)		asonable measures shall be taken to prevent dangerous goods being carried on board inadvertently.		
(d)	repor	operator shall, in accordance with the Technical Instructions, t without delay to the competent authority and the authority e State where the accident or incident occurred:		Paragraph renumbered due to additional paragraph (d). Text moved to end of Section.
	(1)	any incidents or accidents involving dangerous goods; and	MS: re-align with Technical Instructions - amend to refer to dangerous goods incidents and accidents, which are not the same as incidents and accidents involving dangerous goods.	Amended to refer to dangerous goods incidents and accidents as specified by the Technical Instructions.
	(2)	the finding of undeclared or misdeclared dangerous goods discovered in cargo or passengers' baggage.	MS: Passengers are not required to declare dangerous goods in their baggage, so any dangerous goods found are not undeclared or misdeclared. The term 'mail' needs to be included as a	Passenger baggage moved to a separate paragraph and the term 'mail' added as a consequence of a change to the Technical Instructions.

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	result of an amendment to the Technical Instructions.	Note – In the T.I. 'mail' will be included in the reporting requirements as of 2011.
Section II - Operational procedures		
OPS.GEN.100 Ice and other contaminants	MS: request that the rule title be amended to refer to ground procedures. An additional rule is required for in-flight procedures in order to tie into AMC/GM;	Accepted, the rule has been split into two requirements, NCC.OP.185 Ice and other contaminants – ground procedures and NCC.OP.190 Ice and other contaminants – flight procedures
(a) At the commencement of a flight the external surfaces of the aircraft shall be clear of any deposit which might adversely affect its performance or controllability.	 MS: The rule as presented does not have an objective requirement. MS: suggest use of the terms "critical surfaces" (e.g. also in front of the fan blades) and "contaminates," IS (GA): request to amend text to " significantly affect its performance"; MS: request to amend text to state that the aircraft must be clear or contaminants at commencement of take-off; MS: safety issue: request to specify the contaminants (frost, snow, slush, 	 Accepted and re-drafted to include PIC, and operator responsibilities Partly accepted. "Critical surfaces" would need to be more precisely described. The term "contaminates" was accepted. Not accepted. "Significantly" is not precise enough. Accepted, text amended accordingly. Contaminants are specified in

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	 ice) and that the A/C must be free of these before flight; 6. MS: "external surface" could be misinterpreted to exclude contamination inside of engine inlets in front of the fan blades, "external surface" should be deleted. 	the AMC and GM. 6. Accepted. The term "external surfaces" is deleted to ensure that contamination inside engine inlets in front of the fan blades is also covered by this requirement.
(b) The operator shall apply ground de-icing/anti-icing processes whenever determined necessary, on the basis of inspections and weather conditions.	IS (BA): Operator is not always in a position to "apply" de-icing, where this service is provided by a third party.	Not accepted. The requirement to establish procedures has to be addressed to the operator.
OPS.GEN.105 Simulated abnormal situations in flight		
Except in the case of flight instruction provided by a training organisation approved in accordance with Part-OR, when carrying passengers or cargo or when conducting commercial operations the following shall not be simulated:	 MS: Request to exempt test flights and demonstration flights; MS: Request to re-consider "Except", since it would constrain commercial and GA operators. 	The rule has been re-drafted to fit NCC operations. The Cover Regulation proposes that flying training is conducted in accordance with either Part-NCC or Part-NCO, depending on whether the aircraft is complex motor-powered or not
(a) abnormal or emergency situations which require the application of abnormal or emergency procedures; or		
(b) Instrument Meteorological Conditions (IMC) by artificial means.	MS: "Artificial means" is redundant.	The text has been rephrased, the content remained unchanged.

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OPS	5.GEN.110 Carriage of persons		Amended to clarify the intent of the rule.
ALL	AIRCRAFT		
(a)	No person shall be in any part of an aircraft in flight which is not a part designed for the accommodation of persons, unless for the purpose of taking action necessary for the safety of the aircraft or of any animal or goods therein	4. IS (BA): request to amend text to include business jets with baggage areas that are accessible from the passenger cabin: " which is not a part where persons can safely reside or is not designated by the manufacturer as an area where persons are allowed to reside during flight, unless";	Text not transposed - beyond the scope of Part-NCC operations and there is no corresponding SARP.
AER	OPLANES AND HELICOPTERS		Editorial change to fit Part-NCC.
(b)	In the case of aeroplanes and helicopters, persons carried shall be seated where, in the event of an emergency evacuation, they may best assist and not hinder evacuation from the aircraft.		Editorial change to fit Part-NCC. "Person" was changed to "passenger" to clarify the intent of the rule.
(c)	Prior to and during taxiing, take-off and landing, and whenever deemed necessary in the interest of safety by the pilot-in- command, each person on board shall occupy a seat or berth and, except in the case of parachute operations, have his/her safety belt or harness properly secured.	1/ Cabin crew may have to perform duties while the aircraft is taxiing.2/ Some helicopters may use restraint devices other than seat belts.	 Not accepted. This rule refers to passengers carried on board, not to crew. Accepted. Text amended accordingly. "Person" was changed to "passenger" to clarify the intent of the rule.

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 HELICOPTERS (d) A helicopter operator shall specify which aircraft seats may be occupied by one adult and one infant properly secured by a restraint device. 		Accepted. The CRD text addresses multiple occupancy of aircraft seats to enable the seating of one adult with one infant. This part of the text was aligned with CAT.OP.MPA.225.	
OPS.GEN.115 Passenger briefing			
Passengers shall be briefed on the location and use of emergency exits and relevant safety and emergency equipment.	 1/ MS, INDIV: Suggestion that the text is returned to the responsibility of the PIC rule; 2/ MS: Text clarification requested regarding the fact that a passenger briefing should not be limited to the mentioned subjects and regarding as to when and how the briefing has to occur. 3/ MS: request to indicate how the briefing will be presented, including reference to an illustrated safety briefing card; 	As requested by commentators, AMC1 has been upgraded to IR. 1/ Accepted. Text clarifies that this is a PIC rule. 2/ Text modified to comply with Annex 6 Part II and III. The text does not exclude more items from being added to the briefing. 3/ The safety objective is clearly defined. The proposed addition would be an AMC, which however would go far beyond ICAO Annex 6 standards.	
OPS.GEN.120 Securing of passenger cabin and galleys			
(a) Prior to and during taxiing, take-off and landing, all exits and escape paths shall be unobstructed.			

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(b) Prior to and during take-off and landing, and whenever deemed necessary in the interest of safety by the pilot-in-command, all equipment and baggage shall be properly secured.	Suggestion that prior to taxi, only baggage shall be stowed and properly secured.	Not accepted. Loose equipment can be a significant hazard.
OPS.GEN.125 Portable electronic devices		
Portable electronic devices that can adversely affect the performance of the aircraft's systems and equipment shall not be used on board the aircraft.	 1/ IS (GA): Request to publish a listing with PED that adversely affect the systems. 2/ MS: Some equipment that has the potential to affect aircraft systems may be allowed at specific times - such as mobile phones when taxiing, and electronic healthcare devices. 	1/ Information is contained in the AMC.2/ Accepted. Proposed text takes this into account.
OPS.GEN.130 Smoking on board		
ALL AIRCRAFT (a) No person shall be allowed to smoke on board:	MS: request to list all items in (a) and (b) as relevant to all aircraft;	In Part-NCC this rule is applicable to all aircraft.
 while the aircraft is on the ground, unless specifically permitted by the operator in accordance with specified procedures; 	MS: Request to replace "on the ground" with "on the surface" to pay due consideration to seaplanes;	Accepted. Text amended accordingly.
(2) while the aircraft is being refuelled; or		

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(3) whenever the pilot-in-command deems necessary in the interest of safety.		
COMPLEX MOTOR-POWERED AIRCRAFT		Editorial change – text aligned with Part-CAT.
(b) No person shall be allowed to smoke on board a complex motor- powered aircraft:		
 in cargo compartments or other areas where cargo is carried; 		
(2) in those areas of the cabin where oxygen is being supplied;	IS (BA): Request to specify: "in those areas of the cabin where oxygen flow is continuous";	Not accepted. Some oxygen systems are demand systems.
(3) if the operator has declared a flight to be operated as a non- smoking flight; or		Not applicable to Part-NCC.
(4) outside those areas that the operator has designated smoking areas.	MS: there is no rule requiring operators to designate smoking and non-smoking areas;	This is implied in the rule
OPS.GEN.135.A Taxiing of aeroplanes		
Aeroplanes shall only be taxied on the movement area of an aerodrome when the person at the controls is properly qualified to taxi an aeroplane.	 1/ Request to clarify the expression "properly qualified" and "properly qualified to taxi an aeroplane" 	Modified to comply with Annex 6 Part II. AMC has been upgraded to IR. Text has been aligned with

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	 respectively; 2/ Suggestion to substitute "aeroplane" by "aircraft" to cover all types of aircraft; 3/ Request to add a condition under which taxiing in low visibility conditions will be permitted; 	 Part-CAT and addressed in NCC.GEN. 1/ Accepted. The qualifications have been expanded on in the IR. 2/ Not accepted. The rules intentionally only apply to aeroplanes. 3/ When low visibility operations (LVOs) are underway further restriction needs to be addressed in the requirements for
OPS.GEN.140.H Rotor engagement		aerodromes.
A helicopter rotor shall only be turned under power for the purpose of flight with a qualified pilot at the controls.	1/ IS: There may be circumstances when the pilot has to leave the controls while the rotor is turning (e.g. remote landings in a hostile environment). This should be addressed in the operations manual;	1/, 2/ Not accepted, as the requirement is in line with ICAO and is there to prevent accidents. Text aligned with Part-CAT and addressed in NCC.GEN.
	2/ MS: re-align with ICAO Annex 6, 2.2.3.2. to specify that only a qualified pilot can be at the controls while the rotors are turned under power for ground movement or flight – this is after all considered as flight time.	

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OPS.GEN.145 Use o	of aerodromes/operating sites				
	y use aerodromes or operating sites that are of aircraft and operation concerned.				
OPS.GEN.147 Visua	I Flight Rules (VFR) Operating minima			This requirement has transposed - it is cove Part-SERA.	
	les (VFR) flights shall be conducted in accordanc Flight Rules and table 1.	e			
Table 1 – Minimum	visibilities for VFR operations				
Airspace class		ABCDE*	F		G
	Above 900 m (3 000 ft) AMSL or above 300 m (1 000 ft) above terrain, whichever is the higher	At and below 900 m (3 000 ft) AMSL or 300 m (1 000 ft) above terrain, whichever is the higher			
Distance from cloud	1 500 m horizontally 300 m (1 000 ft) vertically	Clear of cloud and in sight of the surface			
Flight visibility	8 km at and above 3 050 m (10 000 ft) AMSL**	5 km***			

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	5 km below 3 050 m (10 000 ft) AMSL**		
*	VMC minima for Class A airspace are included for guidance but do not imply acceptance of VFR flights in Class A airspace.		
**	When the height of the transition altitude is lower than 3 050 m (10 000 ft) AMSL, FL100 should be used in lieu of 10 000 ft.		
***	see (c)(1) below.		
(b)	Special VFR flights shall not be commenced when the visibility is less than 3 km and not otherwise conducted when the visibility is less than 1.5 km.		
HEL	ICOPTERS		
(c)	Helicopters shall be operated in a flight visibility of not less than:		
	(1) 1 500 m during daylight, except when in sight of land, if the helicopter is manoeuvred at a speed that will give adequate opportunity to observe other traffic and any obstacles in time to avoid a collision, the visibility may be reduced to 800 m for short period.		
	(2) 5 000 m during night.		
(d)	In Class G airspace, when flying between helidecks where the overwater sector is less than 10 nm, VFR flights are conducted in		

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accordanc	e with table 2	2.					
Table 2 – Mini airspace	ma for flyin	g between	helidecks lo	ocated in Class	G		
	Day		Ni	ght			
	Height*	Visibility	Height*	Visibility			
Single pilot	300 ft	3 km	500 ft	5 km			
Two pilots	300 ft	2 km**	500 ft	5 km***			
	base shall ow and clear		to allow fli	ght at the spe	cified		
provided			-	ity down to 80 iate structure			
*** Helicopters may be operated in flight visibility down to 1 500 m provided the destination or an intermediate structure are continuously visible.							
OPS.GEN.150 Instrument Flight Rules (IFR) Operating minima							
• •				ng minima for ne to be used.		1/ INDIV (GA): Operators must be able to use commercially available material –	1/ Contained in AMC1-

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			non-commercial operators cannot be expected to establish minima themselves.	NCC.OP.110.	
	(1)	not be lower than those specified by the State in which the aerodrome is located, except when specifically approved by that State; and			
	(2)	require the prior approval of the competent authority in accordance with OPS.SPA.001.LVO.	Clarification requested that not every individual aerodrome minimum requires authority approval, but only the method of establishing such minima;	This is clarified in Part-SPA.	
(b)		ninima referred to in (a) shall take into account any increment sed by the competent authority.	INDIV (GA): This provision appears to be unnecessary.	Accepted. This has not been transposed for Part-NCC.	
(c)		ninima for a specific type of approach and landing procedure pplicable if:			
	(1)	the ground equipment required for the intended procedure is operative;			
	(2)	the aircraft systems required for the type of approach are operative;			
	(3)	the required aircraft performance criteria are met; and			
	(4)	the crew is qualified accordingly.			

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(d)		tablishing the aerodrome operating minima which will apply to particular operation, an operator shall take account of:		
	(1)	the type, performance and handling characteristics of the aircraft;		
	(2)	the composition of the flight crew, their competence and experience;		
	(3)	the dimensions and characteristics of the Final Approach and Take-off Areas (FATOs)/runways which may be selected for use;		
	(4)	the adequacy and performance of the available visual and non-visual ground aids;		
	(5)	the equipment available on the aircraft for the purpose of navigation and/or control of the flight path, as appropriate, during the take-off, the approach, the flare, the landing, the roll-out and the missed approach;		
	(6)	the obstacles in the approach, the missed approach and the climb-out areas required for the execution of contingency procedures and necessary clearance;		
	(7)	the obstacle clearance altitude/height for the instrument approach procedures;		
	(8)	the means to determine and report meteorological conditions; and		

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	(9) the flight technique to be used during the final approach.		
OPS	GEN.155 Selection of alternate aerodromes		Title amended since the rule was split into separate rules for take- off alternates and destination alternates.
TAK	-OFF ALTERNATE AERODROMES		
ALL	AEROPLANES		
(a)	A take-off alternate aerodrome shall be selected and specified in the flight plan if the weather conditions at the aerodrome of departure are at or below the applicable aerodrome operating minima or it would not be possible to return to the aerodrome of departure for other reasons.	A take-off alternate should not be required when the departure aerodrome is at limits – only when it is below limits or 2. IS (GA): please clarify that (a)-(c) refer to IFR ops only;	 1/ Not accepted. This rule is as per ICAO Annex 6 Part II 2/ Accepted. This is clarified in the new text of NCC.OP.155.
(b)	The take-off alternate aerodrome shall be located within the following distance from the aerodrome of departure:	MS: request to define distance for single-engined A/C;	Single-engined aeroplanes are addressed in (a). There are no further requirements on distances foreseen.
	(1) aeroplanes having two power-units. Not more than a distance equivalent to a flight time of one hour at the single- engine cruise speed; and	 MS: These times should be in still air conditions as per EU-OPS 1.295 (b) INDIV: request to define "power unit" to avoid confusion with "electrical power 	1/ Not Accepted. This rule is the same as CAT except extended range operations with two- engined aeroplanes (ETOPS) does not apply. Still air

A: R	ule	B: Summary of comments	C: Reason for change, remarks
		unit";	conditions are used for ETOPS but not for other situations.
			2/ Term "power unit" changed to "engine" in accordance with latest ICAO amendments.
	(2) aeroplanes having three or more power-units. Not more than a distance equivalent to a flight time of two hours at the one-engine inoperative cruise speed.		Editorial change in alignment with ICAO terminology.
(c)	For an aerodrome to be selected as a take-off alternate the available information shall indicate that, at the estimated time of use, the conditions will be at or above the aerodrome operating minima for that operation.	IS (BA): request to align text with OPS.CAT rule to require conditions above the applicable operating minima for a reasonable time <i>before and after</i> estimated time of use.	Not accepted. Text aligned with Annex 6 Part II.
HELI	COPTERS – COMMERCIAL AIR TRANSPORT		Text not transposed - beyond the scope of Part-NCC.
(d)	Helicopters used in commercial air transport shall comply with (a) and (c) above.		
DES	TINATION ALTERNATE AERODROMES		
(e)	For a flight to be conducted in accordance with instrument flight rules (IFR), at least one destination alternate shall be selected and specified in any flight plan, unless:		
	(1) for aeroplanes, the duration of the flight and the available current meteorological information indicates that, at the	MS, INDIV, IS (GA): Definition of "reasonable period" requested. Suggest	Accepted. The new text determines the period for the

A: Rule		B: Summary of comments	C: Reason for change, remarks
	estimated time of arrival at the place of intended landing, and for a reasonable period before and after such time, the approach and landing may be made under visual meteorological conditions; or	1 hour before and 1 hour after estimated time of arrival (ETA);	validity of the meteorological conditions to one hour before and one hour after the estimated time of arrival - as in Part-CAT.
(2)	for helicopters, available current meteorological information indicates that the following meteorological conditions will exist from two hours before to two hours after the estimated time of arrival:		Helicopter-specific rules are in a separate rule.
	 A cloud base of at least 130 metres (m) (400 ft) above the minimum associated with the instrument approach procedure; 		Helicopter-specific rules are in a separate rule.
	(ii) Visibility of at least 1 500 m more than the minimum associated with the procedure; or		Helicopter-specific rules are in a separate rule.
(3)	the place of intended landing is isolated and:	IS (GA): "Isolated" should be defined for aeroplanes and helicopters.	Not accepted. Isolated is defined in the IR, only for aeroplane operations.
	(i) there is no suitable destination alternate;		
	(ii) an instrument approach procedure is prescribed for the aerodrome of intended landing;		
	 (iii) for aeroplanes, available current meteorological information indicates that the following meteorological conditions will exist from two hours before to two hours after the estimated time of arrival: 		

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A: R	ule	B: Summary of comments	C: Reason for change, remarks
	 (A) A cloud base of at least 300 m (1 000 ft) above the minimum associated with the instrument approach procedure; 		
	(B) Visibility of at least 5.5 km or of 4 km more than the minimum associated with the procedure; and		
	(iv) for helicopters, a Point of No Return (PNR) is determined in case of an offshore destination.		
OPS	.GEN.160 Departure and approach procedures	 MS: It should be clear that this applies to IFR ops IS (GA): proportionality – re-align with ICAO Annex 6, Part II and exempt non-commercial operations from using only published, approved procedures; 	 1/ Not accepted. Departure and approach procedures can also be prescribed for VFR flights. 2/ The proposed rule is in compliance with Annex Part II and adds further requirements that do not infringe Annex 6 rules.
(a)	Unless otherwise approved by the State responsible for an aerodrome, an operator shall use the departure and approach procedures established by that State.	MS, IS (GA): Alignment with ICAO, especially referring to the term for the State, requested (to cater for offshore operations);	This has to be a PIC responsibility. Text amended accordingly. The term for the State complies with Annex 6 Part II.
(b)	The pilot-in-command shall only accept an Air Traffic Control (ATC) clearance to deviate from a published departure or arrival route, provided obstacle clearance criteria can be observed and full account is taken of the operating conditions. In any case, the final	IS (GA): ATC can use vectoring altitudes below published obstacle clearance altitudes. Request that ATC is responsible for obstacle clearance when	Accepted. Text amended accordingly.

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A: Rule	B: Summary of comments	C: Reason for change, remarks
approach shall be flown visually or in accordance with the published approach procedures.	PIC follows radar vectors provided by the ATC;	
OPS.GEN.165 Noise abatement		
Operating procedures shall take into account the need to minimise the effect of aircraft noise.	IS (H): request to have the rule apply only to aeroplanes, in line with AMC/GM material to the rule;	Comment accepted. Safety has to take precedence over noise abatement procedure. Editorial improvement.
OPS.GEN.170 Minimum terrain clearance altitudes – IFR flights	MS: re-align with ICAO to refer to minimum obstacle clearance altitudes;	Accepted. Rule title amended.
For each flight to be conducted in accordance with instrument flight rules (IFR), terrain clearance altitudes for the route to be flown shall be specified.		The text has been aligned with ICAO Annex 6. The objective of the rule is that the operator specifies the method to establish minimum flight altitudes; and based on this method, the PIC then establishes the minimum flight altitudes for each flight.
OPS.GEN.175 Minimum flight altitudes		Text not transposed - the content of the rule is addressed in Part- SERA.
An aircraft shall not be flown below minimum altitudes established by the State overflown, except when:		

A: Rule C: Reason for change, **B:** Summary of comments remarks necessary for take-off or landing; or (a) descending in accordance with procedures established by that State (b) subject to demonstration by the operator that the operation does not create a hazard to persons or property on the surface. **OPS.GEN.180** Routes and areas of operation Operations shall be conducted in accordance with any restriction on the Text not transposed - the content routes or the areas of operation imposed by the State overflown. of the rule is addressed in Part-SERA **OPS.GEN.185** Meteorological conditions The pilot-in-command shall not initiate or continue a Visual Flight 1. MS: Request to add "if the weather 1/ Not accepted. The proposed (a) Rules (VFR) flight unless the latest available meteorological wording is clear and realistic. conditions are of such kind that the information indicates that the weather conditions along the route latest available meteorological 2/ Annex 6 Part III 2.6.1 and at the intended destination at the appropriate time will be at or information will be similar to the contains: "except one of purely above the applicable VFR operating minima. previous issued information the pilot local character in VMC", which can fly based upon that information" to should only apply to NCO clarify that the pilot does not have to operations. check the MET every minute. 3-5/ Not accepted. This would be 2. MS: request exception for helicopter below the safety level established operations, as helicopters can be put on by ICAO. the ground at almost any time; 3. IS (GA): request to allow

commencement and continued flight to

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A: Rule	B: Summary of comments	C: Reason for change, remarks
	a destination aerodrome where meteorological conditions are below VFR operating minima, as for IFR;	
	4. IS (GA): proportionality – request exemption for GA where weather information may not be available (e.g. to aerodromes without a terminal area forecast (TAF)), or inaccurate for low- level;	
	5. IS (GA): request to delete "initiate" as, in mountainous regions, weather can change rapidly and pilots are trained to react accordingly;	
(b) A flight in accordance with instrument flight rules (IFR) shall only be initiated or continued towards the planned destination aerodrome when the latest available meteorological information indicates that, at the estimated time of arrival, the weather conditions at the destination, or at least one destination alternate aerodrome are at or above the applicable aerodrome operating minima.	Request to align with ICAO, which states that the weather conditions at the destination <u>and</u> at least one destination alternate have to be at or above the minima;	Not accepted. Annex 6 Part II has been amended and is now similar to the proposed rule.
OPS.GEN.190 Take-off conditions		
Before commencing take-off, the pilot-in-command shall ensure that:		Editorial improvements.

A: R	ule	B: Summary of comments	C: Reason for change, remarks
(a)	according to the information available, the weather at the aerodrome or operating site and, for motor-powered aircraft, the condition of the runway/Final Approach and Take-off Area (FATO) intended to be used, will not prevent a safe take-off and departure; and		
(b)	the visibility/Runway Visual Range (RVR) and the ceiling in the take-off direction are equal to or better than the applicable aerodrome operating minima.		Text has been shortened and only refers to the aerodrome operating minima.
OPS	GEN.195 Approach and landing conditions		
ensu aero conc inter appr	re commencing an approach to land, the pilot-in-command shall are that according to the information available, the weather at the drome or operating site and, for motor-powered aircraft, the lition of the runway/Final Approach and Take-off Area (FATO) anded to be used, will not prevent a safe approach, landing or missed to ach, having regard to any performance information contained in the raft Flight Manual (AFM) and/or the operations manual.		
OPS	.GEN.200 Commencement and continuation of approach		
(a)	An instrument approach shall only be continued below 1 000 ft above the aerodrome on the final approach segment when the reported Runway Visual Range (RVR) is at or above the applicable minima specified for the runway.	1. IS (BA): Request to delete "reported" RVR, since it is also possible that a pilot assesses the RVR.	1/ Not accepted. The pilot cannot assess the RVR when the aircraft is outside of the final approach fix (FAF).

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A: R	A: Rule		B: Summary of comments	C: Reason for change, remarks
(b)	appr the a	after passing 1 000 ft above the aerodrome on the final roach segment, the RVR falls below the applicable minimum, approach may be continued to Decision Altitude/Height (DA/H) inimum Descent Altitude/Height (MDA/H).		
(c)) The approach may be continued below DA/H or MDA/H and the landing may be completed provided that at least one of the following visual references for the intended runway is established at the DA/H or MDA/H and maintained:		shall only continue an approach and	 Accepted. Text aligned. The visual references mentioned in the former NPA rule – which are applicable to non-precision
	(1)	Elements of the approach light system;		approaches, APV and CAT I
	(2)	The threshold;		operations only - have been moved to the AMC. This new AMC
	(3)	The threshold markings;		contains the appropriate visual
	(4)	The threshold lights;		references for all approach and landing operations. This AMC has
	(5)	The threshold identification lights;		been aligned with the AMC in
	(6)	The visual glide slope indicator;		Part-CAT.
	(7)	The touchdown zone or touchdown zone markings;		
	(8)	The touchdown zone lights; or		
	(9)	Runway/Final Approach and Take-off Area (FATO) edge lights.		
OPS	OPS.GEN.205 Fuel and oil supply		 1/ MS: The header discusses both fuel and oil while the details are for fuel only request to change title to "Fuel supply"; 	AMC1 OPS.GEN.205 specifying the procedure to determine the minimum fuel required has been moved to IR level.
			2/ MS, INDIV: Request for a requirement for a fuel reserve stated for	1/ "Oil" has been added in the

A: Rule		B: Summary of comments	C: Reason for change, remarks
		a local / A-to-A flight, e.g. of 10 minutes.	rule text.
			2/ Not accepted. The proposed rule would be below the safety level established by ICAO.
(a)	In compliance with paragraph 2.a.7. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), the following amounts of reserve fuel for visual flight rules (VFR) flights and fuel for instrument flight rules (IFR) flights shall at least be carried.	The rule is overly complex.	Accepted. The rule has been simplified and aligned with Annex 6 Part II.
BALL	LOONS		Not transposed for Part-NCC.
(b)	For flights conducted in accordance with VFR, reserve fuel (gas or ballast) shall not be less than 30 minutes of flight.		
AER	OPLANES		Transposed in a new aircraft- specific rule - NCC.OP.130.
(c)	powered aircraft taking off and landing at the same a aerodrome/operating site and remaining within 50 nautical miles	1. INDIV: request that the rule apply to all A/C and operations;	Rule text simplified and aligned with Annex 6 Part II.
		2. IS (GA): request that amount of fuel reserve be at the PIC's discretion;	1/ Not accepted. Aircraft specific rules apply.
			2/ Not accepted. The proposed rule would be below the safety level established by ICAO.
	(1) 30 minutes fuel at normal cruising altitude by day; or	IS (GA): What is the requirement when	Rule text simplified and aligned

A: Rule			B: Summary of comments	C: Reason for change, remarks
			the flight includes both day and night?	with Annex 6 Part II.
				If part of the flight is night, the night requirements would apply.
	(2)	45 minutes fuel at normal cruising speed by night.	specified for day and cruising speed for winight?	Rule text simplified and aligned with Annex 6 Part II.
				Accepted. Should be cruising altitude.
(d)		lights conducted in accordance with IFR the amount of fuel to arried shall be sufficient:		
	(1)	to fly to the aerodrome/operating site of intended landing, and thereafter to fly 45 minutes at normal cruising altitude, when no alternate is required or no suitable alternate is	INDIV, IS (GA): request 30 min. reserve for Turbo Jet A/C in line with CAT rule, and at normal holding altitude (15 000	Not accepted. The proposed rule would be below the safety level established by ICAO.
		available (i.e. the aerodrome/operating site is isolated and no suitable alternate is available); or	ft);	Rule text simplified and aligned with Annex 6 Part II.
	(2)	when an alternate is required, to fly to and execute an approach and a missed approach at the aerodrome/operating site of intended landing, and thereafter:		Rule text simplified and aligned with Annex 6 Part II.
		(i) to fly to the specified alternate; and		Rule text simplified and aligned with Annex 6 Part II.
		(ii) to fly at least 45 minutes at normal cruising altitude.		Rule text simplified and aligned with Annex 6 Part II.

A: Rule			B: Summary of comments	C: Reason for change, remarks
HELICOPTERS		RS		Transposed in a new aircraft- specific rule - NCC.OP.131. The structure of the new rule text is aligned with the structure for the aeroplane rule.
(e)	power aerod (nm) accore	ot for non-commercial flights with other than complex motor- red aircraft taking off and landing at the same frome/operating site and remaining within 50 nautical miles of that aerodrome/operating site, flights conducted in dance with VFR shall carry reserve fuel not less than 20 tes fuel at best range speed.		
(f)) For flights conducted in accordance with IFR, the amount of fuel to be carried shall be sufficient:			
	(1)	to fly to the aerodrome/operating site of intended landing, and thereafter to fly 30 minutes at holding speed at 450 m (1 500 ft) above the destination aerodrome/operating site under standard temperature conditions and approach and land, when no alternate is required or no suitable alternate is available (i.e. the aerodrome/operating site is isolated and no suitable alternate is available); or		
	(2)	when an alternate is required, to fly to and execute an approach and a missed approach at the aerodrome/operating site of intended landing, and thereafter:		

A: R	lule		B: Summary of comments	C: Reason for change, remarks
	(i) to fly to the specified alternate; and			
	(ii)	to fly 30 minutes at holding speed at 450 m (1 500 ft) above the alternate aerodrome/operating site under standard temperature conditions and approach and land.		
		Refuelling with passengers embarking, on board or	Align text with ICAO concerning the fuelling with AVGAS, or wide-cut fuels;	Accepted. The text differentiates between AVGAS and wide-cut fuels on one side and other types of fuel on the other side.
				Text aligned with Part-CAT.
(a)		shall not be refuelled with passengers embarking, on lisembarking.		Not transposed for Part-NCC.
(b)		aircraft shall not be refuelled when passengers are g, on board or disembarking, unless:	Definition requested for precautions for fuelling during disembarking;	Addressed in the AMC.
	per	s attended by the pilot-in-command or other qualified sonnel ready to initiate and direct an evacuation of the craft; and	Remark that due consideration should be paid to the fact that the whole process shall take place under the authority of the PIC;	It may be the PIC or another qualified person. It should also be noted that Annex 6 Part II Section 3 does not require the PIC.
	ma sup	commercial operations, two-way communication is intained between the personnel involved in the operation pervising the refuelling and the pilot-in-command or other	 1/ Request to replace "maintained" by "shall be established and remain available". 	1/ The text has been aligned with Annex 6 Part II, which uses the term "maintained".
	qua	alified personnel required.	2/ MS: Precautions when refuelling with	2/ This topic is addressed in

A: Rule	B: Summary of comments	C: Reason for change, remarks
	wide-cut fuel should be added	GM1-NCC.OP.160.
OPS.GEN.215 In-flight fuel checks		The text has been amended to reflect the content of ICAO Annex 6 Part II. The new rule contains an operator requirement and better specifies the safety objective of the in-flight fuel management to ensure that after landing the remaining fuel is not less than the planned final reserve fuel.
In-flight fuel checks shall be carried out on each flight at regular intervals.	IS (GA): request exemption for A/C that rely on a dip-stick to test fuel levels (i.e. older/vintage A/C);	Text derived from the new ICAO Fuel policy that will be included in Annex 6 Part II. In accordance with Annex II to the BR, for historic aircraft the rules do not apply and national rules continue to be applicable.
OPS.GEN.220.B Operational limitations - balloons		Text not transposed - beyond the scope of Part-NCC.
(a) A landing with a balloon during night shall not be made, except for emergencies.		
(b) A balloon may take-off during night, provided sufficient fuel is carried for a landing during day.		

A: R	tule	B: Summary of comments	C: Reason for change, remarks
OPS	GGEN.222 Ground proximity detection		
	en undue proximity to the ground is detected, the pilot flying shall nediately take corrective action to establish safe flight conditions.		
9	Section III – Aircraft performance and operating limitations		
OPS	G.GEN.300 Operating limitations		
(a)	During any phase of operation, the loading, the mass and, except for balloons, the centre of gravity (CG) of the aircraft shall comply with any limitation specified in the Aircraft Flight Manual (AFM).	1. MS, INDIV: request to add "or the Operators Manual, if more restrictive", and upgrade AMC material to IR;	Revised to reflect the requirements of Annex 6 Part II 2.3.1 and Part III Section III 3.1.
			Although this would be only relevant for a limited number of operations, the addition of the operations manual has been accepted in the interest of safety.
(b)	An aeroplane shall be operated within the limitations imposed by compliance with the applicable noise certification standards.		
OPS	GEN.305 Weighing		
(a)	The mass and, except for balloons, the CG of an aircraft shall be established by actual weighing prior to initial entry into service.		Balloons reference is not relevant for NCC therefore is deleted, and -flexibility is introduced in the requirement of weighing for NCC.

A: R	A: Rule		B: Summary of comments	C: Reason for change, remarks
				A GM is also added to clarify the intent of the rule
(b)) The accumulated effects of modifications and repairs on the mass and balance shall be accounted for and properly documented. The aircraft shall be reweighed whenever the effect of modifications on the mass and balance is not accurately known.			This subparagraph has been attributed to the operator
(c)	comr	mass and CG of complex motor-powered aircraft used in non- mercial operations and aircraft used in commercial operations be re-established by actual weighing:	 MS: Request to extend (c) to 5 and 10 years; IS (BA): request to weigh every 4.5 years, to facilitate operators, where fleet masses have a 9 year cycle of weighing; 	1-2/ Requirement is proposed to be deleted since the conditions for a reweighing are already sufficiently described in the new subparagraph (-a).
	(1)	at least every 4 years if individual aircraft masses are used; or		
	(2)	at least once every 9 years if aeroplane fleet masses are used.		
(d)	aircra	weighing shall be accomplished by the manufacturer of the aft or by a maintenance organisation approved in accordance Part-M or Part-145.	1. MS, INDIV: Suggestion to move (d) into Part M, since it is dealing with maintenance tasks;	For the time being the requirement is kept. All weighing related provisions will be transferred to regulation (EC) 2042/2003 with the rulemaking task MDM.047.
		.310 Mass and balance system - complex motor-powered sed in non-commercial operations and aircraft used in		

A: Rule	B: Summary of comments	C: Reason for change, remarks
commercial operations		
COMPLEX MOTOR-POWERED AIRCRAFT USED IN NON-COMMERCIAL OPERATIONS AND AIRCRAFT USED IN COMMERCIAL OPERATIONS		
(a) An operator of a complex motor-powered aircraft used in non- commercial operations or an aircraft used in commercial operations shall establish a mass and balance system specifying how the following items are accurately determined for each flight:	 IS (GA); request alleviation for very light jets in private ops, compared with CAT regulation; INDIV (GA): request exemption for ops with a single pilot and no other on board; INDIV (GA): request that flights where costs are shared between the pilot and others on board not be considered as commercial ops, and be exempt from this rule; Suggestion to put the list in an AMC; 	Editorial modifications. 1/ Rules for VLJ are subject of a separate rulemaking task. This comment will be kept to the file. The is no justification why this rule should not apply to a operation with a VLJ. The intent of the rule is that the method for determining the items below is specified. 2/ Exemptions to the safety objective are not accepted. However, means to comply with the rule are in AMC and provide sufficient flexibility to address such operations. 3/ This rule is also applicable to non-commercial operations. 4/ The key elements of the mass and balance system need to be in the IR.

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	(1)	aircraft dry operating mass and CG, if applicable;		
	(2)	mass of the traffic load;		
	(3)	mass of the fuel load;		
	(4)	aircraft loading under the supervision of qualified personnel;	IS (GA): Request to clarify the intention of (a)(4) "under the supervision of qualified personnel" and why it is only added to this item	The intention is that the loading of the aircraft is supervised either by the PIC or by other qualified personnel (e.g. load masters or similar ground personnel).
	(5)	load distribution;		
	(6)	take-off mass, landing mass and zero fuel mass, if applicable;		
	(7)	CG positions, if applicable; and		
	(8)	preparation and disposition of all documentation.		
(b)	shall	nass and balance computation based on electronic calculations be replicable by the flight crew. USED IN COMMERCIAL OPERATIONS	MS: Request to clarify "replicable" in (b);	Text modified to clarify the intent of the requirement.
AIRU	CKAFI	USED IN COMMERCIAE OPERATIONS		

A: Rule		B: Summary of comments	C: Reason for change, remarks
(c)	For commercial operations, mass and balance documentation shall be prepared prior to each flight specifying the load and its distribution.		
OPS	.GEN.315 Performance – general		
(a)	An aircraft shall only be operated if the performance is adequate to comply with the applicable rules of the air and any other restrictions applicable to the flight, the airspace or the aerodromes/operating sites used, taking into account the charting accuracy of any charts/maps used.	MS: request to add - the take-off mass must not exceed the maximum take-off mass specified in the Aeroplane Flight Manual for the pressure altitude and the ambient temperature at the aerodrome at which the take-off is to be made.	Accepted.
(b)	Except when necessary for take-off or landing at an approved operating site, an aircraft shall only be operated over the congested areas of cities, towns or settlements or over an open-air assembly of persons, if it is able to make a landing without undue hazard to the aircraft occupants or to third parties, in the event of a power- unit failure.	 IS (H): request to re-draft (b) as the intent should be to mitigate risks to third parties. Safety of A/C occupants is covered in other rules; MS: Request to delete or rephrase (b), add "and property", and to define "open air assembly of persons"; MS: Need to define "approved operating site"; IS (BA): request to clarify if (b) applies in emergency situations; 	 1/ Accepted. The text transposed should be the one contained in ICAO Annex 2 3.1.2 and 4.6 and link it to performance such that performance is available to comply with these requirements. 2/ Partially accepted. "Open air assembly" is also not further defined in ICAO. 3/ Partially accepted. The text has been amended to try to identify 2 situations (1. Take-off and landing and 2. Overflying) taking into account Annex 6 text. 4/ Please refer to the ERs, which

A: Rule **B:** Summary of comments C: Reason for change, remarks already contain the clause that in emergency situations the PIC can deviate from the rules (7.d.). Text added to better clarify the safety objective as regards mass limitation and to align with Annex 6 Part II. **OPS.GEN.320.A Take-off - complex motor-powered aeroplanes** IS (GA): request to exempt non-This is not required by Annex 6 used in non-commercial operations and aeroplanes used in commercial operations from this rule; Part II. However, it is kept in the commercial operations interest of safety. COMPLEX MOTOR-POWERED AEROPLANES USED IN NON-COMMERCIAL OPERATIONS AND AEROPLANES USED IN COMMERCIAL OPERATIONS When determining the maximum permitted take-off mass, the (a) following shall be taken into account: 1. MS: This requirement is confusing 1/ Accepted. The term (1)the take-off distance shall not exceed the take-off distance available, with a clearway distance not exceeding half of the and not appropriate for some complex "calculated" was added. take-off run available; motor-powered aircraft. Suggest 2/ This term is not defined in EU-"calculated take-off distance/run shall OPS nor ICAO Annex 6, and a not exceed..."; definition has not been added to 2. IS (GA): request to define "clearway" Annex I Definitions. distance": 3/ Not accepted. The appropriate 3. MS: request to also add width of width of a runway is dealt with runway. Certification Specifications (CS) by requirements to the

should require this information in the

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aerodrome operator. The CS is

an initial airworthiness-related

A: R	ule		B: Summary of comments	C: Reason for change, remarks
			AFM;	document and beyond the scope of Part-NCC.
	(2)	the take-off run shall not exceed the take-off run available;		
	(3)	a single value of V_1 shall be used for the rejected and continued take-off; and	 MS: request to clarify that this is not applicable to single-engined A/C. IS (GA, BA): request that this rule applies to single-engined aeroplanes; IS (BA): Make (a)(3) only applicable to commercial operations so that there is no change compared to JAR/EU- OPS1.490 and 1.565. This will allow non-commercial operators to have a V_{STOP} different from V_{GO} as permitted by §6.2.2 to AMC25.1591. 	 1/ 2/ It is clarified that the single value is applicable to such aircraft where a V₁ is appropriate. 3/ For safety considerations, this requirement should be kept also for NCC operations.
	(4)	on a wet or contaminated runway, the take-off mass shall not exceed that permitted for a take-off on a dry runway under the same conditions.	MS: request to redraft the text, which makes no sense in this form;	Not accepted – it is transposed from EU-OPS.
СОМ	PLEX N	MOTOR-POWERED AEROPLANES		
(b)	moto and	e event of a critical engine failure during take-off, complex r-powered aeroplanes shall be able to discontinue the take-off stop within the runway available or, in the case of multi- ned aeroplanes, continue the take-off and clear all obstacles	MS, IS (GA): proportionality - request to exempt CS-23 "normal category" aeroplanes in GA from this rule as specific performance data under one-	Not accepted. This is a safety- critical requirement.

A: Rule	B: Summary of comments	C: Reason for change, remarks
along the flight path by an adequate margin until the aeroplane is in a position to comply with OPS.GEN.325.	engine-inoperative (OEI) is not available;	
OPS.GEN.325 En-route - Critical engine inoperative - complex motor-powered aircraft		
In the event of a critical engine becoming inoperative at any point along the route, a multi-engine complex motor-powered aircraft shall be able to continue the flight to an aerodrome without flying below the minimum obstacle clearance altitude at any point.	1. MS: Request to standardise use of "minimum flight altitude/minimal obstacle clearance altitude" throughout the NPA";	"Adequate" was added to align with Annex 6 Part II which requires that a safe landing can be made.
	 IS (GA): Request to apply only to commercial operations; INDIV: Add landing site. 	 1/ Minimum obstacle clearance altitude is the term used by ICAO Annex 6. Minimum flight altitudes would be higher and would not be stricter than Annex 6 Part II. 2/ Not accepted. Text is aligned with ICAO Annex 6 Part II. 3/ Accepted.
OPS.GEN.330.A Landing - complex motor-powered aeroplanes	2. IS (GA): request to withdraw this rule as the PIC should decide if the chosen landing aerodrome is suitable for a safe operation;	Not accepted. Rule is aligned with Annex 6 Part II.
At any aerodrome, after clearing all obstacles in the approach path by a safe margin, the aeroplane shall be able to land and stop, a seaplane come to a satisfactorily low speed, within the landing distance available. Allowance may be made for expected variations in the approach and landing techniques, if such allowance has not been made in the	 IS (BA): request clarification that the landing factor = 1; IS (GA): request GM regarding variation in approach and landing regarding: temperature, excess 	Text aligned with Annex 6 Part II. 1/ Landing factors have to be applied in accordance with the AFM and/or operations manual.

A: Rule	B: Summary of comments	C: Reason for change, remarks
scheduling of performance data.	 height/speed at threshold, runway slope, unpaved/wet/contaminated runway, non-standard flare techniques, gusts and crosswinds; 3. MS: Replace "by a safe margin" by "from an appropriate screen height"; 4. Add "at an <i>adequate</i> aerodrome"; 5. MS: Request for a clear requirement for all aeroplanes to comply with the weight, altitude and temperature (WAT) limitations for landing. 	 2/ This would be considered in future rulemaking tasks. Please consider providing a more detailed rulemaking proposal, using the form available on the Agency's website: www.easa.europa.eu 3/ Not accepted. This would alter the intent of the IR. 4/ Any aerodrome means the aerodrome of intended landing and any alternate aerodrome. 5/ This is now part of the new requirement NCC.POL.120.
Section IV - Instruments, data and equipment	 1. This subpart is applicable to all aircraft, whatever the date of issuance of the first individual certificate of airworthiness might be. This is a major difference compared to ICAO Annex 6 provisions and to EU/JAR-OPS 1/3 as well as to national rules (e.g. regarding harnesses, TAWS,). The question of cost/benefit of the retrofit must be studied carefully, especially in the case of general aviation or aerial work where situations can be different according to the Member States rules, as the only common basis 	1. Appropriate transition periods have now been established in order to enable smooth implementation of the operations rules affecting installation of additional equipment. This is particularly the case where Member States have filed differences to ICAO standards.

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A: Rule	B: Summary of comments	C: Reason for change, remarks
	(ICAO Annex 6 Part II) may either be a simple recommendation or a standard that takes into account the date of issuance of the first individual certificate of airworthiness.	
	2. In JAR-OPS 3, there used to be one section dealing with the equipment needed and another section dealing with operational procedures. In IR OPS everything is mixed and makes things difficult to understand (ex: oxygen, CVR and FDR - see OPS.GEN.510)	2. Accepted. Purely operational rules have been moved to NCC.GEN or NCC.OP as applicable.
	3. The European Private Helicopter Alliance (EPHA) is strongly opposed to the proposed regulations commented upon herein. It is simply grossly unreasonable to impose such a heavy burden of compliance when no safety case exists. We thus urge EASA to either withdraw these proposals entirely, amend them as suggested, define a maximum take-off mass (MTOM) weight limit below which they would not apply (e.g. 3 175 kg or 2 000 kg), or simply apply the fixed wing proposals to helicopters. Other practical mitigation measures could be exemptions for helicopters under	 3. Partially accepted The rule should be based on and compliant with Annex 6 Part III Section III. The rules need to encompass: offshore operations; provide leeway for other-than offshore operations. The rules on floatation equipment and emergency equipment for over water flight have been adapted to take into account helicopter size in Part-NCC and Part-NCO. Specific requirements

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A: Rule	B: Summary of comments	C: Reason for change, remarks
	2 000 kg MTOM, for non-complex helicopters, or for helicopters in private flight.	for SPO will also be developed.
	Our preferred solution is that EASA adopt option 4C as defined in paragraph 2.9 of Notice of Proposed Amendment (NPA) No 2009-02G.	
	Should EASA mandate equipment that is not currently required under a Member State's present regulations, it is essential that a practical time period of exemption is allowed for equipage to occur. We would suggest that a major item such a floats should have a 25 year compliance period (this being a reasonable life for the current helicopter fleet). ELTs could have a 10 year compliance period. It would be completely unreasonable and disproportionate to demand immediate compliance, especially when there is no immediate perceived safety need.	
	Helicopter owners would be asked to comply with these costly proposals merely because of outdated ICAO standards, whereas light fixed wing aircraft operating in the same manner will have almost no changes nor the	

A: Rule	B: Summary of comments	C: Reason for change, remarks
	resulting costs imposed on them.	
	EASA's policy should be "regulation at a sensible minimum", and "safety in a cost-effective manner", as stated by Peter Hunt, Head of Operating Standards Division of the UK CAA in December 1998.	
	EASA should consider:	
	The proportionality of the proposals as regards private helicopter use; the lack of legal necessity of ICAO compliance; the unreasonableness of ICAO helicopter standards as applied to private operations; the safety benefit of the proposals, if any; the practicality of the equipment which is proposed to become mandatory; the cost of the equipment which is proposed to become mandatory; the need as perceived by a substantial majority of helicopter pilots; the arbitrary and discriminatory nature of parts of the NPA as applied to private helicopters (but not to private aeroplanes).	
	Ref: European Parliament resolution of 3 February 2009 on an Agenda for Sustainable Future in General and Business Aviation (2008/2134(INI).	

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A: Rule	B: Summary of comments	C: Reason for change, remarks
OPS.GEN.400 Instruments and equipment – General		
GENERAL OBJECTIVES		
(a) An aircraft shall be equipped with instruments which will enable the flight crew to:		This ICAO performance-based rule is very generic and may be difficult to implement
		Furthermore, this rule may be seen as a duplication of paragraph 5 Instruments, data and equipment of Annex IV to Regulation (EC) No 216/2008 that this Section implements.
		The various instruments, data and equipment that are required by this Section IV shall suffice to ensure that the Essential Requirements are complied with.
		In order to avoid duplication of the Essential Requirement, (a) has been deleted.
		Consequently the title of the rule has been revised.
(1) control or, in the case of balloons, determine the flight path;		

A: R	ule	B: Summary of comments	C: Reason for change, remarks
	(2) carry out any required procedural manoeuvre; and		
	(3) observe the operating limitations in the expected operating conditions.		
APP	ROVED AND NON-APPROVED EQUIPMENT		General remark:
			A distinction is made for applicable airworthiness requirements depending on whether the aircraft is registered in a EU country or in a third country. A GM is added for clarification
(b)	quipment and instrument required by Part-OPS shall be approved except as specified in (b), and installed in accordance with Part-21	1. Approval shall be also required for equipment that is used to perform a function required by Part-OPS.	1. The statement of compliance is used to encompass equipment required through the function(s)
		2. The approval of equipment shall not be required if "deemed not practical".	it performs. 2. The exceptions to the requirement of (a) are listed in (c). See below for the changes proposed to (c).
(c)	Instruments and equipment required by Part-OPS which do not need to be approved in accordance with Part-21, as well as any additional equipment which is not required by Part-OPS, but is carried on a flight, shall comply with the following:	required by this Part shall be separated	1&2. To improve clarity, (c) has been revised to address equipment not required by this regulation and that does not need to be approved in accordance with Part-21. This will

A: Rule	B: Summary of comments	C: Reason for change, remarks
	board an aircraft should be approved according to Part-21. Reason: Airlines must demonstrate in accordance with Part-21 that all instruments and equipment shall not affect the airworthiness of the aircraft, even in the case of failures or malfunction.	ensure that equipment or instruments that have not been subject to an approval in accordance with Part-21 are used to comply with airworthiness and operations safety requirements.
(1) The information provided by these instruments, equipment or accessories shall not be used by the flight crew to comply with(a);	1. This requirements will prevent the use of GPS devices not approved under Part-21 from being use for navigational purpose (e.g. glider competitions). It is proposed to amend the rule to state that the safe navigation of the aircraft must not be dependent on the use of unapproved instruments.	1. The proposed rule ensures that compliance with Essential Requirements does not rely on non-approved equipment. If equipment is installed in addition to that required to comply with the Essential Requirements, its utilisation is not precluded.
(2) The instruments and equipment shall not affect the airworthiness of the aircraft, even in the case of failures or malfunction.		
ACCESSIBILITY AND POSITIONING OF INSTRUMENTS AND EQUIPMENT		
(d) Instruments and equipment shall be readily operable or accessible from the station where the flight crew member that needs to use it is seated.	1. Paragraph (d) states: "Instruments and equipment shall be readily operable or accessible from the station where the flight crew member that needs to use it is seated."	 The paragraph has been removed. The paragraph has been removed.

A: F	Rule	B: Summary of comments	C: Reason for change, remarks
		We recommend that this paragraph be reworded or eliminated. The location of equipment needing to be readily operable or accessible might require redesign and recertification of existing and approved locations of the emergency equipment. JUSTIFICATION: Paragraph (d) appears to be specifying design requirements for airplanes, which is inappropriate for an operational rule. It should be removed from this NPA.	
		 2. (d) "seated" is not applicable to balloons. Rewrite (d) as follows: "Instruments and equipment shall be readily operable by or accessible to from the station where the flight crew member that needs to use it is seated." 	
(e)	Instruments and equipment used by flight crew members shall be arranged so as to enable them to see the indications for use readily from their station, with the minimum practicable deviation from the position and line of vision which they normally assume when looking forward along the flight path.	 The original intent of the rules was clearer in previous EU-OPS/JAR-OPS 3 texts. It is proposed to revert back to the EU- OPS/JAR-OPS 3 text. 	1. This paragraph is kept in line with ICAO Annex 6 and reworded in line with EU-OPS text.
(f)	All required emergency equipment shall be easily accessible for immediate use.		

A: F	Rule	B: Summary of comments	C: Reason for change, remarks
OPS	6.GEN.405 Equipment for all aircraft		
(a)	Aeroplanes and helicopters shall be equipped with:		
	(1) except in the case of aerobatic flights, at least one hand fire extinguisher:	1. It is not clear why the previously single rule for seats, safety belts and harnesses has been divided between a number of rules (GEN.405, GEN.480 GEN.545, CAT.406 and CAT.482); it might be more appropriate to put all of the elements concerned with seats into a single rule in GEN.480.	For clarification of presentation of the requirements: - OPS.GEN.405 (a) (2), (a) (3) and (e) (2) on seat, berth, seat belts and harnesses, and child restraint devices have been moved to a dedicated rule. - A dedicated rule for hand fire extinguisher has been created incorporating OPS.GEN.405 (a)(1) and (b).
	(i) in the cockpit; and	 In aircraft with very limited cockpit space, the requirement to install a hand fire extinguisher: may not be feasible, 	 Not accepted. The requirement is maintained, in compliance with ICAO standard. Exceptions to the requirement
		 may create an additional safety hazard: losing sight and aircraft control when intentionally or not activated, installation and periodic maintenance of a fire extinguisher on board that 	will be investigated as part of future rulemaking task OPS.066 Operations and equipment for high performance aircraft, to take into account other-than aerobatics operations where high

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	category of aeroplane will cost time and money for the operators (mainly aero- clubs),	G-loads can be met.
	- the use of any fire extinguisher in a cabin of an aircraft is dangerous for all occupants because of the agents used.	
	It is much safer to have a pilot concentrating on the emergency procedures learned and on landing the aircraft as quickly as possible, rather than to try to extinguish the fire.	
	It is proposed to delete the requirement for non-complex aeroplanes involved in non-commercial operations, at least for aeroplanes below 2 000 kg maximum take-off weight (MTOW).	
	2. It is proposed to replace "except in the case of aerobatic flights" by "except where it adversely affects the safety of crew and/or passengers".	
	Add AMC OPS.GEN.405(a)(1) "Adverse effects on the crew's and/or passenger's safety are expected on aerobatic flights, where the hand fire extinguisher may become a hazard due to high G-loads" reason: Aerobatic flights are only one	
	(although the most common) example for possible adverse effects on the	

A: Rule	B: Summary of comments	C: Reason for change, remarks
	crew's safety, as stated in the respective Guidance Material.	
	Launching gliders into a lee wave system can require the tug aircraft to repeatedly climb and descend in the extremely turbulent air associated with strong air flows in the lee of mountains. Once off tow, the glider can climb in the smooth higher air, but the tug immediately returns for a further dose.	
	Following EASA's concept of "hard" and "soft law", the Aerobatic Flights would be subject to the AMC, where other exemptions might be added, in case adverse effects on safety are identified. Advantage: the proposed wording gives the same level of safety but offers more flexibility in case other hazards than the one mentioned in the GM are identified in the future.	
(ii) in each passenger compartment which is separate from the cockpit;		

A: Rule		B: Summary of comments	C: Reason for change, remarks
(2)	a seat or berth for each person older than 24 months;		1. Clarification of the wording, consistent with other paragraphs.
(3)	a seat belt for each seat and restraining belts for each berth;		
(4)	a restraint device for each person younger than 24 months; and	 A baby in an incubator cannot be provided with a restraint device. The proposed text says restraint device for each person younger than 24 months but does not give any indication on where the infant should sit or if double occupancy is permitted. Replace "a restraint device" by "a child restraint device (CRD)". This will make the link with the AMC more understandable. 	 The AMC provides several means of compliance. In this specific case, an alternate means of compliance may be developed. This is proposed to be clarified in future rulemaking task MDM.063 Child restraint devices. Reference to child restraint device clarified in line with AMC.
(5)	spare electrical fuses, of the ratings required for complete circuit protection, for replacement of those fuses that are accessible in flight.	 The requirement shall apply only for fuses that are required for the correct operation of instruments and equipment required by Part-OPS. We feel that accessibility is not the criteria. As NPA OPS 43 was not approved and was just a draft, the criteria should be "for use". Therefore please change sub-paragraph 5 to read: (5) "Spare electrical fuses, of the ratings required for complete circuit protection, 	1&2. In order to address the output of NPA OPS 43 the text has been amended to improve clarity and in order to cover the intent of the rule.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	for replacement of those fuses that are allowed to be changed in flight".	
(b) The type and quantity of extinguishing agent for the required fire extinguishers shall be suitable for the type of fire likely to occur in the compartment where the extinguisher is intended to be used and the hazard of toxic gas concentration in compartments occupied by persons shall be minimised.		
SAILPLANES		Not transposed, beyond the scope of Part-NCC.
(c) Sailplanes shall be equipped in accordance with $(a)(2)$ and $(a)(3)$.		
BALLOONS		Not transposed, beyond the scope of Part-NCC.
(d) Balloons shall be equipped in accordance with (a)(1)(i) and an alternative source of ignition.		
LARGE BALLOONS AND BALLOONS INVOLVED IN COMMERCIAL OPERATIONS		

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A: R	Rule	B: Summary of comments	C: Reason for change, remarks
(e)	 Balloons with a maximum passenger capacity of more than 11 and balloons used in commercial operations shall, in addition to (d), be equipped with: (1) protective gloves for each crew member; (2) crew restraint harness(es); (3) a hook knife; (4) a fire blanket or fire resistant cover; and (5) a drop line of at least 30 metres (m). 		
CAR	RIAGE OF PARACHUTISTS		Not transposed, beyond the scope of Part-NCC.
(f)	Notwithstanding the provisions of $(a)(2)$, in the case of carriage of parachutists, the floor may be used as a seat, provided means are available for the parachutists to hold on.		
OPS	5.GEN.410 Flight instruments and equipment - VFR flights		In order to comply with ICAO the requirement on Mach number indication is added
(a)	When operating under Visual Flight Rules (VFR), sailplanes, aeroplanes, and helicopters shall be equipped with a means of measuring and displaying:		

A: R	ule	B: Summary of comments	C: Reason for change, remarks
	(1) magnetic heading;		
	(2) time, in hours, minutes and seconds;	 For a VFR flight, there is no need for a precision down to the nearest second. Hours and minutes are enough. Delete "shall be equipped with". Justification: A requirement for a time piece equipment could be subject to an installation, which could be very costly. 	 Not accepted. The requirement for seconds is in line with ICAO. Not accepted. An AMC - allows compliance by using a wrist watch for other-than-complex motor-powered aircraft.
	(3) pressure altitude; and		
	(4) indicated air speed.		
(b)	When sailplanes, aeroplanes and helicopters operating under VFR cannot be maintained in a desired attitude without reference to one or more flight instruments, it shall, in addition to those required in (a), be equipped with a means of measuring and displaying:	 This paragraph details the required flight instruments and equipment for various flight regimes. The matter is complex due to the varying requirements of the different types of operation, the flight conditions and the time of day. There is a need to specify the types of flight conditions for which additional instruments shall be installed. See proposal. As written, "cannot be maintained in a desired attitude" refers to a concept of 	 This paragraph is based on ICAO Annex 6 Part II, 2.4.7. ICAO interpretation The additional instruments necessary to fulfil the Essential Requirements may vary depending on the operational conditions. Leaving the uncertainty on which of the instruments listed in (b)(1) to (5) shall be installed, in case operations that will require them

A: Rule	B: Summary of comments	C: Reason for change, remarks
	"IMC rating", which is not applicable, even considered as illegal, in other European countries.	are performed, is not acceptable at this level of Implementing Rule.
	 There is no certainty as to the operational conditions it corresponds to. 3. The LBA requests to stay away from the performance base rule making here. If this system shall be used, OPS.GEN 405 should be phrased in a similar way. However, this does not make much sense at all. 4. The set of rules for instruments VFR, night VFR and IFR are made extremely complex by the policy of inheritance. The combination of rules has to provide for simple types and complex types in General Aviation; it must also cater for simple types and complex types in Commercial Air Transport; and simple and complex types in commercial and non-commercial Aerial Work. The criteria for instruments and equipment vary for simple types within GA, CAT and AW and also for complex types in GA, CAT and AW. There are also differences between: single pilot and two pilots; operations when control can be maintained by reference outside the cockpit and those where it cannot - 	 2. The new proposal is based on the deletion of current (b) and the introduction of more specific conditions of applicability. The intent of the requirement is to cover operations under VFR, that is to say, in VMC for which nevertheless the need to refer to instruments may be foreseen. Alternatively the same additional instruments shall be required for special VFR flights performed below the VMC minima. In order to avoid confusion it is proposed to further specify the operational conditions for which additional flight instruments may be required. 3-5. Refer to 1&2.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	even though all of these are designated as VFR (and even further complicated by the inheritance from VFR to IFR).	
	5. A VFR-flight is by definition a flight when the attitude is maintained by visual references out of the cockpit and in VMC. AOPA-Sweden does not agree to a need of additional instruments during operations in VFR/VMC, where the desired attitude cannot be maintained by one or more instruments. Only a need can be seen for vertical speed indicator during night and direction during VFR on top.	
 (1) vertical speed; (2) turn and slip for aeroplanes, and slip for helicopters; 	 Move (b)(1) and (b)(2) to OPS.GEN.410 (a) to read : "(5) vertical speed only for sailplanes";"(6) turn and slip for aeroplanes and slip for helicopters". Justification: these items are necessary for VFR with sailplanes, aeroplanes and helicopters. 	The slip indication is not mentioned as part of the required instruments for operations under VFR in ICAO Annex 6 Part II, 2.4.3.1, but is nevertheless acknowledge to be strongly recommended for the safe control of the aeroplane in turn manoeuvres in order to maintain symmetrical flight. This is consistent with existing
		This is consistent with existing training requirements.

A: Rul	e	B: Summary of comments	C: Reason for change, remarks
(attitude. In the case of helicopters, two separate means of indicating attitude shall be available; 		
(4) stabilised heading; and		1. Wording changed from "heading" to "direction" for consistency with EU-OPS & CS- 23/25
(5) when power is not adequately supplied to the instruments.	 None of the instruments listed in OPS.GEN.410 (a) has a need of external power, which suggests that this point can be deleted. How can we measure when power is not adequately supplied to the instruments? On some a/c, no power is needed for vertical speed required by §(b)(1) - only air pressure is enough - so §(b)(5) cannot be complied with. Proposal is to keep this requirement but not under §(b)(5), under a new (c) for example, and also to exempt vertical speed from this requirement. 	1. It is proposed to clarify the requirement by changing the text to: "a means of indicating when the supply of power to the gyroscopic instruments is not adequate".
AEROF	PLANES AND HELICOPTERS IN MULTI-PILOT OPERATIONS		
s (Whenever two pilots are required for the operation, aeroplanes hall be equipped with an additional separate means of displaying a)(3), (a)(4), (a)(5) and if applicable (b) (1), (b)(2), (b) (3) and b)(4).		

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A: Rule	B: Summary of comments	C: Reason for change, remarks
BALLOONS		Not transposed, beyond the scope of Part-NCC.
 (d) When operating under VFR, balloons shall be equipped in accordance with (a)(2), (a)(3) and also with: (1) a means of indicating: (i) drift direction; and (ii) envelope temperature; and 		
LARGE BALLOONS AND BALLOONS INVOLVED IN COMMERCIAL OPERATIONS		Not transposed, beyond the scope of Part-NCC.
 (e) Balloons with a maximum passenger capacity of more than 11 and balloons used in commercial operations, when operating under VFR shall, in addition to (d), be equipped with: (1) a means of measuring and displaying: (i) vertical speed; and (ii) pressure for each supply line; 		
OPS.GEN.415 Flight instruments and equipment - VFR night flights and IFR flights	1. Amend the title as follows to take into account the fact that not only VFR night flights and IFR flights are dealt with in	 The proposed rules are developed with the basic assumption that the flight rules

A: Rule	B: Summary of comments	C: Reason for change, remarks
	this paragraph and that the criteria are not only the flight rules but the meteorological conditions: "Flight	for VFR will impose VMC and that flight in IMC will have to be performed under IFR.
	instruments and equipment – Flight at night or in IMC and VFR night flights and IFR flights"	Reducing the requirements in terms of instruments when flying in IMC compared to IFR is
	for IMC flights add the following instruments:	contradictory to ICAO Annex 6 Part II.
	attitude 2) stabilized heading 3) vertical speed 4) a second attitude instrument or a turn and slip indicator with an electrical supply independent from the	2. Comment period of this NPA was extended to allow stakeholders to review its content.
	first one.	3. Accepted.
	2. There are numerous other instruments required by our national safety requirements for general aviation that are missing in this NPA. Drawing the comparison has been really time- consuming. We are not in a position, though, to make deeper comment in such a short time. This is in favour of converting this NPA into an A-NPA to enable stakeholders to give more detailed comments.	A dedicated paragraph is created for operating lights.
	3. Simplify by breaking out the lighting requirements from the instrument requirements into a new section (OPS.GEN.XXX) as there are several	

A: R	Rule	B: Summary of comments	C: Reason for change, remarks
		errors of omission included in the NPA.	
AER	OPLANES AND HELICOPTERS.		
(a)	Aeroplanes and helicopters operating Visual Flight Rules (VFR) night flights and Instrument Flight Rules (IFR) flights shall, in addition to complying with OPS.GEN.410(a), (b), and (c), be equipped with:	 To require the same requirements for night VFR and IFR is not reasonable as some requirements are not relevant for night VFR flight. ICAO Annex 6 Part III now has separate requirements for night VFR for helicopters. The ICAO designation for helicopters should be used; it removes 'flight in controlled airspace' and has three separate requirements ('Day VFR', 'Night VFR' and 'IFR - day and night'). 	 The night VFR requirements are covered under paragraph (b). The additional instruments are required to be fitted only if operations at night are envisaged. The night VFR requirements are covered under paragraph (b). Rules for different aircraft classes have now been drafted.
		2. Fitting IFR equipment in a VFR-only private helicopter that is necessary only in IFR conditions will only serve to encourage pilots to fly in those conditions, thereby breaking the law and being a negative as far as safety is concerned.	
		It is illegal for VFR helicopter pilots to fly IFR unless they have an appropriate rating. In VFR flight such conditions are rarely encountered and are overcoming	

A: Rule		B: Summary of comments	C: Reason for change, remarks
		by applying VFR flying rules	
		3. The compliance with (c) shall be "if applicable".	
(1)	a means of measuring and displaying outside air temperature;		
(2)	a means of preventing malfunction due to either condensation or icing for the means of measuring and displaying indicated air speed;		
(3)	an alternative source of static pressure;	1. The requirement for an alternate source of static pressure is not requested by ICAO Annex 6 Part III Section III General Aviation - what is the justification for this requirement? We draw the attention that currently many single helicopters are not equipped with an alternate source of static pressure because it is not mandated by ICAO and because JAR- OPS 3 (CAT) forbids to operate in VFR night, as it is a limitation of the Performance Class 3.	The requirement for an alternative source of static pressure stems from JAR-OPS 0, where it is specified that this alternate source is for altimeter, airspeed and vertical speed indicators.
(4)	an anti-collision light system;	1. Anti-collision for helicopters already occurs in CAT.410; it appears to be a double requirement.	1. This issue is corrected with the new presentation of the rules.

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A: Rule		B: Summary of comments	C: Reason for change, remarks
(5)	navigation/position lights;		
(6)	a landing light;	1. This differs from JAR-OPS 3 where two light sources were required (landing and search light, the latter adjustable).	1. The requirement is in line with ICAO Annex 6 Part II, 2.4.8 c)
		At least two landing lights; electric torch for each crew member.	
(7)	lighting supplied from the aircraft's electrical system to provide adequate illumination for all instruments and equipment essential to the safe operation of the aircraft;		
(8)	lighting supplied from the aircraft's electrical system to provide illumination in all passenger compartments;		
(9)	an electric torch for each crew member station;	 Lack of clarity of the requirement. It is proposed to align with EU-OPS wording. A carry-on flash-light should be 	1, 2. The requirement is in line with ICAO Annex 6 Part II, 2.4.8 f)
		acceptable for small GA-airplanes with a MTOW <5700 kg.	
(10)	lights to conform with International Regulations for Preventing Collisions at Sea (hereinafter referred to as International Regulations for Preventing Collisions at Sea) if the aircraft is amphibious; and		
(11)	in the case of aeroplanes with speed limitations expressed in		1. Wording has been clarified.

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A: Rule		B: Summary of comments	C: Reason for change, remarks	
	terms of Mach number, a means of indicating Mach number.			
(b)	Aeroplanes operating VFR night flights and IFR flights and helicopters operating IFR flights shall be equipped with a chart holder in an easily readable position which can be illuminated for night operations.	1. This requirement should only be applicable to IFR flights. For VFR flights, AOPA-S does not deem a lighted chart holder necessary and the size of the map will make it impractical to fit in, since the critical parts of the flight will be performed with visual references.	1&2&3. The requirements for chart holder and its illumination at night are only maintained in case of flight under IFR.	
		2. Paragraph (b) is not a certification requirement for night VFR or IFR and should be deleted for non-commercial operations. The variety of suitable cockpit lighting, accessories and layout is captured during certification of individual types, and should not be constrained by this kind of "one-size- fits-all" regulation.		
		3. If the operator and aeroplane are approved for alternate means to provide chart data to the flight crew (like EFB electronic charts) this requirement is not necessary.		
			1. Missing requirements of ICAO Annex 6 Part II, 3.6.5.1 have been introduced.	
			Requires 2 independent altitude	

A: Rule	B: Summary of comments	C: Reason for change, remarks
		measuring and display systems for large and turbojet aeroplanes and an emergency power supply for attitude indicator for aeroplanes over 5 700 kg.
SAILPLANES		Not transposed, beyond the scope of Part-NCC.
(c) Sailplanes operating VFR night flights or IFR flights, shall comply with (a) (4) to (10) inclusive.		
BALLOONS		Not transposed, beyond the scope of Part-NCC.
(d) Balloons operated at night shall, in addition to complying with OPS.GEN.410(d) and (e), as applicable, be equipped with:		
(1) position lights; and		
(2) a means of illuminating all of the instruments used by the flight crew.		
OPS.GEN.420 Flights over water		
SAILPLANES AND BALLOONS		Not transposed, beyond the scope of Part-NCC.

A: Rule			B: Summary of comments	C: Reason for change, remarks
risks to survival of the occupants of the aircraft in the event				
(1)	on b	oard, stowed in a position which is readily accessible from		
(2)	eme	rgency locator transmitters; and		
(3)	equi	pment for making distress signal,		
when operating a flight:		a flight:		
	(i)	over water beyond gliding distance from the shore; or		
	(ii)	where the take-off or approach path is so disposed over water that in the event of a mishap there would be a likelihood of ditching;		
	The risks ditch (1) (2) (3)	The pilot- risks to s ditching, t (1) life (on b the s (2) eme (3) equi operating (i)	 The pilot-in-command of a sailplane or balloon shall determine the risks to survival of the occupants of the aircraft in the event of a ditching, based on which he/she shall determine the carriage of: (1) life jackets, or equivalent floatation devices, for each person on board, stowed in a position which is readily accessible from the seat or berth of the person for whose use it is provided; (2) emergency locator transmitters; and (3) equipment for making distress signal, operating a flight: (i) over water beyond gliding distance from the shore; or (ii) where the take-off or approach path is so disposed over water that in the event of a mishap there would be a 	 The pilot-in-command of a sailplane or balloon shall determine the risks to survival of the occupants of the aircraft in the event of a ditching, based on which he/she shall determine the carriage of: (1) life jackets, or equivalent floatation devices, for each person on board, stowed in a position which is readily accessible from the seat or berth of the person for whose use it is provided; (2) emergency locator transmitters; and (3) equipment for making distress signal, operating a flight: (i) over water beyond gliding distance from the shore; or (ii) where the take-off or approach path is so disposed over water that in the event of a mishap there would be a

A: R	ule		B: Summary of comments	C: Reason for change, remarks
LAN	OPLAN	NES		
(b)	Land	dplanes shall be equipped with (a)(1) when:		
	(1)	flying over water beyond gliding distance from the shore; or	 Proposal: (a)(i) and (b)(1) replace "from the shore" by "to the shore". Justification: Change to take into account wind, the intent being to return to the shore within a certain time. 	1. Distance specified.
	(2)	taking off or landing at an aerodrome or operating site where, in the opinion of the pilot-in-command, the take-off or approach path is so disposed over water that in the event of a mishap there would be a likelihood of a ditching.	 French FFA believes that this proposal is not adapted to non-commercial sports and recreational operations, even for initial flight training. Justifications: Life-jackets for each person on board seems disproportionate and very difficult to implement for the numerous sports and recreational flying organisations (aero-clubs) based on aerodromes or airfields situated next to a shoreline (sea, lakes or ponds). Flight safety statistics do not show specific problems justifying that requirement. FFA proposal: Delete the OPS.GEN.420 (b) (2) requirement for non-commercial operations on non -complex aeroplanes, and at least for non-complex aeroplanes below 2 000 kg MTOW. EPFU is of the opinion that life-jackets 	1.&2. The requirement is in line with ICAO recommendation and is based on pilot assessment under (b)(2) operational conditions.3. Accepted.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	requirement for all persons on board as soon as take-off or landing are partially conducted above water is not necessary on non-commercial operations with non- complex aeroplanes. Justifications: Decades of air operations without life- jackets in those circumstances (i.e. flight over water during take-off or landings) on non-commercial operations on non-complex aeroplane show no flight safety problem, so there is no justification for this requirement.	
	3. Delete "in the event of a mishap" Justification: These words don't bring anything useful; moreover the word 'mishap' is not adapted to regulatory wording.	
SEAPLANES		
(c) When flying over water, seaplanes shall, in addition to (a)(1), be equipped with:		
 equipment for making the sound signals, as prescribed by the International Regulations for Preventing Collisions at Sea; 		
(2) one anchor; and		

A: R	tule	B: Summary of comments	C: Reason for change, remarks
	(3) one sea anchor (drogue), when necessary to assist in manoeuvring.		
AER	OPLANES		1. 50 NM included instead of 100 NM in line with ICAO Annex 6 Part II definition of extended flight over water.
(d)	from land where an emergency landing is possible greater than that corresponding to:	1. "normal" cruising speed is missing as per ICAO Annex 6 Part II standard at 2.4.4.3. Additionally, it is noted that the ICAO definition of 'Extended flight over water' as the distance of 93 km (50 NM) or 30 minutes at normal cruising speed, has been extended to 100 NM, in the proposal (mirrored by EU-OPS).	 The proposed changes have been incorporated, in line with ICAO Annex 6 Part II standard at 2.4.4.3. The wording "from land" is conventional ICAO wording used throughout the regulation.
	(1) 120 minutes at cruising speed or 400 nautical miles (nm), whichever is the lesser, in the case of aeroplanes capable of continuing the flight to an aerodrome with the critical power unit(s) becoming inoperative at any point along the route or planned diversions; or		
	(2) 30 minutes at cruising speed or 100 nm, whichever is the lesser, for all other aeroplanes,		1. Changed to 50 NM in line with ICAO Annex 6 definition of 'extended flight over water'.
	shall determine the risks to survival of the occupants of the aeroplane in the event of a ditching, based on which he/she shall determine the carriage, in addition to (b) or (c) and (a)(3), of:		

Part-NCC	CRST
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A: Rule		B: Summary of comments	C: Reason for change, remarks
(i)	life-saving rafts in sufficient numbers to carry all persons on board, stowed so as to facilitate their ready use in emergency; and		
(ii)	life-saving equipment, including means of sustaining life, as appropriate to the flight to be undertaken.	1. Editorial comment: Page 44: OPS.GEN.420 §(d)(2)(ii) : "is" seems superfluous.	1. Editorial change
HELICOPTERS			
(e) Helicopter	s shall be equipped with (a)(1), when operated in:	1. This proposed legislation seems absolutely disproportionate to the risk. I do not believe there has ever been a case of an R22 or R44 ditching. Surely the kind of requirements that you outline should be confined to helicopters over 3 175 kg. Please don't spoil or make unaffordable the genuine enjoyment of visiting countries outside the UK.	1. 2. & 3. The requirements are in line with JAR-OPS 0 and ICAO Annex 6 Part III.
		Why is EASA trying to impose far more restrictions on helicopters than single-engined fixed wing aircraft?	
		2. For non-commercial operations, safety measures for flights over water should be left entirely at the pilot's discretion. I suggest to apply OPS.GEN.420 to commercial operations	

A: Rule		B: Summary of comments	C: Reason for change, remarks
		only. 3. I agree for commercial flights, but this is ridiculous for private flights in the UK! We're an island for goodness sake with more water than any other country in the EU!	
(1)	Performance Class 1 or 2 on a flight over water at a distance corresponding to more than 10 minutes flying time at normal cruising speed;	 The distance shall be referenced from/to the land. All distances stated as time "from land" should be exchanged with "to land". This since a strong wind from land would carry an aircraft further from land and result in greater distance to land. Furthermore, if the return heading towards land results in a headwind the time to reach land will increase. This scenario would produce a lower level of safety level than anticipated. The requirement is more stringent. Flight to Helgoland can, in accordance with the proposed requirement, only be performed with helicopters fully certified and equipped. The HEMS flights should be excluded from the requirement. 	 Editorial change for clarification. "from land" is a phrase used in ICAO. The comment will be taken into account for the development of GM as part of future rulemaking tasks. HEMS is regulated under SPA.HEMS. Moreover for HEMS operations a CAT AOC is required.
(2)	Performance Class 3 on a flight over water beyond autorotational distance from the land; or	1. OPS.GEN.420 (e)(2) uses the wording "beyond autorotational distance from land" while OPS.GEN.425.H § (b)(2)	 The terms "autorotational distance" or "safe forced landing distance" have different

A: R	Rule	B: Summary of comments	C: Reason for change,
			remarks
		uses the wording: "beyond a safe forced landing distance from land".	definitions and are introduced in the requirements on purpose.
		Proposal: the terms shall be harmonised.	
	(3) Performance Class 2 or 3 when taking off or landing at an aerodrome/operating site where the take-off or approach path is over water.		
(f)	When operated in Performance Class 1 or 2 on a flight over water at a distance corresponding to more than 10 minutes flying time at normal cruising speed or in Performance Class 3 on a flight over water at a distance corresponding to more than three minutes flying time at normal cruising speed, helicopters shall, in addition to	 Because there are a number of complicated conditions in (d), the requirement should be spelled out here: "(f), be equipped with (a)(3) and (d)(i) and (ii)." 	 Editorial change for clarification purpose 3 & 4. The requirement to
	(a)(1), and when not precluded by considerations related to the type of helicopter used, be equipped with (a)(3) and (d).	2. Comments are against the propose rule because:	carry life-rafts in small helicopters may be reviewed based on an analysis of the safety benefit, considering the
		- Lack of proportionality for recreational and private helicopter flights;	necessary time to deploy the raft from a small cabin in a short time compared to the additional
		 Lack of space to accommodate the rafts in light helicopters; 	purchase and maintenance costs. Giving appropriate considerations
		- Risk of inadvertent inflation;	to the comments made, small
		 Risk of adverse consequence when trying to deploy the raft; 	helicopters are understood to be other-than-complex helicopters and therefore the passenger
		- UK CAA statistics shows better record for helicopters than for aeroplanes overwater (single engine aircraft).	threshold is nine or less. The Agency therefore believes that for complex helicopters, a life-

A: Rule	B: Summary of comments	C: Reason for change, remarks
	Proposal: carriage of life-rafts is left at	raft shall be carried.
	pilot's decision.	This text:
	For private helicopter flights in non- complex helicopters it should be the pilot's option to carry a life-raft or not. Reason for objection: proportionality for recreational and private helicopter flights. Where the helicopter is non- complex there is not room in the helicopter for safe storage of an accessible life-raft when only one person is in the helicopter. Successful deployment is unlikely. There is also a danger of inadvertent inflation, which could cause loss of control of the helicopter. Suggested alternative wording: add new sentence at the end of current text. "For non-complex private helicopter flight-life raft carriage	"when not precluded by considerations related to the type of helicopter used" should therefore be interpreted not to apply to NCC and is consequently proposed to be deleted. Other-than complex helicopters will be addressed in Part-NCO and alleviations will be included, as provided in Annex 6 Part III Section III.
	is at the pilots option."	
	3. UK CAA statistics demonstrate that privately operated single-engined helicopters have a substantially better safety record with regard to flight over water than do privately operated single- engined aeroplanes.	
	4. There would be great chance of anyone getting caught or snagged when	

A: R	ule	B: Summary of comments	C: Reason for change, remarks
		trying to do so and be pulled down under the water by the sinking machine. Any fire officer will advise any member of the public to leave a burning building asap and let it burn. Why do you want someone who has escaped a ditching helicopter to go back to it and try and retrieve a life-raft? Any sensible pilot- in-command wears and asks their passengers to wear the life-jackets when flying over large areas of water. This is common sense but shouldn't be mandatory as often it could cause fear in passengers even before they set off. Life-jackets maybe, life-rafts no - they only add a disproportionate amount of weight and are inaccessible when needed. There isn't sufficient room in most small helicopters for anything else. We don't need the extra weight either.	
(g)	The pilot-in-command of a helicopter operated in Performance Class 3 shall determine the risks to survival of the occupants of the helicopter in the event of a ditching, based on which he/she shall determine if the life jackets required in (e) shall be worn by all occupants.	1. The text would be improved by amending as follows: "(g) The pilot-in- command of a helicopter operated in Performance Class 3 shall determine the risks to survival of the occupants of the helicopter in the event of a ditching, based on which he/she shall determine when deciding if the life jackets required in (e)(1) shall be worn by all	1. Accepted

A: Rule		B: Summary of comments	C: Reason for change, remarks	
		occupants."		
ALL	AIRCRAFT			
(h)	Each life jacket or equivalent individual flotation device, when carried in accordance with (b), (c), (d), (e), (f) or (g) shall be equipped with a means of electric illumination for the purpose of facilitating the location of persons.	1. To keep the electric illumination in good shape and working is demanding. The same lighting can be supplied by chemical lights. Electric or chemical illumination should be an option - not a demand.	 & 2. The requirement is based on ICAO. Requirements for sailplanes and balloons are proposed to be re-introduced in line with ICAO. 	
		2. The requirement for life-jackets to be equipped with a means of electric illumination, according to the title, is for all aircraft and yet sailplanes and balloons ((a)) are omitted from the list.		
OPS	5.GEN.425.H Ditching - Helicopters			
(a)	Helicopters operated in Performance Class 1 or 2 on a flight over water in a hostile environment at a distance corresponding to more than 10 minutes flying time at normal cruising speed shall be:	 Comments are against the propose rule because: There is no safety case for these proposals. These proposals would, effectively, prohibit helicopter flights over water without emergency floatation equipment. 	1. & 2. The requirement to install a flotation device on light helicopters with reduced passenger capacity for private operations is reviewed in Part NCO.	

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	 Many helicopters cannot be fitted with floats. 	
	 The cost of fitting floats is very high (price list provided by some commenters) - disproportionate to the risk. 	
	 There is a strong argument that ditching without floats is safer than ditching with floats. 	
	 Specific training for autorotation techniques with floats may be required. 	
	 The risk of uncommanded inflation can create more hazard than potential safety. 	
	 Floats increase the operating costs and decrease the performance of helicopters - fuel efficiency is also reduced. 	
	Recorded proposals are as follows:	
	The requirements for light helicopters should be the same as those for aeroplanes, or:	
	 to apply the requirements OPS.GEN.425.H (a) & (b) to CAT only (with AMC to require wearing lifejackets whilst more than 10 minutes flying time 	

A: Rule	B: Summary of comments	C: Reason for change, remarks
	to land), or	
	- to apply the requirement OPS.GEN.425.H (a) to CAT only and only maintain OPS.GEN.425.H (b) for PC 1 & 2.	
	The decision is left to the private pilot's decision.	
	Compliance is achieved for private flights by wearing life-jackets whilst more than 10 minutes flying time from land when operating over open water.	
	When an IFR helicopter is flying in IMC it is rightly and reasonably considered adequately safe even if IMC conditions extend below 1 000 ft AGL, and potentially to the ground. The risk of flying over land in such conditions is no less, and may be greater, than flying over water. The equipment levels required for IFR certification should therefore enable flight over water for at least 60 minutes without floatation equipment.	
	2. This proposal is prejudiced against non-mainland European helicopter owners/pilots. UK owners/pilots live on an island and very occasionally wish to cross to mainland Europe or to Ireland.	

A: Rule		B: Summary of comments	C: Reason for change, remarks
	(1) designed for landing on water; or		
	(2) certificated in accordance with ditching provisions in the relevant airworthiness code.		
(b)	In addition, helicopters shall comply with (a) or be fitted with emergency flotation equipment when operated in:	 In Part COM it is not always possible to have the helicopter fitted with flotation equipment (because of the specific equipment). Alleviation should be given for helicopters in COM to reach the work area if a flight over water is needed. HEMS flights shall be excluded from the rule. 	 COM will be addressed in SPO.IDE. HEMS are not addressed in NCC.IDE but in SPA.HEMS. Moreover, a HEMS operation would require a CAT AOC.
	 Performance Class 1 or 2 on a flight over water in a non- hostile environment at a distance corresponding to more than 10 minutes flying time at normal cruising speed; 		
	(2) Performance Class 3 on a flight over water beyond a safe forced landing distance from land; or		
	(3) Performance Class 2 when taking off or landing over water, except in the case of Helicopter Emergency Medical Services (HEMS) operations, where for the purpose of minimising exposure, the landing or take-off at a HEMS operating site located in a congested environment is conducted over water.		

A: Rule	B: Summary of comments	C: Reason for change, remarks
OPS.GEN.430 Emergency Locator Transmitter (ELT)	 1. Disproportionate approach toward light aircraft operated for non-commercial purposes. It is the opinion of the UK CAA that portable ELT devices i.e. (ELT(S)) or (ELT(AP)) are the most appropriate device for use in light aircraft. However, there are few products on the market that are approved to EUROCAE Document ED-62 specifications. Those operating non-complex aircraft on non-commercial flights should not be required to carry approved ELT equipment all the time but have available the option of carrying an alternative non-approved survival locator beacon instead. This is in line with the policy for other items of survival equipment such as lifejackets, which do not need to be approved. New text is proposed. The requirement should account for specific operations. Proposal is to reduce the scope to "operated in areas designated by Member States as those in which search 	 1., 2. The requirements are in line with ICAO Annex 6 Part II. 3., 4. The proposal is taken into account for NCO.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	and rescue would be especially difficult".	
	3. It is proposed that a GPS-based personal locator beacon (PLB) is made mandatory for GA pilots rather than an automatic ELT. At least PLB shall be AMC.	
	4. A hand-held Emergency Locator Beacon is a sensible precaution when flying over hostile terrain, particularly water, but it is neither necessary nor justified when flying over populated areas. It should be the pilot's discretion what equipment to carry on any given flight.	
AEROPLANES		
(a) Aeroplanes first issued with an individual certificate of airworthiness before and including 1 July 2008 shall be equipped with an Emergency Locator Transmitter (ELT) of any type.	 The ELT rule does not include the frequency requirement, asking for 121.5 and 406 MHz capability. This, however, is included in AMC OPS.GEN.430 No.2, which is not acceptable. Type shall be defined otherwise there will be different regulations in the Member States. It should possible to satisfy the ELT requirement by carrying a PLB (personal locater baccon) approved by Cospass 	 The frequencies have been included in the rule. Types are defined in AMC. The possibility to carry PLBs as an alternative to ELTs will be addressed for NCO. If the commentator wishes to extend this to NCC is invited to submit a proposal for a future rulemaking task.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	Sarsat and coded for aviation use. 4 If an ELT "of any type" is acceptable, is the aircraft "equipped" if an ELT(S) is carried by the pilot?	installation is approved, the aircraft is considered equipped.
(b) Aeroplanes first issued with an individual certificate of airworthiness after 1 July 2008 shall be equipped with an automatic ELT.	 For small airplanes used in non- commercial operations, this results in an undue installation cost burden. Propose to drop this requirement for small airplanes (e.g. up to 2 000 kg MTOW) used in non-commercial operations. The requirement in Annex 10 Vol V about the remote control in cockpit for ELT (in order to switch between OFF, ARM an ON) should be clarified in this paragraph or in an AMC. 	 The requirements are in line with ICAO Annex 6 Part II. This ELT function is already included in the following documents: ETSO 2C91A (121.5 Mhz) 2C126 (406Mhz) paragraph 2.5 and EUROCAE ED-62 and future ED62(a) reference.
OPS.GEN.435 Survival equipment – Motor-powered aircraft		
ALL AIRCRAFT		
(a) Aircraft operated across areas in which search and rescue would be especially difficult shall be equipped with the following:	 According to the title of the paragraph applicability to MPA only shall be specified. `across areas in which SAR would be especially difficult' is open to interpretation, and therefore lacks legal certainty. Proposal: specify 'over areas 	 Applicability to MPA only is clarified. The requirement is in line with ICAO and the area will be specified by the States.

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A: Rul	e	B: Summary of comments	C: Reason for change, remarks
		designated by Member States as especially difficult for SAR'.	
(1) Signalling equipment to make distress signals;	1. If you fly over the Alps you have to be equipped with signalling equipment. This will interfere with security regulations. If a pilot carries a signal device to his/her aircraft he/she will have problems with the police.	1. The requirement is in line with ICAO and no conflict with safety regulations has been identified.
(3	2) At least one ELT (S); and		
(1	 Additional survival equipment for the route to be flown taking account of the number of persons on board. 		
AEROP	PLANES		
s	Notwithstanding (a)(3), in the case of aeroplanes, the additional survival equipment specified in (a)(3) need not be carried when the aeroplane:		
(1) remains within a distance from an area where search and rescue is not especially difficult, that corresponds to:		
	 (i) 120 minutes at the one engine inoperative cruising speed for aeroplanes capable of continuing the flight to an aerodrome with the critical power unit(s) becoming inoperative at any point along the route or planned diversion routes; or 		

A: R	ule	B: Summary of comments	C: Reason for change, remarks
	(ii) 30 minutes at cruising speed for all other aeroplanes;		
	or,		
	(2) remains within a distance no greater than that corresponding to 90 minutes at cruising speed from an area suitable for making an emergency landing; in the case of a complex motor-powered aircraft certificated in accordance with the applicable airworthiness code issued by the Agency.	1. The applicability of this paragraph to CMPA only is confusing as there is an "or" between (a) and (b).	1. This has been clarified in the redrafted requirements.
OPS.GEN.440 High altitude flights – Oxygen		 Editorial change: should be amended to read "High altitude flights - Supplemental oxygen". 	1. The title of the paragraph is changed in line with CAT.IDE, for consistency.
		2. A/C in parachute operation are excepted. Up to the PIC to determine whether oxygen should be carried on board or not.	2. Current proposal in line with ICAO.
ALL A	AIRCRAFT		
(b)	Aircraft flying above altitudes at which the pressure altitude in the passenger compartments is above 10 000 ft (feet) shall carry enough breathing oxygen to supply:	 Not an OPS requirement. To be moved to Part 21. Editorial change. Suggestion: "When an aircraft is operated such that the pressure altitude 	 Part-21 does not contain a requirement for oxygen equipment and it needs to be addressed in this rule. Editorial change: improvement of the wording for consistency.
		of the passenger compartment is above 10,000ft, enough breathing oxygen	of the wording for consistency within the requirement.

A: Rule		B: Summary of comments	C: Reason for change, remarks
		should be carried to supply.	3. The proposed requirements
		3. Should be 12 500 ft for harmonisation with most 3 rd country rules.	are in line with ICAO Annex 6 Part II.
(1)	in the case of non-pressurised aircraft:		
	 (i) all crew members and at least 10% of the passengers for any period in excess of 30 minutes when the pressure altitude in the passenger compartments will be between 10 000 ft and 13 000 ft; and 	1. Should be 12 500 ft and 14 000 ft respectively.	1. No change - current proposal is in line with ICAO.
	 (ii) all crew members and passengers for any period that the pressure altitude in passenger compartments will be above 13 000 ft. 		
(2)	in the case of pressurised aeroplanes:		
(i)	all crew members and a proportion of the passengers, for any period when, in the event of loss of pressurisation and taking into account the circumstances of the flight, the pressure altitude in the passenger compartment would be above 10 000 ft; and	1. Proportion of passengers to be clearly defined.	1. Accepted. Proportion of the passengers is specified
(ii)	all the occupants of the passenger compartment for no less than 10 minutes, in the case of aeroplanes operated at pressure altitudes above 25 000 ft, or operated below that altitude, but under conditions that will not allow them to descend safely to a pressure altitude of 13 000 ft within four		1. Accepted.

Part-NCC	CRST
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A: Rule		B: Summary of comments	C: Reason for change, remarks
	minutes; and		
(3)	in the case of pressurised helicopters, all crew members and a proportion of the passengers, for any period when, in the event of loss of pressurisation and taking into account the circumstances of the flight, the pressure altitude in the passenger compartment would be above 10 000 ft.	 Editorial change. Alleviation, approved by NAA, for short periods up to 16 000 ft shall be possible, based on the kind of work and the training of the crew flying in mountainous areas. 	1 6. Requirement for pressurised helicopters has been withdrawn - in line with the change made for Part-CAT.
		3. Should be 12 500 ft in order to harmonise with most 3rd country rules.	
		4. 10 000 ft rigid limit is not acceptable. Operations up to 13 000 ft for up to 30 min without the need for supplemental oxygen should be allowed (in line with ICAO). JAR-OPS 3.385 to be inserted.	
		5. Providing oxygen for passengers should only be applicable to CAT operations, and is already covered by OPS.CAT.440 and table 2 OPS.CAT.440.	
		6. Proportion of passengers to be clearly defined.	
(b)	When engaged in performing duties essential to the safe operation of an aircraft in flight, flight crew members shall use breathing oxygen continuously whenever the circumstances prevail for which its supply has been specified in (a).	1. Should be transferred to Operational Procedures (Section II) e.g. OPS.GEN.XXX Use of supplemental oxygen, to be consistent with ICAO and EU-OPS.	 The text remains with this rule, in line with the Part-CAT equivalent. The wording has been modified to align with ICAO

A: R	ule	B: Summary of comments	C: Reason for change, remarks
		2. "the circumstances specified in (a)" could be interpreted as being a requirement to breath oxygen continuously above 10 000 ft because actions of the flight crew are essential to the safe operation and a cabin decompression will lead to a cabin altitude above the 10 000 ft criteria. This is not the intent of the proposed requirement. To be replaced by EU-OPS 1.385.	Annex 6.
(c)	Aircraft operated at altitudes for which the oxygen supply is required in accordance with (a) shall be equipped with oxygen storage and dispensing apparatus.	1. Portable systems should not be excluded. Such systems are used in GA and gliders.	1. Portable oxygen is not explicitly excluded by the proposed rule.
		2. Not an OPS requirement. To be moved to Part-21.	2. There is not requirement in Part-21 to mandate oxygen equipment and it needs to be addressed in this rule.
			Editorial change: improvement of the wording for clarity.
PRES	SSURISED AEROPLANES		
(d)	Pressurised aeroplanes operated at flight altitudes above 25 000 ft, or pressurised aeroplanes used in commercial air transport, shall, in addition, be equipped with a device to provide a warning indication to the flight crew of any loss of pressurisation.	1. Requirement for pressurised aeroplanes used in commercial air transport to be moved to OPS.CAT.440.	1. Editorial change.

A: Rule	B: Summary of comments	C: Reason for change, remarks
OPS.GEN.445 Operations in icing conditions at night		
Aircraft flying in expected or actual icing conditions at night shall be equipped with a means to illuminate or detect the formation of ice. Such illumination shall not cause glare or reflection that would handicap flight crew members in the performance of their duties.	 An objective requirement for Flight in Expected or Known Icing Conditions should be appended to this rule. This goes beyond the ERs and places the responsibility with the PIC for entering icing conditions. This will also be seen by private pilots who will then know to avoid such conditions. Text proposal, perhaps only applicable to CAT and CMPA, to add that "(d) an operator shall establish procedures for flights in expected or actual icing conditions". Requirement only applicable to CAT and COM, and complex. AMC OPS.GEN.445 should be introduced: "For non-complex A/C (excluded CAT and COM) a portable light can be used to illuminate or detect the formation of ice". Title to be amended "in-flight operations []". Equipment designed to only cover the in-flight part of the flight, not the ground part of it (i.e. taxiing). "AEROPLANE and HELICOPTER" sub- title to be added, as no other A/C category is certified to conduct flights in 	 Such objective requirement is already covered by 2.a.5 of Annex IV to the Basic Regulation and does not need to be duplicated. The requirement is maintained for complex aircraft only. The illumination means may also be used on ground for identifying if de-icing is needed. The requirement is maintained for aeroplanes and helicopters only.

A: Rule	B: Summary of comments	C: Reason for change, remarks
	icing conditions.	
OPS.GEN.450 Marking of break-in points	1. Not an OPS requirement; keep design requirements within appropriate documents	1. This requirement is in line with ICAO and does not impose on any aircraft to have break-in areas.
br break-in by rescue crews in II be marked as shown in Som Not over 2 m		
OPS.GEN.455 First-aid kits		
(a) Aeroplanes shall be equipped with first-aid kits in accordance with Table 1 of OPS.GEN.455.	 Powered sailplanes shall be excluded. Exemption shall be allowed for ferry flights where one FAK should be 	1. Not accepted. Powered sailplanes are addressed under NCO.IDE.
	enough. 3. Delete this requirement of a "first-aid kit" on non-commercial and non-	2. The ICAO requirement specifies it is applicable on all flights.
	complex aeroplanes operations, and at least on non-commercial, non-complex aeroplanes below 2 000 kg MTOW.	 3. The requirement is in line with ICAO. 4. The requirement is in line with

A: Rule		B: Summary	of comments	C: Reason for change, remarks
		aircraft where	nt shall not apply to each occupant is forced to eats (e.g. single	ICAO.
		manner does which is not e	a first-aid kit in safe require sufficient space, asily available in smaller minor change is required.	
			tion: no requirement for a r small airplanes and craft.	
Table 1 of OPS.GEN.4	155			
Number of passenger seats installed	Number of first- aid kits required	the ICAO SAR that will be ap	r of kits does not match Ps in Annexe 6, Chapter 6 plicable from November	1. Number of kits updated according to ICAO Annex 6 Part I Amdt 33.
0 to 99	1	2009.		
100 to 199	2			
200 to 299	3			
300 and more	4			

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(b)	 Balloons. (1) Balloons shall be equipped with one first-aid kit. (2) In the case of balloons with a maximum passenger capacity of more than 11, a second first-aid kit shall be carried in the retrieval vehicle. 		Not transposed, beyond the scope of Part-NCC.
(c)	The first-aid kits shall be readily accessible for use.		
(d)	First-aid kits shall be maintained.	 First-aid kits shall be maintained and replenished at regular intervals. Justification: in line with JAR-OPS. 	1. Change of wording and format.
OPS	5.GEN.460 Airborne Collision Avoidance System (ACAS) II		
ALL	AIRCRAFT		
(a)	Whenever Airborne Collision Avoidance System (ACAS) II is installed, it shall be used in normal conditions during flight in a mode that enables Resolution Advisories (RAs) to be produced for the pilot flying when undue proximity to another aircraft is detected.	1. Paragraphs (a) and (b) are operational rules, not equipment requirements and they should be transferred to Subpart A Section II under Operational Procedures.	1. This operational rule is now addressed in NCC.OP.210. The text has been aligned with Part- CAT, CAT.OP.MPA.295. It should be further noted that ACAS is also addressed in a European airspace rule in AUR.ACAS, which will be published in the near

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			future. The text is also aligned with these rules. The OPS rule is maintained to cater for situations in which the aircraft is operated outside of European airspace.
(b)	When an RA is produced by ACAS II, the pilot flying shall immediately take the corrective action indicated by the RA, even if this is in conflict with an Air Traffic Control (ATC) instruction. The aircraft shall be promptly returned to the terms of the ATC instructions or clearance when the situation is resolved.	 It is proposed to realign with EU-OPS 1.398 as EU-OPS is not only clearer, but also contains an important prerequisite for initiating any corrective action indicated by the RA, that is "unless doing so would jeopardise the safety of the aeroplane". Suggest: "When the situation is resolved the aircraft will thereafter be 	 Accepted, Text aligned with EU-OPS. The content of this proposal has been transposed. - 4. Noted. Text aligned with CAT and AUR.ACAS. Additional clarification to manage priority with TAWS alert has been
		flown in accordance with the previously received and acknowledged ATC instructions or clearance."	introduced. 5. Grammar has been corrected.
		3. To further increase safety, a clause should be added that TAWS alerts or warnings to avoid terrain or obstacles take precedence over ACAS II RA instructions.	
		4. To do not delay urgent HEMS missions, helicopters usually are allowed to cross centre-line or approach- /landing sectors during IFR traffic on the aerodrome. ATC advises all aircraft crew about the helicopter traffic within their	

A: R	ule	B: Summary of comments	C: Reason for change, remarks
		approach or take-off sectors to avoid unessential ACAS advices. Following the rule (b) HEMS mission are forced to make a detour and will be unacceptable delayed.	
		The last sentence of (b) is proposed to be deleted.	
(c)	prescribed ACAS ATC communications are specified.		
(d)	when the conflict is resolved the aeroplane is promptly returned to the terms of the ATC instructions or clearance.		
AER	PLANES		
(c)	Turbine-powered aeroplanes shall be equipped with an ACAS with a minimum performance level of at least ACAS II, when having:		
	 for non-commercial operations a maximum certificated take- off mass exceeding 15 000 kg (kilograms) or a maximum passenger seating configuration of more than 30; or 		
	(2) for commercial operations a maximum certificated take-off mass exceeding 5 700 kg or a maximum passenger seating configuration of more than 19.		
	GEN.465.A Terrain Awareness Warning System (TAWS) - oplanes	1. Small airplanes designed and certified to different regulations i.e. CAR-3 (preceding FAR/CS 23) were only limited to a MTOW of 12 500 lbs with no	 The criterion of the number of seats is in line with ICAO Annex 6.

A: R	lule	B: Summary of comments	C: Reason for change, remarks	
		limitation on the number of seats. It is therefore unreasonable to mandate airplanes designed and certified to these earlier regulations that may have a seating capacity of one more than is required in the modern regulations to also comply with the latest regulatory requirements, purely based on the number of seats.		
(a)	Turbine-powered aeroplanes with a maximum certificated take-off mass exceeding 5 700 kg, or maximum passenger seating configuration of more than nine, shall be equipped with a TAWS that meets the requirements for Class A equipment.	 The requirement for Class A is not in line with EU-OPS 1.665. Stick to EU-OPS 1.665. The more than 9 criteria is challenge. Why not more than 19? This rule requires a higher standard of TAWS equipment than is required by ICAO Annex 6 Part II. The requirement in OPS.GEN should be Class B TAWS as the basic standard applicable to non- CAT operations. Class A TAWS is restricted to OPS.CAT and therefore the requirement should be moved to Subpart B. 	 1.&3. The proposal is in line with NPA-OPS 39B draft conclusions for CAT.IDE. For NCC, the applicability criteria are proposed to be aligned with ICAO Annex 6. 2. The criterion of more than 9 is in line with current ICAO and EU- OPS requirements. 	
(b)	Reciprocating-engined-powered aeroplanes with a maximum certificated take-off mass exceeding 5 700 kg, or a maximum passenger seating configuration of more than nine and used in commercial air transport, shall be equipped with a TAWS that			

A: Rule		B: Summary of comments	C: Reason for change, remarks
	meets the requirement for Class B equipment.		
(c)	A TAWS shall provide, automatically, a timely and distinctive warning to the pilot flying, of:		1. The specifications on TAWS functions are included in the Class A & B definitions and therefore paragraph (c) is proposed to be removed.
	(1) sink rate;		
	(2) ground proximity;		
	(3) altitude loss after take-off or go-around;		
	(4) incorrect landing configuration; and		
	(5) downward glide-slope deviation.		
OPS	GEN.470.A Means for emergency evacuation - Aeroplanes	1. Requirement to be deleted as it is already addressed in the corresponding CS-25 requirement, which is less stringent by the way. As CS-25 cannot be amended retro-actively, this subject should be addressed by a Safety Directive, if required.	1. This rule has been withdrawn as it is not part of ICAO Annex 6.

A: R	tule	B: Summary of comments	C: Reason for change, remarks
(a)	Aeroplanes with passenger emergency exit sill heights of more than 1.83 metres (six feet) above the ground and aeroplanes with a separate emergency exit for the flight crew of more than 1.83 metres (six feet) above the ground, shall have <u>a</u> means to enable passengers and crew at each exit, to reach the ground safely in an emergency		
(b)	Notwithstanding, such means need not be provided at over-wing exits if the designated place on the aeroplane structure at which the escape route terminates is less than 1.83 metres (six) from the ground, with the aeroplane on the ground, the landing gear extended and the flaps in the take-off or landing position, whichever flap position is higher from the ground.		
	 after the collapse of, or failure to extend of, one or more legs of the landing gear, for aeroplanes type certificated after 31 March 2000; and 		
	(2) when the aeroplane has its landing gear extended, for all other aeroplanes.		
(d)	Assisting means for emergency evacuation that deploy automatically shall be armed during all phases of flight, including taxiing.		
OPS	G.GEN.475 Emergency lighting – Aeroplanes and helicopters	1. The requirement should be moved to OPS.CAT. There is no requirement for emergency lighting in ICAO Annex 6 Part II or Part III Section III.	1. This rule has been withdrawn as it is not part of ICAO Annex 6.

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A: F	ule	B: Summary of comments	C: Reason for change, remarks
AER	OPLANES		
(a)	Aeroplanes with a maximum passenger seating configuration of more than nine shall be equipped with an emergency lighting system with an independent power supply to facilitate the evacuation of the aeroplane.		
(b)	For aeroplanes with a maximum passenger seating configuration of more than 19, the emergency lighting system shall include:		
	(1) sources of general cabin illumination;		
	(2) internal lighting in floor level emergency exit areas;		
	(3) illuminated emergency exit marking and locating signs;		
	(4) in the case of aeroplanes for which the application for the type certificate or equivalent document was filed before 1 May 1972, when flying by night, exterior emergency lighting at all over-wing exits and at exits where descent assist means are required;		
	(5) in the case of aeroplanes for which the application for the type certificate or equivalent document was filed after 30 April 1972, when flying by night, exterior emergency lighting at all passenger emergency exits; and		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	(6) in the case of aeroplanes for which the type certificate was first issued after 31 December 1957, floor proximity emergency escape path marking systems in the passenger compartments.		
(c)	For aeroplanes with a maximum passenger seating configuration of 19 or less and issued with a type certificate in accordance with the European Aviation Safety Agency's (hereinafter referred to as the Agency) airworthiness codes, the emergency lighting system shall include (a)(2)(i)(b)(1), b)(2)(a)(2)(ii) and b)(3)(a)(2)(iii).		1. Erroneous references have been corrected. Editorial change.
(d)	For aeroplanes with a maximum passenger seating configuration of 19 or less and not issued with a type certificate in accordance with the applicable airworthiness codes, the emergency lighting system shall include (a)(2)(i).		1. Erroneous references have been corrected. Reference to airworthiness codes is clarified.
HEL	ICOPTERS		
(e)	Helicopters with a maximum passenger seating configuration of more than 19 shall be equipped with:		
	(1) an emergency lighting system having an independent power supply to provide a source of general cabin illumination to facilitate the evacuation of the helicopter; and		
	(2) illuminated emergency exit marking and locating signs.		
(f)	Helicopters operating over water in commercial air transport operations shall be equipped with emergency exit illumination.		

A: Rule	B: Summary of comments	C: Reason for change, remarks
OPS.GEN.480 Seat belts and harnesses	 It is not clear why the previously single rule for seats, safety belts and harnesses has been divided between a number of rules (GEN.405, GEN.480 GEN.545, CAT.406 and CAT.482); it might be more appropriate to put all of the elements concerned with seats into a single rule in GEN.480. The below requirements are design requirements that should not be included in OPS rules. Our comment is to highlight the different terms used between "safety belt with shoulder harness", "safety harness", "safety belt with diagonal shoulder strap" since it may cause confusion. Our understanding is that a "safety belt with shoulder harness" equals a "safety belt with diagonal shoulder strap" because the word "harness" is singular (plural is harnesses). Clarification that "safety belt with shoulder harness" and "safety belt with diagonal shoulder strap" have only one band on the torso should be part of an AMC or GM. This AMC or GM should also provide some more clarification on what a "safety harness" 	 For clarification of presentation of the requirements: provisions on seat belts and harnesses have been moved into a dedicated requirement for restraint systems and child restraint devices. a dedicated rule for the hand fire extinguisher has been created, incorporating OPS.GEN.405 (a)(1) and (b). Proposed rules are in line with ICAO Annex 6 requirements. For consistency, with CAT.IDE, only the term "harness" is no longer used. A definition is proposed for upper torso restraint systems (UTR).

A: R	Rule	B: Summary of comments	C: Reason for change, remarks
		is compared to the two other terms.	
ALL	AIRCRAFT EXCEPT BALLOONS		
(a)	All aircraft, excluding balloons, used in commercial operations and complex motor-powered aircraft shall be equipped with.		
	(1) a safety belt with shoulder harness incorporating a device which will automatically restrain the occupant's torso in the event of rapid deceleration on each flight crew seat; and	1. The wording "safety belt with shoulder harness", "safety harness", "safety belt with diagonal shoulder strap" is very confusing. It is not clear whether "safety belt with shoulder harness" = "safety belt with diagonal shoulder strap".	 The text is updated to refer to "UTR". The requirement is in line with ICAO Annex 6 Part II Section III with regards to the safety harness.
	(2) a safety harness on the seats for the minimum required cabin crew of all aircraft with a maximum passenger seating configuration of more than 19, which were first issued with an individual certificate of airworthiness after 31 December 1980.		
(b)	Safety belts with shoulder harnesses shall have a single point release.	1. A performance objective is suggested: "Safety belts with shoulder harnesses shall be easy to release" with an associated AMC for the single point release feature. This will avoid unnecessary cost of replacing existing approved seat belts.	1. The comment will be taken into account for future rulemaking task OPS.065 Review of equipment requirements.

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A: F	Rule	B: Summary of comments	C: Reason for change, remarks
HEL	ICOPTERS		
(c)	Helicopters used in commercial operations and complex motor- powered helicopters first issued with an individual certificate of airworthiness after 31 July 1999, shall be equipped with a safety belt with diagonal shoulder strap or safety harness for each passenger over the age of 24 months.	3. Our comment is to highlight the different terms used between "safety belt with shoulder harness", "safety harness", "safety belt with diagonal shoulder strap" since it may cause confusion. Our understanding is that a " safety belt with shoulder harness" equals a "safety belt with diagonal shoulder strap" because the word harness is singular (plural is harnesses). Clarification that "safety belt with shoulder harness" and "safety belt with diagonal shoulder strap" have only one band on the torso should be part of an AMC or GM. This AMC or GM should also provide some more clarification on what a "safety harness" is compared to the two other terms.	 For clarification of presentation of the requirements provisions on seat belts and harnesses have been moved into a dedicated requirement for restraint systems and child restraint devices. Proposed rules are in line with ICAO Annex 6 requirements. For consistency, with CAT.IDE, only the term "harness" is no longer used. A definition is proposed for upper torso restraint systems (UTR).
OPS	5.GEN.485.A Crash axes and crowbars - Aeroplanes		-
(a)	Aeroplanes with a maximum certificated take-off mass exceeding 5 700 kg, or with a maximum passenger seating configuration of more than nine, shall be equipped with one crash axe or crowbar located on the flight deck.		

A: R	ule	B: Summary of comments	C: Reason for change, remarks
(b)	When the aeroplane has a maximum passenger seating configuration of more than 200, an additional crash axe or crowbar shall be carried and located in or near the most rearward galley area.		
OPS	.GEN.490 Flight data recorder - Aeroplanes and Helicopters		
AERO	DPLANES		
(a)	Aeroplanes: (1) with a maximum certificated take-off mass exceeding 5 700 kg and first issued with an individual certificate of airworthiness after 1 January 2005; and	1. OPS.GEN.490 states 2005 while the AMC states 2010. There should be a single date and it should be postponed to avoid costly retrofits.	 Regulatory justification: ICAO Annex 6 Part II Section 3 states: "3.6.3.1.2.2 All aeroplanes of a maximum certificated take-off mass of over 5 700 kg for which the individual certificate of airworthiness is first issued on or after 1 January 2005 shall be equipped with a Type IA FDR." However, there was no requirement in the European regulation for this, and sufficient notice should be given to
			industry to allow it to comply. Two years' notice as a minimum seems to be reasonable for the industry. Since the OPS

A: Rule	B: Summary of comments	C: Reason for change, remarks
		Regulation should enter into force on 8 April 2012, it is proposed to mandate flight parameter recording for aeroplanes with an individual certificate of airworthiness issued on or after 1 January 2016 to be consistent with requirements for CAT.
(2) with a maximum certificated take-off mass exceeding 27 000 kg and first issued with an individual certificate of airworthiness after 31 December 1988,		
shall be equipped with a flight data recorder (FDR) that uses a digital	1. The FDR does not usually contain a method to fully retrieve data from its storage medium: some additional information is needed, in particular the parameters frame layout.	1. Editorial justification
method of recording and storing data and for which a method of readily retrieving that data from the storage medium is available.		ICAO Annex 6 Part II prohibits the use of non-digital recording medium:
		"3.6.3.1.3.1 The use of engraving metal foil FDRs shall be discontinued by 1 January 1995."
		"3.6.3.1.3.3The use of photographic film FDRs shall be discontinued".
		Flight data recorders do not contain themselves a method to retrieve data. The "method of readily retrieving data" from the storage medium is implied by the

A: Rule	B: Summary of comments	C: Reason for change, remarks
		description of the inspection of the flight recorder systems in Appendix 3.1, paragraph 7.2:
		"d) the readout facility shall have the necessary software to accurately convert the recorded values to engineering units and to determine the status of discrete signals;"
(b) The FDR for aeroplanes shall be capable of retaining data recorduring at least the last 25 hours.	rded 1. The requirements for the specific parameters to be recorded that were in EU-OPS must not be 'downgraded' to what is effectively only a recommendation as the NPA only addresses them in the AMC section. If these are not required as part of the rule it is likely that any standardisation currently in place in Europe will be lost. The tables need moving back to the rule.	1. Safety justification The nature of the data to record should at least be described succinctly, since some information is essential for safety investigations. However, the exhaustive list of parameters and the recording quality requirements (range, accuracy, resolution, sampling rate) can stay in the AMC.
		In ICAO Annex 6 Part II, regarding the information recorded:
		" 3.6.3.1.1.1 Types I and IA FDRs shall record the parameters required to determine accurately the aeroplane flight path, speed, attitude, engine power,

A: Rule	B: Summary of comments	C: Reason for change, remarks
		configuration and operation."
		Regarding duration of the recording:
		"3.6.3.1.4 Duration
		All FDRs shall be capable of retaining the information recorded during at least the last 25 hours of their operation."
HELICOPTERS	1. For GEN: apply dates and weights	1. Regulatory justification
according to ICAO Annes	according to ICAO Annex 6.	ICAO Annex 6 Part III states in Section 3:
		"4.7.1.2.1 All helicopters of a maximum certificated take-off mass of over 3 180 kg for which the individual certificate of airworthiness is first issued on or after 1 January 2016 shall be equipped with a Type IVA FDR"
		"4.7.1.2.2 All helicopters of a maximum certificated take-off mass of over 7 000 kg, or having a passenger seating configuration of more than nineteen, for which the individual certificate of airworthiness is first issued on or after 1 January 1989 shall be

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		equipped with a Type IV FDR." However there is currently no requirement in European regulation, and sufficient notice should be given to industry to allow it to comply when the new regulation comes into force. Two years' notice as a minimum seems to be a reasonable for industry. Since the OPS Regulation should enter into force on 8 April 2012, it is proposed to mandate flight parameter recording for aeroplanes with an individual certificate of airworthiness issued on or after 1 January 2016 to be consistent with requirements for CAT.
(c) Helicopters:		
 with a maximum certificated take-off mass exceeding 3 175 kg and first issued with an individual certificate of airworthiness after 1 January 2005; 		
(2) with a maximum certificated take-off mass exceeding 7 000 kg and first issued with an individual certificate of airworthiness after 1 August 1999 up to and including 31		

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	December 2004; and		
	(3) with a maximum certificated take-off mass exceeding 7 000 kg and first issued with an individual certificate of airworthiness after 31 December 1988 up to and including 31 July 1999,		
and s	be equipped with an FDR which uses a digital method of recording toring data and has a method of retrieving that data from the ge medium available.		
(d)	The FDR for helicopters shall be capable of retaining data recorded during at least:		
	 the last eight hours, for helicopters referred to in (c)(1) and (c)(2); 		
	(2) the last five hours, for helicopters referred to in $(c)(3)$; and		
	(3) the last 10 hours, for helicopters with a maximum certificated take-off mass exceeding 3 175 kg and first issued with an individual certificate of airworthiness after 31 December 2009.		
AERC	PLANES AND HELICOPTERS		
(e)	Data shall be obtained from aircraft sources which enable accurate correlation with information displayed to the flight crew.		

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(f)	The FDR shall automatically start to record the data prior to the aircraft being capable of moving under its own power and shall stop automatically after the aircraft is incapable of moving under its own power.		
(g)	The FDR shall have a device to assist in locating it in water.		
OPS	G.GEN.495 Cockpit voice recorder - Aeroplanes and Helicopters		
AER	OPLANES		
(a)	Aeroplanes with a maximum certificated take-off mass exceeding 27 000 kg and first issued with an individual certificate of airworthiness after 31 December 1986, shall be equipped with a Cockpit Voice Recorder (CVR).		This is a new requirement that industry may have to take into account and retrofit should be avoided. Two years' notice seems to be reasonable for industry. Since the OPS Regulation should enter into force on 8 April 2012, it is proposed to mandate audio recording for aeroplanes with an individual certificate of airworthiness issued on or after 1 January 2016 to be consistent with requirements for CAT aeroplanes and flight data recording. ICAO Annex 6 part II states:

A: Rule	B: Summary of comments	C: Reason for change, remarks
		"3.6.3.2.1.1 All turbine-engined aeroplanes for which a type certificate is first issued on or after 1 January 2016 and required to be operated by more than one pilot shall be equipped with either a CVR or a CARS."
		"3.6.3.2.1.3 All aeroplanes of a maximum certificated take-off mass of over 27 000 kg for which the individual certificate of airworthiness is first issued on or after 1 January 1987 shall be equipped with a CVR".
(b) The CVR shall be capable of retaining data recorded during at leas	t:	
 (1) the preceding two hours, for aeroplanes when the individucertificate of airworthiness has been issued after 1 Janua 2003; or 		Regarding CVR recording duration, ICAO Annex 6 Part II Section 3 states:
(2) the preceding 30 minutes, for all other aeroplanes.		"3.6.3.2.3.1 All CVRs shall be capable of retaining the information recorded during at least the last 30 minutes of their operation."
		And:
		"3.6.3.2.3.2 From 1 January 2016, all CVRs shall be capable

A: Rule	B: Summary of comments	C: Reason for change, remarks
		of retaining the information recorded during at least the last two hours of their operation."
		Since the CVR will only be mandated for aeroplanes with an individual certificate of airworthiness issued in or after 2016, a recording duration of 2 hours can be mandated.
HELICOPTERS		
(c) Helicopters with a maximum certificated take-off mass exceed	-	Regulatory justification:
000 kg and first issued with an individual certificate of airworth after 31 December 1986, shall be equipped with a CVR.	iness	ICAO Annex 6 Part III states in Section 3:
		"4.7.2.1.1 All helicopters of a maximum certificated take-off mass of over 7 000 kg for which the individual certificate of airworthiness is first issued on or after 1 January 1987 shall be equipped with a CVR,"
		"4.7.2.1.3 All helicopters of a maximum certificated take-off mass of over 7 000 kg for which the individual certificate of airworthiness was first issued before 1 January 1987 shall be

A: Rule	B: Summary of comments	C: Reason for change, remarks
		equipped with a CVR."
		However, this is a new requirement that industry may have to take into account and retrofit should be avoided. Two years' notice seems to be reasonable for industry. Since the OPS Regulation should enter into force on 8 April 2012, it is proposed to mandate audio recording for aeroplanes with an individual certificate of airworthiness issued on or after 1 January 2016 to be consistent with requirements for CAT helicopters and flight data recording.
(d) The CVR shall be capable of retaining data recorded during at least:	1. The times required for retaining data recorded by a CVR are incorrect and should reflect those in ICAO Annex 6 Part III Section III.	 Safety justification: Annex 6 Part III, Section 3, states : "4.7.2.3.1 A CVR shall be capable of retaining the information recorded during at least the last 30 minutes of its operation.
		4.7.2.3.2 From 1 January 2016, all helicopters required to be equipped with a CVR shall be equipped with a CVR capable of

A: Rule	B: Summary of comments	C: Reason for change, remarks
		retaining the information recorded during the last two hours of its operation."
		The extension to 2 hours of recording duration for the CVR is of significant importance for accident investigations. Since the carriage of a CVR will only be applicable for helicopters with an individual certificate of airworthiness issued in or after 2016, this should not pose a problem. In addition, such a duration will be consistent with the required durations in CAT.IDE.H.
(1) the preceding one hour, for helicopters when first issued with an individual certificate of airworthiness after 31 July 1999; or		
(2) the preceding 30 minutes, for all other helicopters.		
AEROPLANES AND HELICOPTERS		
(d) The CVR shall start to record automatically prior to the aircraft moving under its own power and shall continue to record until the termination of the flight when the aircraft is no longer capable of moving under its own power.		Regulatory justification: This is now a standard in ICAO Annex 6 Part III:

A: F	Rule	B: Summary of comments	C: Reason for change, remarks
			"3.1.1 The CVR shall start to record prior to the helicopter moving under its own power and record continuously until the termination of the flight when the helicopter is no longer capable of moving under its own power. In addition, depending on the availability of electrical power, the CVR shall start to record as early as possible during the cockpit checks prior to engine start at the beginning of the flight until the cockpit checks immediately following engine shutdown at the end of the flight."
(f)	The CVR shall have a device to assist in locating it in water.		
OPS	6.GEN.500 Data link recording - Aeroplanes and Helicopters		
(a)	Aeroplanes and helicopters first issued with an individual certificate of airworthiness on or after 8 April 2012, which have the capability to operate data link communications and are required to be equipped with a cockpit voice recorder, shall be equipped with a means of recording the following, where applicable:	1. Time should be left for operators to comply. April 2012 is not achievable. In addition it is not consistent with Commission Regulation EC 29/2009.	 Regulatory and safety justifications: Commission Regulation (EC) No 29/2009 states in Article 3 that: "operators shall ensure that aircraft operating flights referred to in Article 1(3) () have the

Part-NCC | CRST 30 Aug 2011 A: Rule **B:** Summary of comments **C:** Reason for change, remarks capability to operate the data link services". It will be applicable: - as from 7 February 2013 for those aircraft issued with an individual certificate of airworthiness after 01 January 2011, - as from 5 February 2015 for other aircraft. In order to avoid the loss of essential communications that used to be transmitted via VHF, it is proposed to mandate recording of data link communication consistently with voice recording. ICAO Annex 6 Part II Section 3 states: "3.6.3.3.1.1 All aeroplanes for which the individual certificate of airworthiness is first issued on or after 1 January 2016, which utilize any of the data link communications applications (...) and are required to carry a cockpit voice recorder (CVR), shall record on a flight recorder, all data link communications messages".

A: Rule	B: Summary of comments	C: Reason for change, remarks
		In addition, 3.6.3.3.1.2 states a similar requirement for "All aeroplanes which are modified on or after 1 January 2016" to install data link communication applications.
		2/ (b) to (e) make reference to a recorder. In addition, NPA 2009- 11 contains a draft ETSO for stand-alone data link recorders. In order to make clear that the recording function should be performed by any crash- protected recorder, be it a CVR, an FDR or a standalone data link recorder, it is proposed to replace "means of recording" by "recorder".
 data link communication messages related to air tr services communications to and from the aircraft; 	raffic	
(2) information that enables correlation to any associated reconstructed related to data link communications and stored separation from the aircraft;		Safety justification: This requirement should be kept in order to enable investigators to correlate airborne data link recordings with data link communications recorded on the ground by ATM systems.

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	(3) information on the time and priority of data link communications messages, taking into account the system's architecture.		
(b)	The recorder shall use a digital method of recording and storing data and information and a method for retrieving that data. The recording method shall be such as to allow the data to match the data recorded on the ground.		Safety justification: This requirement should be kept in order to enable investigators to correlate airborne data link recordings with data link communications recorded on the ground by ATM systems.
(c)	The recorder shall be capable of retaining data recorded during at least the preceding two hours of operation.		Regulatory justification: ICAO Annex 6 Part II, Section 3 states: "3.6.3.3.2 Duration The minimum recording duration shall be equal to the duration of the CVR".
(d)	The recorder shall have a device to assist in locating it in water.		
(e)	The recorder shall start to record automatically prior to the aeroplane or helicopter moving under its own power and shall continue to record until the termination of the flight when the aeroplane or helicopter is no longer capable of moving under its own power.		It is consistent with the requirement on audio (CVR) recording to start during the cockpit checks prior to engine start and last until the cockpit checks immediately following

A: Rule **B:** Summary of comments **C:** Reason for change, remarks engine shutdown. A requirement for combination recorders has been added. Regulatory justification: ICAO Annex 6 Part II, Section 3 states: "3.6.3.4.6 Combination recorders Recommendation. – All aeroplanes of a maximum certificated take-off mass over 5 700 kg, required to be equipped with an FDR and a CVR, may alternatively be equipped with two combination recorders (FDR/CVR)." The case of an aeroplane with an MCTOM between 5 700 kg and 27 000 kg (only the FDR is mandatory) has been added here. **OPS.GEN.505** Preservation of FDR and CVR recordings -Moved to NCC.GEN.106 **Aeroplanes and Helicopters** (a) The pilot-in-command shall be responsible for ensuring that during A paragraph on the use and The rule in NCC.GEN.150 takes flight, means installed on board for recording FDR, CVR and data preservation of data link recordings also into account the should be added preservation of data link link data are not:

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A: R	ule		B: Summary of comments	C: Reason for change, remarks
				recordings.
	(1)	disabled;		
	(2)	switched off; or		
	(3)	intentionally erased in the event of an accident or an incident which is subject to mandatory reporting.		
(b)		der to preserve the data recorded for investigating an accident cident which is subject to mandatory reporting:		
	(1)	the CVR may be switched off during flight by the pilot-in- command, if he/she believes that the recorded data would otherwise be erased automatically;	ICAO Annex 6 prohibits the switching off of flight recorders.	Accepted. The IR does not allow that the CVR is switched off. This is also aligned with ICAO Annex 6 Part II, Section 3, which states:
				"3.6.3.4.2.1 Flight recorders shall not be switched off during flight time".
	(2)	recorders shall be de-activated upon completion of the flight; and		
	(3)	the recorders shall not be re-activated without the investigating authority's agreement.		

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A: F	Rule	B: Summary of comments	C: Reason for change, remarks
(c)	The operator shall preserve the original recorded data referred to in (b) for a period of 60 days, unless otherwise directed by the investigating authority.		ICAO Annex 6 Part II, Section 3, paragraph 3.6.3.4.3 states: "3.6.3.4.3 Flight recorder records The pilot-in-command, and/or the owner/operator, shall ensure, to the extent possible, in the event the aeroplane becomes involved in an accident or incident, the preservation of all related flight recorder records, and if necessary the associated flight recorders, and their retention in safe custody pending their disposition as determined in accordance with Annex 13." However, it does not state how long the operator shall preserve the data. It is proposed to keep the current wording in order to remain consistent with CAT.
(d)	Operational checks and evaluations of recordings from the FDR and CVR systems shall be conducted to ensure the continued serviceability of the recorders.	A paragraph on the use and preservation of data link recordings should be added.	Regulatory justification: Considering the advent of data link recording as mentioned in NCC.IDE.A.170, the requirements should also regulate the use and preservation of data link recordings in the same manner

A: Rule	B: Summary of comments	C: Reason for change, remarks
		as is done for CVR and FDR recordings.
		ICAO Annex 6 Part II, Section 3 states:
		"3.6.3.4.4 Continued serviceability
		Operational checks and evaluations of recordings from the flight recorder systems shall be conducted to ensure the continued serviceability of the recorders."
(e) The operator shall make available any flight recorder recording that has been preserved, if so determined by the competent authority.		
OPS.GEN.510 Use of FDR and CVR recordings - Aeroplanes and Helicopters		
Without prejudice to national criminal law:		
(a) Cockpit Voice Recorder (CVR) recordings shall not be used for purposes other than for the investigation of an accident or incident which is subject to mandatory reporting, except with the consent of all crew members concerned; and		Safety justification The protection of CVR recordings against any misuse is extremely important, since it guarantees the protection of flight crew privacy. It is essential for maintaining a "just culture",

A: Rule	B: Summary of comments	C: Reason for change, remarks
		which has proved to be highly beneficial for safety.
		Therefore this provision should be maintained.
(b) Flight Data Recorder (FDR) recordings shall not be used for purposes other than for the investigation of an accident or incident which is subject to mandatory reporting, except when such records are:		Safety justification The protection of FDR recordings against any misuse is extremely important, since it guarantees the protection of flight crew privacy. It is essential for maintaining a "just culture", which has proved to be highly beneficial for safety.
(1) used by the operator for airworthiness or maintenance		Therefore this provision should be maintained.
purposes only;		
(2) de-identified; or		
(3) disclosed under secure procedures.		
OPS.GEN.515 Microphones - Aeroplanes and Helicopters		Moved to NCC.OP

Part-NCC		CRST	30 Aug 2011
A: R	ule	B: Summary of comments	C: Reason for change, remarks
AER	OPLANES		
(a)	Flight crew members on flight deck duty of complex motor-powered aeroplanes shall communicate through boom or throat microphones or equivalent, when flying below the transition level/altitude.		
HELI	COPTERS		
(b)	Flight crew members on flight deck duty of complex motor-powered helicopters and helicopters used in commercial operations shall wear a headset with boom microphone or equivalent and use it as the primary device for all communications.		
OPS	GEN.520 Flight crew interphone system		
(a)	Aircraft operated by more than one flight crew member shall be equipped with a flight crew interphone system, including headsets and microphones for use by all flight crew members.	 This rule inadvertently makes intercom systems necessary in two- seater training sailplanes and powered sailplanes. They are not necessary in those environments due low noise levels. Interphone may be needed in single crew powered aircraft, regardless of use if cockpit noise requires it. Passenger(s) or student(s) use interphone systems as required to 	1.&2.&3. The rule has been modified in line with JAR-OPS 0.685. Comments related to small aircraft will be taken into account in NCO.

A: R	Rule	B: Summary of comments	C: Reason for change, remarks
(b)	When a radio communication system is required, and in addition to (a), the aircraft shall be equipped with a transmit button on the flight controls at each pilot's station.	 achieve the flight objectives. 3. Crews of aircraft certified with overhead loudspeaker(s) shall not be forced to use headsets. However, they may use additional headsets if available. 1. Some microlight aircraft used for training use handheld radios and are not designed for two-pilot operation. They should also be exempted. 	1. Microlight aircraft fall under Annex II of the Basic Regulation and are not covered by this regulation.
OPS	G.GEN.525 Communication equipment		
(a)	Aircraft operated under Visual Flight Rules (VFR) as a controlled flight, under Instrument Flight Rules (IFR) or at night shall be provided with radio communication equipment. Such equipment shall be capable of conducting two-way communication with those aeronautical stations and on those frequencies prescribed by the appropriate authority.	 Communication equipment is not required for VFR flight when not controlled: clarification required. There is no justification to require two-way radio communication for VFR night flights. If the VFR night flight takes place in areas where communication is not required 	 The reference to "controlled flight" is replaced by "when required by airspace requirements". The requirement for communication equipment at night is in line with ICAO Annex 6.
(b)	The radio communication equipment required in (a) shall provide for communication on the aeronautical emergency frequency.		1. additional requirements in line
			with ICAO Annex 6 Part II, 3.7.1 applicable to NCC have been

A: Rule	B: Summary of comments	C: Reason for change, remarks
		added:
		 capability of conducting two- way communication for aerodrome control purposes;
		 capability of receiving meteorological information at any time during flight.
OPS.GEN.530 Pressure-altitude-reporting transponder		
Aircraft shall be equipped with a pressure-altitude-reporting Secondary Surveillance Radar (SSR) transponder when required by the airspace requirements.	1. A pressure-altitude SSR transponder shall be required for all aeroplanes and helicopters to enable ACAS-equipped aircraft to generate resolution advisories.	13. The requirement for the SSR transponder is in line with ICAO and ensures ACAS capability regardless of the transponder-carriage rules
	2. It should be added that the air traffic provider can grant exemptions depending on the traffic situation for aircraft not fitted with such a transponder.	applying to the airspace in which either aircraft is flying.
	3. Now a pressure-altitude-reporting transponder is prescribed for all flights, irrespective of the conditions of flight, when required by the airspace requirements - OPS.GEN.525.	
OPS.GEN.535 Navigation equipment		

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A: R	tule	B: Summary of comments	C: Reason for change, remarks
(a)	An aircraft shall be provided with navigation equipment which will enable it to proceed in accordance with:	 This unconditional requirement on navigation equipment is not consistent with the communication equipment requirement. The rule needs to be more specific in terms of required navigation equipment, either here, or in the AMC. For an IFR flight, it requires equipment as an aid to landing, but has no navigation requirements. 	 The requirement is alleviated for flight under VFR accomplished by visual reference to landmarks. & 3. Text is in line with ICAO. Equipment specifications given for CAT were considered not suitable for NCC.
	(1) the flight plan; and		
	(2) the applicable airspace requirements.		
(b)	The number of navigation equipment shall be such that in the event of failure of one item of equipment during the flight, the remaining equipment will enable the aircraft to comply with (a).	1. It should not be a requirement that the aircraft can always continue according to the original plan in case of equipment malfunction. It should be sufficient that the pilot has planned for an alternative course of action since otherwise equipment duplication will often be required when this is not necessary.	1. The wording is modified.

A: Rule		B: Summary of comments	C: Reason for change, remarks
pro poir sha it i	Instrument Flight Rules (IFR) flights, an aircraft shall be ovided with navigation equipment that provides guidance to a fint from which a visual landing can be performed. This equipment all be capable of providing guidance for each aerodrome at which is intended to land in IFR and for any designated alternate rodromes.	1. This rule prevents aircraft conducting a part of their flight in IMC, even if there is no intention to perform an instrument approach, without such equipment. It is proposed to return to ICAO text.	1. Accepted.
Complex motor-powered aeroplanes		1. It is not clear why this only applies to aeroplanes. The 'A' should be removed from the title.	1. There is no ICAO requirement for other aircraft.
		This shall not be limited to complex motor-powered aeroplanes as well.	2. This is proposed to be kept under NCC.IDE.
		 Not an equipment specification - include these requirements in Part OR, or in OPS Section II operational procedures. 	3. Not accepted. This cannot be specified at IR level. It's up to
		3. AIRAC Cycles for electronic navigation data and hard copy prints may be different (e.g. issue dates.) This creates in practice SAFA findings in some occasions. Therefore it is recommended to define which of the sources are the primary ones for the intended operation.	the operator, depending on the type of data systems and providers he/she uses, to define in the operations manual what is the primary source and criteria for timely up-dating of navigation data.
		4. How can an individual owner of an airplane comply with this paragraph?	4. See comment on paragraph (c).
		This is not a proportional rule.	

A: Rule		B: Summary of comments	C: Reason for change, remarks
(a)	An operator shall only use electronic navigation data products which support a navigation application that meets standards of integrity which are adequate for the intended use of the data.		
(b)	When the electronic navigation data products support a navigation application needed for an operation requiring a specific approval as per OPS.SPA, the operator shall demonstrate to the competent authority that the process applied and the delivered products meet standards of integrity which are adequate for the intended use of the data.		
(c)	An operator shall continuously monitor both the process and the products,	1. Many operators outsource electronic navigation or chart database management to third party providers. Directly monitoring the process of third party providers is not efficient and should be done by established compliance monitoring procedures and practices.	1. Proposed text accepted.
(d)	An operator shall ensure the timely distribution and insertion of current and unaltered electronic navigation data to all aircraft that require it.		
OPS.GEN.545 Cabin Crew Seats			
	n seat for the minimum required cabin crew member shall be forward earward facing within 15° of the longitudinal axis of the aircraft.	1. Add "during take-off and landing".	1. The requirement is in line with ICAO.

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A: R	tule	B: Summary of comments	C: Reason for change, remarks
OPS	GGEN.550 Minimum equipment for flight		
(a)	A flight of a complex motor-powered aircraft or an aircraft involved in commercial operations shall not be commenced when any of the aircraft instruments, items of equipment or functions are inoperative, except if:		1. The intent of this rule is to implement the Essential Requirement in 2.a.3 (iii) of Annex IV to the Basic Regulation, where it is required that :
			"the pilot in command must be satisfied that:
			(i) the aircraft is airworthy as specified in point 6;
			[]
			(iii) instruments and equipment as specified in point 5 required for the execution of that flight are installed in the aircraft and are operative, unless waived by the applicable Minimum Equipment List (MEL) or equivalent document;
			[]″
			The point 5 mentioned above is the paragraph providing the Essential Requirements for instruments, data and equipment.

A: R	ule	B: Summary of comments	C: Reason for change, remarks
			Therefore the purpose of this proposed rule is to clarify which are the legal means of releasing an aircraft with inoperative instruments, items of equipment or functions that are required by NCC.IDE.
	(1) such the aircraft instrument, item of equipment or function is part of the operator minimum equipment list (MEL); or		
	(2) the aircraft has been subject to a permit to fly issued by the competent authority or organisations approved in accordance with Part-21; or		
	(3) the aircraft instrument, item of equipment or function is not required for the safe operation of the aircraft.		
(b)	The MEL shall not deviate from Airworthiness Directives or Safety Directives issued or adopted by the Agency when these Directives exclude the MEL alleviation.		
(c)	Any instrument or item of equipment that has been installed in an aircraft and becomes inoperative shall not be removed thereof, unless:		
	(1) it is replaced by an operative instrument or equipment; or		

A: Rule		B: Summary of comments	C: Reason for change, remarks
(2)	it is specifically permitted by the MEL; or		
(3)	the aircraft has been subject to a permit to fly issued by the competent authority or approved organisations.		
	Section V - Manuals, Logs and Records		Section deleted and incorporated in NCC.GEN.
OPS.GEN. aircraft	600 Documents and information to be carried on all		Relevant content moved to NCC.GEN.145.
(a) On flight:	any aircraft, the following documents shall be carried on each		
(1)	the Aircraft Flight Manual or equivalent documents;		
(2)	the Certificate of Airworthiness;		
(3)	the Certificate of Registration;		
(4)	the original or copy of the Noise Certificate, if applicable;		
(5)	the original or copy of the third party liability Insurance Certificate;		
(6)	the journey log book for the aircraft;		
(7)	current and suitable aeronautical charts for the route of the proposed flight and all routes along which it is reasonable to expect that the flight may be diverted;		
(8)	procedures and visual signals for use by intercepting and intercepted aircraft which shall be easily accessible to the flight crew; and		

			•		
A: R	A: Rule		B: Summary of comments	C: Reason for change, remarks	
	(9)	any other documentation which may be pertinent to the flight or is required by the States concerned with the flight.			
(b)	motor the s naution docur	ion-commercial flights with sailplanes, other than complex r-powered aeroplanes and helicopters taking off and landing at same aerodrome/operating site and remaining within 50 cal miles (nm) of that aerodrome/operating site, the ments and information referred to in OPS.GEN.600(a)(1) - (6) be retained at the aerodrome/operating site.			
(c)		on-commercial flights with balloons, the documents referred to $S.GEN.600(a)(1) - (6)$ may be carried in the retrieve vehicle.			
OPS.GEN.605 Documents and information to be carried on non- commercial flights with complex motor-powered aircraft and aircraft used in commercial operations		al flights with complex motor-powered aircraft and		Relevant content moved to NCC.GEN.145.	
(a)	powe used	ddition to OPS.GEN.600(a), on flights with complex motor- red aircraft used in non-commercial operations and aircraft in commercial operations, the following documents and nation shall be carried on each flight:			
	(1)	the Aircraft Radio Licence;			
	(2)	the Aircraft Technical Log in accordance with Part-M;			
	(3)	details of the filed Air Traffic Service (ATS) flight plan, if applicable;			
	(4)	information concerning search and rescue services for the area of the intended flight;			
	(5)	a certified true copy of the operator certificate if the aircraft is used in commercial operations and a copy of the			

A: R	ule		B: Summary of comments	C: Reason for change, remarks
		declaration if the aircraft is used in non-commercial operations;		
	(6)	portions of the operations manual relevant to the duties of the crew;		
	(7)	the Minimum Equipment List (MEL) or equivalent document; and		
	(8)	the following operational flight information:		
		(i) the operational flight plan, if applicable;		
		(ii) appropriate notice to airmen (NOTAM)/aeronautical information service (AIS) briefing documentation;		
		(iii) appropriate meteorological information;		
		 (iv) notification of special categories of passenger and special loads including dangerous goods, if applicable; and 		
		(v) mass and balance documentation; if applicable.		
b)	Notw	vithstanding paragraphs OPS.GEN.600(a) and OPS.GEN.605(a),		
	(1)	on commercial flights other than commercial air transport remaining within 50 nm of the aerodrome/operating site of departure and returning to the aerodrome/operating site of departure on the same day;		
	(2)	on commercial flights with sailplanes taking off and landing at the same aerodrome/operating site and remaining within 50 nm of that aerodrome/operating site; or		
	(3)	on commercial air transport flights by day and over routes navigated by reference to visual landmarks, with:		

A: Rule			B: Summary of comments	C: Reason for change, remarks
	(i)	other than complex motor-powered aeroplanes taking off and landing at the same aerodrome/operating site; or		
	(ii)	other than complex motor-powered helicopters and helicopters having a maximum passenger seating configuration of 9 or less engaged in flight operations conducted within a local area specified in the operations manual;		
	OPS.GEN.6	ments and information referred to in paragraphs $500(a)(1) - (6)$ and OPS.GEN.605(a)(1), (5) and (8)(ii)- e retained at the aerodrome/operating site.		
(c)	paragraph	ercial flights with balloons, the documents referred to in OPS.GEN.600(a)(1) - (6) and OPS.GEN.605(a)(1), (5) -(iii) may be carried in the retrieve vehicle.		
OPS	GEN.610 J	lourney log book		Covered in ORO.MLR.
		e aircraft, its crew and each journey shall be retained for ries of flights in the form of a journey log book.		
OPS	GEN.615 I	Production of documentation and records		Relevant content moved to NCC.GEN.145.
bein	g requested	mand shall make available within a reasonable time of to do so by the competent authority, the documentation arried on board.		
		Section VI – Security		Section deleted and incorporated

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A: Rule	B: Summary of comments	C: Reason for change, remarks
		in Subpart A.
OPS.GEN.700 Disruptive Passenger Behaviour		Text not transposed - already covered in the ER and by Regulation (EC) No 300/2008.
When deemed necessary, the pilot-in-command shall take appropriate measures to reduce the risk to flight safety emanating from potentially disruptive passengers hindering crew members from performing their duties or not complying with crew member instructions.		
OPS.GEN.705 Reporting acts of unlawful interference		Moved to NCC.GEN.106(e).
Following an act of unlawful interference on board an aircraft, the pilot- in-command or, if unable, the operator shall submit, without delay, a report to the competent authority in the State of the operator in compliance with its national civil aviation security programme, and shall inform the designated local authority.		

Part-NCC – AMC/GM

A: F	A: Rule		B: Summary of comments	C: Reason for change, remarks
	Sul	opart A – General operating and fight rules - AMC/GM Section I - General requirements		
GM	OPS.	GEN.015 Pilot-in-command responsibilities and authority		
GEN	NERAL			
216 com and	In accordance with paragraph 1.c. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), the pilot-in- command must be responsible for the operation and safety of the aircraft and for the safety of all crew members, passengers and cargo on board. This should be understood as including the following:		Clarification requested of the responsibility between boarding of crew and first movement of A/C and taxiing, and the responsibility starting when "entering the aircraft" or "after closing doors";	Accepted. The inserted text clarifies the moment from and until when the PIC takes responsibility
1.	 The safety of all crew members, passengers and cargo on board, as soon as he/she arrives on board, until he/she leaves the aircraft at the end of the flight; 			
2.	The a. b. c.	operation and safety of the aircraft: from the moment it is first ready to move for the purpose of flight until the moment it comes to rest at the end of the flight and the engine(s) used as primary propulsion unit(s) is/are shut down, for aeroplanes; when the rotors are turning, for helicopters; from the moment the launch procedure is started until the	 MS: request re-alignment with EU- OPS 1.085 and edit 2.a. " move, for the purpose of flight under its own power, for the purpose of ground repositioning or taxiing prior to take off"; INDIV: request to edit 2.b. to 	 Not accepted. The comment should have been addressed to the rule. 2.a : "to move" includes taxiing or ground positioning. Accepted. It clarifies the most critical time of the move of the helicopter. The same was

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A: Rule	·	B: Summary of comments	C: Reason for change, remarks
d.	aircraft comes to rest at the end of the flight, for sailplanes; and from the moment the inflating of the envelope is started until the envelope is deflated, for balloons.	"when the rotors start turning for the purpose of flight until the rotors come to a complete stop after flight";	executed for aeroplanes above.
AMC1 C authori	OPS.GEN.015(a)(5) Pilot-in-command responsibilities and ity		Text not transposed - the IR has not been transposed.
ADMISS	TION TO THE COCKPIT/PILOT COMPARTMENT		
1. ir compart operatio 2. a	t-in-command should ensure that: n the interests of safety, admission to the cockpit/pilot ment does not cause distraction and/or interfere with the flight on; and all persons carried in the cockpit/pilot compartment are made with the relevant restrictions and safety procedures.		
AMC2 C authori	OPS.GEN.015(a)(5) Pilot-in-command responsibilities and ity		Text not transposed - beyond the scope of Part-NCC.
ADMISS TRANSP	ION TO COCKPIT/PILOT COMPARTMENT – COMMERCIAL AIR ORT		
	hly the following persons may be admitted to or carried in the ckpit/pilot compartment: An operating crew member; A representative of the competent authority, if it is required for the performance of his/her official duties;		

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c. A person authorised by the operator;d. Passengers, in the case of balloons with no s	enarate nilot		
compartment.			
 In the case of other than complex motor-powered admitted to or carried in the cockpit/pilot comp carried in a pilot seat. 			
GM OPS.GEN.015(b) Pilot-in-command responsibilit authority	ies and		
AUTHORITY TO REFUSE CARRIAGE OR DISEMBARK			
This may include:	IS (GA): Request to insert a definition	Text amended to refer to "passengers who have special needs that cannot be provided on	
 special categories of passengers; or persons that appear to be under the influence of all 	or a list of "special categories of passengers";		
 persons that appear to be under the influence of alc AMC OPS.GEN.015(c) Pilot-in-command responsibilities 		the aircraft".	
authority REPORTING OF HAZARDOUS FLIGHT CONDITIONS			
1. These reports should include any detail which ma the safety of other aircraft.	y be pertinent to		
 Such reports should be made whenever any conditions are encountered or observed: a. severe turbulence; 	of the following Clarify that the points under (2) also have to be reported to air traffic services;	That is the base requirement in NCC.GEN.106(c) that the AMC supports.	

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A: Rule			B: Summary of comments	C: Reason for change, remarks
	b.	severe icing;		
	с.	severe mountain wave;		
	d.	thunderstorms, with or without hail, which are obscured, embedded, widespread or in squall lines;		
	e.	heavy dust storm or heavy sandstorm;		
	f.	volcanic ash cloud;		
	g.	unusual and/or increasing volcanic activity or a volcanic eruption.		
3.	shear, comm other	other meteorological conditions not listed above, e.g. wind , are encountered which, in the opinion of the pilot-in- and, may affect the safety or markedly affect the efficiency of aircraft operations, the pilot-in-command should advise the priate air traffic services unit as soon as practicable.		
AMC OPS.GEN.015(d) Pilot-in-command responsibilities and authority		GEN.015(d) Pilot-in-command responsibilities and		
MITI	MITIGATING MEASURES		MS: request to redraft and align with comments made to OPS.GEN.015 (d) and state here that the GM applies "in respect of fatigue only";	Text was changed accordingly.
The use of additional crew members and controlled rest during flight are considered to be adequate mitigating measures.				
	GM OPS.GEN.015(d) Pilot-in-command responsibilities and authority			

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A: Rule		B: Summary of comments	C: Reason for change, remarks
MITIGATING MEASURES - CONTROLLED REST		It should be clarified that the mentioned mitigation measure only refers to fatigue;	Accepted. Subtitle amended accordingly.
1.	This Guidance Material (GM) addresses controlled rest taken by the minimum certificated flight crew. It is not related to planned in flight rest by members of an augmented crew.		
2.	Although flight crew members should stay alert at all times during flight, unexpected fatigue can occur as a result of sleep disturbance and circadian disruption. To cover for this unexpected fatigue, and to regain a high level of alertness, a controlled rest procedure in the cockpit, organised by the pilot-in-command may be used, if workload permits. 'Controlled rest' means a period of time 'off task' that may include actual sleep. The use of controlled rest has been shown to significantly increase the levels of alertness during the later phases of flight, particularly after the top of descent, and is considered to be good use of Crew Resource Management (CRM) principles. Controlled rest should be used in conjunction with other on board fatigue management countermeasures such as physical exercise, bright cockpit illumination at appropriate times, balanced eating and drinking, and intellectual activity.		The term "cockpit" has been replaced (in general) with "flight crew compartment" and is in line with the BR.
3.	Controlled rest taken in this way should not be considered to be part of a rest period for the purposes of calculating flight time limitations, nor used to justify any duty period. Controlled rest may be used to manage both sudden unexpected fatigue and fatigue which is expected to become more severe during higher workload periods later in the flight. Controlled rest is not related to fatigue management which is planned before flight.		

		Part-NCC	CRST	30 Aug 2011
A: F	Rule		B: Summary of comments	C: Reason for change, remarks
4.	needs	olled rest periods should be agreed according to individual s and the accepted principles of CRM; where the involvement of abin crew is required, consideration should be given to their oad.		
5.		applying controlled rest procedures, the pilot-in-command d ensure that:		
	a.	the other flight crew member/s is/are adequately briefed to carry out the duties of the resting flight crew member;		
	b.	one flight crew member is fully able to exercise control of the aircraft at all times; and		
	c.	any system intervention which would normally require a cross-check according to multi-crew principles is avoided until the resting flight crew member resumes his/her duties.		
6.	Contr	olled rest procedures should satisfy the following criteria:		
	a.	Only one flight crew member at a time should take rest at his/her station; the harness should be used and the seat positioned to minimise unintentional interference with the controls;		
	b.	The rest period should be no longer than 45 minutes (in order to limit any actual sleep to approximately 30 minutes) to limit deep sleep and associated long recovery time (sleep inertia);		
	C.	After this 45-minute period, there should be a recovery period of 20 minutes during which sole control of the aircraft should not be entrusted to the flight crew member during		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
	his/her recovery period;		
d.	In the case of two-crew operations, means should be established to ensure that the non-resting flight crew member remains alert. This may include:		Text added to improve clarity.
	i. appropriate alarm systems;		
	ii. on board systems to monitor flight crew activity; and		
	iii. frequent cabin crew checks. In this case, the pilot-in- command should inform the senior cabin crew member of the intention of the flight crew member to take controlled rest, and of the time of the end of that rest; frequent contact should be established between the non-resting flight crew member and the cabin crew by communication means, and the cabin crew should check that the resting flight crew member is alert at the end of the period;		
e.	There should be a minimum of 20 minutes between two subsequent controlled rest periods in order to overcome the effects of sleep inertia and allow for adequate briefing;		
f.	If necessary, a flight crew member may take more than one rest period, if time permits, on longer sectors, subject to the restrictions above;		
g.	Controlled rest periods should terminate at least 30 minutes before the top of descent.		

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A: R	Rule		B: Summary of comments	C: Reason for change, remarks	
	C OPS. hority	GEN.015(e)(3) Pilot-in-command responsibilities and		Text not transposed - beyond the scope of Part-NCC.	
PRO	TECTI	VE CLOTHING – BALLOONS			
Prot	ective	clothing should include:			
1.	long	g sleeves and trousers made out of natural fibres;			
2.	sto	ut footwear; and			
3.	glov	ves.			
ΑΜΟ	C1 OP	S.GEN.020(a) Crew responsibilities		Text not transposed - to be transposed in ORO.FTL.	
FAT	IGUE R	RISK MANAGEMENT			
1. In accordance with paragraph 7.f. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), no crew member must allow their task achievement/decision making to deteriorate to the extent that flight safety is endangered because of the effects of fatigue. Before commencing a flight, a crew member should take the following factors into account, as relevant, to assess the risk of personal fatigue:		2008 (Essential requirements for air operations), no crew ber must allow their task achievement/decision making to riorate to the extent that flight safety is endangered because of effects of fatigue. Before commencing a flight, a crew member Id take the following factors into account, as relevant, to assess			
	a.	Fatigue accumulated over a period of time and time since last rest;			
	b.	Sleep deprivation (i.e. arising from interruption of the normal sleep/wake cycle);			
	c.	Activity carried out prior to a flight, including physical			

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		activity;		
	d.	Number or series of flights (sectors) planned or accomplished;		
	e.	Duration of the planned flight activity;		
	f.	The time of day/night at which the flight is scheduled to begin and time flown at night;		
	g.	The pattern of working and sleeping relative to the circadian rhythm (24-hour physiological cycle);		
	h.	Number and direction of time zones crossed;		
	i.	Operational characteristics of the planned flight (e.g. weather conditions, familiarity with the route/destination, single/multi-pilot operation, service limitations of the aircraft, crew complement, such as reduced number of crew members, etc.).		
2.	facilit	members should make optimum use of the opportunities and ties for rest provided and plan and use their rest periods opriately.		
АМС	2 OPS	5.GEN.020(a) Crew responsibilities		
		G OF ANY OCCURRENCE RELATED TO THE SAFETY OF THE AND ITS OCCUPANTS		
		a crew member makes use of the applicable reporting systems, he report should be communicated to the pilot-in-command.		

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GM OPS.GEN.020(a) Crew responsibilities			
 GENERAL In accordance with paragraph 7.g. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), a crew member must not perform allocated duties on board an aircraft when under the influence of psychoactive substances or alcohol or when unfit due to injury, fatigue, medication, sickness or other similar causes. This should be understood as including the following: 1. Effects of deep water diving and blood donation, and allowing for a certain time period between these activities and returning to flying; 	MS: Extra guidance requested for crews to include a minimum length	The UK Diving Medical Advisory Committee http://www.dmac-	
Certain time period between these activities and returning to hying,	during which flying should not occur after deep-water-diving and blood donation (24 hrs);	diving.org/guidance/DMAC07.pdf suggests that in certain circumstances the period between diving and flying should be 48 hrs. The UK CAA http://www.caa.co.uk/docs/33/CA P686.pdf suggests 48 hrs for diving and 24 hrs for blood donation.	
2. Without prejudice to more restrictive national regulations, the consumption of alcohol while on duty or less than 8 hours prior to the commencement of duties, and commencing a flight duty period with a blood alcohol level in excess of 0.2 per mille.			

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EXE	C OPS.GEN.030 Transport of dangerous goods MPTION AND APPROVAL PROCEDURES OF THE TECHNICAL TRUCTIONS	Amendment is required to clarify that the title does not mean 'approval of the Technical Instructions'	Title amended to refer to the 'provisions of the Technical Instructions'. This entire section has become Guidance Material (GM1- NCC.GEN.155.	
1.	The Technical Instructions provide that in certain circumstances dangerous goods, which are normally forbidden on an aircraft, may be carried. These circumstances include cases of extreme urgency or when other forms of transport are inappropriate or when full compliance with the prescribed requirements is contrary to the public interest. In these circumstances all the States concerned may grant exemptions from the provisions of the Technical Instructions provided that every effort is made to achieve an overall level of safety which is equivalent to that provided by the Technical Instructions. Although exemptions are most likely to be granted for the carriage of dangerous goods which are not permitted in normal circumstances, they may also be granted in other circumstances, such as when the packaging to be used is not provided for by the appropriate packing method or the quantity in the packaging is greater than that permitted. The Instructions also make provision for some dangerous goods to be carried when an approval has been granted only by the State of Origin, providing specific conditions which are laid down in the Technical Instructions are met.			
2.	The States concerned are those of origin, transit, overflight and destination of the consignment and that of the operator.			

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3.	are g inten again opera	Technical Instructions provide that exemptions and approvals granted by the "appropriate national authority", which is ded to be the authority responsible for the particular aspect ast which the exemption or approval is being sought. The ator should ensure all relevant conditions on an exemption or oval are met.		
4.		exemption or approval referred to above is in addition to the oval required by OPS.SPA.001.DG.		
	C ops. Ieral	GEN.030(b) Transport of dangerous goods		AMC not transposed – it is already mentioned in the ICAO Technical Instructions.
1.		erous goods required to be on board an aircraft in accordance airworthiness and operational requirements are those which are		
	a.	the airworthiness of the aircraft;		
	b. the safe operation of the aircraft; or			
	c. the health of passengers or crew.			
		Such dangerous goods include, but are not limited to:		
		i. batteries;		

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ii. fire extinguishers;		
iii. first-aid kits;		
iv. insecticides/air fresheners;		
v. life-saving appliances; and		
vi. portable oxygen supplies.		
2. Articles and substances for specialised purposes are those connected with certain specialised aerial tasks or medical service operations. They should be carried in accordance with procedures and instructions contained in the operations manual and in accordance with officially recognised standards or national legislation. These procedures should be based on the ICAO Technical Instructions where possible but, as a minimum, they should mitigate the risks associated with their carriage.		
AMC OPS.GEN.030(d)(1)Dangerous godos incident and accident		

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-	orting IERAL			
1.	Any type of dangerous goods incident or accident should be reported, irrespective of whether the dangerous goods are contained in cargo, mail, passengers' baggage or crew baggage.	MS: request to delete first sentence as it is already required in the IR;	Not accepted. The first sentence adds elements to the IR: "irrespective of". Text concerning operators' stores included to accommodate an amendment to the Technical Instructions.	
2.	The first report should be dispatched within 72 hours of the event. It may be sent by any means, including e-mail, telephone or fax. This report should include the details that are known at that time, under the headings identified in paragraph 3. If necessary, a subsequent report should be made as soon as possible giving all the details that were not known at the time the first report was sent. If a report has been made verbally, written confirmation should be sent as soon as possible.	MS: It is proposed that 72 hours is too long a period in some circumstances, such as when there has been no contact with other emergency response services. Request to add that the operator should take immediate action where this is required;	The current text does not prohibit earlier reporting and SPA.DG addresses notification to emergency services. In addition, Note 5 of the form states that the form must be sent as soon as possible. Therefore, it is not proposed to amend this text.	
3.	The first and any subsequent report should be as precise as possible and contain such of the following data that are relevant:			
	 Date of the incident or accident or the finding of undeclared or misdeclared dangerous goods; 			
	b. Location, the flight number and flight date;			

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с.	Description of the goods and the reference number of the air waybill, pouch, baggage tag, ticket, etc;			
d.	Proper shipping name (including the technical name, if appropriate) and UN/ID number, when known;			
e.	Class or division and any subsidiary risk;			
f.	Type of packaging, and the packaging specification marking on it;			
g.	Quantity;			
h.	Name and address of the shipper, passenger, etc;			
i.	Any other relevant details;			
j.	Suspected cause of the incident or accident;			
k.	Action taken;			
١.	Any other reporting action taken; and			
m.	Name, title, address and telephone number of the person making the report.			

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4.	Copies of relevant attached to a repo		any pho	otographs taken should be	Amend text to say that any documents and photographs should be attached to "the report" rather than "a report".	Text amended as proposed.
5.		serious incident	t or i	t may also constitute an ncident. The criteria for be met.		
6.	6. Dangerous Goods Reporting Form:				MS: It was proposed that the content of the form has to be mentioned and that it be made clear that the form is an example, otherwise it would appear that other formats, including electronic forms, would not be possible.	Text amended to address the comments.
DA	DANGEROUS GOODS OCCURRENCE REPORT DGOR No:			DGOR No:		
1.	1. Operator:2. Date of Occurrence:3. Local time of occurrence:					
4	4. Flight date: 5. Fligh		light No:			

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6. Departure aerod	lrome:	7. De	stination aerodrome:		
8. Aircraft type:		9. Aiı	craft registration:		
10. Location of occ	urrence:	11. C	rigin of the goods:		
12. Description of t damage, etc.(if necessary c	the occurrence, incl	_			
13. Proper shipping (including the tech	-		14. UN/ID No (when known):		
15.Class/Division (when known):	16. Subsidiary risk(s):	17. Packi ng group :	18 Category (Class 7 only):		
19. Type of packaging:	20.Packaging specification	21. No of	22. Quantity (or transport index, if		

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marking:	packa ges:	applicable):				
23. Reference No of Airway Bill:						
24. Reference No of courier pouc	h, baggage ta	g, or passenger ticket:				
25. Name and address of shipper	, agent, pass	enger, etc.:				
26. Other relevant information (i action taken):	ncluding susp	ected cause, any				
27. Name and title of person man report:	king 28. T	elephone No:				
29. Company:	30. F	Reporters ref:				
31. Address:	32. 5	Signature:				

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	33. Date:			
Description of the occurrence (continue	ation)			

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Note	es for c	ompletion of the form:		
1.	irre	type of dangerous goods occurrence must be reported, spective of whether the dangerous goods are contained in go, mail or baggage.		
2.	relate seriou purpo	ngerous goods accident is an occurrence associated with and ed to the transport of dangerous goods which results in fatal or us injury to a person or major property damage. For this ose serious injury is an injury which is sustained by a person in accident and which:		
	a.	requires hospitalisation for more than 48 hours, commencing within 7 days from the date the injury was received; or		
	b.	results in a fracture of any bones (except simple fractures of fingers, toes or nose); or		
	с.	involves lacerations which cause severe haemorrhage, nerve, muscle or tendon damage; or		
	d.	involves injury to any internal organ; or		
	e.	involves second or third degree burns, or any burns affecting more than 5 % of the body surface; or		
	f.	involves verified exposure to infectious substances or injurious radiation. A dangerous goods accident may also be		

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	an aircraft accident; in which case the normal procedure for reporting of air accidents must be followed.		
3.	A dangerous goods incident is an occurrence, other than a dangerous goods accident, associated with and related to the transport of dangerous goods, not necessarily occurring on board an aircraft, which results in injury to a person, property damage, fire, breakage, spillage, leakage of fluid or radiation or other evidence that the integrity of the packaging has not been maintained. Any occurrence relating to the transport of dangerous goods which seriously jeopardises the aircraft or its occupants is also deemed to constitute a dangerous goods incident.		
4.	This form should also be used to report any occasion when undeclared or misdeclared dangerous goods are discovered in cargo, mail or unaccompanied baggage or when accompanied baggage contains dangerous goods which passengers or crew are not permitted to take on aircraft.		
5.	An initial report, which may be made by any means, must be dispatched within 72 hours of the occurrence, to the competent authority of the State (a) of the operator; and (b) in which the incident occurred, unless exceptional circumstances prevent this. This occurrence report form, duly completed, must be sent as soon as possible, even if all the information is not available.		
6.	Copies of all relevant documents and any photographs should be attached to this report.		
7.	Any further information, or any information not included in the initial report, must be sent as soon as possible to authorities identified in		

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paragraph 5.		
 Providing it is safe to do so, all dangerous goods, pad documents, etc., relating to the occurrence must be retai after the initial report has been sent to the authorities ide paragraph 5 and they have indicated whether or not thes continue to be retained. 	ned until after `identified in paragraph 5', ntified in although no reason was provided.	text since it is important in some
AMC OPS.GEN.030(d)(2) Dangerous goods incident and acc reporting REPORTING OF UNDECLARED OR MISDECLARED GOODS	cident	Text transposed in the above AMC.
An operator should also report the finding of undeclared or misded dangerous goods. The first report should be dispatched within 72 the discovery and include the details that are known at that time. necessary, a subsequent report must be made as soon as possible whatever additional information has been established.	hours of If	
GM OPS.GEN.030 Transport of dangerous goods GENERAL		
1. The requirements to transport dangerous goods by ai accordance with the 2007-2008 Edition of the Technic Instructions for the Safe Transport of Dangerous Goo published by decision of the Council of the Internation	cal reference to an out-of-date edition ods by Air the ICAO Technical Instructions.	1/ The rule now contains a dynamic reference to the Technical Instructions; the term "Technical Instructions" is defined

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Aviation Organization. (ICAO Doc 9284-AN/905.) is irrespective of whether an approval to carry dangerous goods in accordance with OPS.SPA.001.DG is held.	either say 'The requirementsare' or 'The requirementis'.	in Annex I. 2/ Typographical error corrected.
2. Dangerous goods referred to in OPS.GEN.030(b) are those mentioned in Part 1 of the ICAO Technical Instructions.		Deleted due to inclusion of text in GM1-NCC.GEN.155.
3. Baggage separated from its owner includes lost or improperly routed baggage which is carried by an operator for the purpose of returning it to its owner.		Deleted as no longer referred to in NCC.GEN.155.
Section II - Operational procedures		
AMC1 OPS.GEN.100 Ice and other contaminants		Text aligned with Part-CAT.
FLIGHT IN EXPECTED OR ACTUAL ICING CONDITIONS	 1/ MS: in light of the suggested changes to OPS.GEN.100, request to amend title of AMC "flight in expected or actual icing conditions – aeroplane and helicopter"; 2/ IS (BA): move the AMC to a more appropriate place as its relevance goes beyond OPS.GEN.100; 3/ The agreed text of NPA-OPS 50 should be included in the AMC 	 1/ Not accepted, provisions for Part-NCC apply in any case only to H and A; 2/ Not accepted. This is a means to comply with the safety objective established in the IR; 3/ This JAA material is followed up in rulemaking task MDM.045.
1. In accordance with paragraph 2.a.5 of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), in case of	MS: request to delete reference to BR and insert reference to GEN.OPS.445	Accepted. The first sentence is

A: F	Rule		B: Summary of comments	C: Reason for change, remarks
	flight into known or expected icing conditions, the aircraft must be certified, equipped and/or treated to operate safely in such conditions. The procedures to be established by the operator should take account of the design, the equipment, the configuration of the aircraft and the necessary training. For these reasons, different aircraft types operated by the same company may require the development of different procedures. In every case the relevant limitations are those which are defined in the Aircraft Flight Manual (AFM) and other documents produced by the manufacturer.		regarding additional conditions at night;	now part of an IR.
2.	The operator should ensure that the procedures take account of the following:			
	a.	The equipment and instruments which must be serviceable for flight in icing conditions;		
	b.	The limitations on flight in icing conditions for each phase of flight. These limitations may be imposed by the aircraft's de- icing or anti-icing equipment or the necessary performance corrections which have to be made;		
	c.	The criteria the flight crew should use to assess the effect of icing on the performance and/or controllability of the aircraft;		
	d.	The means by which the flight crew detects, by visual cues or the use of the aircraft's ice detection system, that the flight is entering icing conditions; and		
	e.	The action to be taken by the flight crew in a deteriorating situation (which may develop rapidly) resulting in an adverse affect on the performance and/or controllability of the		

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	aircraft, due to:		
	 the failure of the aircraft's anti-icing or de-icing equipment to control a build-up of ice; and/or 		
	ii. ice build-up on unprotected areas.		
3.	Training for dispatch and flight in expected or actual icing conditions. The content of the operations manual should reflect the training, both conversion and recurrent, which flight crew, cabin crew and all other relevant operational personnel require in order to comply with the procedures for dispatch and flight in icing conditions:		
	a. For the flight crew, the training should include:		
	 instruction in how to recognise, from weather reports or forecasts which are available before flight commences or during flight, the risks of encountering icing conditions along the planned route and on how to modify, as necessary, the departure and in-flight routes or profiles; 		
	ii. instruction in the operational and performance limitations or margins;		
	iii. the use of in-flight ice detection, anti-icing and de-icing systems in both normal and abnormal operation; and		
	 instruction in the differing intensities and forms of ice accretion and the consequent action which should be taken. 		

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b.	For the cabin crew, the training should include;		
	 awareness of the conditions likely to produce surface contamination; and 		
	ii. the need to inform the flight crew of significant ice accretion.		
AMC2 O	PS.GEN.100 Ice and other contaminants		This text has been transposed as GM2-NCC.OP.185. The text is aligned with Part-CAT.
DE-ICIN	G/ANTI-ICING	MS: in the light of comments to OPS.GEN.100, and content of this AMC, request to amend title to add "ground procedures", and clarify that it applies only to aeroplanes;	Not accepted. This AMC also applies to H
ma	-icing and/or anti-icing procedures should take into account nufacturer's recommendations, including those that are type-ecific and cover:		
a.	contamination checks, including detection of clear ice and under-wing frost. Limits on the thickness/area of contamination published in the AFM or other manufacturers' documentation should be followed;		
b.	procedures to be followed if de-icing and/or anti-icing procedures are interrupted or unsuccessful;		
с.	post treatment checks;		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
d.	pre take-off checks;		
e.	pre take-off contamination checks;		
f.	the recording of any incidents relating to de-icing and/or anti- icing; and		
g.	the responsibilities of all personnel involved in de-icing and/or anti-icing.		
2. Ope	erator's procedures should ensure that:		
a.	when aircraft surfaces are contaminated by ice, frost, slush or snow, they are de-iced prior to take-off; according to the prevailing conditions. Removal of contaminants may be performed with mechanical tools, fluids (including hot water), infra-red heat or forced air, taking account of aircraft type- specific requirements;		
b.	 account is taken of the wing skin temperature versus Outside Air Temperature (OAT), as this may affect: i. the need to carry out aircraft de-icing and/or anti-icing; and/or ii. the performance of the de-icing/anti-icing fluids; 		
C.	when freezing precipitation occurs or there is a risk of freezing precipitation occurring which would contaminate the surfaces at the time of take-off, aircraft surfaces should be anti-iced. If both de-icing and anti-icing are required, the procedure may be performed in a one or two-step process, depending upon weather conditions, available equipment,		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	available fluids and the desired hold-over time (HoT). One- step de-icing/anti-icing means that de-icing and anti-icing are carried out at the same time, using a mixture of de- icing/anti-icing fluid and water. Two-step de-icing/anti-icing means that de-icing and anti-icing are carried out in two separate steps. The aircraft is first de-iced using heated water only or a heated mixture of de-icing/anti-icing fluid and water. After completion of the de-icing operation a layer of a mixture of de-icing/anti-icing fluid and water, or of de- icing/anti-icing fluid only, is to be sprayed over the aircraft surfaces. The second step will be applied before the first step fluid freezes, typically within three minutes and, if necessary, area by area;		
d.	when an aircraft is anti-iced and a longer HoT is needed/desired, the use of a less diluted Type II or Type IV fluid should be considered;		
e.	all restrictions relative to OAT and fluid application (including, but not necessarily limited to, temperature and pressure) published by the fluid manufacturer and/or aircraft manufacturer, are followed and procedures, limitations and recommendations to prevent the formation of fluid residues are followed;		
f.	during conditions conducive to aircraft icing on the ground or after de-icing and/or anti-icing, an aircraft is not dispatched for departure unless it has been given a contamination check or a post treatment check by a trained and qualified person. This check should cover all treated surfaces of the aircraft and be performed from points offering sufficient accessibility		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	to these parts. To ensure that there is no clear ice on suspect areas, it may be necessary to make a physical check (e.g. tactile);		
g.	the required entry is made in the technical log;		
h.	the pilot-in-command continually monitors the environmental situation after the performed treatment. Prior to take-off he/she performs a pre-take-off check, which is an assessment of whether the applied HoT is still appropriate. This pre-take-off check includes, but is not limited to, factors such as precipitation, wind and OAT;		
i.	if any doubt exists as to whether a deposit may adversely affect the aircraft's performance and/or controllability characteristics, the pilot-in-command should arrange for a pre-take-off contamination check to be performed in order to verify that the aircraft's surfaces are free of contamination. Special methods and/or equipment may be necessary to perform this check, especially at night time or in extremely adverse weather conditions. If this check cannot be performed just before take-off, re-treatment should be applied;		
j.	when re-treatment is necessary, any residue of the previous treatment should be removed and a completely new de- icing/anti-icing treatment should be applied; and		
k.	when a Ground Ice Detection System (GIDS) is used to perform an aircraft surfaces check prior to and/or after a		

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		treatment, the use of GIDS by suitably trained personnel should be part of the procedure.		
3.	Spec	ial operational considerations:		
	a.	When using thickened de-icing/anti-icing fluids, the operator should consider a two-step de-icing/anti-icing procedure, the first step preferably with hot water and/or un-thickened fluids;		
	b.	The use of de-icing/anti-icing fluids should be in accordance with the aircraft manufacturer's documentation. This is particularly true for thickened fluids to assure sufficient flow- off during take-off;		
	c.	The operator should comply with any type-specific operational requirement(s), such as an aircraft mass decrease and/or a take-off speed increase associated with a fluid application;		
	d.	The operator should take into account any flight handling procedures (stick force, rotation speed and rate, take-off speed, aircraft attitude etc.) laid down by the aircraft manufacturer when associated with a fluid application;		
	e.	The limitations or handling procedures resulting from subparagraphs c and/or d above should be part of the flight crew pre-take-off briefing.		

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4.	Comr	nunications:		
	a.	Before aircraft treatment. When the aircraft is to be treated with the flight crew on board, the flight and personnel involved in the operation should confirm the fluid to be used, the extent of treatment required and any aircraft type- specific procedure(s) to be used. Any other information needed to apply the HoT tables should be exchanged;		
	b.	Anti-icing code. The operator's procedures should include an anti-icing code, which indicates the treatment the aircraft has received. This code provides the flight crew with the minimum details necessary to estimate a HoT and confirms that the aircraft is free of contamination;		
	C.	After Treatment. Before reconfiguring or moving the aircraft, the flight crew should receive a confirmation from the personnel involved in the operation that all de-icing and/or anti-icing operations are complete and that all personnel and equipment are clear of the aircraft.		
5.	manu diagr and t the t guide	over protection. The operator should publish in the operations ual, when required, the HoTs in the form of a table or a am, to account for the various types of ground icing conditions the different types and concentrations of fluids used. However, imes of protection shown in these tables are to be used as elines only and are normally used in conjunction with the pre- off check.	IS (BA): request to add " The operator should publish and update in the";	Accepted. Text amended.

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6.	icing flight are in	ng. The operator's initial and recurrent de-icing and/or anti- training programmes (including communication training) for crew and those of its personnel involved in the operation who wolved in de-icing and/or anti-icing should include additional ng if any of the following is introduced: A new method, procedure and/or technique; A new type of fluid and/or equipment; A new type of aircraft.		
7.	icing, the o	acting. When the operator contracts training on de-icing/anti- the operator should ensure that the contractor complies with perator's training/qualification procedures, together with any ic procedures in respect of:		
	a.	de-icing and/or anti-icing methods and procedures;		
	b.	fluids to be used, including precautions for storage and preparation for use;		
	с.	specific aircraft requirements (e.g. no-spray areas, propeller/engine de-icing, APU operation etc.); and		
	d.	checking and communications procedures.		
8.	Specia	al maintenance considerations:		
	a.	General. The operator should take proper account of the possible side-effects of fluid use. Such effects may include, but are not necessarily limited to, dried and/or re-hydrated residues, corrosion and the removal of lubricants;		

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b.	Special considerations regarding residues of dried fluids. The operator should establish procedures to prevent or detect and remove residues of dried fluid. If necessary the operator should establish appropriate inspection intervals based on the recommendations of the airframe manufacturers and/or the operator's own experience:		
	 Dried fluid residues. Dried fluid residues could occur when surfaces have been treated and the aircraft has not subsequently been flown and has not been subject to precipitation. The fluid may then have dried on the surfaces; 		
	 Re-hydrated fluid residues. Repetitive application of thickened de-icing/anti-icing fluids may lead to the subsequent formation/build up of a dried residue in aerodynamically quiet areas, such as cavities and gaps. This residue may re-hydrate if exposed to high humidity conditions, precipitation, washing, etc., and increase to many times its original size/volume. This residue will freeze if exposed to conditions at or below zero degrees Celsius. This may cause moving parts, such as elevators, ailerons, and flap actuating mechanisms to stiffen or jam in-flight. Re-hydrated residues may also form on exterior surfaces, which can reduce lift, increase drag and stall speed. Re-hydrated residues may also collect inside control surfaces to flight controls. Residues may also collect in hidden areas, such as around flight control hinges, pulleys, grommets, 		

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		on cables and in gaps;		
		iii. Operators are strongly recommended to obtain information about the fluid dry-out and re-hydration characteristics from the fluid manufacturers and to select products with optimised characteristics;		
		 Additional information should be obtained from fluid manufacturers for handling, storage, application and testing of their products. 		
GM1	OPS.	GEN.100 Ice and other contaminants		Transposed as GM1-NCC.OP.185. The text is aligned with Part-CAT.
TERM	INOLO	DGY		
	-	y. Terms used in the context of de-icing/anti-icing should be blowing meaning:		
1.	Anti-i follow	cing fluid. Anti-icing fluid includes, but is not limited to, the ing:		
	a.	Type I fluid if heated to min 60 °C at the nozzle;		
	b.	Mixture of water and Type I fluid if heated to min 60 °C at the nozzle;		
	c.	Type II fluid;		
	d.	Mixture of water and Type II fluid;		
	e.	Type III fluid;		

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	f. Mixture of water and Type III fluid;		
	g. Type IV fluid;		
	h. Mixture of water and Type IV fluid.		
	On uncontaminated aircraft surfaces Type II, III and IV anti-icing fluids are normally applied unheated.		
2.	Clear ice. A coating of ice, generally clear and smooth, but with some air pockets. It forms on exposed objects, the temperatures of which are at, below or slightly above the freezing temperature, by the freezing of super-cooled drizzle, droplets or raindrops.		
3.	Conditions conducive to aircraft icing on the ground (e.g. freezing fog, freezing precipitation, frost, rain or high humidity (on cold soaked wings), snow or mixed rain and snow).		
4.	Contamination. Contamination, in this context, is understood as being all forms of frozen or semi-frozen moisture, such as frost, snow, slush or ice.		
5.	Contamination check. Check of aircraft for contamination to establish the need for de-icing.		
6.	De-icing fluid. Such fluid includes, but is not limited to, the following:		
	a. Heated water;		
	b. Type I fluid;		
	c. Mixture of water and Type I fluid;		

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	d. Type II fluid;			
	e. Mixture of water and	Type II fluid;		
	f. Type III fluid;			
	g. Mixture of water and	Type III fluid;		
	h. Type IV fluid;			
	i. Mixture of water and	Type IV fluid.		
	De-icing fluid is normally ap efficiency.	plied heated to ensure maximum		
7.	De-icing/anti-icing. This is the combination of de-icing and anti-icing performed in either one or two steps.			
8.	Ground Ice Detection System (GIDS). System used during aircraft ground operations to inform the personnel involved in the operation and/or the flight crew about the presence of frost, ice, snow or slush on the aircraft surfaces.			
9.	Lowest Operational Use Temperature (LOUT). The lowest temperature at which a fluid has been tested and certified as acceptable in accordance with the appropriate aerodynamic acceptance test whilst still maintaining a freezing point buffer of not less than:			
	a. 10°C for a Type I de	-icing/anti-icing fluid; or		
	b. 7°C for Type II, III o	r IV de-icing/anti-icing fluids.		
10.		xternal check of the aircraft after de-icing t accomplished from suitably elevated		

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	or ot	ervation points (e.g. from the de-icing/anti-icing equipment itself ther elevated equipment) to ensure that the aircraft is free from frost, ice, snow, or slush.		
11.	 Pre-take-off check. An assessment normally performed from within the flight deck, to validate the applied HoT. 			
12.	12. Pre-take-off contamination check. A check of the treated surfaces for contamination, performed when the HoT has been exceeded or if any doubt exists regarding the continued effectiveness of the applied anti-icing treatment. It is normally accomplished externally, just before commencement of the take-off run.			
GM2	GM2 OPS.GEN.100 Ice and other contaminants			Transposed as GM1-NCC.OP.185. The text is aligned with Part-CAT.
ANT	I-ICIN	IG CODES		
1.	The f a. b. c.	following are examples of anti-icing codes: "Type I" at (start time) – To be used if anti-icing treatment has been performed with a Type I fluid; "Type II/100" at (start time) – To be used if anti-icing treatment has been performed with undiluted Type II fluid; "Type II/75" at (start time) – To be used if anti-icing treatment has been performed with a mixture of 75 % Type II fluid and 25 % water; "Type IV/50" at (start time) – To be used if anti-icing treatment has been performed with a mixture of 50 % Type	Non-EU NAA: Recommend adopting AEA guidelines	AEA guidelines could be used as alternative means of compliance (as foreseen by Part-ARO), provided that all criteria are fulfilled.

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IV fluid and 50 % water.		
2. When a two-step de-icing/anti-icing operation has been carried out, the anti-icing code should be determined by the second step fluid. Fluid brand names may be included, if desired.		
GM3 OPS.GEN.100 Ice and other contaminants		Transposed as GM3-NCC.OP.185.
		The text is aligned with Part-CAT.
DE-ICING/ANTI-ICING		
Further guidance material on this issue is given in the International Civil Aviation Organization (ICAO) Manual of Aircraft Ground De-icing/Anti-icing Operations (Doc 9640) (hereinafter referred to as the ICAO Manual of Aircraft Ground De-icing/Anti-icing Operations).		
1. General:		
a. Any deposit of frost, ice, snow or slush on the external surfaces of an aircraft may drastically affect its flying qualities because of reduced aerodynamic lift, increased drag, modified stability and control characteristics. Furthermore, freezing deposits may cause moving parts, such as elevators, ailerons, flap actuating mechanism etc., to jam and create a potentially hazardous condition. Propeller/engine/Auxiliary Power Unit (APU)/systems performance may deteriorate due to the presence of frozen contaminants to blades, intakes and components. Also, engine operation may be seriously affected		

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	by the ingestion of snow or ice, thereby causing engine stall or compressor damage. In addition, ice/frost may form on certain external surfaces (e.g. wing upper and lower surfaces, etc.) due to the effects of cold fuel/structures, even in ambient temperatures well above zero degrees centigrade;		
b.	Procedures established by the operator for de-icing and/or anti-icing are intended to ensure that the aircraft is clear of contamination so that degradation of aerodynamic characteristics or mechanical interference will not occur and, following anti-icing, to maintain the airframe in that condition during the appropriate HoT;		
C.	Under certain meteorological conditions, de-icing and/or anti- icing procedures may be ineffective in provided sufficient protection for continued operations. Examples of these conditions are freezing rain, ice pellets and hail, heavy snow, high wind velocity, fast dropping OAT or any time when freezing precipitation with high water content is present. No HoT guidelines exist for these conditions;		
d.	Material for establishing operational procedures can be found, for example, in:		
	i. ICAO Annex 3, Meteorological Service for International Air Navigation;		
	ii. ICAO Manual of Aircraft Ground De-icing/Anti-icing Operations;		

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	iii.	ISO 11075 Aircraft De-icing/anti-icing fluids ISO type I ;		
	iv.	ISO 11076 Aircraft De-icing/anti-icing methods with fluids2;		
	v.	ISO 11077 Aerospace Self propelled de-icing/anti- icing vehicles Functional requirements2;		
	vi.	ISO 11078 Aircraft - De-icing/anti-icing fluids ISO types II, III and IV2;		
	vii.	AEA "Recommendations for de-icing/anti-icing of aircraft on the ground";		
	viii.	AEA "Training recommendations and background information for de-icing/anti-icing of aircraft on the ground";		
	ix.	EUROCAE ED-104A Minimum Operational Performance Specification for Ground Ice Detection Systems;		
	x.	SAE AS5681 Minimum Operational Performance Specification for Remote On-Ground Ice Detection Systems;		
	xi.	SAE ARP4737 Aircraft - De-icing/anti-icing methods;		
	xii.	SAE AMS1424 De-icing/anti-Icing Fluid, Aircraft, SAE Type I;		

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	xiii. SAE AMS1428 Fluid, Aircraft De-icing/anti-Icing, Non- Newtonian, (Pseudoplastic), SAE Types II, III, and IV;		
	xiv. SAE ARP1971 Aircraft De-icing Vehicle - Self-Propelled, Large and Small Capacity;		
	xv. SAE ARP5149 Training Programme Guidelines for De- icing/anti-icing of Aircraft on Ground; and		
	xvi. ARP5646 Quality Program Guidelines for De-icing/anti- icing of Aircraft on the Ground.	MS: request to accurately reference the document as "SAE ARP5646";	Accepted. Text modified accordingly for consistency.
2. Fluic	is:		
а.	Type I fluid: Due to its properties, Type I fluid forms a thin, liquid-wetting film on surfaces to which it is applied which, under certain weather conditions, gives a very limited HoT. With this type of fluid, increasing the concentration of fluid in the fluid/water mix does not provide any extension in HoT;		
b.	Type II and Type IV fluids contain thickeners which enable the fluid to form a thicker liquid-wetting film on surfaces to which it is applied. Generally, this fluid provides a longer HoT than Type I fluids in similar conditions. With this type of fluid, the HoT can be extended by increasing the ratio of fluid in the fluid/water mix;		
C.	Type III fluid is a thickened fluid especially intended for use on aircrafts with low rotation speeds;		

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	d.	Fluids used for de-icing and/or anti-icing should be acceptable to the operator and the aircraft manufacturer. These fluids normally conform to specifications such as SAE AMS1424, SAE AMS1428 or equivalent. Use of non-conforming fluids is not recommended due to their characteristics being unknown. The anti-icing and aerodynamic properties of thickened fluids may be seriously degraded by, for example, inappropriate storage, treatment, application, application equipment and age.		
3.	Hold-	over protection:		
	a.	Hold-over protection is achieved by a layer of anti-icing fluid remaining on and protecting aircraft surfaces for a period of time. With a one-step de-icing/anti-icing procedure, the HoT begins at the commencement of de-icing/anti-icing. With a two-step procedure, the HoT begins at the commencement of the second (anti-icing) step. The hold-over protection runs out:		
		 at the commencement of the take-off roll (due to aerodynamic shedding of fluid); or 		
		 when frozen deposits start to form or accumulate on treated aircraft surfaces, thereby indicating the loss of effectiveness of the fluid; 		
	b.	The duration of hold-over protection may vary depending on the influence of factors other than those specified in the HoT tables. Guidance should be provided by the operator to take		

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	account of such factors which may include:		
	 atmospheric conditions, e.g. exact type and rate of precipitation, wind direction and velocity, relative humidity and solar radiation; and 		
	 the aircraft and its surroundings, such as aircraft component inclination angle, contour and surface roughness, surface temperature, operation in close proximity to other aircrafts (jet or propeller blast) and ground equipment and structures; 		
с.	HoTs are not meant to imply that flight is safe in the prevailing conditions if the specified HoT has not been exceeded. Certain meteorological conditions, such as freezing drizzle or freezing rain, may be beyond the certification envelope of the aircraft;		
d.	References to usable HoT tables may be found in the AEA "Recommendations for de-icing/anti-icing of aircraft on the ground".		
AMC OPS	.GEN.110 Carriage of persons		Text transposed as AMC1- NCC.OP.140
SEATS WHICH PERMIT DIRECT ACCESS TO EMERGENCY EXITS			
Persons who are allocated seats which permit direct access to emergency exits should appear to be reasonably fit, strong and able to assist the rapid evacuation of the aircraft in an emergency after an appropriate briefing by the crew.			

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GM1 OPS.GEN.110 Carriage of persons		Text not transposed - beyond the scope of Part-NCC.
GENERAL – COMMERCIAL OPERATIONS OTHER THAN COMMERCIAL AIR TRANSPORT		
The carriage of operational personnel indispensable to the performance of a task and carried on a flight taking place immediately before, during or immediately after and directly associated with a specialised task, is not considered Commercial Air Transport. Except for parachute operations, the number of persons carried should not exceed six, excluding crew members.		
GM OPS.GEN.110 Carriage of persons		Text aligned with Part-CAT.
MEANING OF DIRECT ACCESS 'Direct access' means a seat from which a passenger can proceed directly to the exit without entering an aisle or passing around an obstruction.	IS (BA): "direct access" is not used in OPS.GEN.110 – request to delete this GM;	The term "direct access" is used in AMC1-NCC.OP.140.
AMC1 OPS.GEN.115 Passenger briefing		Upgraded to IR.
SAFETY AND EMERGENCY EQUIPMENT		
Relevant safety and emergency equipment includes:	1. MS: Request to upgrade to IR;	1/ Accepted. Upgraded to IR.
1. seat belts or harnesses;	2. MS: request to delete "relevant" -	2/ Accepted. The word "relevant"
2. life jackets;	some equipment may not be present depending on operation/aircraft;	has been deleted.
3. oxygen equipment;		

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4.	ра	issenge	er emergency briefing cards; and		
5.	otl	her em	ergency equipment.		
AM	C2 OF	PS.GEN	I.115 Passenger briefing		Text not transposed - beyond the scope of Part-NCC.
MOT	TOR-P	OWERE	ED AIRCRAFT – COMMERCIAL AIR TRANSPORT		
1.	Befo	ore take	e-off, passengers should be given a demonstration on:		
	a.		use of safety belts and/or safety harnesses, including how asten and unfasten safety belts and/or safety harnesses;		
	b.	the	location and use of oxygen equipment, if required;		
	с.	the	location and use of life jackets, if required; and		
	d.		helicopters, the location and use of life-rafts and survival s, if required.		
2.	Passengers should be given a briefing, if applicable, on the following items:		s should be given a briefing, if applicable, on the following		
	a.	Befo	ore take-off:		
		i.	smoking regulations;		
		ii.	back of the seat to be in the upright position and tray table stowed before take-off and landing;		
		iii.	location of emergency exits;		
		iv.	extinguishing all smoking materials whenever oxygen is being used;		

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	۷.	location and use of floor proximity escape path markings;		
	vi.	stowage of hand baggage;		
	vii.	restrictions on the use of portable electronic devices; and		
	viii.	location and contents of the safety briefing card.		
b.	Afte	r take-off:		
	i.	smoking regulations;		
	ii.	use of safety belts and/or safety harnesses; and		
	iii.	safety benefits of having seat belts fastened when seated irrespective of seat belt sign illumination.		
с.	Befo	pre landing:		
	i.	smoking regulations;		
	ii.	use of safety belts and/or safety harnesses;		
	iii.	back of the seat to be in the upright position and tray table stowed before take-off and landing;		
	iv.	re-stowage of hand baggage; and		
	v.	restrictions on the use of portable electronic devices.		
d.	Afte	r landing:		
	i.	smoking regulations; and		
	ii.	use of safety belts and/or safety harnesses.		

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3.	The briefing may be conducted verbally, through the use of audio- visual equipment, or a combination of both.		
4.	In addition to paragraph 3.a.6. of Annex IV to Regulation (EC) No 216/2008 (Essential requirements for air operations), passengers should also be instructed in the use and location of the required survival equipment.		
АМС	C3 OPS.GEN.115 Passenger briefing		Text not transposed - beyond the scope of Part-NCC.
CAB	SENGER SAFETY TRAINING – MOTOR-POWERED AIRCRAFT WHERE NO IN CREW IS REQUIRED TO BE CARRIED – COMMERCIAL AIR NSPORT		
1.	An operator should establish a training programme for passengers covering safety and emergency procedures, including AMC1 CAT OPS.GEN.115 1. and 2. for a given type of aircraft.		
2.	Passengers who have been trained according to this programme and have flown on the aircraft type within the last 90 days may be carried on board without receiving a briefing/demonstration as required by AMC1 CAT OPS.GEN.115.		
АМС	C4 OPS.GEN.115.B Passenger briefing		Text not transposed - beyond the scope of Part-NCC.
GEN	ERAL - BALLOONS		

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1.		re and after take-off and landing, passengers should be given a ing, relevant to the phase of flight, on the following items:		
	a.	The use of safety and emergency equipment, such as:i. landing hand-holds; andii. the items mentioned in AMC1 OPS.GEN.115, where applicable.		
	b.	Wearing of suitable clothing;		
	C.	Smoking regulations;		
	d.	Stowage of baggage;		
	e.	The importance to remain inside the basket at all times;		
	f.	The landing positions to be assumed to minimise the effect of the impact upon an emergency landing.		
2.		re take-off, passengers should be given a demonstration on 4 OPS.GEN.115.B 1.a. and 1.f.		
АМС	C OPS.	GEN.120.B Securing of passenger cabin and galleys		Text not transposed - beyond the scope of Part-NCC.
CAR	RIAGE	AND STOWAGE OF HAND BAGGAGE - BALLOONS		

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1.	 Only the following items should be considered for carriage as hand baggage during a balloon flight, provided they can be stowed so that they do not pose any hazard during flight or when carrying out emergency procedures: a. Camera equipment; b. Binoculars. 		
2.	In the few minutes preceding the landing and on indication by the pilot-in-command, passengers should stow cameras, binoculars, etc. preferably in their adapted bag/case. Passengers should not keep bulky objects attached around the neck by straps.		
AMC OPS.GEN.125 Portable electronic devices			Based on several comments received within the internal consultation, this AMS has been shortened and limited to explanatory information about the reasons for interferences and recommendations to the operator. The newly proposed GM does not anymore contain out-dated information. A new AMC / GM on portable electronic devices will be developed in a forthcoming rulemaking task.
GEN	ERAL – COMMERCIAL AIR TRANSPORT		

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1. Scope: This AMC addresses Portable Electronic Devices (PEDs), which are not approved equipment permanently installed in the aircraft. Systems and equipment approved and installed in the aircraft will need to satisfy applicable certification requirements and related operating restrictions. Similarly, this AMC does not apply to permitted medical equipment which meets applicable requirements.		
2. Restrictions on use of PEDs by passengers: If an operator permits passengers to use PEDs on board its aircraft, procedures should be in place to control their use. It is the responsibility of the operator to ensure that all aircraft crew and ground agents are trained to enforce the restrictions on this equipment consistent with these procedures. These procedures should ensure the following:		
a. Cell phones and other transmitting devices are not used and are switched off from the time at the start of the flight when the passengers have boarded and all doors have been closed until the end of the flight when a passenger door has been opened.		
The pilot-in-command may permit the use of cell phones when the aircraft is stationary during prolonged departure delays provided that sufficient time is available to check the cabin before the flight proceeds. Similarly, after landing, the pilot-in-command may authorise cell phone use in the event of a prolonged delay for a parking/gate position (even though doors are closed and the engines are running).		
This paragraph does not apply to a PED where the sole means of transmission is identified as a low power transmitting device compliant		

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with the "Bluetooth" Standard.		
This paragraph may not apply to systems installed in the aircraft for the use of cell phones in-flight;		
b. PEDs that are not transmitting devices are disconnected from any in-seat electrical power supply, switched off and stowed during taxiing, take-off, approach and landing, and during abnormal or emergency conditions.		
This restriction does not apply to permitted medical equipment.		
This restriction applies to equipment carried on by the passenger or loaned to the passenger by the aircraft operator.		
In the case of a PED where the sole means of transmission is identified as a low power "Bluetooth" transmitter, it may be considered to be a non- intentional transmitter and may be used during non-critical phases of flight as allowed by this paragraph;		
c. Necessary announcements are made both prior to and during boarding of the aircraft so that passengers may be reminded of the restrictions applicable to cell phones and other transmitting devices before fastening their seat belts;		
d. Cabin crew monitor passenger use of equipment during the flight and, where necessary, ensure suspect equipment is switched off. The cabin crew should be particularly alert to passenger misuse of equipment which has a built-in cell phone. Furthermore, if turbulence is encountered		

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and the crew determine that loose items could present a hazard, instructions will be given that these should be stowed;		
e. Appropriate flight crew to cabin crew co-ordination exists to deal with interference or other PED safety related problems;		
f. Crew are aware of the proper means to switch off in-seat power supplies used for PEDs;		
g. Check-in and ground handling staff, as well as flight and cabin crews, are aware of the safety issues and restrictions concerning PEDs;		
h. Occurrences are reported to the responsible authority of suspected or confirmed interference which has potential safety implications. Where possible, to assist follow-up technical investigation, reports should describe the offending device, identify the brand name and model number, its location in the aircraft at the time of the occurrence, interference symptoms, and the results of actions taken by the crew.		
The co-operation of the device owner should be sought by obtaining contact details.		
3. Restrictions on the use of PEDs by cabin crew:		
PEDs provided to assist cabin crew in their duties should be switched off and stowed during taxiing, take-off, approach and landing, unless tests have been performed which confirm that these PEDs are not a source of unacceptable interference or other safety hazard. Cabin crew should observe the same restrictions for cell phone usage as applicable to passengers.		

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4. Restrictions	on use of PEDs by flight crew:		
in compliance with manual of the aircr	ssist the flight crew in their duties will need to be used the procedures and conditions stated in the operations raft operator. Such equipment will need to be switched ing all phases of flight unless:		
	peen performed which confirm that these PEDs are not ptable interference or distraction;		
b. the PEDs do	o not pose a loose-item risk or other hazard, and		
c. the conditio manual.	ns for their use in-flight are stated in the operations		
transmitting device loading route inform loading). Otherwise	ews should avoid using cell phones and other es during critical pre-flight procedures (e.g. when mation into navigation systems or when monitoring fuel e, flight crews and other persons involved in dispatching ed to observe the same restrictions as passengers.		
This restriction does not preclude use of a cell phone by the flight crew to deal with an emergency although reliance should not be predicated on a cell phone for this purpose.			
5. Other preca	utions:		
together wir carried on b	tems which do not pose a loose item risk, PEDs, th any accessories such as spare batteries or cables, board an aircraft for crew or passenger use, should be th suitable stowage facilities.		
b. Where in-se	eat electrical power supplies are available for passenger		

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use, information cards giving safety instructions should be provided.		
GM OPS.GEN.125 Portable electronic devices		
GENERAL - COMMERCIAL AIR TRANSPORT		
1. General:		
The use of PEDs on board aircraft by crew members and passengers presents a source of uncontrolled electro-magnetic radiation with the risk of adverse interference effects to aircraft systems. Given that a civil aircraft flying at high altitude and high speed in busy airspace is in an obviously hazardous environment, and given that many of the onboard systems are safety devices intended to reduce the risks of that environment to tolerable levels, then anything that degrades the effectiveness of those systems will increase the exposure of the aircraft to the hazards. Consequently, the aircraft operator needs to take measures that will reduce the risks to acceptable limits.		
PEDs fall into two main categories; non-intentional transmitters and intentional transmitters. The first category includes, but is not limited to, computing equipment, cameras, radio receivers, audio and video reproducers, electronic games and toys, together with portable, non- transmitting devices intended to assist crew members in their duties. Intentional transmitters are transmitting devices such as remote control equipment (which may include some toys), two-way radios, cell phones and satellite phones. In periods between transmissions, an intentional		

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transmitter may radiate interference as a non-intentional transmitter.		
2. Non-intentional transmitting PEDs:		
PEDs that are non-intentional transmitters will radiate emissions from internal oscillators and processor clocks, some types of motor, and power supply converters. The radio frequencies involved may fall in the bands used for aeronautical radio services, and emission levels may be sufficient to affect aircraft radio receivers through their antennas. Use of a PED on the flight deck presents a particular risk to those navigation systems having antenna systems located in the radome.		
3. Intentional transmitting PEDs:		
PEDs that are intentional transmitters may induce interference directly into aircraft equipment, wiring or components with sufficient power to adversely affect the proper functioning of aircraft systems. Many aircraft have non-metallic floors and internal doors that present no barrier to prevent the transmission from penetrating to the avionics equipment bays and to the flight deck. Tests have shown that demonstrated susceptibility levels of aircraft equipment, particularly equipment qualified to earlier standards can easily be exceeded.		
a. Cell phones:		
The rapid growth in cell phone usage has presented the most significant risk to aircraft safety from PED interference. Cell phones are both non- intentional and intentional transmitting PEDs, operating on spot channel		

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frequencies in the bands of approximately 415 MHz, 900 MHz or 1800 MHz. (Some regions of the world use slightly different bands). Most use digital modulation but analogue types are still in use. Their maximum transmitted power is in the range of typically one to five watts. The actual power transmitted at a particular time is controlled by the cellular network and may vary from 20 mW to maximum rated power of the cell phone depending on quality of the link between the cell phone and the network. Even in standby mode, a cell phone transmits periodically to register and re-register with the cellular network and to maintain contact with a base station.		
The transmitted power and precise radio frequency of an operating cell phone is dependent on the traffic on the network, the distance of the cell phone from the nearest base station, and any obstacles or attenuation in the signal path. An aircraft on the ground at an airport is likely to be in close proximity to a base station resulting in a strong link between that station and an onboard cell phone. Under these circumstances, the cell phone would seek a free channel in the assigned communication band and its output power would be set by the network to a low level sufficient to maintain the link. Interference levels would, as a result, be low and probably harmless but this cannot be guaranteed. Closing of the aircraft doors increases attenuation in the signal path, and as the aircraft increases its distance from the base station, the output power setting of the cell phone is increased, eventually to its maximum rating. The risk of interference is then at its greatest. At altitude, the cell phone will transmit periodically attempting to register with the cellular network. The quality of the link is likely to be poor and the cell phone will radiate maximum power in these circumstances. Furthermore, since it is likely to be in line-of-sight range of multiple base stations, some degradation of the network operation may result and actual communication may not be possible.		

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The effect of this type of functioning is that, when the aircraft is on the ground near a base station, the interference risk can be low but not negligible, and it will increase as the aircraft taxies and then climbs away from the network base stations.		
The simultaneous use in an aircraft of several cell phones will result in transmissions at different radio frequencies leading to a more complex interference environment.		
b. Private Mobile Radios (PMRs):		
Private mobile radios conforming to the PMR 446 standard are now available to the general public for use as two-way radios without the need of a licence. These radios operate in the 446 MHz band and with sufficient power when transmitting to present an unacceptable interference risk in aircraft. Similarly, other types of two-way radios including those operating in the citizens' band present an unacceptable interference risk.		
c. Wireless Area Networks:		
Wireless Local Area Network (WLAN) is an evolving technology offering wireless data communications, replacing Ethernet cables, for computing information exchange with a range of about 100 metres.		
Standards are being developed for WLAN such as the IEEE 802.11 and some future PEDs are likely to have this capability. WLAN uses radio transmissions of low power in the 2.4 GHz band with consideration being given to use of the 5 GHz band. WLAN transmissions do not need to be licensed.		

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Similarly, Wireless Personal Area Network (WPAN) is an emerging technology offering wireless data and audio communications, with a range of about 10 metres. "Bluetooth" is the name given to one example of a WPAN technology. WPAN also uses unlicensed, very low power radio transmissions in the 2.4 GHz band. Bluetooth will be incorporated into many classes of PED and passengers are likely to bring them on board aircraft expecting to use such devices during the flight. Studies have been completed which show that the interference risk in aircraft from PEDs with a Bluetooth transmitter is sufficiently low to permit their use during non- critical phases of flight i.e. Bluetooth devices need be subject only to the general restrictions applied to non-intentional transmitters.		
4. In-seat Power Supplies:		
Many aircraft now offer an electrical supply at each passenger seat primarily for the purpose of operating laptop computers. These computers have safety devices to protect against over-charging of their re- chargeable batteries. Other types of PED might not have such protection and might be fitted (possibly incorrectly) with standard, non-chargeable batteries. Overcharging of batteries, or attempts to charge standard batteries, could cause them to fail in a dangerous manner with fire, smoke and fumes risks. It is the responsibility of the aircraft operator to ensure that PEDs connected to the in-seat supply do not present any additional hazard to persons on board the aircraft or to the aircraft itself. Safeguards include issuing passengers with information cards giving safety instructions for using the in-seat supplies and the restrictions for charging or handling batteries. The availability of a means to terminate and isolate such electrical supplies together with appropriate cabin crew procedures will be required as a condition of approval and use of in-seat power		

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A: Rule	B: Summary of comments	C: Reason for change, remarks
supplies.		
5. Interference levels and effects:		
a. Aircraft Equipment Qualification:		
To qualify for approval, equipment to be installed in aircraft has to demonstrate that it is not susceptible to prescribed levels of radiated interference irrespective of the source, and that it will not radiate unacceptable interference. The levels were originally set to ensure equipment could co-exist in the aircraft without mutual interference. For example, for an equipment susceptibility test prior to 1985, the maximum field strength of radiated interference was set at only 0.1 volts per metre. The risk of an uncontrolled interference source within the aircraft was not addressed by earlier standards. Recognising the inadequacy of the earlier standards, the tests have become progressively more severe primarily to protect against external threats such as broadcast transmitters, radars, and satellite uplinks. For critical equipment, the susceptibility tests now involve field strengths of 200 volts per metre or more. However, even the latest standards permit a low level of immunity for some equipment. Many aircraft, including newly manufactured aircraft, still have systems and equipment qualified to earlier standards.		
b. Interference Levels:		
Studies have confirmed that the levels and radio frequencies of radiated interference from non-intentional transmitters are such that aircraft radio receivers can be affected. Over the years, many reports have been		

A: Rule	B: Summary of comments	C: Reason for change, remarks
received by the authorities concerning such interference.		
For an intentional transmitter such as a cell phone, an obvious risk is recognised even though the cell phone is not transmitting in the aeronautical frequency band. Applying fundamental principles, the maximum field strength E in volts per metre of the transmission at a distance D from a cell phone transmitting P Watts of radio frequency power in a free, unobstructed space, can be estimated using the equation;		
$E = 7 \sqrt{P}$ divided by the distance D		
(The strengths of electric and magnetic fields that exist in close proximity to the transmitting antenna (i.e. distances less than one wavelength and known as the near field) are not considered in this simple explanation.)		
Thus, for a two watt cell phone, the maximum field strength in free space at one metre distance is approximately 10 volts per metre, and at 100 metres distance, approximately 0.1 volts per metre.		
However, in the confines of a metallic aircraft fuselage, complex propagation paths arise due to reflections from the metallic structure which can lead to signal cancellation or re-enforcement at different locations in the aircraft. Although the free space equation does not give reliable results under these conditions, tests have shown that the field strength of the interfering cell phone transmission, at maximum power, will exceed by a significant margin the levels used in susceptibility tests for avionic equipment qualified to earlier standards. Similarly, these tests have shown that interference levels would vary by relatively small changes of location of a cell phone and that persons obstructing the		

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transmission path reduce the interference.		
c. Effects:		
Reports of interference are increasing but it has been difficult to positively confirm in all cases that a PED has been the actual cause of a problem. This is due to the difficulty in replicating the conditions that existed at the time of the occurrence due to the multiple factors involved (e.g. geographical location of the aircraft, system operating modes, interference frequency and intensity, source location in the aircraft, and path attenuation). Cell phones have been positively identified as the cause of degraded communications and of false baggage compartment smoke warnings. Cell phones have been strongly implicated in other spurious cockpit/pilot compartment warnings, corrupted instrument displays, and pressurisation system malfunctions.		
Although the total number of reports is relatively low considering the aircraft flight hours involved, the potential severity of the effects of interference means that the problem cannot be ignored.		
As a general conclusion, interference can result in:		
i. malfunctioning of multiple systems;		
ii. false warnings of unsafe conditions;		
iii. increased work load for the flight crew and the possibility of invoking emergency drills;		
iv. reduced crew confidence in protection systems which may then be ignored during a genuine warning;		

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v. distraction of the flight crew from their normal duties;		
vi. noise in the flight crew headphones; and/or		
vii. hidden failures of safety systems with loss of protection.		
6. Recommendations:		
a. Aircraft operators should consider installing detectors in their aircraft, which together with suitable procedures can assist the cabin crew to detect unauthorised transmissions from commonly used types of cell phone.		
 b. Aircraft operators should seek the assistance of airport operators for the display of safety notices at aircraft boarding points reminding passengers to switch off cell phones and other transmitting devices. 		
AMC OPS.GEN.135.A Taxiing of aeroplanes		Text upgraded to IR, NCC.GEN.120, and aligned with CAT.
QUALIFIED PERSONNEL		
A qualified person is either a flight crew member or a person designated by the operator that is:		
1. competent to taxi;		
2. qualified to use the radio telephone if radio communications are required;		

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3. has received instruction from a competent person on operational procedures, aerodrome layout, and where appropriate, information on routes, signs, marking, lights, ATC signals and instructions, phraseology and procedures; and		
4. able to conform to the operational standards required for safe aircraft movement at the aerodrome.		
GM OPS.GEN.140.H Rotor engagement		Text transposed as GM1- NCC.GEN.125.
		Completely redrafted based on comments received during the internal consultation. The new text describes the intent of the rule and provides further guidance for the rotor engagement for the purpose of flight and for maintenance.
QUALIFIED PERSONNEL		
The intent of this paragraph is to ensure that the pilot remains at the controls when the rotors are turning under power whilst not preventing ground runs being conducted by qualified personnel other than flight crew. The operator should ensure that the qualification of personnel, other than flight crew, which are authorised to conduct ground runs, is described in the appropriate manual.		
AMC1 OPS.GEN.145 Use of aerodromes/operating sites		

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A: F	A: Rule		B: Summary of comments	C: Reason for change, remarks
USE	e of (OPERATING SITES		
1.	1. The pilot-in-command should have available from a pre-survey or other publication, for each operating site to be used, diagrams or ground and aerial photographs, depiction (pictorial) and description of:			
	a.	the overall dimensions of the operating site;		
	b.	location and height of relevant obstacles to approach and take-off profiles, and in the manoeuvring area;		
	c.	approach and take-off flight paths;		
	d.	surface condition (blowing dust/snow/sand);		
	e.	provision of control of third parties on the ground (if applicable);		
	f.	lighting (if applicable);		
	g.	procedure for activating operating site in accordance with national regulations (if applicable);		
	h.	other useful information, for example appropriate Air Traffic Services (ATS) agency and frequency; and		
	i.	site suitability with reference to available aircraft performance.		
2.	ma fror	sites which are not pre-surveyed, the pilot-in-command should ke, from the air and, in the case of balloons, also prior to take-off m the ground, a judgement on the suitability of a site. At least C OPS.GEN.145 1.a. to 1.e.inclusive and 1.i. should be		Also f. should be considered especially for night operations.

Tart Nee		50 Aug 2011
A: Rule	B: Summary of comments	C: Reason for change, remarks
considered.		
AMC2 OPS.GEN.145 Use of aerodromes/operating sites		Text not transposed - beyond the scope of Part-NCC.
USE OF OPERATING SITES - COMMERCIAL AIR TRANSPORT		
1. When defining adequate operating sites for use for the type(s) of aircraft and operation(s) concerned, an operator should take account of the following:		
a. An adequate site is a site which the operator considers to be satisfactory, taking account of the applicable performance requirements and site characteristics;		
b. The operator should have in place a procedure for the survey of sites by a competent person. Such a procedure should take account for possible changes to the site characteristics which may have taken place since last surveyed;		
c. Sites which are pre-surveyed should be specifically indicated in the operator's operations manual. The operations manual should contain diagrams or/and ground and aerial photographs, and depiction (pictorial) and description of:		
i. the overall dimensions of the site;		
 ii. location and height of relevant obstacles to approach and take-off profiles, and in the manoeuvring area; 		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
iii.	approach and take-off flight paths;		
iv.	surface condition (blowing dust/snow/sand);		
٧.	adequacy with reference to aircraft performance;		
vi.	provision of control of third parties on the ground (if applicable);		
vii.	procedure for activating site with land owner or controlling authority;		
viii.	other useful information, for example appropriate Air Traffic Services (ATS) agency and frequency; and		
ix.	lighting (if applicable).		
place a proce case of ballo the suitabilit	ites which are not pre-surveyed, the operator should have in edure which enables the pilot to make, from the air or, in the ons, also prior to take-off, from the ground, a judgement on by of a site. As a minimum, the items listed in 1.c.i. to vi. ould be considered.		
conducted, u	ations to non pre-surveyed sites by night should not be unless the operator is approved to do so in accordance with A.001.HEMS.		
AMC3 OPS.	GEN.145.H Use of aerodromes/operating sites		Text not transposed - beyond the scope of Part-NCC.
HELICOPTER	RS – USE OF HELIDECKS – COMMERCIAL AIR TRANSPORT		
	content of the operations manual relating to the use of should contain the listing of helideck limitations in a Helideck		

A: Rule	B: Summary of comments	C: Reason for change, remarks
Limitations List (HLL) and a pictorial representation (template) of each helideck showing all necessary information of a permanent nature. The HLL will show, and be amended as necessary to indicate, the most recent status of each helideck concerning non-compliance with International Civil Aviation Organization (ICAO) Annex 14 Volume 2, limitations, warnings, cautions or other comments of operational importance. An example of a typical template is shown in figure 1.		
2. In order to ensure that the safety of flights is not compromised, the operator should obtain relevant information and details for compilation of the HLL, and the pictorial representation, from the owner/operator of the helideck.		
3. When listing helidecks, if more than one name of the helideck exists, the most common name should be used; other names should also be included. After renaming a helideck, the old name should be included in the HLL for the ensuing six months.		
4. All helideck limitations should be included in the HLL. Helidecks without limitations should also be listed. With complex installations and combinations of installations (e.g. co-locations), a separate listing in the HLL, accompanied by diagrams where necessary, may be required.		
5. Each helideck should be assessed (based on limitations, warnings, cautions or comments) to determine its acceptability with respect to the following which, as a minimum, should cover the factors listed below:		
a. The physical characteristics of the helideck;		

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b. safeg		preservation of obstacle protected surfaces is the most basic or all flights. These surfaces are:		
	i.	the minimum 210° Obstacle Free Surface (OFS);		
	ii.	the 150° Limited Obstacle Surface (LOS); and		
	iii.	the minimum 180° falling "5:1" - gradient with respect to significant obstacles. If this is infringed or if an adjacent installation or vessel infringes the obstacle clearance surfaces or criteria related to a helideck, an assessment should be made to determine any possible negative effect which may lead to operating restrictions;		
c.	Mark	ing and lighting:		
	i.	Adequate perimeter lighting;		
	ii.	Adequate floodlighting;		
	iii.	Status lights (note: for night and day operations e.g. Aldis Lamp);		
	iv.	Dominant obstacle paint schemes and lighting;		
	٧.	Helideck markings;		
	vi.	General installation lighting levels. Any limited authorisation in this respect should be annotated "daylight only operations" on the HLL;		
d.	Deck	surface:		
	i.	Surface friction;		
	ii.	Helideck net;		

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	iii.	Drainage system;		
	iv.	Deck edge netting;		
	v.	Tie down system;		
	vi.	Cleaning of all contaminants;		
e.	Envir	conment:		
	i.	Foreign Object Damage;		
	ii.	Physical turbulence generators;		
	iii.	Bird control,		
	iv.	Air quality degradation due to exhaust emissions, hot gas vents or cold gas vents;		
	v.	Adjacent helidecks may need to be included in air quality assessment;		
f.	Resc	ue and fire fighting:		
	i.	Primary and complementary media types, quantities, capacity and systems personal protective equipment and clothing, breathing apparatus;		
	ii.	Crash box;		
g.	Com	munications & navigation:		
	i.	Aeronautical Radio(s);		
	ii.	R/T call sign to match helideck name and side identification which should be simple and unique;		
	iii.	NDB or equivalent (as appropriate);		

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	iv.	Radio log;		
	۷.	Light signal (e.g. Aldis Lamp);		
h.	Fuell	ing facilities;		
i.	Addit	tional operational and handling equipment:		
	i.	Windsock;		
	ii.	Wind recording;		
	iii.	Deck motion recording and reporting where applicable;		
	iv.	Passenger briefing system;		
	٧.	Chocks;		
	vi.	Tie downs;		
	vii.	Weighing scales.		
j.	Perso	onnel:		
		deck staff (e.g. Helicopter Landing Officer/Helicopter Deck Id fire fighters etc.).		
visit	informa . During	nelidecks about which there is incomplete information, 'limited' tion may be issued by the operator prior to the first helicopter subsequent operations and before fully used, information athered and the following procedures should apply:		
а.	Picto i.	rial (static) representation: Templates (see Figure 1) should be available, to be filled out during flight preparation on the basis of the information		

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ule		B: Summary of comments	C: Reason for change, remarks
	given by the helideck owner/operator and flight crew observations;		
ii.	Where possible, suitably annotated photographs may be used until the HLL and template has been completed;		
iii.	Until the HLL and template has been completed, operational restrictions (e.g. performance, routing etc.) may be applied;		
iv.	Any previous inspection reports should be obtained by the operator;		
v.	An inspection of the helideck should be carried out to verify the content of the completed HLL and template, following which the helideck may be fully used for operations;		
With ving:	reference to the above, the HLL should contain at least the		
i.	HLL revision date and number;		
ii.	Generic list of helideck motion limitations;		
iii.	Name of helideck;		
iv.	'D'-value of the helideck;		
۷.	Limitations, warnings, cautions and comments;		
The t v):	emplate should contain at least the following (see example		
Insta	llation/Vessel name;		
	 ii. iii. iv. v. With ving: i. ii. iv. v. The t v): 	given by the helideck owner/operator and flight crew observations; ii. Where possible, suitably annotated photographs may be used until the HLL and template has been completed; iii. Until the HLL and template has been completed, operational restrictions (e.g. performance, routing etc.) may be applied; iv. Any previous inspection reports should be obtained by the operator; v. An inspection of the helideck should be carried out to verify the content of the completed HLL and template, following which the helideck may be fully used for operations; With reference to the above, the HLL should contain at least the ring: i. HLL revision date and number; ii. Generic list of helideck motion limitations; iii. Name of helideck; iv. 'D'-value of the helideck; v. Limitations, warnings, cautions and comments; The template should contain at least the following (see example	given by the helideck owner/operator and flight crew observations; ii. Where possible, suitably annotated photographs may be used until the HLL and template has been completed; iii. Until the HLL and template has been completed, operational restrictions (e.g. performance, routing etc.) may be applied; iv. Any previous inspection reports should be obtained by the operator; v. An inspection of the helideck should be carried out to verify the content of the completed HLL and template, following which the helideck may be fully used for operations; With reference to the above, the HLL should contain at least the ring: i. HLL revision date and number; ii. Generic list of helideck motion limitations; iii. Name of helideck; iv. 'D'-value of the helideck; v. Limitations, warnings, cautions and comments; The template should contain at least the following (see example);

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ii.	R/T call sign;		
iii.	Helideck Identification Marking;		
iv.	Side Panel Identification Marking;		
v.	Helideck elevation;		
vi.	Maximum installation/vessel height;		
vii.	'D' Value;		
viii.	Type of installation/vessel;A.Fixed mannedB.Fixed unmannedC.Ship type (e.g. diving support vessel)D.Semi-submersibleE.Jack-up		
ix.	Name of owner/operator;		
x.	Geographical position;		

_		Part-NCC	CRST	30 Aug 2011
A: Ru	ıle		B: Summary of comments	C: Reason for change, remarks
xi.	Com/Nav Fi	requencies and Ident;		
	ations showir	wing preferably looking into the helideck with ng location of derrick, masts, cranes, flare stack, turbine side identification panels, windsock etc.;		
		rawing, chart orientation from the general drawing, to he plan view will also show the 210 degree bisector ees true;		
xiv.	Type of fue	lling:		
	Α.	Pressure and Gravity		
	В.	Pressure only		
	C.	Gravity only		
	D.	None		
xv.	Type and na	ature of fire fighting equipment;		
xvi.	Availability	of GPU;		
xvii.	Deck headii	ng;		
xviii.	Maximum a	llowable mass;		

Karestack Jamery Charress	Part-NCC	CRST	30 Aug 2011
Turbina sh.		B: Summary of comments	C: Reason for change, remarks
Intelline (: (3) GPU : (4) deck head.: Max Mass: 7 status light: (5) revision date 1. Fixed manned; fixed unmanned; small ship; large ship; semi-submersible; jack-te 2. NAM_AMOGO etc.	ıр.		
 Pressure/gravity; pressure; gravity; no. Yes; no; 28V DC. Yes; no. 			
AMC4 OPS.GEN.145 Use of aerodromes/operatin	g sites		Text not transposed - beyond the scope of Part-NCC.
USE OF OPERATING SITES – COMMERCIAL OPERATIO COMMERCIAL AIR TRANSPORT	NS OTHER THAN		
When using operating sites, the operator should take OPS.GEN.145.	account of AMC2		
GM1 OPS.GEN.145 Use of aerodromes/operating	sites		Text transferred to AMC to Annex I Definitions.
ADEQUATE AERODROME			
At the expected time of use, the adequate aerodrome and equipped with necessary ancillary services such a lighting, communications, weather reporting, navaids services.	s ATS, sufficient		
GM2 OPS.GEN.145 Use of aerodromes/operating	sites		Text transposed as GM1-

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		NCC.OP.100	
 PUBLICATIONS 'Other publication' mentioned in AMC OPS.GEN.145 refers to publication means, such as: 1. (Military) Aeronautical Information Publication; 2. Visual Flight Rules (VFR) Guides; 3. commercially available aeronautical publications(e.g. Jeppesen, AERAD, Fugawi); and 4. non-commercially available publications. 	 1/ MS: suggested a text clarification since civil AIPs should be included. 2/ IS: requested deletion of examples. EASA should not make reference to commercial documentation. 	 1/ Partially accepted. By putting 'military' in brackets it implied both; however the text is improved. 2/ Accepted 	
GM3 OPS.GEN.145 Use of aerodromes/operating sites		Text not transposed - beyond the scope of Part-NCC.	
GUIDANCE DOCUMENTS - COMMERCIAL AIR TRANSPORT			
Guidance on standards and criteria for the design of aerodromes are contained in:			
1. ICAO Annex 14 Aerodromes ; and			
2. ICAO Heliport Manual (Doc 9261-AN/903).			
AMC OPS.GEN.147(c)(1) Visual Flight Rules (VFR) Operating minima		Text not transposed - the rule is covered by Part-SERA.	

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ADVISORY SPEEDS IN REDUCED VISIBILITY When flight with a visibility of less than 5 km is permitted, the forward visibility should not be less than the distance travelled by the helicopter in 30 seconds so as to allow adequate opportunity to see and avoid obstacles (see table below).				
Visibility (m)	Advisory speed (kts)			
800	50			
1 500	100			
2 000	120			
AMC1 OPS.GEN.150 Instrument Flight Rules (IFR) operating minima SPECIFYING AERODROME MINIMA An acceptable method of specifying aerodrome operating minima may be		INDIV: request to clarify if a Jeppesen approach chart is sufficient;		That would be an acceptable means to comply with the rule.
through the use of commercially available information.				
AMC2 OPS.GEN.150 Instrument Flight Rules (IFR) operating minima				
AERODROME MINIMA – GENERAL				
1. The aerodrome operating minima should not be lower than the values given in AMC3 OPS.GEN.150 3. and AMC4 OPS.GEN.150.				

A: Rule	B: Summary of comments	C: Reason for change, remarks	
2. All approaches should be flown as Stabilised Approaches (SAp). Different procedures may be used for a particular approach to a particular runway.			
3. All non-precision approaches should be flown using the Continuous Descent Final Approach (CDFA) technique. Different procedures may be used for a particular approach to a particular runway. When calculating the minima in accordance with AMC4 OPS.GEN.150, the applicable minimum Runway Visual Range (RVR) should be increased by 200 metres (m) for Category A/B aeroplanes and by 400 m for Category C/D aeroplanes for approaches not flown using the CDFA technique, provided the resulting RVR/Converted Meteorological Visibility (CMV) value does not exceed 5000 m. SAp or CDFA should be used as soon as facilities are improved to allow these techniques.	1. IS (GA): request that the RVR is not increased to 200/400 m for non- commercial operations where CDFA is not used. No safety case for the increase;	Not accepted. NPA flown with the CDFA technique are considered to be safer than the conventional technique. The text was maintained and aligned with CAT.	
AMC3 OPS.GEN.150 Instrument Flight Rules (IFR) operating minima			
AERODROME MINIMA – TAKE-OFF MINIMA	Upgrade this AMC to an IR, put this AMC in Part SPA, respectively;	The text for low visibility take-off (LVTO) requiring a specific approval in accordance with Part- SPA has not been transposed, since such operations are now covered by SPA.LVO.	
1. General:	1/ Suggestion to replace "discontinued take-off" by "rejected take-off";	 Accepted. Text aligned. Not accepted in the interest of safety. 	
	2/ IS (GA): request for exemption for		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
		GA, in order to continue with current system of ops in accordance with LVTO minima, but without LVTO approval. No safety case to answer;	
a.	Take-off minima should be expressed as visibility or RVR limits, taking into account all relevant factors for each aerodrome planned to be used and aircraft characteristics. Depending on the obstacle situation on departure and/or forced landing, additional conditions (e.g. ceiling) should be specified;	MS: edit to align with the intention of EU/JAR-OPS - last sentence should be replaced with: "Depending on the obstacle situation on departure and/or forced landing, where there is a specific need to see and avoid obstacles on departure and/or for a forced landing, additional conditions (e.g. ceiling) should be specified";	Text aligned with the CAT text.
b.	The pilot-in-command should not commence take-off unless the weather conditions at the aerodrome of departure are equal to or better than applicable minima for landing at that aerodrome unless a suitable take-off alternate aerodrome is available;	 INDIV: request to clarify if non- commercial helicopter operations require a take-off alternate aerodrome, as this is not required in OPS.GEN.155 – request to align AMC and IR material consistently; INDIV: request to add " unless a suitable take-off alternate aerodrome/operating site" to cater for helicopter ops; 	 1/ 2/ take-off alternate aerodrome are not required for NCC helicopter operations. However, to comply with this rule the PIC should not commence a take-off unless the weather conditions at the at the aerodrome of departure are equa to or better than applicable minima for landing at that aerodrome
c.	When the reported meteorological visibility is below that required for take-off and RVR is not reported, a take-off should only be commenced if the pilot-in-command can determine that the visibility along the take-off runway/area is		

		Part-NCC	CRST	30 Aug 2011
A: R	ule		B: Summary of comments	C: Reason for change, remarks
		equal to or better than the required minimum;		
	d.	When no reported meteorological visibility or RVR is available, a take-off should only be commenced if the pilot-in-command can determine that the RVR/visibility along the take-off runway/area is equal to or better than the required minimum.	Pilot assessment of RVR: text clarification required concerning required RVR for initial take-off run can be replaced by pilot assessment;	Rule text is sufficiently clear.
2.	Visua	al reference:		
	a.	The take-off minima should be selected to ensure sufficient guidance to control the aircraft in the event of both a discontinued take-off in adverse circumstances and a continued take-off after failure of the critical power unit;		
	b.	For night operations, ground lighting should be available to illuminate the runway/Final Approach and Take-Off Area (FATO) and any obstacles.		
3.	Requ	iired RVR/visibility:		
	a.	Aeroplanes:		
		i. For multi-engined aeroplanes whose performance is such that, in the event of a critical power unit failure at any point during take-off, the aeroplane can either stop or continue the take-off to a height of 1 500 ft above the aerodrome while clearing obstacles by the required margins, the take-off minima specified by an operator should be expressed as RVR/visibility values not lower than those given in Table 1a, except as provided in		

				Part-NCC	С	RST	30 Aug 2011
A: Rule					B	: Summary of comments	C: Reason for change, remarks
	3.a.iv.						
Table 1a of AM	C3 OPS.GEN.150	RVR/visibility for	r take-off				The amended table only provides values for take-off operations not requiring an LVTO approval.
TAKE-OFF RVI	R/VISIBILITY						
Facilities	RVR/Visibility	(Note 2)					
	Category A, B aeroplanes	and C	Category D	aeroplanes			
	Without LVTO approval in accordance with OPS.SPA.001. LVO	With LVTO approval in accordance with OPS.SPA.001. LVO	ithout LVTO approval in accordance with OPS.SPA.0 01.LVO	With LVTO approval in accordance with OPS.SPA.00 .LVO			
Nil (Day only)	500 m	500 m	500 m	500 m			
Runway edge lighting and/or centreline lighting	400 m (Note 1)	250 m (Note 1)	400 m (Note 1)	300 m (Note 1)			
Runway	400 m	200 m	400 m	250 m			

							-
A: Rule	A: Rule					: Summary of comments	C: Reason for change, remarks
edge and centreline lighting							
Runway edge and centreline lighting and multiple RV information	/R	150 m (Note 3)	400 m	200 m (Note 3)			
	For night operations ights are required.	s at least runwa	y edge and rur	nway end			
ļ	The reported RVR/v part of the take-off assessment.		epresentative o replaced by pi				
1	<i>Note 3: The required RVR value should be achieved for all of the relevant RVR reporting points with the exception given in Note 2 of Table 1a.</i>						
ii. For multi-engined aeroplanes whose performance is such that they cannot comply with the performance conditions in 3.a.i., in the event of a critical power unit failure, there may be a need to re-land immediately and to see and avoid obstacles in the take-off area. Such aeroplanes may be operated to the following take-off minima provided they are able to comply with the applicable obstacle clearance criteria, assuming engine failure at the height specified. The take-off minima				Formance I power unit mediately and area. Such ng take-off with the ming engine			

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A: Rule			B: Summary of comments	C: Reason for change, remarks
from which off flight pa may not be Table 1a or	the One-Engine-Ind ath can be construct lower than either c	be based upon the height operative (OEI) net take- ed. The RVR minima used of the values given in oval in accordance with		
	Table 2a of AMC3 OPS.GEN.150 Assumed engine failure height above the runway versus RVR/visibility			The amended table only provides values for take-off operations not requiring an LVTO approval.
TAKE-OFF RVR/VISIBILITY				
Assumed engine failure height above the take- off runway	RVR/Visibility (Note 2)			
	Without LVTO approval in accordance with OPS.SPA.001.L VO	With LVTO approval in accordance with OPS.SPA.001.LVO		
< 50 ft	400 m	200 m		
51 – 100 ft	400 m	300 m		
101 – 150 ft	400 m	400 m		
151 – 200 ft	500 m	500 m		

				1	50 //ag 2011	
A: Rule				B: Summary of comments	C: Reason for change, remarks	
201 - 300) ft	1 000 m	1 000 m			
> 300 ft		1 500 m	1 500 m			
		(Note 1)	(Note 1)			
	1 500 m is als be constructe		ive take-off flight path can			
	The reported part of the tai assessment.		resentative of the initial eplaced by pilot			
	iii. When reported RVR or meteorological visibility is not available, the pilot-in-command should not commence take-off unless he/she can determine that the actual conditions meet the applicable take-off minima.					
	•	cors approved in accord PA.001.LVO may reduce	ance with e the take-off minima to:			
	 A. 125 m RVR (Category A, B and C aeroplanes) or 150 m RVR (Category D aeroplanes) provided the following criteria are met: 			This text is transposed in SPA.LVO.		
	1. L	ow visibility procedures	are in force;			
	n		entreline lights, spaced 15 nsity edge lights, spaced ration;			

A: Rule			B: Summary of comments	C: Reason for change, remarks
		 Flight crew members have satisfactorily completed training in a flight simulator; 		
		 A 90 m visual segment is available from the cockpit at the start of the take-off run; 		
		5. The required RVR value has been achieved for all of the relevant RVR reporting points.		
	В.	less than 125 m (Category A, B and C aeroplanes) or 150 m (Category D aeroplanes) but not lower than 75 m, provided runway protection and facilities equivalent to Category III landing operations are available, when using either:		
		1. an approved lateral guidance system; or		
		 an approved Head Up Display/Head Up Landing System (HUD/HUDLS) for take-off. 		
b.	Helio	opters:		
	i.	For Performance Class 1 operations, an operator should specify an RVR and Visibility (RVR/VIS) as take-off minima in accordance with Table 1h;		
Table 1h o	f AMC3	OPS.GEN.150 RVR/visibility for take-off		The amended table only provides values for take-off operations not requiring an LVTO approval.

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A: Rule			B: Summary of comments	C: Reason for change, remarks
TAKE-OFF RVR/VISIBILITY				
Onshore aerodromes with	RVR/Visibility			
Instrument Flight Rules (IFR) departure procedures	Without LVTO approval in accordance with OPS.SPA.001.LV O	With LVTO approval in accordance with OPS.SPA.001.LV O		
No lighting and no markings (day only)	400 m or the rejected take-off distance, whichever is the greater	250 m or the rejected take-off distance, whichever is the greater		
No markings (night)	800 m	800 m		
Runway edge/FATO lighting and centreline marking	400 m	200 m		
Runway edge/FATO lighting, centreline marking and RVR information	400 m	150 m		
Offshore helideck				
Two-pilot operations	400 m (Note 1)	250 m (Note 1)		
Single-pilot operations	500 m (Note 1)	500 m (Note 1)		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
Note 1:	The take-off flight path must be free of obstacles.		
	 For Performance Class 2 operations onshore, the pilot- in-command should operate to take-off minima of 800 m RVR/VIS and remain clear of cloud during the take- off manoeuvre until reaching Performance Class 1 capabilities; 		
	iii. For Performance Class 2 operations offshore, the pilot- in-command should operate to minima not less that that for Performance Class 1 and remain clear of cloud during the take-off manoeuvre until reaching Performance Class 1 capabilities. (Refer to Note 1 of Table 1h of AMC3 OPS.GEN.150);		
	 Table 1 of AMC11 OPS.GEN.150, for converting reported meteorological visibility to RVR, should not be used for calculating take-off minima. 		
AMC4 OP minima	S.GEN.150 Instrument Flight Rules (IFR) operating	Suggestion to upgrade this AMC to an IR;	Accepted. In the interest of safety, this AMC has been upgraded to the IR level, NCC.OP.111.
	ME MINIMA – NON-PRECISION, CATEGORY I AND APPROACHES RTICAL GUIDANCE		
appr	Non-Precision Approach (NPA) operation is an instrument roach using any of the facilities described in Table 1 of AMC4 G.GEN.150, with a Minimum Descent Height (MDH) or Decision	Suggestion to have two different texts for non-precision approaches and CAT I approaches;	This is the case.

A: F	tule	B: Summary of comments	C: Reason for change, remarks
	Height (DH) not lower than 250 ft and an RVR/CMV of not less than 750 m for aeroplanes and 600 m for helicopters.		
2.	A Category I approach operation is a precision instrument approach and landing using Instrument Landing System (ILS), Microwave Landing System (MLS), GPS Landing System (GLS) (Global Navigation Satellite System (GNSS)/Ground-Based Augmentation System (GBAS)) or Precision Approach Radar (PAR) with a DH not lower than 200 ft and an RVR of not less than 550 m for aeroplanes and 500 m for helicopters.		
3.	An approach Procedure with Vertical Guidance (APV) is an instrument approach which utilises lateral and vertical guidance, but does not meet the requirements established for precision approach and landing operations, with a DH not lower than 250 ft and an RVR of not less than 600 m.	INDIV: Request to address DH for APV, request to allow an APV down to 200ft; Eurocontrol, MS: as approach classification is currently under review in ICAO, request to permit LPV (APV SBAS) with DH 200 ft. Definition should be changed to: "An Approach Procedure with Vertical guidance (APV) is an instrument approach which uses lateral and vertical guidance using SBAS or Barometric VNAV with a DH not lower than 200 ft and an RVR of not less than 550 m for aeroplanes and 500 m for helicopters";	1/, 2/ APV are defined in Annex I. The minimum DH is 250 ft. LPV flown to the CAT I minima are considered as CAT I operations.
	4. The DH to be used for an approach should be the highest of:a. the minimum height to which the approach aid can		

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A: R	A: Rule					ary of comments	C: Reason for change, remarks	
		be used without	t the required visual r	eference;				
	b.	the Obstacle Cle aircraft;	earance Height (OCH)) for the category of				
	c.	the published a	pproach procedure DI	H, where applicable;				
	d.	200 ft for Categ	Jory I approach opera	tions;				
	e.	the system min	imum in Table 1 of Al	MC4 OPS.GEN.150; or				
	f. the lowest decision height specified in the AFM or equivalent document, if stated.							
5.	The I	MDH for an approa	ach should be the hig	hest of:				
	a.	The OCH for the	e category of aircraft;					
	b.	The system mir	imum in Table 1 of A	MC4 OPS.GEN.150; or				
	с.	The minimum d	escent height specifie	ed in the AFM, if stated.				
Tabl	e 1 of	AMC4 OPS.GEN.1	50 System minima ve	s facilities				
SYS	STEM	MINIMA				trol: Request to amend V minima to 250 ft, as for	1/ Values amended and updated with GNSS approaches.	
Fac	cility		Lowest DH/MDH		VOR/DME;		2/ Accepted. System minima for	
			Aeroplanes	Helicopters		request to insert figures as anes, for helicopters, under	aeroplane and helicopter operations are aligned.	
Loc DM		with or without	250 ft	250 ft		AV/LNAV and VOR/DME;		
SRA (terminating at ½ nm)		ninating at ½	250 ft	250 ft		-		

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A: Rule			B: Summ	ary of comments	C: Reason for change, remarks
SRA (terminating at 1 nm)	300 ft	300 ft			
SRA (terminating at 2 nm or more)	350 ft	350 ft		-	
RNAV/LNAV	300 ft	n/a		-	
VOR	300 ft	300 ft		-	
VOR/DME	250 ft	250 ft		_	
NDB	350 ft	300 ft		_	
NDB/DME	300 ft	n/a		_	
VDF (QDM & QCH)	350 ft	300 ft			
AMC5 OPS.GEN.150 Instruminima AERODROME MINIMA – CRIT				quest to consider that a ly specified for precision s;	Not accepted. DA/H can also be applied to certain APV operations and to NPA flown with the CDFA technique.
detailed in Table 3 of approach grouping), th	or the lowest allowable v f AMC6 OPS.GEN.150.A (ne instrument approach sh quirements and associated	applicable to each ould meet at least			
and including 4.	oaches with designated ve 5 degrees for Category A a s for Category C and D aer	nd B aeroplanes			

	Part-NCC		30 Aug 2011
A: Rule		B: Summary of comments	C: Reason for change, remarks
	the facilities are:		
	i. ILS/MLS/GLS/PAR; or		
	ii. APV; and		
	 where the final approach track is offset by not more than 15 degrees for Category A and B aeroplanes and by not more than 5 degrees for Category C and D aeroplanes; 		
b.	Instrument approaches flown using the CDFA technique with a nominal vertical profile, up to and including 4.5 degrees for Category A and B aeroplanes and 3.77 degrees for Category C and D aeroplanes, where the facilities are NDB, NDB/DME, VOR, VOR/DME, LLZ, LLZ/DME, VDF, SRA or RNAV/LNAV, with a final approach segment of at least 3 Nautical Miles (nm), which also fulfil the following criteria:		
	 The final approach track is offset by not more than 15 degrees for Category A and B aeroplanes and by not more than 5 degrees for Category C and D aeroplanes; 		
	 The Final Approach Fix (FAF), or another appropriate fix where descent is initiated is available, or distance to THR is available by FMS/RNAV or DME; 		
	iii. If the Missed Approach Point (MAPt) is determined by timing, the distance from FAF to THR is < 8 nm.		
c.	Instrument approaches where the facilities are NDB, NDB/DME, VOR, VOR/DME, LLZ, LLZ/DME, VDF, SRA or RNAV/LNAV, not fulfilling the criteria in 1.b. of AMC5 OPS.GEN.150, or with an MDH \geq 1 200 ft.		

A: Rule	B: Summary of comments	C: Reason for change, remarks
2. The missed approach, after an approach has been flown using the CDFA technique, should be executed when reaching the Decision Altitude/Height (DA/H) or the MAPt, whichever occurs first. The lateral part of the missed approach procedure should be flown via the MAPt unless otherwise stated on the approach chart.		
AMC6 OPS.GEN.150.A Instrument Flight Rules (IFR) operating minima	Request to publish all requirements of MDH/DH/RVR as IR;	After assessment with the Review Group, it was decided to keep the text as AMC.
AERODROME MINIMA – DETERMINATION OF RVR/CMV/VISIBILITY MINIMA FOR CATEGORY I APPROACH PROCEDURES WITH VERTICAL GUIDANCE AND NON-PRECISION APPROACH OPERATIONS – AEROPLANES		
1. The minimum RVR/CMV/visibility should be the highest of the values derived from Table 2 of AMC6-OPS.GEN.150.A (RVR/CMV vs DH/MDH) and Table 3 of AMC6 OPS.GEN.150.A (Minimum and maximum applicable RVR/CMV for all instrument approaches down to Category I minima (lower and upper cut-off limits), but not greater than the maximum values shown in Table 3 of AMC6 A OPS.GEN.150, where applicable.	IS (GA): request that current practices be permitted to continue, of non-commercial single-pilot operations, landing with RVR of 550 m without autopilot, providing touchdown zone (TDZ)/centreline lighting is available;	Text aligned with CAT rules.
2. The values in Table 2 of AMC6 OPS.GEN.150.A (RVR/CMV vs DH/MDH) are derived from the following formula:		
$\frac{\text{Required RVR/visibility (m)} = [(DH/MDH (ft) \times 0.3048)/tana] - \text{length of approach lights (m)}$		
Note 1:a is the calculation angle, being a default value of 3.00 degrees increasing in steps of 0.10 degrees for each line in Table 2 of AMC6 OPS.GEN.150.A up to 3.77 degrees and		

		Part-NCC	CRST	30 Aug 2011
A: Ru	ule		B: Summary of comments	C: Reason for change, remarks
		then remains constant.		
Note	2:	The values derived from the above formula have been rounded to the nearest 50 m up to a value of 800 m RVR and thereafter to the nearest 100 m.		
Note	3:	The DH/MDH intervals in Table 2 of AMC6 OPS.GEN.150.A have been selected to avoid anomalies caused by the rounding of the calculated Obstacle Clearance Altitude (OCA)/H. The height intervals are 10 feet up to a DH/MDH of 300 feet, 20 feet up to a DH/MDH of 760 feet and then 50 feet for DH/MDH above 760 feet.		
	The formula may be used with the actual approach slope and/or the actual length of the approach lights for a particular runway. This formula may also be used to calculate the applicable RVR for special (one-off) approach operations in accordance with AMC8 A OPS.GEN.150 4. The formula may also be used to calculate the applicable RVR value for approaches with approach slopes greater than 4.5 degrees.			
	If the approach is flown with a level flight segment at or above MDA/H, 200 m should be added for Category A and B aeroplanes and 400 m for Category C and D aeroplanes to the minimum RVR/CMV value resulting from the application of Table 2 of AMC6 OPS.GEN.150.A (RVR/CMV vs DH/MDH) and Table 3 of AMC6 OPS.GEN.150.A (Minimum and maximum applicable RVR/CMV for all instrument approaches down to Category I minima (lower and upper cut-off limits)The added value corresponds to the time/distance required to establish the aeroplane on the final descent.		IS (BA): Request to harmonise item 4 with AMC2 150 item 3 and amend text: " (lower and upper cut-off limits) provided the resulting RVR/CMV value does not exceed 5 000 m";	Accepted.

			-	5		
A: F	Rule		B: Summary of comments	C: Reason for change, remarks		
5.		XVR of less than 750 m as indicated in Table 2 of AMC6 GEN.150.A (RVR/CMV vs DH/MDH) may be used: for Category I approach operations to runways with Full Approach Light System (FALS), Runway Touchdown Zone Lights (RTZL) and Runway Centreline Lights (RCLL), provided	Request to correct (5)(a) to "DH is not less than 200 ft";	Accepted.		
	b. c.	that the DH is not more than 200 ft; for Category I approach operations to runways without RTZL and RCLL when using an approved HUDLS, or equivalent approved system, or when conducting a coupled approach or flight-director-flown approach to a DH equal to or greater than 200 ft. The ILS must not be promulgated as a restricted facility. The equivalent system could for instance be an approved HUD which is not certificated as a landing system, but is able to provide adequate guidance cues. Other devices may also be suitable, such as Enhanced/Synthetic Vision Systems (E/SVS) or other hybrids of such devices; or for APV approach operations to runways with FALS, RTZL and RCLL when using an approved HUD.				
6.	OPS.	values lower than those given in Table 2 of AMC6 GEN.150.A (RVR/CMV vs DH/MDH) may be used for HUDLS and land operations in accordance with Part OPS.SPA.LVO.				
7.	appro runwa RCLL)	visual aids comprise standard runway day markings and bach and runway lighting (runway edge lights, threshold lights, ay end lights and in some cases also touch-down zone and/or). The approach light configurations acceptable are classified listed in Table 1 of AMC6 OPS.GEN.150.A (Approach light ms).				

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A: F	Rule		B: Summary of comments	C: Reason for change, remarks
8.	values relevant to a Basic Ap used on runways where the	nents in AMC6 OPS.GEN.150.A 7. RVR proach Lighting System (BALS) may be approach lights are restricted in length or water, but where at least one cross-		
9.	and approach lights is re-	any operation where credit for runway quired, the lights must be on and ovided for in Table 1 of AMC12		
Tab	ble 1 of AMC6 OPS.GEN.150.4	A Approach light systems		
CL	CLASS OF FACILITY LENGTH, CONFIGURATION		Eurocontrol: ICAO CAT I lighting requirement length = 900 m, not	Accepted. The new table only provides the applicable values for
	ll Approach Landing System ALS)	Ch Landing System Lighting System (HIALS ≥ 720 m) Distance Coded Centreline, Barrette Centreline	720 m (the latter is taken from JAR- OPS);	clarity.
		FAA: ALSF1, ALSF2, SSALR, MALSR, high or medium intensity and/or flashing lights, 720 m or more		
	ermediate Approach Light tem (IALS) ICAO: Simple Approach Lighting System (HIALS 420 – 719 m) Single Source, Barrette			
		FAA: MALSF, MALS, SALS/SALSF, SSALF, SSALS, high or medium		

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A: Rule					B: 9	Summary of comments	C: Reason for change, remarks	
			ensity and/or 1 9 m	flashing lights,	420 -	-		
Basic Approach Light System (BALS)			y other Approa IALS, MIALS o A: ODALS, hig ensity or flash	r ALS 210-419 h or medium	m)			
No Approach Lig	No Approach Light System (NALS) Any other Approach Lighting S (HIALS, MIALS or ALS <210 m Approach Lights							
Table 2 of AMC6	OPS.GEN.15	D.A RVR/CN	1V vs DH/MDH					
DH or MDH	CLASS O	CLASS OF LIGHTING FACILITY				MS: request to correct errors in table	Accepted.	
	FALS	IALS	BALS	BALS NALS			and use figures in table 5 of Appendix 1 to EU-OPS 1.430(d);	
ft	m							
200 - 210	550	750	1 000	1 200				
211 - 220 550 80		800	1 000	1 200				
221 - 230	550	800	1 000	1 200				
231 - 240	550	800	1 000	1 200				
241 - 250	550	800	1 000	1 300				

A: Rule					B: Summary of comments	C: Reason for change, rem
251 - 260	600	800	1 100	1 300		
261 - 280	600	900	1 100	1 300		
281 - 300	650	900	1 200	1 400		
301 - 320	700	1 000	1 200	1 400		
321 - 340	800	1 100	1 300	1 500		
341 - 360	900	1 200	1 400	1 600		
361 - 380	1 000	1 300	1 500	1 700		
381 - 400	1 100	1 400	1 600	1 800		
401 - 420	1 200	1 500	1 700	1 900		
421 - 440	1 300	1 600	1 800	2 000		
441 - 460	1 400	1 700	1 800	2 100		
461 - 480	1 500	1 800	1 900	2 200		
481 - 500	1 500	1 800	2 000	2 300		
501 - 521	1 600	1 900	2 100	2 400		
521 - 540	1 700	2 000	2 200	2 400		
541 - 560	1 800	2 100	2 300	2 500		

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A: Rule						B: Summary of comments	C: Reason for change, remarks
561 - 580	1 900	2 200	2 400	2 600			
581 - 600	2 000	2 300	2 500	2 700			
601 - 620	2 100	2 400	2 600	2 800			
621 - 640	2 200	2 500	2 700	2 900			
641 - 660	2 300	2 600	2 800	3 000			
661 - 680	2 400	2 700	2 900	3 100			
681- 700	2 500	2 800	3 000	3 200			
701 - 720	2 600	2 900	3 100	3 300			
721 - 740	2 700	3 000	3 200	3 400			
741 - 760	2 700	3 000	3 300	3 500			
761 - 800	2 900	3 200	3 400	3 600			
801 - 850	3 100	3 400	3 600	3 800			
851 - 900	3 300	3 600	3 800	4 000			
901 - 950	3 600	3 900	4 100	4 300			
951 - 1 000	3 800	4 100	4 300	4 500			
1 001 - 1 100	4 100	4 400	4 600	4 900			

A: Rule		//					
						B: Summary of comments	C: Reason for change
1 101 – 1 200	4 600	4 900	5 000	5 000)		
1 201 and above	e 5 000	5 000	5 000	5 000)		
	OPS.GEN.15 IV for all ins (lower and	trument ap	proaches do				
	RVR/CMV	AEROPL/	ANE CATEG	ORY			
ONDITIONS	(m)	Α	В	С	D		
, ,	Min	According	to Table 2	of AMC6A O	PS.GEN.15	0	
GLS, PAR and	Max	1 500	1 500	2 400	2 400		
NDB,	Min	750	750	750	750		

					•			-
A: Rule	: Rule						: Summary of comments	C: Reason for change, remark
1.b.								
For NDB,	Min	1 000	1 000	1 200	1 200			
NDB/DME, VOR, VOR/DME, LLZ, LLZ/DME, VDF, SRA, RNAV/LNAV: not fulfilling the criteria in AMC5 OPS.GEN.150 1.b., or with a DH or MDH \geq 1200 ft	Max	if flown usi an add-on in Table 2	ng the CDFA of 200/400 of AMC6 OP	of AMC6 OPS A technique, m applies to S.GEN.150.A eding 5000	otherwise the value but not to	s		
approache and its AM a. An OPS pro	es should be IC material: RVR of less t S.GEN.150.A	ations, the calculated in than 800 m as may be used the following	n accordanc s indicated i l for Catego	n Table 2 of ry I approacl	.GEN.150 AMC6 nes			
i.		autopilot, cou Igated as res	•	ILS or MLS v	vhich is			

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A: Rule					B: Sun	nmary of con	nments	C: Reason for	change, remarks
		DLS (including, whe		æ,					
		CLL are not available e less than 600 m;	•	ım					
OPS.GEN.150.A marked RTZL and RCLL wh	OPS.GEN.150.A may be used for APV operations to runways with FALS, RTZL and RCLL when using an approved HUDLS, or equivalent approved system, or when conducting a coupled approach to a DH equal to or								
AMC7 OPS.GEN.1 minima	.50.H Instrume	nt Flight Rules (1	(FR)operatin	g					
AERODROME MINI MINIMA FOR CATE GUIDANCE AND NO	GORY I, APPROA	CH PROCEDURES	WITH VERTIC						
Performance		ches by helico he minima given	•						
Table 1 of AMC7 O	PS.GEN.150.H O	nshore non-precisi	on approach r	ninima					
ONSHORE NON-PRECISION APPROACH MINIMA (Notes 5, 6 and 7)									
MDH (ft)	MDH (ft) Facilities/RVR (m)								
	Full	Intermediate	Basic	Nil					

A: Rule	A: Rule						of comments	C: Reason for change,
		(Note 1)	(Note 2)	(Note 3)	(Note	4)		
250 - 29	9	600	800	1 000	1 000			
300 - 44	9	800	1 000	1 000	1 000			
450 and a	above	1 000	1 000	1 000	1 000			
Note 1:	of HI/M	I approach light	FATO/runway ma ts, FATO/runway y end lights. Ligh	edge lights, thr				
Note 2:	ote 2: Intermediate facilities comprise FATO/runway markings, 420 - 719 m of HI/MI approach lights, FATO/runway edge lights, threshold lights and FATO/runway end lights. Lights must be on.				hts,			
Note 3:	<i>Te 3:</i> Basic facilities comprise FATO/runway markings, <420 m of HI/MI approach lights, any length of LI approach lights, FATO/runway edge lights, threshold lights and FATO/runway end lights. Lights must be on.				,			
Note 4:	e 4: Nil approach light facilities comprise FATO/runway markings, FATO/runway edge lights, threshold lights, FATO/runway end lights or no lights at all.							
Note 5:	a nomii descent	nal descent slop t slopes will usu	plicable to conver e of not greater t ally require that also visible at th	han 4 degrees. visual glide slop	Greater			

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A: Rule		B: Summary of comments	C: Reason for change, remarks
Note 6:	The above figures are either reported RVR or CMV.		
Note 7:	The MDH mentioned in Table 1 of AMC7 OPS.GEN.150.H refers to the initial calculation of MDH. When selecting the associated RVR, there is no need to take account of a rounding up to the nearest ten feet, which may be done for operational purposes, e.g. conversion to MDA.		
a.	Where the missed approach point is within ½ nm of the landing threshold, the approach minima given for full facilities may be used regardless of the length of approach lighting available. However, FATO/runway edge lights, threshold lights, end lights and FATO/runway markings are still required;		
b.	For night operations, ground lighting must be available to illuminate the FATO/runway and any obstacles;		
c.	For single-pilot operations, the minimum RVR is 800 m or the minima in Table 1 of AMC7 OPS.GEN.150.H, whichever is higher.		
Cla	Category I approaches by helicopters operated in Performance ss 1 or 2, the minima given in Table 2 of AMC7 OPS.GEN.150.H ould apply:		
Table 2 c	of AMC7 OPS.GEN.150.H Onshore precision approach minima – Category I		

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A: Rule						В:
ONSHOR	RE PRECIS	SION APPROA	.CH MINIMA – CATE	GORY I (Note	es 5, 6 and 3	7)
MDH (ft	t)	Facilities/	RVR (m)			
		Full (Note 1)	Intermediate (Note 2)	Basic (Note 3)	Nil (Note 4)	
200		500	600	700	1 000	
201 - 25	50	550	650	750	1 000	
251 - 30	00	600	700	800	1 000	
301 and	above	750	800	900	1 000	
Note 1:	of HI/M	II approach lig	se FATO/runway ma ghts, FATO/runway way end lights. Ligh	edge lights, t	threshold	
Note 2:	719 m	of HI/MI appı	es comprise FATO/ru roach lights, FATO/r FATO/runway end li	unway edge	lights,	
Note 3:	HI/MI a FATO/r	approach light	rise FATO/runway m ts, any length of LI o ights, threshold ligh ust be on.	approach ligh	nts,	

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A: Rule		B: Summary of comments	C: Reason for change, remarks
Note 4:	Nil approach light facilities comprise FATO/runway markings, FATO/runway edge lights, threshold lights, FATO/runway end lights or no lights at all.		
Note 5:	The above figures are either the reported RVR or CMV.		
Note 6:	The table is applicable to conventional approaches with a glide slope up to and including 4 degrees.		
Note 7:	The MDH mentioned in Table 2 of AMC7 OPS.GEN.150.H refers to the initial calculation of MDH. When selecting the associated RVR, there is no need to take account of a rounding up to the nearest ten feet, which may be done for operational purposes, e.g. conversion to DA.		
a.	For night operations, ground lighting must be available to illuminate the FATO/runway and any obstacles;		
b.	For-single pilot operations, the minimum RVR should be calculated based on OPS.GEN.150 and its AMC material. An RVR of less than 800 m should not be used except when using a suitable autopilot coupled to an ILS or MLS, in which case normal minima apply. The DH applied should not be less than 1.25 times the minimum use height for the autopilot.		
AMC8 OF minima	PS.GEN.150.A Instrument Flight Rules (IFR) operating	Request to upgrade to an IR;	Accepted. In the interest of safety is was decided to upgrade this AMC to the IR level, NCC.OP.112.

		Part-NCC	CRST	30 Aug 2011
A: F	Rule		B: Summary of comments	C: Reason for change, remarks
AER	RODRO	DME MINIMA – CIRCLING – AEROPLANES		
1.	The	MDH for circling should be the highest of:		
	a.	the published circling OCH for the aeroplane category;		
	b.	the minimum circling height derived from Table 1 of AMC8 OPS.GEN.150.A; or		
	c.	the DH/MDH of the preceding instrument approach procedure.		
2.	aero	MDA for circling should be calculated by adding the published odrome elevation to the MDH, as determined by AMC8 S.GEN.150.A 1.		
3.	The	minimum visibility for circling should be the higher of:		
	a.	the circling visibility for the aeroplane category, if published;		
	b.	the minimum visibility derived from Table 1 of AMC8 OPS.GEN.150.A; or		
	c.	the RVR/CMV derived from Tables 2 and 3 of AMC6 OPS.GEN.150.A for the preceding instrument approach procedure.		
4.	and mai abo taki	withstanding the requirements in AMC8 OPS.GEN.150.A 3. above limited to locations where there is a clear public interest to intain current operations, the visibility may not be increased we the values derived from Table 1 of AMC8 OPS.GEN.150.A, ing into account the operator's experience, training programme flight crew qualification.		

				Part-NCC	C CRST	30 Aug 2011
A: Rule					B: Summary of comments	C: Reason for change, remarks
Table 1 of AMC8 (vs. aero	OPS.GEN.150 oplane catego		visibility and	MDH for circling		
	AEROPLA	ANE CATEGO	RY			
	Α	В	С	D		
MDH (ft)	400	500	600	700		
Minimum meteorological visibility (m)	1 500	1 600	2 400	3 600		
-	n prescribed this paragrap		accepted pro	ocedure within th	e	
AMC9 OPS.GEN. minima	150.H Instr	ument Fligh	t Rules (IFF	R) operating		In the interest of safety is was decided to upgrade this AMC to the IR level, NCC.OP.113.
AERODROME MIN	IMA – ONSH	ORE CIRCLIN	G – HELICOP	TERS		
instrument	approach, to	bring an airc	raft into posi	sual phase of a tion for landing o for a straight i	1	
-	•	MDH should r not less than a		an 250 ft, and th	e	

	Part	CRST	30 Aug 2011	
A: Rule		B: Summary of comments	C: Reason for change, remarks	
3. Visual manoeuvring (circling) wir procedure within the meaning of	-	cepted		
AMC10 OPS.GEN.150 Instrument F minima	light Rules (IFR) operating			
AERODROME MINIMA – VISUAL APPRO	ACH			
An RVR of less than 800 m should not	be used for a visual approach.			
AMC11 OPS.GEN.150 Instrument F minima	light Rules (IFR) operating			
AERODROME MINIMA – CONVERSION VISIBILITY TO RVR/CMV	OF REPORTED METEOROLOGIC	CAL		
 A conversion from meteorological be used for calculating take-off n when reported RVR is available above the maximum value assess "RVR more than 1 500 m", it i value for the purpose of this para 	ninima, Category II or III mini e. If the RVR is reported as sed by the aerodrome operator is not considered to be a rep	ma or being r, e.g.		
2. For all other circumstances, Tabl be used.	le 1 of AMC11 OPS.GEN.150 s	should		
Table 1 of AMC11 OPS.GEN.150 Conve RVR/CMV	rsion of meteorological visibilit			
	/CMV = reported eorological visibility x			

		Pa	CRST	30 Aug 2011	
A: Rule			B: Summary of comments	C: Reason for change, remarks	
	Day	Night			
HI approach and runway lighting	1.5	2.0			
Any type of lighting installation other than above	1.0	1.5			
No lighting	1.0	n/a			
MINIMA – EFFECT	AMC12 OPS.GEN.150 Instrument Flight Rules (IFR) operating minima AERODROME MINIMA – EFFECT ON LANDING MINIMA OF TEMPORARILY FAILED OR DOWNGRADED GROUND EQUIPMENT				Text aligned with the CAT rule.
1. These instructions are intended for use both pre-flight and in-flight. It is however not expected that the pilot-in-command would consult such instructions after passing the outer marker or equivalent position. If failures of ground aids are announced at such a late stage, the approach could be continued at the pilot-in-command's discretion. If failures are announced before such a late stage in the approach, their effect on the approach should be considered as described in Table 1 of AMC12 OPS.GEN.150, and the approach may have to be abandoned to allow this to happen.				"outer marker or equivalent position 1 000 ft above the aerodrome";	Accepted.
 Conditions applicable to Table Multiple failures of in Table 1 of AMC1 	runway/ FATO ligl 2 OPS.GEN.150 m	nts other than indi ay not be accepta	ıble;	1/ IS (BA): request that CAT II and CAT III operations be handled in Subpart SPA;	Accepted. Operations requiring an approval in accordance with SPA.LVO are covered in Part-SPA.
b. Deficiencies of app	roach and runway	/FATO lights are t	reated		

		Part-NCC	CRST	30 Aug 2011
A: Rule			B: Summary of comments	C: Reason for change, remarks
FATO/run permitteo d. Failures o	II or III operations. A comb way lights and RVR assess d; other than ILS affect RVR or 5.GEN.150 Failed or downgr	ment equipment is not nly and not DH.		Table aligned with the CAT text.
FAILED OR DOWNGRADED EQUIPMENT ILS Standby Transmitter	DOWNGRADED EQUIPMENT Category I APV & Non-Press ILS Standby			
Outer Marker	No effect if replaced by equivalent position	APV – not applicable NPA with FAF: no effect unless used as FAF. If the FAF cannot be identified (e.g. no method available for timing of descent), non- precision operations cannot be conducted		
Middle Marker	No effect	No effect unless used as MAPt		

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A: Rule			B: Summary of comments	C: Reason for change, remarks
RVR Assessment Systems	On runways equipped with 2 or more RVR Assessment Units; one may be inoperative	No effect		
Approach lights	Not permitted	Minima as for NALS		
Approach lights except the last 210 m	Not permitted	Minima as for BALS		
Approach lights except the last 420 m	No effect	Minima as for IALS		
Standby power for approach lights	No effect			
Edge lights, threshold lights and runway end lights	Day – no effect Night – not permitted			
Centreline lights	No effect if F/D, HUDLS or auto-land otherwise RVR 750 m	No effect		
Centreline lights spacing increased to 30 m	No effect			

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A: Rule			B: Summary of comments	C: Reason for change, remarks
lights	No effect if F/D, HUDLS or auto-land otherwise RVR 750 m	No effect		
Taxiway light No effect system				
GM1 OPS.GEN.150.A Instrument Flight Rules (IFR) operating minima				
AERODROME MINIMA – AEROPLANE CATEGORIES				The amended text is also applicable to helicopters.
1. The criteria taken into consideration for the classification of aeroplanes by categories is the indicated air speed at threshold (V_{AT}) which is equal to the stalling speed (V_{SO}) multiplied by 1.3 or V_{S1G} multiplied by 1.23 in the landing configuration at the maximum certificated landing mass. If both V_{SO} and V_{S1G} are available, the higher resulting V_{AT} should be used. The aeroplane categories corresponding to V_{AT} values are in the Table 1 of GM1 OPS.GEN.150.A.				
Table 1 of GM1 OPS.GEN.150.A Aeroplane categories corresponding to V_{AT} values				
AEROPLANE CATEGO	DRY V _{AT}			
А	Less than 91 kts			
В	91 – 120 kts			

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A: Rule	e			B: Summary of comments	C: Reason for change, remarks
С		121 – 140 kts			
D		141 – 165 kts			
E		166 – 210 kts			
 The landing configuration which is to be taken into consideration should be defined by the operator or by the aeroplane manufacturer. 					
 3. Permanent change of category (maximum landing mass): a. An operator may impose a permanent, lower, landing mass, and use this mass for determining the V_{AT}; b. The category defined for a given aeroplane should be a permanent value and thus independent of the changing conditions of day-to-day operations; c. The category should be stated in the operations manual, where required. 			ver, landing mass, ; e should be a the changing		Not accepted. This text has not been transposed because it does not qualify for a GM.
GM2 OF minima	PS.GEN.150.A Instrun a	nent Flight Rules (IFF	R) operating		
AERODROME MINIMA - CONTINUOUS DESCENT FINAL APPROACH (CD – AEROPLANES		APPROACH (CDFA)		The NPA text confused APV operations with non-precision approach operations flown with the CDFA technique. The amended text provides a clear distinction between these operations.	

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A: I	Rule		B: Summary of comments	C: Reason for change, remarks		
1.	Intro	oduction:				
	a.	Controlled Flight Into Terrain (CFIT) is a major causal category of accident and hull loss in commercial aviation. Most CFIT accidents occur in the final approach segment of non-precision approaches; the use of stabilised-approach criteria on a continuous descent with a constant, pre- determined vertical path is seen as a major improvement in safety during the conduct of such approaches. Operators should ensure that the following techniques are adopted as widely as possible, for all approaches;				
	b.	The elimination of level flight segments at MDA close to the ground during approaches, and the avoidance of major changes in attitude and power/thrust close to the runway which can destabilise approaches, are seen as ways to reduce operational risks significantly;				
	c.	For completeness this guidance also includes criteria which should be considered to ensure the stability of an approach (in terms of the aeroplanes energy and approach-path control);				
	d.	The term CDFA has been selected to cover a technique for any type of non-precision approach;				
	e.	Non-precision approaches operated other than using a constant pre-determined vertical path or when the facility requirements and associated conditions do not meet the conditions specified in 2.d., RVR penalties apply. However,				

ary of comments C: Reason for change, remarks
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A: F	Rule		B: Summary of comments	C: Reason for change, remarks
		vii. The Approach profile affords reduced noise levels;		
		viii. The technique affords procedural integration with APV approach operations;		
		 ix. When used and the approach is flown in a stabilised manner is the safest approach technique for all approach operations. 		
2.	CDFA	\:		
	a.	Continuous Descent Final Approach. A specific technique for flying the final approach segment of a non-precision instrument approach procedure as a continuous descent, without level-off, from an altitude/height at or above the final approach fix altitude/height to a point approximately 15 m (50 ft) above the landing runway threshold or the point where the flare manoeuvre should begin for the type of aircraft flown;		
	b.	An approach is only suitable for application of CDFA technique when it is flown along a pre-determined vertical approach slope which follows a:		
		 Designated Vertical Profile: A continuous vertical approach profile which forms part of the approach procedure design. APV is considered to be an approach with a designated vertical profile; or a 		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
	 ii. Nominal Vertical Profile: A vertical profile not formin part of the approach procedure design, but which can be flown as a continuous descent. The nominal vert profile information may be published or displayed (a the approach chart) to the pilot by depicting the nominal slope or range/distance vs height. Approact with a nominal vertical profile are considered to be: A. NDB, NDB/DME; B. VOR, VOR/DME; C. LLZ, LLZ/DME; D. VDF, SRA or 	an cical on ches	
с.	E. RNAV/LNAV; Stabilised Approach (SAp). An approach which is flown in controlled and appropriate manner in terms of configurat energy and control of the flight path from a pre-determin point or altitude/height down to a point 50 feet above the threshold or the point where the flare manoeuvre is initia if higher:	rion, ned e	
	 The control of the descent path is not the only consideration when using the CDFA technique. Cont of the aeroplane's configuration and energy is also to the safe conduct of an approach; 		
	 The control of the flight path, described above as or the requirements for conducting an SAp, should not confused with the path requirements for using the 0 	t be	

A: Rule		B: Summary of comments	C: Reason for change, remarks
	technique. The pre-determined path requirements for conducting an SAp are established by the operator and published in the operations manual part B; guidance for conducting SAp operations is given in 5.;		
	 iii. The predetermined approach slope requirements for applying the CDFA technique are established by the following: A. The instrument-procedure design when the approach has a designated vertical profile; 		
	 B. The published `nominal' slope information when the approach has a nominal vertical profile; C. The designated final-approach segment minimum of 3 nm, and maximum, when using timing techniques, of 8 nm; 		
	 iv. A Stabilised Approach will never have any level segment of flight at DA/H (or MDA/H as applicable). This enhances safety by mandating a prompt go-around manoeuvre at DA/H (or MDA/H); 		
	 An approach using the CDFA technique will always be flown as an SAp, since this is a requirement for applying CDFA; however, an SAp does not have to be flown using the CDFA technique, for example a visual approach; 		
d.	Approach with a designated vertical profile using the CDFA technique:		

A: Rule			B: Summary of comments	C: Reason for change, remarks
	i.	The optimum angle for the approach slope is 3 degrees, and the gradient should preferably not exceed 6.5 percent which equates to a slope of 3.77 degrees, (400 ft/nm) for procedures intended for conventional aeroplane types/classes and/or operations. In any case, conventional approach slopes should be limited to 4.5 degrees for Category A and B aeroplanes and 3.77 degrees for Category C and D aeroplanes, which are the upper limits for applying the CDFA technique. A 4.5 degree approach slope is the upper limit for certification of conventional aeroplanes;		
	ii.	The approach is to be flown utilising operational flight techniques and on board navigation system(s) and navigation aids to ensure it can be flown on the desired vertical path and track in a stabilised manner, without significant vertical path changes during the final- segment descent to the runway. APV is included;		
	iii.	The approach is flown to a DA/H;		
	iv.	No MAPt is published for these procedures;		
e.		roach with a nominal vertical profile using the CDFA inique:		
	i.	The optimum angle for the approach slope is 3 degrees, and the gradient should preferably not exceed 6.5 percent which equates to a slope of 3.77 degrees, (400 ft/nm) for procedures intended for conventional aeroplane types / class and / or operations. In any case, conventional approaches should be limited to 4.5		

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		degrees for Category A and B aeroplanes and 3.77 degrees for Category C and D aeroplanes, which are the upper limits for applying CDFA technique. A 4.5 degree approach slope is the upper limit for certification of conventional aeroplanes;		
	ii.	The approach procedure should meet at least the following facility requirements and associated conditions. NDB, NDB/DME, VOR, VOR/DME, LLZ, LLZ/DME, VDF, SRA, RNAV/LNAV) and fulfil the following criteria:		
		A. The final approach track off-set \leq 5 degrees except for Category A and B aeroplanes, where the approach-track off-set is \leq 15 degrees; and		
		 A FAF, or another appropriate fix where descent initiated is available; and 		
		C. The distance from the FAF to the Threshold (THR) is less than or equal to 8 nm in the case of timing; or		
		D. The distance to the THR is available by FMS/RNAV or DME; or		
		E. The minimum final-segment of the designated constant angle approach path should not be less than 3 nm from the THR unless approved by the authority;		
	iii.	CDFA may also be applied utilising the following:		
		A. RNAV/LNAV with altitude/height cross checks		

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	against positions or distances from the THR; or B. Height crosscheck compared with DME distance values;		
	iv. The approach is flown to a DA/H;		
	v. The approach is flown as an SAp.		
	Generally, an MAPt is published for these procedures.		
3. Oper	rational procedures:		
a.	An MAPt should be specified to apply CDFA with a nominal vertical profile as for any non-precision approach;		
b.	The flight techniques associated with CDFA employ the use of a predetermined approach slope. The approach, in addition, is flown in a stabilised manner, in terms of configuration, energy and control of the flight path. The approach should be flown to a DA/H at which the decision to land or go-around is made immediately. This approach technique should be used when conducting:		
	all NPAs meeting the specified CDFA criteria in 2.d.; andall approaches categorised as APV;		
c.	The flight techniques and operational procedures prescribed above should always be applied; in particular with regard to		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
	control of the descent path and the stability of the aeroplane on the approach prior to reaching MDA/H. Level flight at MDA/H should be avoided as far as practicable. In addition appropriate procedures and training should be established and implemented to facilitate the applicable elements of 4., 5. and 6. Particular emphasis should be placed on 4.h., 5.a. to g. and 8.d.;		
d.	In cases where the CDFA technique is not used with high MDA/H, it may be appropriate to make an early descent to MDA/H with appropriate safeguards to include the above training requirements, as applicable, and the application of a significantly higher RVR/Visibility;		
e.	For circling approaches (Visual Manoeuvring), all the applicable criteria with respect to the stability of the final descent path to the runway should apply. In particular, the control of the desired final nominal descent path to the threshold should be conducted to facilitate the techniques described in 4. and 5.:		
	 Stabilisation during the final straight-in segment for a circling approach should ideally be accomplished by 1000 ft above aerodrome elevation for turbo-jet aeroplanes; 		
	ii. For a circling approach where the landing runway threshold and appropriate visual landing aids may be visually acquired from a point on the designated or published procedure (prescribed tracks), stabilisation should be achieved not later than 500 ft above		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	aerodrome elevation. It is however recommended that the aeroplane be stabilised when passing 1000 ft above aerodrome elevation;		
111.	When a low-level final turning manoeuvre is required in order to align the aeroplane visually with the landing runway, a height of 300 ft above the runway threshold elevation, or aerodrome elevation as appropriate, should be considered as the lowest height for approach stabilisation with wings level;		
iv.	Dependent upon aeroplane type/class the operator may specify an appropriately higher minimum stabilisation height for circling approach operations;		
v.	The operator should specify in the operations manual the procedures and instructions for conducting circling approaches, including at least:		
	A. the minimum required visual reference;		
	 B. the corresponding actions for each segment of the circling manoeuvre; 		
	C. the relevant go-around actions if the required visual reference is lost; and		
	 D. the visual reference requirements for any operations with a prescribed track circling manoeuvre to include the MDA/H and any published MAPt; 		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
f.	Visual approach. All the applicable criteria with respect to the stability of the final descent path to the runway should apply to the operation of visual approaches. In particular, the control of the desired final nominal descent path to the threshold should be conducted to facilitate the appropriate techniques and procedures described in 6. and 7.:		
	 Stabilisation during the final straight-in segment for a visual approach should ideally be accomplished by 500 ft above runway threshold elevation for turbo-jet aeroplanes; 		
	 When a low level final turning manoeuvre is required in order to align the aeroplane with the landing runway, a minimum height of 300 ft above the runway threshold elevation (or aerodrome elevation as appropriate) should be considered as the lowest height for visual approach stabilisation with wings level; 		
	Dependent upon aeroplane type/class, the operator may specify an appropriately higher minimum stabilisation height for visual approach operations;		
	 iv. The operator should specify in the operations manual the procedures and instructions for conducting visual approaches to include at least: 		
	A. the minimum required visual reference;B. the corresponding actions if the required visual reference is lost during a visual approach		

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A: Rule		B: Summary of comments	C: Reason for change, remarks
	manoeuvre; and		
	C. the appropriate go-around actions;		
g.	The control of the descent path using the CDFA technique ensures that the descent path to the runway threshold is flown using either:		
	 a variable descent rate or flight path angle to maintain the desired path, which may be verified by appropriate crosschecks; 		
	 a pre-computed constant rate of descent from the FAF, or other appropriate fix which is able to define a descent point and/or from the final approach segment step-down fix; or 		
	iii. vertical guidance, including APV;		
	The above techniques also support a common method for the implementation of flight-director-guided or auto-coupled RNAV/VNAV or GLS approaches;		
h.	The manoeuvre associated with the vertical profile of the missed approach should be initiated not later than reaching the MAPt or the DA/H specified for the approach, whichever occurs first. The lateral part of the missed approach procedure must be flown via the MAPt unless otherwise stated on the approach chart;		
i.	In case the CDFA technique is not used the approach should be flown to an altitude/height at or above the MDA/H where a level flight segment at or above MDA/H may be flown to the		

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A: Ru	le			B: Summary of comments	C: Reason for change, remarks
		MAP	rt;		
j	j.	appr ensu subs	ase the CDFA technique is not used when flying an roach, an operator should implement procedures to ure that early descent to the MDA/H will not result in a sequent flight below MDA/H without adequate visual rence. These procedures could include:		
		i.	awareness of radio altimeter information with reference to the approach profile;		
		ii.	enhanced Ground Proximity Warning System and/or Terrain Awareness information;		
		iii.	limitation of rate of descent;		
		iv.	limitation of the number of repeated approaches;		
		v.	safeguards against too early descents with prolonged flight at MDA/H; and		
		vi.	specification of visual requirements for the descent from the MDA/H.		
4. F	Flight	techr	niques:		
ā	a.	non	CDFA technique can be used on almost any published -precision approach when the control of the descent path ded by either:		
		i.	a recommended descent rate, based on estimated ground speed, which may be provided on the approach		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	chart; or ii. the descent path as depicted on the chart;		
b.	In order to facilitate the requirement of 4.a.ii., the operator should either provide charts which depict the appropriate cross check altitudes/heights with the corresponding appropriate range information, or such information should be calculated and provided to the flight crew in an appropriate and usable format;		
c.	For approaches flown coupled to a designated descent path using computed electronic glide-slope guidance (normally a 3 degree path), the descent path should be appropriately coded in the flight management system data base and the specified navigational accuracy (RNP) should be determined and maintained throughout the operation of the approach;		
d.	With an actual or estimated ground speed, a nominal vertical profile and required descent rate, the approach should be flown by crossing the FAF configured and on-speed. The tabulated or required descent rate is established and flown to not less than the DA/H, observing any step-down crossing altitudes if applicable;		
e.	To assure the appropriate descent path is flown, the pilot not flying should announce crossing altitudes as published fixes and other designated points are crossed, giving the appropriate altitude or height for the appropriate range as depicted on the chart. The pilot flying should promptly adjust the rate of descent as appropriate;		

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f.	With the required visual reference requirements established, the aeroplane should be in position to continue descent through the DA/H or MDA/H with little or no adjustment to attitude or thrust/power;		
g.	When applying CDFA on an approach with a nominal vertical profile to a DA/H, it may be necessary to apply an add-on to the published minima (vertical profile only) to ensure sufficient obstacle clearance. The add-on, if applicable, should be published in the operations manual – (Aerodrome Operating Minima). However, the resulting procedure minimum will still be referred to as the DA/H for the approach;		
h.	Operators should establish a procedure to ensure that an appropriate callout (automatic or oral) is made when the aeroplane is approaching DA/H. If the required visual references are not established at DA/H, the missed approach procedure is to be executed promptly. Visual contact with the ground alone is not sufficient for continuation of the approach. With certain combinations of DA/H, RVR and approach slope, the required visual references may not be achieved at the DA/H in spite of the RVR being at or above the minimum required for the conduct of the approach. The safety benefits of CDFA are negated if prompt go-around action is not initiated;		
i.	The following bracketing conditions in relation to angle of bank, rate of descent and thrust/power management are considered to be suitable for most aeroplane types/class to		

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	ensure the predetermined vertical path approach is conducted in a stabilised manner:		
	i. Bank angle: As prescribed in the operations manual, should generally be less than 30 degrees;		
	ii. Rate of descent (ROD): The target ROD should not exceed 1000 fpm). The ROD should deviate by no more than + 300 fpm from the target ROD. Prolonged rates of descent which differ from the target ROD by more than 300 fpm indicate that the vertical path is not being maintained in a stabilised manner. The ROD should not exceed 1200 fpm, except under exceptional circumstances which have been anticipated and briefed prior to commencing the approach; for example, a strong tailwind. Zero rate of descent may be used when the descent path needs to be regained from below the profile. The target ROD may need to be initiated prior to reaching the required descent point (typically 0.3 nm before the descent point, dependent upon ground speed, which may vary for each type/class of aeroplane). (Refer to 4.i.iii.);		
	iii. Thrust/power management: The limits of thrust/power and the appropriate range should be specified in the operations manual Part B or equivalent documents;		
j.	Transient corrections/overshoots: The above-specified range of corrections should normally be used to make occasional momentary adjustments in order to maintain the desired path and energy of the aeroplane. Frequent or sustained		

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		overshoots should require the approach to be abandoned and a go-around initiated. A correction philosophy should be applied similar to that described in 5.;		
	k.	The relevant elements of 4. should, in addition, be applied to approaches not flown using the CDFA technique; the procedures thus developed, thereby ensure a controlled flight path to MDA/H. Dependent upon the number of step down fixes and the aeroplane type/class, the aeroplane should be appropriately configured to ensure safe control of the flight path prior to the final descent to MDA/H.		
5.		lisation of energy/speed and configuration of the aeroplane on pproach:		
	a.	The control of the descent path is not the only consideration. Control of the aeroplane's configuration and energy is also vital to the safe conduct of an approach;		
	b.	The approach should be considered to be fully stabilised when the aeroplane is:		
		i. tracking on the required approach path and profile;		
		ii. in the required configuration and attitude;		
		iii. flying with the required rate of descent and speed; and		
		iv. flying with the appropriate thrust/power and trim;		
	C.	The following flight path control criteria should be met and maintained when the aeroplane passes the gates described in 6. and 7.;		

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d.	The aeroplane is considered established on the required approach path at the appropriate energy for stable flight using the CDFA technique when:		
	 it is tracking on the required approach path with the correct track set, approach aids tuned and identified as appropriate to the approach type flown and on the required vertical profile; and 		
	 it is at the appropriate attitude and speed for the required target ROD with the appropriate thrust/power and trim; 		
e.	It is recommended to compensate for strong wind/gusts on approach by speed increments given in the operations manual. To detect windshear and magnitude of winds aloft, all available aeroplane equipment such as FMS, INS, etc. should be used;		
f.	It is recommended that stabilisation during any straight-in approach without visual reference to the ground should be achieved at the latest when passing 1 000 ft above runway threshold elevation. For approaches with a designated vertical profile applying CDFA, a later stabilisation in speed may be acceptable if higher than normal approach speeds are required by ATC procedures or allowed by the operations manual. Stabilisation should, however, be achieved not later than 500 ft above runway threshold elevation;		
g.	For approaches where the pilot has visual reference with the ground, stabilisation should be achieved not later than 500 ft above aerodrome elevation. However, it is recommended that		

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		the aeroplane should be stabilised when passing 1000 ft above runway threshold elevation;		
	h.	The relevant elements of 5. should, in addition, be applied to approaches not flown using the CDFA technique; the procedures thus developed ensure that a controlled and stable path to MDA/H is achieved. Dependent upon the number of step down fixes and the aeroplane type/class, the aeroplane should be appropriately configured to ensure safe and stable flight prior to the final descent to MDA/H.		
6.	CDFA OPS.(comb pitch incluc	I reference and path-control below MDA/H when not using the technique. In addition to the requirements stated in GEN.150 and its AMC material the pilot should have attained a ination of visual cues to safely control the aeroplane in roll and to maintain the final approach path to landing. This should be ded in the standard operating procedures and reflected in the standard.		
7.	•	ational procedures and instructions for using the CDFA ique or not:		
	a.	The operator should establish procedures and instructions for flying approaches using the CDFA technique and not. These procedures should be included in the operations manual and should include the duties of the flight crew during the conduct of such operations:		
		 The operator should publish in the operations manual the requirements stated in 4. and 5., as appropriate to the aeroplane type or class to be operated; 		

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	ii. The checklists should be completed as early as practicable and preferably before commencing final descent towards the DA/H;		
ь.	The operator's manuals should at least specify the maximum ROD for each aeroplane type/class operated and the required visual reference to continue the approach below: i. the DA/H, when applying CDFA; ii. the MDA/H, when not applying CDFA;		
c.	The operator should establish procedures which prohibit level flight at MDA/H without the flight crew having obtained the required visual references. It is not the intention to prohibit level flight at MDA/H when conducting a circling approach, which does not come within the definition of the CDFA technique;		
d.	 The operator should provide the flight crew with unambiguous details of the technique used (CDFA or not). The corresponding relevant minima should include: i. type of decision, whether DA/H or MDA/H; ii. MAPt as applicable; and iii. appropriate RVR/Visibility for the approach classification and aeroplane category; 		
e.	Specific types/class of aeroplane, in particular certain Performance Class B and Class C aeroplanes, may be unable to comply fully with the requirements of this guidance relating to the operation of CDFA. This problem arises		

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	because some aeroplanes must not be configured fully into the landing configuration until required visual references are obtained for landing, because of inadequate missed approach performance engine out. For such aeroplanes, the operator should either:		
	 obtain approval from the authority for an appropriate modification to the stipulated procedures and flight techniques prescribed herein; or 		
	increase the required minimum RVR to ensure the aeroplane will be operated safely during the configuration change on the final approach path to landing.		
8. Trair	ning:		
a.	The operator should ensure that, prior to using the CDFA technique or not (as appropriate), each flight crew member undertakes the appropriate training and checking as required by Part OR.OPS.FC. Such training should cover the techniques and procedures appropriate to the operation which are stipulated in 4. and 5. The operator's proficiency check, if applicable, should include at least one approach to a landing or go-around as appropriate using the CDFA technique or not. The approach should be operated to the lowest appropriate DA/H or MDA/H as appropriate; and, if conducted in a simulator, the approach should be operated to the lowest approved RVR. The approach is not in addition to any manoeuvre currently required by either Part-FCL or Part-OPS. The requirement may be fulfilled by undertaking any		

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	currently required approach (engine out or otherwise) other than a precision approach, whilst using the CDFA technique;		
b.	The policy for the establishment of constant predetermined vertical path and approach stability are to be enforced both during initial and recurrent pilot training and checking. The relevant training procedures and instructions should be documented in the operations manual;		
c.	The training should emphasise the need to establish and facilitate joint crew procedures and CRM to enable accurate descent path control and the requirement to establish the aeroplane in a stable condition as required by the operator's operational procedures. If barometric vertical navigation is used, the crews should be trained in the errors associated with these systems;		
d.	During training, emphasis should be placed on the flight crew's need to:		
	 maintain situational awareness at all times, in particular with reference to the required vertical and horizontal profile; 		
	ii. ensure good communication channels throughout the approach;		
	 ensure accurate descent-path control particularly during any manually-flown descent phase. The non- operating/non-handling pilot should facilitate good flight path control by: 		

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	 A. communicating any altitude/height crosschecks prior to the actual passing of the range/altitude or height crosscheck; 		
	 B. prompting, as appropriate, changes to the target ROD; and 		
	C. monitoring flight path control below DA/MDA.		
iv.	understand the actions to be taken if the MAPt is reached prior to the MDA/H;		
v.	ensure that the decision to go-around must, at the latest, have been taken upon reaching the DA/H or MDA/H;		
vi.	ensure that prompt go-around action is taken immediately when reaching DA/H if the required visual reference has not been obtained as there may be no obstacle protection if the go-around manoeuvre is delayed;		
vii.	understand the significance of using the CDFA technique to a DA/H with an associated MAPt and the implications of early go-around manoeuvres; and		
viii.	understand the possible loss of the required visual reference (due to pitch-change/climb) when not using the CDFA technique for aeroplane types/classes which require a late change of configuration and/or speed to ensure the aeroplane is in the appropriate landing configuration;		

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e.		l specific training when not using the CDFA e with level flight at or above MDA/H:		
	i. The	training should detail:		
	Α.	the need to facilitate CRM; with appropriate flight crew communication in particular;		
	В.	the additional known safety risks associated with the `dive-and-drive' approach philosophy which may be associated with non-CDFA;		
	C.	the use of DA/H during approaches flown using the CDFA technique;		
	D.	the significance of the MDA/H and the MAPt where appropriate;		
	E.	the actions to be taken at the MAPt and the need to ensure that the aeroplane remains in a stable condition and on the nominal and appropriate vertical profile until the landing;		
	F.	the reasons for increased RVR/Visibility minima when compared to the application of CDFA;		
	G.	the possible increased obstacle infringement risk when undertaking level flight at MDA/H without the required visual references;		
	Н.	the need to accomplish a prompt go-around manoeuvre if the required visual reference is lost;		
	I.	the increased risk of an unstable final approach and an associated unsafe landing if a rushed		

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			approach is attempted either from:		
			 inappropriate and close-in acquisition of the required visual reference; or 		
			unstable aeroplane energy and or flight path control; and		
		J.	The increased risk of CFIT (see introduction).		
9.	Appro	oaches requ	uiring level flights:		
	a.	•	edures which are flown with level flight at/or above hould be listed in the operations manual;		
	b.	approach being B a will depei	s should classify aerodromes where there are es which require level flight at/or above MDA/H as and C categorised. Such aerodrome categorisation nd upon the operator's experience, operational t, training programme(s) and flight crew ion(s).		
	3 OPS. nima	.GEN.150./	A Instrument Flight Rules (IFR) operating		
AER	ODRO	ME MINIMA	- CIRCLING - AEROPLANES		
1.	Term	iinology: XL	S = ILS/MLS/GLS etc.	Eurocontrol : request to include precision approach radar (PAR), as it is otherwise referred to alongside XLS throughout the NPA ;	Accepted.

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2.	is to applic	I manoeuvring (circling). The purpose of this guidance material provide operators with supplemental information regarding the cation of aerodrome operating minima in relation to circling paches.			
3.	Condu a.	uct of flight – General: The MDH and OCH included in the procedure are referenced			
	b.	to aerodrome elevation; The MDA is referenced to mean sea level;			
	с.	For these procedures, the applicable visibility is the meteorological VIS.			
4.		ument approach followed by visual manoeuvring (circling) ut prescribed tracks:			
	a.	When the aeroplane is on the initial instrument approach, before visual reference is stabilised, but not below MDA/H - the aeroplane should follow the corresponding instrument approach procedure until the appropriate instrument MAPt is reached;			
	b.	At the beginning of the level flight phase at or above the MDA/H, the instrument approach track determined by radio navigation aids, RNAV, RNP or XLS should be maintained until:			
		 the pilot estimates that, in all probability, visual contact with the runway of intended landing or the runway environment will be maintained during the entire 			

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		circling procedure;		
	ii.	the pilot estimates that the aeroplane is within the circling area before commencing circling; and		
	iii.	the pilot is able to determine the aeroplane's position in relation to the runway of intended landing with the aid of the appropriate external references;		
C.	con the	en reaching the published instrument MAPt and the ditions stipulated in 4.b., are unable to be established by pilot, a missed approach should be carried out in ordance with that instrument approach procedure;		
d.	inst run req	er the aeroplane has left the track of the initial (letdown) trument approach, the flight phase outbound from the way should be limited to an appropriate distance, which is uired to align the aeroplane onto the final approach. Such noeuvres should be conducted to enable the aeroplane:		
	i.	to attain a controlled and stable descent path to the intended landing runway; and		
	ii.	remain within the circling area and in such way that visual contact with the runway of intended landing or runway environment is maintained at all times;		
e.	-	ht manoeuvres should be carried out at an altitude/height t is not less than the circling MDA/H;		
f.	thre	scent below MDA/H should not be initiated until the eshold of the runway to be used has been appropriately ntified and the aeroplane is in a position to continue with a		

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		normal rate of descent and land within the touchdown zone.		
5.		ument approach followed by a visual manoeuvring (circling) prescribed track:		
	а.	The aeroplane should remain on the initial instrument approach or letdown procedure until one of the following is reached:		
		 The prescribed divergence point to commence circling on the prescribed track; or 		
		ii. The appropriate initial instrument MAPt;		
	b.	The aeroplane should be established on the instrument approach track determined by the radio navigation aids, RNAV, RNP, or XLS in level flight at or above the MDA/H at or by the circling manoeuvre divergence point;		
	C.	If the divergence point is reached before the required visual reference is acquired, a missed approach should be initiated not later than the initial instrument approach MAPt and completed in accordance with the initial instrument approach procedure;		
	d.	When commencing the prescribed track-circling manoeuvre at the published divergence point, the subsequent manoeuvres should be conducted to comply with the published routing and promulgated heights/altitudes;		

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e	 Unless otherwise specified, once the aeroplane is established on the prescribed track(s), the promulgated visual reference should not be required to be maintained unless: i. required by the State; or ii. the Circling MAPt (if published) is reached; 		
f.	If the prescribed track-circling manoeuvre has a published MAPt and the required visual reference has not been obtained a missed approach should be executed in accordance with 6.b. and 6.c.;		
g	. Subsequent further descent below MDA/H should only commence when the required visual reference is obtained;		
h	. Unless otherwise specified in the procedure, final descent should not be initiated from MDA/H until the threshold of the intended landing runway has been appropriately identified and the aeroplane is in a position to continue with a normal rate of descent and land within the touchdown zone.		
6. M	lissed approach:		
a	 Missed approach during instrument approach prior to circling: i. If the decision to carry out a missed approach is taken when the aeroplane is positioned on the instrument approach track defined by radio-navigation aids RNAV, RNP, or XLS, and before commencing the circling manoeuvre, the published missed approach for the 		

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	instrument approach should be followed;		
	 If the instrument approach procedure is carried out with the aid of an XLS or an SAp, the MAPt associated with an XLS procedure without glide path (GP-out procedure) or the SAp, where applicable, should be used; 		
b.	If a prescribed missed approach is published for the circling manoeuvre, this overrides the manoeuvres prescribed below;		
C.	If visual reference is lost while circling to land after the aeroplane has departed from the initial instrument approach track, the missed approach specified for that particular instrument approach should be followed. It is expected that the pilot will make an initial climbing turn toward the intended landing runway and continue overhead the aerodrome where the pilot will establish the aeroplane in a climb on the instrument missed approach track;		
d.	 The aeroplane should not leave the visual manoeuvring (circling) area, which is obstacle protected, unless: i. established on the appropriate missed approach track; or ii. at Minimum Sector Altitude (MSA); 		
e.	 All turns should (see Note 1 below) be made in the same direction and the aeroplane should remain within the circling protected area while climbing to either: i. the altitude assigned to any published circling missed approach manoeuvre if applicable; 		

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	ii.	the altitude assigned to the missed approach of the initial instrument approach;		
	iii.	the Minimum Sector Altitude (MSA);		
	iv.	the Minimum Holding Altitude (MHA) applicable for transition to a holding facility or fix, or continue to climb to a Minimum Safe Altitude; or		
	٧.	as directed by ATS/Control (C).		
Note 1:	the c align appr	n the go-around is commenced on the "downwind" leg of fircling manoeuvre, an "S" turn may be undertaken to the aeroplane on the initial instrument approach missed oach path, provided the aeroplane remains within the ected circling area.		
Note 2:	adeq man	pilot-in-command should be responsible for ensuring wate terrain clearance during the above-stipulated peuvres, particularly during the execution of a missed oach initiated by ATS;		
f.	in n to e cou is lo dire	as much as the circling manoeuvre may be accomplished nore than one direction, different patterns will be required establish the aeroplane on the prescribed missed approach rse depending on its position at the time visual reference ost. In particular, all turns are to be in the prescribed ection if this is restricted, e.g. to the west/east (left or of hand) to remain within the protected circling area;		
g.	run	missed approach procedure is promulgated for the way (XX) onto which the aeroplane is conducting a circling roach and the aeroplane has commenced a manoeuvre to		

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	align with the runway; the missed approach for this direction may be accomplished. The ATS should be informed of the intention to fly the promulgated missed approach procedure for runway XX;		
h.	When the option described in 6.g. is undertaken the pilot-in- command should whenever possible advise at the earliest opportunity, the ATS/C of the intended go-around procedure. This dialogue should, if possible occur during the initial approach phase and include the intended missed approach to be flown and the level off altitude;		
i.	In addition to 6.h., the pilot-in-command should advise ATS/C when any go-around has commenced, the height/altitude the aeroplane is climbing to and the position the aeroplane is proceeding towards and or heading the aeroplane is established on.		
GM4 OPS minima	.GEN.150.H Instrument Flight Rules (IFR) operating		
AERODROME MINIMA - ONSHORE AERODROME DEPARTURE PROCEDURES - HELICOPTERS		INDIV: request clarification when taking off in fog with a visibility of 150 m – would the fog be considered as cloud?	Accepted. This GM was not transposed.
The cloud base and visibility should be such as to allow the helicopter to be clear of cloud at Take-off Decision Point (TDP), and for the pilot flying to remain in sight of the surface until reaching the minimum speed for flight in instrument meteorological conditions given in the AFM.			
GM OPS.G minima	GEN.150(b) Instrument Flight Rules (IFR) operating		Not transposed – beyond the scope of Part-NCC.

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INCREMENTS IMPOSED BY THE COMPETENT AUTHORITY Additional increments to the published minima may be imposed by the competent authorities to take into account special operations, such as downwind approaches and single-pilot operations.		al increments to the published minima may be imposed by the ent authorities to take into account special operations, such as nd approaches and single-pilot operations.		
		PS.GEN.155.H Selection of alternate aerodromes		
1.	th a. b. c. d.	 PNR. Prior to a PNR, onshore alternates should be used; Mechanical reliability of critical control systems and critical components should be considered and taken into account when determining the suitability of the alternate; OEI performance capability should be attainable prior to arrival at the alternate; To the extent possible, deck availability should be guaranteed; and 	Under 1.e. weather observer should be acceptable to the competent authority;	Beyond the scope of the rule of air operations and the presumption of an observer would preclude the use of automatic weather reporting systems.
2.	en	Weather information must be reliable and accurate. If shore alternates should not be used when it is possible to carry ough fuel to have an onshore alternate. Offshore alternates should it be used in a hostile environment.		

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3.	The landing technique specified in the AFM following control system failure may preclude the nomination of certain helidecks as alternate aerodromes.			
GM 1	L OPS.GEN.155.A(a)(3) Selection of alternate aerodromes		Text upgraded to IR, NCC.OP.105.	
The if the	ATED AERODROME – AEROPLANES destination aerodrome could be considered as an isolated aerodrome e fuel required to the nearest adequate destination alternate drome is more than:	INDIV: Request to provide correct GM number and put the definition of "Isolated Aerodrome" in the cover regulation;	The term is only used in this context and therefore explained in a GM.	
1.	for aeroplanes with reciprocating engines, fuel to fly for 45 minutes plus 15 % of the flight time planned to be spent at cruising level or two hours, whichever is less; or			
2.	for aeroplanes with turbine engines, fuel to fly for two hours at normal cruise consumption above the destination aerodrome, including final reserve fuel.			
GM2 OPS.GEN.155.H Selection of alternate aerodromes			Text not transposed - beyond the scope of Part-NCC.	
OFF	SHORE ALTERNATES – HELICOPTERS – COMMERCIAL AIR TRANSPORT			
stan Mete	The procedures contained in AMC OPS.CAT.155.H(a)(1) are ther-critical. Consequently, meteorological data conforming to the dards contained in the Regional Air Navigation Plan and ICAO Annex 3 eorological Service for International Air Navigation has been specified. he following meteorological data is point specific, caution should be			

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exerc	ised when associating it with nearby aerodromes (or helidecks).		
2.	Meteorological Reports (METARs):		
a.	Routine and special meteorological observations at offshore installations should be made during periods and at a frequency agreed between the meteorological authority and the operator concerned. They should comply with the requirements contained in the meteorological section of the ICAO Regional Air Navigation Plan , and should conform to the standards and recommended practices, including the desirable accuracy of observations, promulgated in ICAO Annex 3 Meteorological Service for International Air Navigation.		
	Routine and selected special reports are exchanged between prological offices in the METAR or SPECI code forms prescribed by orld Meteorological Organisation.		
3.	Aerodrome Forecasts (TAFS):		
opera less t incluc of one may l opera instal	The aerodrome forecast consists of a concise statement of the or average meteorological conditions expected at an aerodrome or ting site during a specified period of validity, which is normally not han nine hours, or more than 24 hours in duration. The forecast les surface wind, visibility, weather and cloud, and expected changes e or more of these elements during the period. Additional elements be included as agreed between the meteorological authority and the itors concerned. Where these forecasts relate to offshore lations, barometric pressure and temperature should be included to ate the planning of helicopter landing and take-off performance.		

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b. Aerodrome forecasts are most commonly exchanged in the TAF code form, and the detailed description of an aerodrome forecast is promulgated in the ICAO Regional Air Navigation Plan and also in ICAO Annex 3 Meteorological Service for International Air Navigation, together with the operationally desirable accuracy elements. In particular, the observed cloud height should remain within $\pm 30\%$ of the forecast value in 70% of cases, and the observed visibility should remain within $\pm 30\%$ of the forecast value in 80% 0f cases.		
4. Landing Forecasts (TRENDS):		
a. The landing forecast consists of a concise statement of the mean or average meteorological conditions expected at an aerodrome or operating site during the two-hour period immediately following the time of issue. It contains surface wind, visibility, significant weather and cloud elements, and other significant information, such as barometric pressure and temperature, as may be agreed between the meteorological authority and the operators concerned.		
b. The detailed description of the landing forecast is promulgated in the ICAO Regional Air Navigation Plan and also in ICAO Annex 3 Meteorological Service for International Air Navigation, together with the operationally desirable accuracy of the forecast elements. In particular, the value of the observed cloud height and visibility elements should remain within +/-30% of the forecast values in 90% of the cases.		
c. Landing forecasts most commonly take the form of routine or special selected meteorological reports in the METAR code, to which either the code words "NOSIG", i.e. no significant change expected; "BECMG" (becoming), or "TEMPO" (temporarily), followed by the expected change,		

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are added. The two-hour period of validity commences at the time of the meteorological report.		
5. When operating off shore, any spare payload capacity should be used to carry additional fuel if it would facilitate the use of an onshore alternate.		
AMC OPS.GEN.165.A Noise abatement		
NOISE ABATEMENT PROCEDURES - AEROPLANES	 1/ Request to add a statement that NAP should only be applied when noise benefits can be expected; 2/ IS (GA): Request to insert ICAO PANS-OPS (Doc 8168), Volume I, Part V, Chapter 2, 2.1.3. and PANS-ATM (Doc 4444) Chapter 7, 7.2.5. (safety case); 	1/ This is already contained in the safety objective in the IR.2/ The GM to this rule provides a reference to the ICAO document.
1. The operator's noise abatement procedures for departure and arrival/approach for each aircraft type, which should be designed to be simple and safe to operate with no significant increase in crew workload during critical phases of flight.		
2. A pilot-in-command should follow noise abatement procedures whenever they would not have a detrimental effect on aircraft safety.	Clarification requested;	This clause is included in the proposed revision to the IR.

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GM	OPS.GEN.165.A Noise abatement		
_	SE ABATEMENT PROCEDURES - COMPLEX MOTOR-POWERED OPLANES – COMMERCIAL AIR TRANSPORT		
1.	For each aeroplane type only two departure procedures should be defined, in accordance with Part I Section 7 of ICAO PANS-OPS Volume 1 (Doc 8168-OPS/611), as follows:		
	 Noise Abatement Departure Procedure one (NADP 1) designed to meet the close-in noise abatement objective; 		
	 b. Noise Abatement Departure Procedure two (NADP 2) designed to meet the distant noise abatement objective; 		
	 In addition, each NADP climb profile can only have one sequence of actions. 		
2.	This GM addresses only the vertical profile of the departure procedure. Lateral track has to comply with the Standard Instrument Departure (SID).		
3.	"Climb profile" means the vertical path of the NADP as it results from the pilot's actions (engine power reduction, acceleration, slats/flaps retraction).		
4.	"Sequence of actions" means the order and the timing in which these pilot's actions are done.		
5.	Example:		
	For a given aeroplane type, when establishing the distant NADP, an		

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ace ret dif	erator should choose either to reduce power first and then celerate, or to accelerate first and then wait until slats/flaps are tracted before reducing power. The two methods constitute two ferent sequences of actions. For an aeroplane type, each of the o departure climb profiles may be defined by:		
a.	one sequence of actions (one for close-in, one for distant);		
b.	two Above Aerodrome Level (AAL) altitudes/heights:		
C.	the altitude of the first pilot's action (generally power reduction with or without acceleration). This altitude should not be less than 800 ft AAL; or		
d.	the altitude of the end of the noise abatement procedure. This altitude should usually not be more than 3 000 ft AAL.		
Flig pe alt no	ese two altitudes may be runway specific when the aeroplane ght Management System (FMS) has the relevant function which rmits the crew to change thrust reduction and/or acceleration titude/height. If the aeroplane is not FMS equipped or the FMS is t fitted with the relevant function, two fixed heights should be fined and used for each of the two NADPs.		
AMC OF	PS.GEN.170 Minimum terrain clearance altitudes	Request to delete this AMC since it might lead to confusion;	Not accepted. Text provides a mean to comply with the safety objective.
GENERA	۱L		
	rcially available information specifying minimum terrain clearance s may be used.		

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АМ	C OPS.	GEN.175 Minimum flight altitudes		Text not transposed - the IR is covered by Part-SERA.
COM	IMERCI	AL OPERATIONS OTHER THAN COMMERCIAL AIR TRANSPORT		
1.		specifying minimum flight altitudes, the operator should take int of AMC OPS.CAT.170.		
2.	perfo	specifying minimum altitudes for operations that can only be rmed below the minimum altitudes, it should be taken into int that the aircraft:		
	a.	should not be flown over congested areas; and		
	b.	should not expose persons or property on the surface to risk of injury or damage.		
3.		consideration should be given to environmental factors, such as exposure.		
4.		dures should be developed based on a risk assessment in dance with OPS.COM.270 giving due regard to factors, such as:		
	a.	type of operation;		
	b.	terrain to be overflown;		
	с.	obstacle situation;		
	d.	weather conditions, visibility, wind, turbulence;		
	e.	lighting conditions;		
	f.	crew experience;		

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	g.	crew familiarity with the area;		
	h.	crew training;		
	i.	aircraft performance;		
	j.	number of persons on board; and		
	k.	any attached or underslung equipment or load.		
5.		wing the risk assessment, the Operations Manual (OM) should de Standard Operating Procedures (SOPs), such as:		
	a.	method of operational control of such operations;		
	b.	crew minimum experience;		
	с.	flight planning;		
	d.	pre-survey of the route and/or take-off/landing site;		
	e.	weather limitations;		
	f.	minimum altitudes;		
	g.	performance;		
	h.	(safe) forced landing areas;		
	i.	navigation; and		
	j.	personal protective equipment.		
GM	OPS.G	GEN.175 Minimum flight altitudes		Text not transposed - the content is covered by Part-SERA.
DES	DESCENDING BELOW SPECIFIED MINIMUM FLIGHT ALTITUDES			

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1.	The operator may have to obtain permission from the authority of the State in which it intends to conduct operations involving flights below minimum altitudes.		
2.	The State overflown may specify procedures for certain commercial operations (e.g. helicopter operations or photo-flights) or flight instruction (e.g. training of emergency landings) which allow the operator or the approved training organisation to descend below the specified minimum flight altitudes.		
AM	C OPS.GEN.180.H Routes and areas of operation		Text not transposed - beyond the scope of Part-NCC.
	ICOPTER COASTAL TRANSIT OPERATIONS – COMMERCIAL AIR NSPORT		
tran	helicopters operated in Performance Class 3 and conducting coastal sit operations, the width of the coastal corridor and the equipment ied, should be consistent with the conditions prevailing at the time.		
GM	OPS.GEN.180.H Routes and areas of operation		Text not transposed - beyond the scope of Part-NCC.
	ICOPTER COASTAL TRANSIT OPERATIONS – COMMERCIAL AIR NSPORT		
fron guic	A helicopter operating overwater in Performance Class 3 has to e certain equipment fitted. This equipment varies with the distance n land that the helicopter is expected to operate. The aim of this lance material is to discuss that distance, bring into focus what fit is uired and to clarify the operator's responsibility, when a decision is		

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made to conduct coastal transit operations.		
2. In the case of operations north of 45N or south of 45S, the coastal corridor facility may or may not be available in a particular state, as it is related to the State definition of open sea area as described in the definition of hostile environment.		
3. Where the term coastal transit is used, it means the conduct of operations overwater within the coastal corridor in conditions where there is reasonable expectation that; the flight can be conducted safely in the conditions prevailing; and, following an engine failure, a safe forced landing and successful evacuation can be achieved; and survival of the crew and passengers can be assured until rescue is effected.		
4. Coastal corridor is a variable distance from the coastline to a maximum distance corresponding to three minutes flying at normal cruising speed.		
5. Establishing the width of the coastal corridor:		
a. The distance from land of coastal transit, is defined the boundary of a corridor that extends from the land, to a maximum distance of up to three minutes at normal cruising speed (approximately five to six nm). Land in this context includes sustainable ice and, where the coastal region includes islands, the surrounding waters may be included in the corridor and aggregated with the coast and each other. Coastal transit need not be applied to inland waterways, estuary crossing or river transit:		
i. In some areas, the formation of ice is such that it can be possible to land, or force land, without hazard to the		

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		helicopter or occupants. The operator may regard the definition of the "land" extends to these areas.		
	ii.	In view of the fact that such featureless and flat white surfaces could present a hazard and could lead to white-out conditions, the definition of land does not extend to flights over ice fields in OPS.CAT.410(b)(2)(ii) and OPS.CAT.418.H.		
wide.	ions i A num	width of the corridor is variable from not safe to conduct n the conditions prevailing, to the maximum of three minutes nber of factors will, on the day, indicate if it can be used - and can be. These factors will include but not be restricted to:		
	i.	the meteorological conditions prevailing in the corridor;		
	ii.	the instrument fit of the aircraft;		
	iii.	the certification of the aircraft - particularly with regard to floats;		
	iv.	the sea state;		
	v.	the temperature of the water;		
	vi.	the time to rescue; and		
	vii.	the survival equipment carried.		
These	can b	e broadly divided into the following three functional groups:		
А.	Those	e which meet the requirement for safe flying – i. and ii;		
B. evacua		e which meet the requirement for a safe forced landing and - i., ii., iii. and iv;		
C. landing		e which meet the requirement for survival following a forced successful evacuation – i., iv., v., vi. and vi;		

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6.	Requirement for safe flying:		
a.	It is generally recognised that when flying out of sight of land in certain meteorological conditions, such as occur in high pressure weather patterns (goldfish bowl - no horizon, light winds and low visibility), the absence of a basic panel (and training) can lead to disorientation. In addition, lack of depth perception in these conditions demands the use of a radio altimeter with an audio voice warning as an added safety benefit - particularly when autorotation to the surface of the water may be required.		
b.	In these conditions a helicopter, without the required instruments and radio altimeter, should be confined to a corridor in which a pilot can maintain reference using the visual cues on the land.		
7.	Requirement for a safe forced landing and evacuation:		
a.	Weather and sea state both affect the outcome of an autorotation following an engine failure. It is recognised that the measurement of sea state is problematical and when assessing such conditions, good judgement has to be exercised by the operator and the commander.		
b.	Where floats have been certificated only for emergency use (and not for ditching), operations should be limited to those sea states which meet the requirement for such use - where a safe evacuation is possible.		
of re occu gene	ching certification requires compliance with a comprehensive number equirements relating to rotorcraft water entry, flotation and trim, apant egress and occupant survival. Emergency flotation systems, erally fitted to smaller CS-27 rotorcraft, are approved against a broad irement that the equipment should perform its intended function and		

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not hazard the rotorcraft or its occupants. In practice, the most significant difference between ditching and emergency flotation systems is substantiation of the water entry phase. Ditching requirements call for water entry procedures and techniques to be established and promulgated in the AFM. The fuselage/flotation equipment should thereafter be shown to be able to withstand loads under defined water entry conditions which relate to these procedures. For emergency flotation equipment, there is no requirement to define the water entry technique and no specific conditions defined for the structural substantiation.)			
8. a.	Requirements for survival: Survival of crew members and passengers, following a successful autorotation and evacuation, is dependant on the clothing worn, the equipment carried and worn, the temperature of the sea and the sea state (see GM OPS.CAT.H.426). Search and rescue response/capability consistent with the anticipated exposure should be available before the conditions in the corridor can be considered non-hostile.		
b.	Coastal Transit can be conducted (including north of 45N and south of 45S - when the definition of open sea areas allows) provided the requirements of GM OPS.GEN.180.H 7. and 8. are met and the conditions for a non-hostile coastal corridor are satisfied.		
AMC1 OPS.GEN.185 Meteorological conditions			
CONTINUATION OF A FLIGHT In the case of in-flight re-planning, continuation of a flight refers to the point from which a revised flight plan applies.		MS: request to place this as a GM;	Accepted. Text classified as GM.

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AMC2 OPS.GEN.185 Meteorological conditions		
EVALUATION OF METEOROLOGICAL CONDITIONS Pilots should carefully evaluate the available meteorological information relevant to the proposed flight, such as applicable surface observations, temperatures aloft, terminal and area forecasts, AIRMETs, SIGMETs, and pilot reports. The ultimate decision whether, when, and where to make the flight rests with the pilot-in-command. A pilot also should continue to	Request to add "winds aloft";	Accepted. Text amended.
re-evaluate changing weather conditions. AMC3 OPS.GEN.185 Meteorological conditions		Text not transposed - beyond the scope of Part-NCC.
GENERAL – COMMERCIAL AIR TRANSPORT		
In addition to AMC 1 and 2 OPS.GEN.185, a flight according to Instrument Flight Rules (IFR) should only be continued beyond:		
1. the decision point when using the reduced contingency fuel procedure; or		
2. the pre-determined point when using the pre-determined point procedure,		
when available information indicates that the expected weather conditions at the time of arrival at the destination and/or required alternate aerodromes are at, or above, the applicable aerodrome operating minima.		
AMC OPS.GEN.190.B Take-off conditions		Text not transposed - beyond the

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		scope of Part-NCC.
FACILITIES AT THE TAKE-OFF SITE – BALLOONS At the take-off site an anemometer should be provided by the operator.		
AMC OPS.GEN.195 Approach and landing conditions		
LANDING DISTANCE/FATO SUITABILITY The in-flight determination of the landing distance/FATO suitability should be based on the latest available report, preferably not more than 30 minutes before the expected landing time.	Request to clarify what "suitability" means;	NCC.OP.195 addresses the need to ensure that a safe approach and landing would not be prevented. The AMC links back to that requirement. Deleted text is not needed because the AMC already states that the latest available report should be taken into account.
AMC OPS.GEN.200 Commencement and continuation of approach		Text upgraded to IR level.
GENERAL		
1. The RVR should not be less than the applicable minima.		
2. Where RVR is not available, RVR values may be derived by converting the reported visibility (CMV) in accordance with AMC11 OPS.GEN.150.		

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3.	If the MDA/H is above 500 ft above the aerodrome, the operator should establish an altitude/height, below which the approach should not be continued if the RVR/CMV is less than the applicable minima. This altitude/height should be at or above MDA/H + 500 ft but not above the FAF altitude/height.		
4. c.	 The touchdown zone RVR is always prevailing over the other RVR values. If reported and relevant, the mid point and stop end RVR are also controlling. The minimum RVR should be at least: a. 125 m for the mid-point; or b. the RVR required for the touchdown zone; and 75 m for the stop-end. For aeroplanes equipped with a rollout guidance or control system, the minimum RVR value for the mid-point is 75 m. 'Relevant' means that part of the runway used during the high speed phase of the landing down to a speed of approximately 60 knots. 	IS (BA): additional text provided for clarity to cover the minimum mid- point RVR and minimum stop-end RVR;	Not accepted. The Agency supported by the RG preferred to maintain the text. Text aligned with the CAT rule.
AM	C1 OPS.GEN.205 Fuel and oil supply	Request to review the objective requirement of the AMC and the prescriptive requirement of the rule;	This complements the IR; therefore it was moved from AMC to the IR.
FUE	E PLANNING - NON-COMMERCIAL OPERATIONS		
1. cons	In computing the fuel and oil required, the following should be sidered:		

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a.	Meteorological conditions forecast;			
b.	Expected air traffic control routings and traffic delays;			
c.	For IFR flight, one instrument approach at the destination aerodrome, including a missed approach;			
d.	The procedures for loss of pressurisation, where applicable, or failure of one power-unit while en route;			
e. increa	Any other conditions that may delay the landing of the aircraft or ase fuel and/or oil consumption.			
	Nothing precludes amendment of a flight plan in-flight, in order to an the flight to another destination, provided that the all rements can be complied with from the point where the flight is re- red.			
AMC	2 OPS.GEN.205.B Fuel and oil supply		Text not transposed - beyond the scope of Part-NCC.	
FUEL	PLANNING – BALLOONS – COMMERCIAL AIR TRANSPORT			
In addition to AMC2 B OPS.GEN.205, for flights conducted in accordance with visual flight rules, reserve fuel (gas or ballast) should not be less than to allow:				
1.	45 minutes flight in mountainous areas; or			
2.	One-hour flight if the take-off is at night.			
AMC:	3 OPS.GEN.205 Fuel and oil supply		Text not transposed - beyond the scope of Part-NCC.	

e	B: Summary of comments	C: Reason for change, remarks
	·	C. Reason for change, remarks
VE FUEL - COMMERCIAL OPERATIONS OTHER THAN COMMERCIAL ANSPORT		
Notwithstanding AMC3 OPS.GEN.205.A and AMC4 OPS.GEN.205.H hts remaining within 25 NM of the aerodrome/operating site of ure and with operating flight crew on board only, reserve fuel not be less than: for aeroplanes, 20 minutes fuel at normal cruising altitude; and for belicopters, 10 minutes fuel at best range speed		
The operator should demonstrate to the competent authority that nount of reserve fuel in accordance with 1 is essential for carrying specialised task.		
The operator should specify in the OM:		
the type of activity where such reduced reserve fuel may be used;		
methods of reading and calculating the remaining fuel; and		
SOPs.		
Refuelling facilities should be available at the aerodrome/operating		
Refuelling should be performed between each flight.		
OPS.GEN.205 Fuel and oil supply		
	ANSPORT Notwithstanding AMC3 OPS.GEN.205.A and AMC4 OPS.GEN.205.H hts remaining within 25 NM of the aerodrome/operating site of ure and with operating flight crew on board only, reserve fuel not be less than: for aeroplanes, 20 minutes fuel at normal cruising altitude; and for helicopters, 10 minutes fuel at best range speed. The operator should demonstrate to the competent authority that nount of reserve fuel in accordance with 1 is essential for carrying specialised task. The operator should specify in the OM: the type of activity where such reduced reserve fuel may be used; methods of reading and calculating the remaining fuel; and SOPs. Refuelling facilities should be available at the aerodrome/operating Refuelling should be performed between each flight.	ANSPORT Notwithstanding AMC3 OPS.GEN.205.A and AMC4 OPS.GEN.205.H this remaining within 25 NM of the aerodrome/operating site of ure and with operating flight crew on board only, reserve fuel not be less than: for aeroplanes, 20 minutes fuel at normal cruising altitude; and for helicopters, 10 minutes fuel at best range speed. The operator should demonstrate to the competent authority that iount of reserve fuel in accordance with 1 is essential for carrying ipecialised task. The operator should specify in the OM: the type of activity where such reduced reserve fuel may be used; methods of reading and calculating the remaining fuel; and SOPs. Refuelling facilities should be available at the aerodrome/operating Refuelling should be performed between each flight.

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	ELLING/DEFUELLING PROCEDURES - COMMERCIAL OPERATIONS R THAN COMMERCIAL AIR TRANSPORT		Text not transposed - beyond the scope of Part-NCC.	
1. speci	The operator should establish refuelling/defuelling procedures fying:			
a.	the fuelling sites and equipment that may be used for fuelling the aircraft;			
b.	the fuel quality for fuelling the aircraft;			
c.	fire precautions and preparedness;			
d.	the transport and storage of fuel in the operators care according to established standards;			
e.	fuelling with engines/rotors running, if applicable; and			
f.	in-flight refuelling, if applicable.			
2. hazaı	These procedures should take into account the minimisation of fire rds and adequate protection of the natural environment.			
	OPS.GEN.210 Refuelling with passengers embarking, on d or disembarking	MS: Request to put this material in the IR;	Not accepted. This is a means to comply with the safety objective established in the IR.	
GENERAL		Request to prohibit activities like cleaning or catering while re-/de- fuelling because it poses a risk during evacuation;	Not accepted. This is an AMC and cannot contain additional IR.	
		Suggestion to prohibit defueling with passengers;		

ule			
		B: Summary of comments	C: Reason for change, remarks
		IS (GA): proportionality - request that this material apply only to commercial operations, certainly not for small GA aircraft;	Not accepted. The safety rule also applies to NCC operations and is not dependent on the size of the aircraft.
р	ositioned so as to be immediately available in the event of a	INDIV: request that the appropriate scale be qualified by a risk assessment;	The operator should certainly consider this item within its SMS. However, Part D of Annex Va of the Basic Regulation does allow the operator to use an aerodrome with a lower RFFS category except in the case of emergency (which does not apply in this case).
i. ii	 One qualified person should remain at a specified location during fuelling operations with passengers on board. This qualified person should be capable of handling emergency procedures concerning fire protection and fire-fighting, handling communications and initiating and directing an evacuation; A two-way communication should be established and should remain available by the aeroplane's intercommunication system or other suitable means between the personnel involved in the operation supervising the refuelling and the pilot-in-command or other qualified personnel on board the aeroplane; 	 IS (BA): Request to consider that there are aircraft types that do not have a two-way intercom system (add "if available"); INDIV: Request to specify "sufficient" and "qualified personnel"; 	Text not transposed - it is already covered in the IR. 1/ The text already includes "or other suitable means" to account for such cases. 2/ The term "sufficient" has been deleted. The operator is best placed to determine numbers and qualification requirements for its personnel.
	a. F p fi b. F i.	 a. Fire fighting facilities of the appropriate scale should be positioned so as to be immediately available in the event of a fire, when using operating sites; b. For aeroplanes: One qualified person should remain at a specified location during fuelling operations with passengers on board. This qualified person should be capable of handling emergency procedures concerning fire protection and fire-fighting, handling communications and initiating and directing an evacuation; A two-way communication should be established and should remain available by the aeroplane's intercommunication system or other suitable means between the personnel involved in the operation supervising the refuelling and the pilot-in-command or other qualified 	 b. For aeroplanes: i. One qualified person should remain at a specified location during fuelling operations with passengers on board. This qualified person should be capable of handling emergency procedures concerning fire protection and fire-fighting, handling communications and initiating and directing an evacuation; ii. A two-way communication should be established and should remain available by the aeroplane's inter-communication system or other suitable means between the personnel involved in the operation supervising the refuelling and the pilot-in-command or other qualified personnel on board the aeroplane;

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		re/defuelling will take place;		
	iv.	'Fasten Seat Belts' signs should be off;		
	٧.	'NO SMOKING' signs should be on, together with interior lighting to enable emergency exits to be identified;		
	vi.	Passengers should be instructed to unfasten their seat belts and refrain from smoking;		
	vii.	Sufficient qualified personnel or the minimum required number of cabin crew, as applicable, should be on board and be prepared for an immediate emergency evacuation;		
	viii.	If the presence of fuel vapour is detected inside the aeroplane, or any other hazard arises during re/defuelling, fuelling should be stopped immediately;		
	ix.	The ground area beneath the exits intended for emergency evacuation and slide deployment areas should be kept clear;		
	х.	Provision should be made for a safe and rapid evacuation;		
c.	For	helicopters:		
	i.	Door(s) on the refuelling side of the helicopter should remain closed;		
	ii.	Door(s) on the non-refuelling side of the helicopter should remain open, weather permitting;		
	iii.	Sufficient personnel should be immediately available to		

A: R	ule			B: Summary of comments	C: Reason for change, remarks
			move passengers clear of the helicopter in the event of a fire.		
		iv.	Sufficient qualified personnel should be on board and be prepared for an immediate emergency evacuation;		
		v.	If the presence of fuel vapour is detected inside the helicopter, or any other hazard arises during re/defuelling, fuelling should be stopped immediately;		
		vi.	The ground area beneath the exits intended for emergency evacuation and slide deployment areas should be kept clear;		
		vii.	Provision should be made for a safe and rapid evacuation.		
2.	activiti should	ies ar be	efuelling with passengers on board, ground servicing nd work inside the aircraft, such as catering and cleaning, conducted in such a manner that they do not create a that the aisles and emergency doors are unobstructed.		
3.		ency	ment of integral aircraft stairs or the opening of exits as a prerequisite to refuelling is not necessarily		
			210 Refuelling with passengers embarking, on barking		
REFU	JELLING	G/DEF	UELLING WITH WIDE-CUT FUEL		

A: F	Rule	B: Summary of comments	C: Reason for change, remarks
1.	'Wide-cut fuel' (designated JET B, JP-4 or AVTAG) is an aviation turbine fuel that falls between gasoline and kerosene in the distillation range and consequently, compared to kerosene (JET A or JET A1), it has the properties of higher volatility (vapour pressure), lower flash point and lower freezing point.		
2.	Wherever possible, an operator should avoid the use of wide-cut fuel types. If a situation arises such that only wide-cut fuels are available for refuelling/defuelling, operators should be aware that mixtures of wide-cut fuels and kerosene turbine fuels can result in the air/fuel mixture in the tank being in the combustible range at ambient temperatures. The extra precautions set out below are advisable to avoid arcing in the tank due to electrostatic discharge. The risk of this type of arcing can be minimised by the use of a static dissipation additive in the fuel. When this additive is present in the proportions stated in the fuel specification, the normal fuelling precautions set out below are considered adequate.		
3.	Wide-cut fuel is considered to be "involved" when it is being supplied or when it is already present in aircraft fuel tanks.		
4.	When wide-cut fuel has been used, this should be recorded in the technical log. The next two uplifts of fuel should be treated as though they too involved the use of wide-cut fuel.		
5.	When refuelling/defuelling with turbine fuels not containing a static dissipater, and where wide-cut fuels are involved, a substantial reduction on fuelling flow rate is advisable. Reduced flow rate, as recommended by fuel suppliers and/or aeroplane manufacturers, has the following benefits:		

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	a.	It allows more time for any static charge build-up in the fuelling equipment to dissipate before the fuel enters the tank;		
	b.	It reduces any charge which may build up due to splashing;		
	c.	Until the fuel inlet point is immersed, it reduces misting in the tank and consequently the extension of the flammable range of the fuel.		
6.	equip aerop quote	flow rate reduction necessary is dependent upon the fuelling oment in use and the type of filtration employed on the plane fuelling distribution system. It is difficult, therefore, to a precise flow rates. Reduction in flow rate is advisable whether oure fuelling or over-wing fuelling is employed.		
7.	that t	over-wing fuelling, splashing should be avoided by making sure the delivery nozzle extends as far as practicable into the tank. on should be exercised to avoid damaging bag tanks with the e.		
		GEN.210 Refuelling with passengers embarking, on lisembarking		Text not transposed - beyond the scope of Part-NCC.
REFUELLING/DEFUELLING PROCEDURES - COMMERCIAL OPERATIONS OTHER THAN COMMERCIAL AIR TRANSPORT		-		
The OM should contain procedures, including the following:		ould contain procedures, including the following:		
1.	Fue	l quality:		
a.	Doc	umentation of fuel received;		

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b.	Sampling;		
с.	Fuel grade;		
d.	Installation, storage and dispensing processes;		
e.	Labelling;		
f. conta	Checking and testing, as appropriate, of fuel specification, age and imination.		
2.	Fuelling while the engines are running:		
a.	Safety precautions;		
b.	One pilot at the controls.		
3.	Transport and storage of fuel:		
a.	Operators fuel installation;		
b.	Mobile storage (drums, cans, tanks);		
c.	Transportation in, on or under the aircraft (dangerous goods).		
4.	Fuelling safety:		
a.	Electrical bonding;		
b.	Public protection;		
c.	Control of access to storage areas;		
d.	Fire safety in fuel farm and storage areas;		
e.	Fire safety in mobile fuellers, fuelling pits, and fuelling cabinets;		
f.	Training of fuelling personnel in fire safety;		
g.	Fire code.		

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5.	Environment:		
a.	Precautionary measures;		
b.	Fuel spills;		
c.	Clean up;		
d.	Reporting;		
	OPS.GEN.210 Refuelling with passengers embarking, on d or disembarking	Request to delete this GM since Annex 14 does not address flight operations, but design of aerodromes;	This GM is transposed from Annex 6 Part II.
	RAFT REFUELLING PROVISIONS AND GUIDANCE ON SAFE		
(Aero and g	sions concerning aircraft refuelling are contained in Volume I odrome Design and Operations) of the ICAO Annex 14 (Aerodromes), guidance on safe refuelling practices is contained in Parts 1 and 8 of CAO Airport Services Manual (Doc9137).		
GM OPS.GEN.220.B Operational limitations - balloons			Text not transposed - beyond the scope of Part-NCC.
BALL	OON NIGHT FLIGHT		
overs cond land. cond planr	risk of collision with overhead lines is considerable and cannot be stated. The risk is considerably increased during night flights in itions of failing light and visibility when there is increasing pressure to A number of incidents have occurred in the late evening in just such itions, and may have been avoided had an earlier landing been ned. It is intended by the rule that night landings for this reason shall be allowed.		

A: Rule	B: Summary of comments	C: Reason for change, remarks
Section III – Aircraft performance and operating limitations		
AMC1 OPS.GEN.305 Weighing		Text aligned with the CAT text.
1. New aircraft that have been weighed at the factory may be placed into operation without reweighing if the mass and balance records have been adjusted for alterations or modifications to the aircraft. Aircraft transferred from one community operator to another do not have to be weighed prior to use by the receiving operator, unless more than 4 years have elapsed since the last weighing.		
2. The mass and centre of gravity (CG) of an aircraft should be revised whenever the cumulative changes to the dry operating mass exceed \pm 0.5 % of the maximum landing mass or for aeroplanes the cumulative change in CG position exceeds 0.5 % of the mean aerodynamic chord. This may be done by weighing the aircraft or by calculation.		
AMC2 OPS.GEN.305.A Weighing		Text not transposed - beyond the scope of Part-NCC.
FLEET MASS AND CG POSITION FOR AEROPLANES USED IN COMMERCIAL AIR TRANSPORT		
1. For a group of aeroplanes of the same model and configuration, an average dry operating mass and CG position may be used as the fleet		

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mass and CG position, provided that:					
 a. the dry operating mass of an individual ae by more than ±0.5 % of the maximum str from the established dry operating fleet m 	ructural landing mass				
b. the CG position of an individual aeroplane than ±0.5 % of the mean aerodynamic ch					
2. The operator should verify that, after an e configuration change or after weighing, the aerop tolerances above.					
3. To obtain fleet values, the operator should between two fleet mass evaluations, a certain nu specified in the Table below. "n" is the number of using fleet values. Those aeroplanes in the fleet v weighed for the longest time should be selected f					
Table 1 of AMC2 OPS.GEN.305.A Weighing					
Number of aeroplanes in the fleet	Minimum number of weig	nings			
2 or 3 n					
4 to 9	(n + 3)/2				
10 or more	(n + 51)/10				
4. The interval between two fleet mass evalu 48 months.	ations should not exceed				

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5. The fleet values should be updated at least at the end of each fleet mass evaluation.		
6. Aeroplanes which have not been weighed since the last fleet mass evaluation can be kept in a fleet operated with fleet values, provided that the individual values are revised by calculation and stay within the tolerances above. If these individual values no longer fall within the tolerances, the operator should determine new fleet values or operate aeroplanes not falling within the limits with their individual values.		
7. If an individual aeroplane dry operating mass is within the fleet mass tolerance but its CG position exceeds the tolerance, the aeroplane may be operated under the applicable dry operating fleet mass but with an individual CG position.		
8. Aeroplanes for which no mean aerodynamic chord has been published should be operated with their individual mass and CG position values. They may be operated under the dry operating fleet mass and CG position, provided that this can be justified by a study.		
GM OPS.GEN.305.A Weighing		Deleted since it relates to a deleted AMC.
MAXIMUM STRUCTURAL LANDING MASS AEROPLANE		
Maximum Structural Landing Mass is the maximum permissible total aeroplane mass upon landing under normal circumstances.		
AMC OPS.GEN.310(a)(1) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		General remark: Values for standard masses have

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		been upgraded at IR level	
DRY OPERATING MASS To calculate the dry operating mass and the associated CG of the aircraft, the operator should take into account the mass of all operating items and crew members, and the influence of their position on the aircraft CG. This should be done by weighing or using the standard masses of 85 kg for flight and technical crew members and 75 kg for cabin crew members, including hand baggage. Account shall be taken of any additional baggage. On flights where crew masses, including hand baggage, are expected to exceed the standard crew masses, the actual mass of the crew should be determined by weighing.	MS: the figures suggest an assumption that cabin crew are female (lighter), flight and technical crew are male. Request to distinguish between male/female crew members rather than crew type;	This will be addressed in rulemaking task OPS.027	
AMC1 OPS.GEN.310(a)(2) Mass and balance system- complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations			
TRAFFIC LOAD			
Traffic load should be determined by actual weighing or using standard masses for passengers, persons other than crew members and baggage.			
AMC2 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations			

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MASS VALUES FOR PASSENGERS/PERSONS OTHER THAN CREW MEMBERS ¹ AND BAGGAGE	 1/ Request to amend the text considering a significant change in number of PAX, baggage or similar and the actual weighing in those cases; 2/ IS (BA): Request to add more guidance on how to check the baggage (weighing facilities are not always available at aerodromes used by BA); request to increase to "19 seats available", to include larger business jets. Suggested text for a new AMC is given in comment #6868; Request to extend the possibility of deducting 6kg to all operations; 	 Accepted, Operators will get the possibility to propose an alternative means of compliance in order to comply with the objective of proportionality. Noted This will be further assessed in rulemaking task OPS.027. 	
1. When			
 a. the number of passenger seats available is: i. less than 10 for aeroplanes; or ii. less than 6 for helicopters; or 	INDIV: request to change 1.a.ii. to 7 passengers, to accommodate current models such as AW19 and EC130B4 with 7 seating capacity. Suggest grandfather rights be applied to older	Noted: Current values are based on existing rules. New values should be substantiated; therefore this	

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		models;	could be part of the rulemaking task OPS.027.	
b. the	number of passengers is less than 11 for balloons,			
	passenger mass may be calculated on the basis of a statement by, or on behalf of, each passenger, adding to it a predetermined mass to account for hand baggage and clothing.			
	The predetermined mass for hand baggage and clothing should be established by the operator on the basis of studies relevant to his particular operation. In any case, it should not be less than:			
i.	4 kg for clothing; and			
ii.	6 kg for hand baggage.			
The passengers' stated mass and the mass of passengers' clothing and hand baggage should be checked prior to boarding and adjusted, if necessary.				
personal be	en determining the actual mass by weighing, passengers' elongings and hand baggage should be included. Such weighing conducted immediately prior to boarding the aircraft.			
Tables 1 ar baggage ar carried by	en using standard mass values, the standard mass values in nd 2 below should be used. The standard masses include hand nd, for helicopters, the mass of any infant below 24 months an adult on one passenger seat. Infants occupying separate seats are considered as children.			

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Table 1 of AMC2 OP values for passenge			nce system - Mass			
Passenger seats	20 and mor	е	30 and more			
	Male Female		`All adult'			
All flights except holiday charters	88 kg	70 kg	84 kg			
Holiday charters*	83 kg	69 kg	76 kg			
Children	35 kg	35 kg	35 kg			
travel package. On one or more charter travelling, all or in p purposes. The holid than 5 % of passen revenue carriage of company personnel	such flights the er rer(s) for the carr part by air, on a re ay charter mass v ger seats installed certain passenge , tour operators' s cc. can be included	ntire passenger iage of passenger ound- or circle- values apply pro d in the aircraft rs. Categories o staff, represent	trip basis for holiday ovided that not more are used for the non of passengers such as			
Table 2 of AMC2 OP values for passenge			nce system - Mass			
Passenger seats	1 – 5	6 - 9	10 - 19	MS: request to permit deduction of 6kg as per note to Table 2, to all	Operators will have the possibility to propose an alternative means	

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Male	104 kg	96 kg	92 kg	excluding larger aeroplanes from this rule;	future rulemaking task OPS.027 will also address standard mass values.	
Female	86 kg	78 kg	74 kg			
Children	35 kg	35 kg	35 kg			
On aeroplanes flights with 19 passenger seats or less and all helicopter flights where no hand baggage is carried in the cabin or where hand baggage is accounted for separately, 6 kg may be deducted from the figures in Table 2 above. The following items are not considered hand baggage: an overcoat, an umbrella, a small handbag or purse, reading material or a small camera.			n or where hand educted from the considered hand			
For helicopter operations in which a survival suit is provided to passengers, 3 kg should be added to the passenger mass value.						
4. Where the total number of passenger seats available on the aircraft is 20 or more, the standard mass values for checked baggage of Table 3 should be used.						
	2 OPS.GEN.310(a)(2) I gage - 20 or more seats		ice system - Mass			
Type of flight Baggage standard mass		MS: request to insert figures from	Not accepted: Table is already			
Domestic		11 kg		JAR-OPS 3.620(g) for helicopter ops with 20 or more seats, by adding to	applicable for aircraft, not only for aeroplanes.	
Within the Eur	ropean region	13 kg		fourth row "all other and all helicopter ops";		
Intercontinent	tal	15 kg				

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All other	13 kg			
ern ezn ezn ezn ezn ezn ezn ezn ezn ezn ez	ed within the			
Figure 1 - European region				
Domestic flight means a flight with borders of one State.	origin and destination within the			
Flights within the European reg domestic flights, whose origin and specified above.				

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	Intercontinental flights are flights beyond the European region with origin and destination in different continents.		
	For aircraft with 19 passenger seats or less, the mass of checked baggage should be determined by weighing.		
	For aircraft with 19 passenger seats or less used in non-commercial operations, the mass of checked baggage may also be calculated on the basis of a statement by, or on behalf of, each passenger. Where this is impractical, a minimum standard mass value of 13 kg should be used. The mass of checked baggage should be checked prior to loading and increased, if necessary.		
5.	The operator should determine the actual mass of passengers or checked baggage by weighing or add adequate mass increments whenever it can be expected that a significant number of passengers, including hand baggage, or checked baggage exceeds the standard masses.		
6.	Other standard masses may be used provided they are calculated on the basis of a detailed weighing survey plan and a reliable statistical analysis method is applied. The standard mass values should only be used in circumstances comparable with those under which the survey was conducted. Where these standard masses exceed those in Tables 1 - 3, then such higher values should be used.	Request to add "detailed weighing survey plan approved by the competent Authority";	Text not transposed - beyond the scope of Part-NCC.
AMC3 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations			Text not transposed - beyond the scope of Part-NCC.

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SPECIAL STANDARD MASSES FOR TRAFFIC LOAD In addition to standard masses for passengers/persons other than crew members and checked baggage, an operator may use standard mass values for other load items. These standard masses should be calculated on the basis of a detailed evaluation of the mass of the items.		
AMC4 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
PROCEDURE FOR ESTABLISHING REVISED STANDARD MASS VALUES FOR PASSENGERS AND BAGGAGE FOR AIRCRAFT USED IN COMMERCIAL AIR TRANSPORT		
1. Passengers		
a. Weight sampling method. The average mass of passengers and their hand baggage should be determined by weighing, taking random samples. The selection of random samples should by nature and extent be representative of the passenger volume, considering the type of operation, the frequency of flights on various routes, in/outbound flights, applicable season and seat capacity of the aircraft.		
i. Sample size. The survey plan should cover the weighing of at least the greatest of:		
A. A number of passengers calculated from a pilot sample, using normal statistical procedures and based on a relative confidence range (accuracy) of 1 % for all adult and 2 % for separate male and female		

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average masses; and		
B. For aircraft:		
1. With a passenger seating capacity of 40 or more, a total of 2 000 passengers; or		
2. With a passenger seating capacity of less than 40, a total number of 50 multiplied by the passenger seating capacity.		
b. Passenger masses should include the mass of the passengers' belongings which are carried when entering the aircraft. When taking random samples of passenger masses, infants should be weighted together with the accompanying adult.		
c. The location for the weighing of passengers should be selected as close as possible to the aircraft, at a point where a change in the passenger mass by disposing of or by acquiring more personal belongings is unlikely to occur before passengers board the aircraft.		
d. Weighing machines used for passenger weighing should have a capacity of at least 150 kg. The mass should be displayed at minimum graduations of 500 g. The weighing machine should have an accuracy of at least 0,5 % or 200 g whichever is greater.		
e. For each flight included in the survey the mass of the passengers, the corresponding passenger category (i.e. male/female/children) and the flight number should be recorded.		
2. Checked baggage The statistical procedure for determining revised standard baggage mass		

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values based on average baggage masses of the minimum required sample size should comply with paragraph (a) for passengers above. For baggage, the relative confidence range (accuracy) amounts to 1 %. A minimum of 2000 pieces of checked baggage should be weighed.			
3. Determination of revised standard checked baggage	l mass values for passengers and		
a. To ensure that, in preference to the determined by weighing, the use of revise passengers and checked baggage does not safety, a statistical analysis should be can generate average mass values for passent other data.	ed standard mass values for ot adversely affect operational rried out. Such an analysis should		
b. On aeroplanes with 20 or more passenger seats, these averages should apply as revised standard male and female mass values.			
c. On smaller aeroplanes, the following increments should be added to the average passenger mass to obtain the revised standard mass values:			
Table 1 of AMC4 OPS.GEN.310(a)(2)	Mass and balance system		
Number of passenger seats	Required mass increment		
1 – 5	16 kg		
6 - 9	8 kg		

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10 – 19 4 kg		
Alternatively, all adult revised standard (average) mass values may be applied on aeroplanes with 30 or more passenger seats. Revised standard (average) checked baggage mass values are applicable to aircraft with 20 or more passenger seats.		
d. All adult revised standard mass values should be based on a male/female ratio of 80/20 in respect of all flights except holiday charters which are 50/50. A different ratio on specific flights or routes may be used, provided supporting data shows that the alternative male/female ratio covers at least 84 % of the actual male/female ratios on a sample of at least 100 representative flights		
e. The resulting average mass values should be rounded to the nearest whole number in kg. Checked baggage mass values should be rounded to the nearest 0,5 kg figure, as appropriate.		
f. When operating on similar routes or networks, operators may pool their weighing surveys provided that in addition to the joint weighing survey results, results from individual operators participating in the joint survey are separately indicated in order to validate the joint survey results.		
GM1 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
ADJUSTMENT OF STANDARD MASSES FOR AIRCRAFT USED IN COMMERCIAL AIR TRANSPORT		

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When standard mass values are used, AMC2 OPS.GEN.310(a)(2) 5. states that the operator should identify and adjust the passenger and checked baggage masses in cases where significant numbers of passengers or quantities of baggage are suspected of exceeding the standard values. This implies that the operations manual should contain appropriate directives to ensure that:		
1. Check-in, operations and cabin staff and loading personnel report or take appropriate action when a flight is identified as carrying a significant number of passengers whose masses, including hand baggage, are expected to exceed the standard passenger mass, and/or groups of passengers carrying exceptionally heavy baggage (e.g. military personnel or sports teams); and		
2. On small aircraft, where the risks of overload and/or CG errors are the greatest, pilots pay special attention to the load and its distribution and make proper adjustments.		
GM2 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
STATISTICAL EVALUATION OF PASSENGERS AND BAGGAGE DATA FOR AIRCRAFT USED IN COMMERCIAL AIR TRANSPORT		
1. Sample size.		
a. For calculating the required sample size it is necessary to make an estimate of the standard deviation on the basis of standard deviations calculated for similar populations or for preliminary surveys. The precision		

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of a sample estimate is calculated for 95% reliability or 'significance', i.e. there is a 95% probability that the true value falls within the specified confidence interval around the estimated value. This standard deviation value is also used for calculating the standard passenger mass.		
b. As a consequence, for the parameters of mass distribution, i.e. mean and standard deviation, three cases have to be distinguished:		
i. μ , σ = the true values of the average passenger mass and standard deviation, which are unknown and which are to be estimated by weighing passenger samples.		
ii. $\mu', \sigma' =$ the 'a priori' estimates of the average passenger mass and the standard deviation, i.e. values resulting from an earlier survey, which are needed to determine the current sample size.		
iii. $x, s =$ the estimates for the current true values of m and s, calculated from the sample.		
The sa mple هاتدون الم n then be calculated using the following formula:		
where:		
n = number of passengers to be weighed (sample size)		
e'r = allowed relative confidence range (accuracy) for the estimate of μ by x (see also equation in paragraph 3). The allowed relative confidence range specifies the accuracy to be achieved when estimating the true mean. For example, if it is proposed to estimate the true mean to within \pm 1%, then e'r will be 1 in the above formula.		
= value from the Gaussian distribution for 95% significance level of the resulting confidence interval.		

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2. Calculation of average mass and standard deviation. If the sample of passengers weighed is drawn at random, then the arithmetic mean of the sample (x) is an unbiased estimate of the true average mass (μ) of the population		
Arithmetic mean of sample where:		
xj = n_{xj} mass values of individual passengers (sampling units).		
Is ■ Statulard deviation where: n – 1		
xj - = deviation of the individual value from the sample mean.		
3. Checking the accuracy of the sample mean. The accuracy (confidence range) which can be ascribed to the sample mean as an indicator of the true mean is a function of the standard deviation of the sample which has to be checked after the sample has been evaluated. This is the formula:		
whereby er should not exceed 1% for an all adult average mass and not exceed 2% for an average male and/or female mass. The result of this calculation gives the relative accuracy of the estimate of μ at the 95% significance level. This means that with 95% probability, the true average mass μ within the interval:		

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4. Example of determination of the required sample size and average passenger mass.		
a. Introduction. Standard passenger mass values for mass and balance purposes require passenger weighing programs be carried out. The following example shows the various steps required for establishing the sample size and evaluating the sample data. It is provided primarily for those who are not well versed in statistical computations. All mass figures used throughout the example are entirely fictitious.		
b. Determination of required sample size. For calculating the required sample size, estimates of the standard (average) passenger mass and the standard deviation are needed. The 'a priori' estimates from an earlier survey may be used for this purpose. If such estimates are not available, a small representative sample of about 100 passengers has to be weighed so that the required values can be calculated. The latter has been assumed for the example.		
Step 1: estimated average passenger mass.		
<u>n x_j (kg)</u>		
1 79.9		
2 68.1		
3 77.9		
4 74.5		
5 _ 54.1		
6 x 62.2		

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	7	89.3				
	8	108.7				
	85	63.2				
	85 85 1	75.4				
]=1	6 071.6				
Step 2:	estimated	l standard deviatio	n.			
<u>n</u>	xj	(xj - x)	(xj – x)2			
1	79.9	+9.3	86.49			
2	68.1	-2.5	6.25			
3	77.9	+7.3	53.29			
4	74.5	+3.9	15.21			
5	54.1	-16.5	272.25			
6	62.2	-8.4	70.56			
7	89.3	+18.7	349.69			
8	108.7	+38.1	1 451.61			
85	63.2	-7.4	54.76			
<u>86</u>	75.4	-4.8	23.04			

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$\frac{1}{1} = \frac{1}{n} = \frac{\sum x \beta}{n} = \frac{0.71}{86} = \frac{6}{86}$ = 70.6 kg $\sigma' = \sqrt{\frac{24}{86} + 40}$ $\sigma' = 20.20 kg$		
Step 3: required sample size.		
The required number of passengers to be weighed should be such that the confide $(1 - 96 + 30 + 100)^{2}$ does not exceed 1% as specified in paragraph 3. $n \ge (1 + 30 + 20 + 20 + 20 + 100)^{2}$ $n \ge 3145$ The result shows that at least 3 145 passengers have to be weighed to achieve the required accuracy. If e'r is chosen as 2 % the result would be $n \ge 786$.		
Step 4: after having established the required sample size a plan for weighing the passengers is to be worked out.		
c. Determination of the passenger average mass.		
Step 1: Having collected the required number of passenger mass values, the average passenger mass can be calculated. For the purpose of this example it has been assumed that 3 180 passengers were weighed. The		

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	sum of the individual masses amounts to 231 186.2 kg. 231188 2 kg $\frac{5}{2} \times 1^{180}$ 231188 2 kg $\frac{1}{2} \times 1$ 231188 2 kg $\frac{1}{2} \times 1$ 231188 2 kg $\frac{1}{2} \times 1$ 231188 2 kg $\overline{x} = 72.7$ kg		
	Step 2: calculation of the standard deviation. For calculating the standard deviation the method shown in paragraph 4.2 step 2 should be applied. $\begin{array}{r} & & & \\ & & & & \\ & & & \\ & & & & \\ & & & & \\ & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ &$		
	Step 3: 1.96 a curation of the accuracy of the sample 9, = $\frac{1.96 \times 15 \times 31 \times 100}{\sqrt{3180 \times 72.7}}$ er = 0.73 % Step 4: calculation of the confidence range of the sample $\overline{x} \pm \frac{meR. \times 8}{\sqrt{1}}$		

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72.7 ± 0.5 kg The result of this calculation shows that there is a 95% probability of the actual mean for all passengers lying within the range 72.2 kg to 73.2 kg.		
GM3 OPS.GEN.310(a)(2) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
GUIDANCE ON PASSENGER WEIGHING SURVEYS FOR AIRCRAFT USED IN COMMERCIAL AIR TRANSPORT		
1. Information to the competent authority. An operator should advise the competent authority about the intent of the passenger weighing survey and explain the survey plan in general terms.		
2. Detailed survey plan.		
a. An operator should establish and submit to the competent authority a detailed weighing survey plan that is fully representative of the operation, i.e. the network or route under consideration and the survey should involve the weighing of an adequate number of passengers.		
b. A representative survey plan means a weighing plan specified in terms of weighing locations, dates and flight numbers giving a reasonable reflection of the operator's timetable and/or area of operation.		

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c. The minimum number of passengers to be weighed is the highest of the following:		
 The number that follows from the means of compliance that the sample should be representative of the total operation to which the results will be applied; this will often prove to be the overriding requirement; or 		
 ii. The number that follows from the statistical requirement specifying the accuracy of the resulting mean values which should be at least 2% for male and female standard masses and 1% for all adult standard masses, where applicable. The required sample size can be estimated on the basis of a pilot sample (at least 100 passengers) or from a previous survey. If analysis of the results of the survey indicates that the requirements on the accuracy of the mean values for male or female standard masses or all adult standard masses, as applicable, are not met, an additional number of representative passengers should be weighed in order to satisfy the statistical requirements. 		
d. To avoid unrealistically small samples a minimum sample size of 2 000 passengers (males + females) is also required, except for small aircraft where in view of the burden of the large number of flights to be weighed to cover 2 000 passengers, a lesser number is considered acceptable.		
 Execution of weighing programme. At the beginning of the weighing programme it is important to note, and to account for, the data requirements of the weighing survey report (see 6. below). 		

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b.	As far as is practicable, the weighing programme should be conducted in accordance with the specified survey plan.		
c.	Passengers and all their personal belongings should be weighed as close as possible to the boarding point and the mass, as well as the associated passenger category (male/female/child), should be recorded.		
4.	Analysis of results of weighing survey.		
route flight from	The data of the weighing survey should be analysed as explained in OPS.GEN.310(a)(2). To obtain an insight to variations per flight, per etc. this analysis should be carried out in several stages, i.e. by to proute, by area, inbound/outbound, etc. Significant deviations the weighing survey plan should be explained as well as their ible effect(s) on the results.		
5.	Results of the weighing survey.		
and a be ju mass prop OPS. near value seats pass	results of the weighing survey should be summarised. Conclusions any proposed deviations from published standard mass values should astified. The results of a passenger weighing survey are average ses for passengers, including hand baggage, which may lead to osals to adjust the standard mass values given in AMC2 GEN.310(a)(2) Tables 1 and 2. These averages, rounded to the est whole number may, in principle, be applied as standard mass es for males and females on aircraft with 20 and more passenger s. Because of variations in actual passenger masses, the total enger load also varies and statistical analysis indicates that the risk of nificant overload becomes unacceptable for aircraft with less that 20		

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seats. This is the reason for passenger mass increments on small aircraft.		
b. The average masses of males and females differ by some 15 kg or more and because of uncertainties in the male/female ratio the variation of the total passenger load is greater if all adult standard masses are used than when using separate male and female standard masses. Statistical analysis indicates that the use of all adult standard mass values should be limited to aircrafts with 30 passenger seats or more.		
c. Standard mass values for all adults must be based on the averages for males and females found in the sample, taking into account a reference male/female ratio of 80/20 for all flights except holiday charters where a ratio of 50/50 applies. An operator may, based on the data from his weighing programme, or by proving a different male/female ratio, apply for approval of a different ratio on specific routes or flights.		
6. Weighing survey report		
The weighing survey report, reflecting the content of paragraphs 1–5 above, should be prepared in a standard format as follows:		
WEIGHING SURVEY REPORT		
1 Introduction Objective and brief description of the weighing survey.		
2 Weighing survey plan		
Discussion of the selected flight number, airports, dates, etc.		
Determination of the minimum number of passengers to be weighe	d.	

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S	Survey plan.			
3 A	Analysis and discussion of weighing survey results			
9	Significant deviations from survey plan (if any).			
\ \	Variations in means and standard deviations in the network.			
[] [Discussion of the (summary of) results.			
4 9	Summary of results and conclusions			
٩	Main results and conclusions.			
F	Proposed deviations from published standard mass values.			
Attach	nment 1			
Applic	cable summer and/or winter timetables or flight programmes.			
Attach	nment 2			
-	ning results per flight (showing individual passenger masses and se ard deviations per flight, per route, per area and for the total netwo			
motor-p	S.GEN.310(a)(3) Mass and balance system - complex oowered aircraft used in non-commercial operations and used in commercial operations		<u>.</u>	
FUEL LOA	AD			
	s of the fuel load should be determined by using its actual relative or a standard relative density.			
	.GEN.310(a)(3) Mass and balance system - complex oowered aircraft used in non-commercial operations and			

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aircraft used in commercial operations		
FUEL DENSITY		
1. If the actual fuel density is not known, the operator may use standard fuel density values for determining the mass of the fuel load. Such standard values should be based on current fuel density measurements for the airports or areas concerned.		
 2. Typical fuel density values are: a. Gasoline (piston engine fuel) - 0.71 b. JET A1 (Jet fuel JP 1) - 0.79 c. JET B (Jet fuel JP 4) - 0.76 d. Oil - 0.88 		
AMC OPS.GEN.310(a)(4) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		
LOADING - STRUCTURAL LIMITS		
The loading should take into account additional structural limits such as the floor strength limitations, the maximum load per running metre, the maximum mass per cargo compartment, and/or the maximum seating limits as well as in-flight changes in loading (e.g. hoist operations).		
AMC OPS.GEN.310(a)(7) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and		

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CG LIMITS – OPERATIONAL CG ENVELOPE - COMMERCIAL AIR TRANSPORT		
Unless seat allocation is applied and the effects of the number of persons per seat row, of cargo in individual cargo compartments and of fuel in individual tanks is accounted for in the balance calculation, operational margins should be applied to the certificated CG envelope. In determining the CG margins, possible deviations from the assumed load distribution should be considered. Passengers should be evenly distributed in the cabin. Operator procedures should fully account for the worst case variation in CG travel during flight caused by passenger/crew movement and fuel consumption/transfer.		
GM OPS.GEN. 310(a)(7) Mass and balance system - complex motor-powered aircraft used in non-commercial operations and aircraft used in commercial operations		
CG LIMITS – OPERATINAL CG ENVELOPE - COMMERCIAL AIR TRANSPORT		
In the Certificate Limitations section of the Aircraft Flight Manual, forward and aft CG limits are specified. These limits ensure that the certification stability and control criteria are met throughout the whole flight and allow the proper trim setting for take-off. An operator should ensure that these limits are observed by defining operational procedures or a CG envelope which compensates for deviations and errors as listed below:		
1. Deviations of actual CG at empty or operating mass from published values due, for example, to weighing errors, unaccounted modifications		

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and/or equipment variations.		
2. Deviations in fuel distribution in tanks from the applicable schedule.		
3. Deviations in the distribution of baggage and cargo in the various compartments as compared with the assumed load distribution as well as inaccuracies in the actual mass of baggage and cargo.		
4. Deviations in actual passenger seating from the seating distribution assumed when preparing the mass and balance documentation. Large CG errors may occur when 'free seating' (freedom of passengers to select any seat when entering the aircraft) is permitted. Although in most cases reasonably even longitudinal passenger seating can be expected, there is a risk of an extreme forward or aft seat selection causing very large and unacceptable CG errors (assuming that the balance calculation is done on the basis of an assumed even distribution). The largest errors may occur at a load factor of approximately 50% if all passengers are seated in either the forward or aft half of the cabin. Statistical analysis indicates that the risk of such extreme seating adversely affecting the CG is greatest on small aircraft.		
5. Deviations of the actual CG of cargo and passenger load within individual cargo compartments or cabin sections from the normally assumed mid position.		
6. Deviations of the CG caused by gear and flap positions and by application of the prescribed fuel usage procedure (unless already covered by the certified limits).		

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7. Deviations caused by in-flight movement of cabin crew equipment and passengers.	v, galley	
AMC OPS.GEN.310(a)(8) Mass and balance system - commotor-powered aircraft used in non-commercial operate aircraft used in commercial operations	-	
DOCUMENTATION - COMPLEX MOTOR-POWERED AIRCRAFT UNNON-COMMERCIAL OPERATIONS	JSED IN	
The mass and balance computation may be available in flight documents or separate systems and may include standard loa		
AMC OPS.GEN.310(a)(8) and (b) Mass and balance syst complex motor-powered aircraft used in non-commerci operations and aircraft used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
DOCUMENTATION - COMMERCIAL OPERATIONS		
1. Mass and balance documentation should contain the fo	ollowing:	
a. Aircraft registration and type;		
b. Flight identification number and date;		
c. Pilot-in-command;		
d. Person who prepared the information;		
e. Dry operating mass and corresponding CG of the aircra	aft;	
f. Mass of the fuel at take-off and mass of trip fuel;		
g. Mass of consumables other than fuel;		

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A: R	le	B: Summary of comments	C: Reason for change, remarks
h.	Load components including passengers, baggage, freight and ballast;		
i.	Take-off Mass, Landing Mass and Zero Fuel Mass;		
j.	Load distribution;		
k.	Applicable aircraft CG positions; and		
١.	The limiting mass and CG values.		
for ex balar	For Performance Class B aeroplanes and for helicopters, the CG on may not need to be on the mass and balance documentation, if, cample, the load distribution is in accordance with a pre-calculated ice table or if it can be shown that for the planned operations a ct balance can be ensured, whatever the real load is.		
3.	The mass and balance documentation should:		
a.	enable the pilot in command to determine that the load and its distribution are within the mass and balance limits of the aircraft; and		
b. meth	include advise to the pilot in command whenever a non-standard od has been used for determining the mass of the load.		
3. docu	The information above may be available in flight planning ments or mass and balance systems.		
the n opera	Any last minute change should be brought to the attention of the in-command and entered in the flight planning documents containing hass and balance information and mass and balance systems. The ator should specify the maximum last minute change allowed in enger numbers or hold load. New mass and balance documentation		

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should be prepared if this maximum number is exceeded.		
5. Where mass and balance documentation is generated by a computerised mass and balance system, the operator should verify the integrity of the output data at intervals not exceeding 6 months.		
6. A copy of the final mass and balance documentation may be sent to aircraft via data or may be made available to the pilot-in-command by other means for its acceptance.		
7. The person supervising the loading of the aircraft should confirm by hand signature or equivalent that the load and its distribution are in accordance with the mass and balance documentation given to the pilot in command. The pilot in command should indicate his acceptance by hand signature or equivalent.		
AMC OPS.GEN.315.B(b) Performance - general		Text not transposed - beyond the scope of Part-NCC.
BALLOON TAKE-OFF/LANDING IN CONGESTED AREAS		
A balloon, when becalmed over a congested area, should land within that congested area such that third parties on the ground, passengers and crew are not endangered.		
GM OPS.GEN.315.B(b) Performance - general		Text not transposed - beyond the scope of Part-NCC.
APPROVED OPERATING SITE FOR BALLOONS		
In approving congested sites for take-off of balloons, the competent		

authority should consider the following: 1. availability of performance data to determine the climb-out performance of the balloon, taking into account the take-off area and the prevailing meteorological conditions; 2. the surrounding area should permit a safe forced landing; and		·	
1. availability of performance data to determine the climb-out performance of the balloon, taking into account the take-off area and the prevailing meteorological conditions; 2. the surrounding area should permit a safe forced landing; and	A: Rule	B: Summary of comments	C: Reason for change, remarks
performance of the balloon, taking into account the take-off area and the prevailing meteorological conditions; 2. the surrounding area should permit a safe forced landing; and	authority should consider the following:		
	performance of the balloon, taking into account the take-off area and the		
3. the performance of the balloon should be such that a continuous	2. the surrounding area should permit a safe forced landing; and		
climb-out to the minimum safe altitude is ensured.			
AMC1 OPS.GEN.320.A(a) Take-off - complex motor-powered aeroplanes used in non-commercial operations and aeroplanes used in commercial operations	aeroplanes used in non-commercial operations and aeroplanes		
TAKE-OFF MASS - COMPLEX MOTOR-POWERED AEROPLANES AND AEROPLANES USED IN COMMERCIAL OPERATIONS			
The following should be considered for determining the maximum take-off mass:			
1. the pressure altitude at the aerodrome;	1. the pressure altitude at the aerodrome;		
2. the ambient temperature at the aerodrome;	2. the ambient temperature at the aerodrome;		
3. the runway surface condition and the type of runway surface;	3. the runway surface condition and the type of runway surface;		

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4.	the runway slope in the direction of take-off;		
5.	not more than 50% of the reported head-wind component or not less than 150% of the reported tailwind component; and	IS (BA): request better clarification to include or not include forecasted/expected gusts in the performance calculation;	Gusts are a temporary occurrence and should not be considered in the planning.
6.	the loss, if any, of runway length due to alignment of the aeroplane prior to take-off (for performance class A and class C aeroplanes an example is provided in appendix 2 to AMC OPS.CAT.A.316(a)(4)).	MS: delete reference to CAT;	Accepted. Reference deleted.
aer	C2 OPS.GEN.320.A(a) Take-off- complex motor-powered oplanes used in non-commercial operations and aeroplanes d in commercial operations		
CON	ITAMINATED RUNWAY PERFORMANCE DATA		
mar avai	and contaminated runway performance data, if made available by the nufacturer, should be taken into account. If such data is not made lable, the operator should account for wet and contaminated runway ditions by using the best information available.		
aer	1 OPS.GEN.320.A(a) Take-off - complex motor-powered oplanes used in non-commercial operations and aeroplanes d in commercial operations		
Ope	IWAY SURFACE CONDITION ration on runways contaminated with water, slush, snow or ice implies ertainties with regard to runway friction and contaminant drag and		

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therefore to the achievable performance and control of the aeroplane during take-off or landing, since the actual conditions may not completely match the assumptions on which the performance information is based. In the case of a contaminated runway, the first option for the pilot in command is to wait until the runway is cleared. If this is impracticable, he may consider a take-off or landing, provided that he has applied the applicable performance adjustments, and any further safety measures he/she considers justified under the prevailing conditions. The excess runway length available including the criticality of the overrun area should also be considered.		
AMC1 OPS.GEN.320.A(b) Take-off - complex motor-powered aeroplanes used in non-commercial operations and aeroplanes used in commercial operations		Text not transposed - beyond the scope of Part-NCC.
CONTINGENCY PROCEDURES FOR OBSTACLES CLEARANCES WITH ONE ENGINE INOPERATIVE – PERFORMANCE CLASS A AND CLASS C AEROPLANES IN COMMERCIAL AIR TRANSPORT OPERATIONS		
In the case of multi-engined aeroplanes, an operator should establish contingency procedures to provide a safe route, avoiding obstacles, to enable the aeroplane in the case of one engine inoperative to either comply with the en-route requirements or land at either the aerodrome of departure or at a take-off alternate aerodrome.		
GM1 OPS.GEN.320.A(b) Take-off - complex motor-powered aeroplanes used in non-commercial operations and aeroplanes used in commercial operations		Text not transposed - beyond the scope of Part-NCC.

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CONTINGENCY PROCEDURES FOR OBSTACLES CLEARANCES WITH ONE ENGINE INOPERATIVE – PERFORMANCE CLASS A AEROPLANES IN COMMERCIAL AIR TRANSPORT OPERATIONS		
Engine failure procedures for performance class A aeroplanes. If these procedures are based on an engine failure route that differs from the all engine departure route or SID normal departure, a "deviation point" can be identified where the engine failure route deviates from the normal departure route. Adequate obstacle clearance along the normal departure with failure of the critical engine at the deviation point will normally be available. However, in certain situations the obstacle clearance along the normal departure route may be marginal and should be checked to ensure that, in case of an engine failure after the deviation point, a flight can safely proceed along the normal departure.		
AMC2 OPS.GEN.320.A(b) Take-off - complex motor-powered aeroplanes used in non-commercial operations and aeroplanes used in commercial operations		
ADEQUATE MARGIN		
The adequate margin should be defined in the operations manual.		
GM2 OPS.GEN.320.A(b) Take-off - complex motor-powered aeroplanes used in non-commercial operations and aeroplanes used in commercial operations	MS: Request to clarify the reference to ICAO Annex as ICAO regulates commercial operations only. Request margins be stated here;	The intent of this GM is that the margin explained in Annex 6 Part I can be used for NCC operations.
ADEQUATE MARGIN		

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1. "An adequate margin" is illustrated by the appropriate examples included in Attachment C to ICAO Annex 6, Part I.		
2. Critical power-unit is the power-unit failure of which gives the most adverse effect on the aircraft characteristics relative to the case under consideration. On some aircraft there may be more than one equally critical power-unit. In this case, the expression "the critical power-unit" means one of those critical power-units (ICAO Annex 8).		Text not transposed - the term "critical power unit" is replaced with "critical engine" in this regulation.
GM OPS.GEN.325 One power-unit inoperative - complex motor- powered aircraft		Text not transposed - the GM refers to CAT AMC/GM material that is not included in Part-NCC.
HIGH TERRAIN OR OBSTACLE ANALYSIS		
Further guidance material can be found in the applicable acceptable means of compliances with OPS.CAT.340.A and OPS.CAT.365.H.		
AMC OPS.GEN.330.A Landing - complex motor-powered aeroplanes		
ALLOWANCES		
These allowances should be stated in the operations manual.		
Section IV - Instruments, data and equipment		
GM OPS.GEN.400(b) Instruments and equipments - General		
APPROVED EQUIPMENT		Deleted.

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shou	equipment approval in OPS.GEN.400(b) means that the equipment uld have an authorisation or an approval in accordance with Part-21 . European Technical Standards Order (ETSO) authorisation).		
GM:	1 OPS.GEN.400(c) Instruments and equipments - General		
NON	I-APPROVED EQUIPMENT		
1.	The provision of this paragraph does not exempt the item of equipment from complying with Part-21 if the instrument or equipment is installed in the aircraft. In this case, the installation should be approved as required in Part-21 and should comply with the applicable airworthiness codes.		
2.	The functionality of non-installed instruments and equipment required by Part-OPS which do not need an equipment approval should be checked against recognised industry standards appropriated for the intended purpose. The operator is responsible for ensuring the maintenance of these instruments and equipment.		
3.	The failure of additional non-installed instruments or equipment not required by Part-OPS or the airworthiness codes or any applicable airspace requirements should not adversely affect the airworthiness and/or the safe operation of the aircraft. Examples are the following:		
	a. Instruments supplying additional flight information (e.g. stand- alone Global Positioning System (GPS));		
	 Some aerial work equipment (e.g. some mission dedicated radios, wire cutters); 		

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c. Non-installed passenger entertainment equipment.		
GM2 OPS.GEN.400(c) Instruments and equipments - General		
LIST OF NON-APPROVED EQUIPMENT		
The following items are typical examples of equipment which do not need an equipment approval:		This is upgraded at IR level.
1. Electric torch;		
2. Accurate time piece;		
3. Child restraint devices		
4. Chart holder;		
5. First aid kits;		
6. Megaphones;		
7. Survival and signalling equipment;		
8. Sea anchors and equipment for mooring.GM OPS.GEN.400(e) Instruments and equipments - General		
ACCESSIBILITY AND POSITIONING OF INSTRUMENTS AND EQUIPMENT		
This requirement implies that whenever an instrument is required in an aircraft operated in a multi-crew environment, the instrument needs to be visible from each flight crew station.	This is not sensible guidance in all aircraft – in particular training aircraft with tandem seating. Replication	The text now refers to `single instrument', for clarity.

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	should be an acceptable alternative.	
GM OPS.GEN.405(a)(1) Equipment for all aircraft		
HAND FIRE EXTINGUISHERS		
 In the case of other than complex motor-powered aircraft, a hand fire extinguisher is considered to be necessary due to the development of toxic fumes when light composite materials catch fire. 	 The GM is not the proper place to give the rational for the provision requiring hand fire extinguisher. Furthermore composite materials can be found in any part of any aircraft. Proposal : delete paragraph 1. 	1. Deleted - not applicable to NCC.IDE.
2. For aerobatic flights, the hand fire extinguishers may become a hazard due to high G-loads.	1. MS: The GM is not the proper place to give the rational for the provision exempting hand fire extinguisher for aerobatic flights. Proposal: delete paragraph 2 and put a definition of aerobatic flights in OPS.GEN.010 Definitions "Any flights including manoeuvres other than"	1. Deleted - not applicable to NCC.IDE.
AMC OPS.GEN.405(a)(2) Equipment for all aircraft		
SEATS FOR MINIMUM REQUIRED CABIN CREW		

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close	e to tł	the minimum required cabin crew members should be located ne emergency exits and where cabin crew members can best sengers in the event of an emergency evacuation.		
АМС	C OPS	GGEN.405(a)(4) Equipment for all aircraft	1. MS, IND: Downgrade some of the standards to Guidance Material to allow updates of standards as they evolve.	1. Text already accommodates future evolution of standards.
RESTRAIN DEVICES FOR PERSON YOUNGER THAN 24 MONTHS - CHILD RESTRAINT DEVICES (CRD)				
1.	A ch	ild restraint device (CRD) is considered to be acceptable if:		
	a.	It is a 'supplementary loop belt' manufactured with the same techniques and the same materials of the approved safety belts; or		
	b.	It complies with paragraph (b) below.		
2.		ided the CRD can be installed properly on the respective aircraft , the following CRDs are considered "acceptable":		
	a.	CRDs approved for use in aircraft by a competent authority on the basis of a technical standard and marked accordingly.		
	b.	CRDs approved for use in motor vehicles according to the UN standard ECE R 44, -03 or later series of amendments; or		
	c.	CRDs approved for use in motor vehicles and aircraft according to Canadian CMVSS 213/213.1; or		
	d.	CRDs approved for use in motor vehicles and aircraft according to US FMVSS No 213 and are manufactured to these standards		

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	on or after February 26, 1985. US approved CRDs manufactured after this date must bear the following labels in red letters:		
	i. "THIS CHILD RESTRAINT SYSTEM CONFORMS TO ALL APPLICABLE FEDERAL MOTOR VEHICLE SAFETY STANDARDS" and		
	ii. "THIS RESTRAINT IS CERTIFIED FOR USE IN MOTOR VEHICLES AND AIRCRAFT".		
e.	CRDs qualified for use in aircraft according to the German "Qualification Procedure for Child Restraint Systems for Use in Aircraft" (TÜV Doc.: TÜV/958-01/2001).		
f.	Devices approved for use in cars, manufactured and tested to standards equivalent to those listed above. The device must be marked with an associated qualification sign, which shows the name of the qualification organisation and a specific identification number, related to the associated qualification project. The qualifying organisation should be a competent and independent organisation that is acceptable to the competent authority.		
3. Lo	cation		
a.	Forward facing CRDs may be installed on both forward and rearward facing passenger seats but only when fitted in the same direction as the passenger seat on which it is positioned. Rearward facing CRDs can only be installed on forward facing passenger seats. A CRD may not be installed within the radius of action of an airbag, unless it is obvious that the airbag is de-		

A: Rule		B: Summary of comments	C: Reason for change, remarks
	activated or it can be demonstrated that there is no negative impact from the airbag.		
b.	A child in a restraint device should be located as near to a floor level exit as feasible.		
C.	A child in a restraint device should not hinder evacuation for any passenger.		
d.	For complex motor-powered aircraft involved in commercial air transport, a child in a restraint device should neither be located in the row (where rows are existing) leading to an emergency exit nor located in a row immediately forward or aft of an emergency exit. A window passenger seat is the preferred location. An aisle passenger seat or a cross aisle passenger seat is not recommended. Other locations may be acceptable provided the access of neighbour passengers to the nearest aisle is not obstructed by the CRD.	1. IND: Cross-aisle seats with a bulkhead immediately in front are the normal locations for baby bassinette positions and are often requested by passengers with CRDs. I can see a case for CRDs not being in cross-aisles that are directly between exits i.e. without a bulkhead, as these form part of the evacuation route.	1. Not transposed in NCC.IDE – CAT requirement.
e.	For complex motor-powered aircraft involved in commercial air transport, in general, only one CRD per row segment is recommended. More than one CRD per row segment is allowed if the children are from the same family or travelling group provided the children are accompanied by a responsible person sitting next to them. A row segment is the fraction of a row separated by two aisles or by one aisle and the aircraft fuselage.		
4. Inst	callation		

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a.	CRDs should only be installed on a suitable aircraft seat with the type of connecting device they are approved or qualified for. E.g., CRDs to be connected by a three point harness only (most rearward facing baby CRDs currently available) should not be attached to an aircraft seat with a lap belt only, a CRD designed to be attached to a vehicle seat by means of rigid bar lower anchorages (ISO-FIX or US equivalent) only, should only be used on aircraft seats that are equipped with such connecting devices and should not be attached by the aircraft seat lap belt. The method of connecting should be the one shown in the manufacturer's instructions provided with each CRD.		
b.	All safety and installation instructions must be followed carefully by the responsible person accompanying the infant. For aircraft involved in commercial air transport, cabin crew should prohibit the use of any inadequately installed CRD or not qualified seat.		
c.	If a forward facing CRD with a rigid backrest is to be fastened by a lap belt, the restraint device should be fastened when the backrest of the passenger seat on which it rests is in a reclined position. Thereafter, the backrest is to be positioned upright. This procedure ensures better tightening of the CRD on the aircraft seat if the aircraft seat is reclinable.		
d.	The buckle of the adult safety belt must be easily accessible for both opening and closing, and must be in line with the seat belt halves (not canted) after tightening.		

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	e.	Forward facing restraint devices with an integral harness must not be installed such that the adult safety belt is secured over the child.		
5.	Ope	eration		
	a.	Each CRD should remain secured to a passenger seat during all phases of flight, unless it is properly stowed when not in use.		
	b.	Where a CRD is adjustable in recline it must be in an upright position for all occasions when passenger restraint devices are required.		
AMC fligh		5.GEN.410(a)(2) Flight instruments and equipment - VFR		
MEA	NS FO	OR MEASURING AND DISPLAYING THE TIME		
1.	com time	other than complex motor-powered aircraft not involved in imercial operations, a means of measuring and displaying the e in hours, minutes and seconds may be a wrist watch capable of same functions.		
2.	com cloc	complex motor-powered aircraft, an acceptable means of apliance with OPS.GEN.410(a)(2) should be considered to be a k displaying hours, minutes and seconds, with a sweep-second ater or digital presentation.		
AM0 fligh		5.GEN.410(a)(3) Flight instruments and equipment - VFR		

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CALIBRATION OF PRESSURE ALTITU	THE MEANS FOR MEASURING AND DISPLAYING JDE		
be calibrate	nent measuring and displaying pressure altitude should ed in feet (ft), with a sub-scale setting, calibrated in s/millibars, adjustable for any barometric pressure likely ring flight.		1. "sensitive type" added.
2. In the case acceptable.	of sailplanes and balloons, calibration in metres (m) is		
AMC OPS.GEN.410(a)(4) Flight instruments and equipment - VFR flights			
	THE INSTRUMENT INDICATING AIR SPEED - OPLANES AND HELICOPTERS		
The instrument indicating air speed should be calibrated in knots (kt). In the case of sailplanes with a maximum certificated take-off mass below 2 000 kg and aeroplanes other than complex motor-powered aeroplanes with a maximum certificated take-off mass below 2 000 kg, calibration in kilometres (km) per hour is acceptable.			
AMC OPS.GEN.410(d)(1)(i) Flight instruments and equipment - VFR flights			
MEANS OF INDICA	ATING DRIFT DIRECTION - BALLOONS		

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The drift direction may be determined by using a map and reference to visual landmarks.		
ALTIMETERS - AEROPLANES		
The altimeters of aeroplanes operating VFR flights when the aircraft cannot be maintained in a desired attitude without reference to one or more flight instruments, and of IFR flights, should have counter drum- pointer or equivalent presentation.	1. IND: Exemptions granted by NAA to operate aircraft equipped with counter drum-pointer presentation (e.g. MEP PA23 Aztec, PA34 Seneca, PN68 Partenavia) with operational restrictions to 10 000 ft operating altitude. Aircraft operate in public transport (PT) /commercial air transport (CAT), aerial work (AW) (flight training) and privately. The above cases shall remain acceptable, based on 10 000 ft altitude limitation (similar to RVSM, B- RNAV,etc).	1. An exception is introduced for unpressurised aeroplanes operating below 10 000 ft.
	Only when an aeroplane with a three pointer type altimeter flies above 10 000 feet is there any possibility of confusion being caused by the third pointer (which displays units of tens of thousands of feet). Therefore, the undoubted safety benefit of a counter drum-pointer altimeter only comes into effect when an aircraft flies above 10 000 feet. If an aeroplane is	

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	subject to an operational limit whereby it does not fly above 10 000 feet, altimeters other than those with a counter drum-pointer indicator would provide the pilot with an equivalent presentation to that achieved with a counter drum-pointer altimeter. A safety level equivalent to that attained by OPS 1.652(c) would be achieved.	
AMC OPS.GEN.410(b)(4) and OPS.GEN.415(a) Flight instruments and equipment - VFR flights and flight instruments and equipment - VFR night flights and IFR flights		
HEADING INDICATOR - HELICOPTERS		
Stabilised heading should be achieved for VFR flights by a gyroscopic direction indicator, whereas for IFR flights, this should be achieved through a magnetic gyroscopic direction indicator.		
AMC OPS.GEN.410 and OPS.GEN.415 Flight instruments and equipment - VFR flights and flight instruments and equipment - VFR night flights and IFR flights		
INTEGRATED INSTRUMENTS		
1. Individual equipment requirements may be met by combinations of instruments or by integrated flight systems or by a combination of parameters on electronic displays, provided that the information so		

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	available to each required pilot is not less than the required in the applicable operational requirements, and the equivalent safety of the installation has been shown during type certification approval of the aircraft for the intended type of operation.		
2.	The means of measuring and indicating turn and slip, aircraft attitude and stabilised aircraft heading may be met by combinations of instruments or by integrated flight director systems, provided that the safeguards against total failure, inherent in the three separate instruments, are retained.		
AMC OPS.GEN.410(c) and OPS.GEN.415(a) Flight instruments and equipment - VFR flights and flight instruments and equipment - VFR night flights and IFR flights			
-	TI-PILOT OPERATIONS - DUPLICATE INSTRUMENTS - AEROPLANES HELICOPTERS		
Duplicate instruments include separate displays for each pilot and separate selectors or other associated equipment where appropriate.			
	COPS.GEN.415(a)(1) Flight instruments and equipment - VFR It flights and IFR flights		
OUT	SIDE AIR TEMPERATURE		
1.	The instrument should be calibrated in degrees Celsius.	1. IND: This part cannot be applicable for third country aircraft, because if you change the OAT-meter from Fahrenheit to Celsius, the aircraft is	1. Not accepted. Calibration in degrees Celsius is maintained.

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	not airworthy anymore by the originator. "Should" shall be interpreted as "should".	
2. A means of indicating outside air temperature may be an air temperature indicator which provides indications that are convertible to outside air temperature.		
AMC OPS.GEN.415.A(a)(3) Flight instruments and equipment - VFR night flights and IFR flights		1. This AMC has been upgraded to IR level.
ALTERNATIVE SOURCE OF STATIC PRESSURE - AEROPLANES		
1. Aeroplanes should have two independent static pressure systems.		
2. However, in the case of propeller driven aeroplanes with a maximum certificated take-off mass of 5 700 kilograms (kg) or less, one static pressure and one alternative source of static pressure may be allowed.		
GM OPS.GEN.415(a)(5) Flight instruments and equipment - VFR night flights and IFR flights		
NAVIGATION LIGHTS		
Specifications for navigation lights are contained in Appendix 1 to International Civil Aviation Organization (ICAO) Annex 6, Part II.		

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AMC OPS.GEN.415.H(a)(6) Flight instruments and equipment - VFR night flights and IFR flights		
LANDING LIGHT - HELICOPTERS		
The landing light should be trainable, at least in the vertical plane		
AMC OPS.GEN.415(b) Flight instruments and equipment - VFR night flights and IFR flights		
CHART HOLDER		
An acceptable means of compliance with the chart holder requirement would be to display a pre-composed chart on an Electronic Flight Bag (EFB).		
AMC OPS.GEN.415(d) Flight instruments and equipment - VFR night flights and IFR flights		
BALLOON LIGHTS		
1. The position lights should be one steady aviation white position light, and one flashing aviation red position light (or flashing aviation white) with an effective flash frequency of at least 40, but not more than 100, cycles per minute.		
2. Both lights should have 360 degrees horizontal coverage and should be visible for at least 2 miles (approximately 3.22 km) under clear		

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atmospheric conditions.		
3. The white light should be located not more than 20 ft below the basket, trapeze, or other means for carrying occupants. The flashing red or white light should be located not less than seven ft or more than 10 ft below the steady white light.		
4. There should be a means to retract and store the lights.		
AMC OPS.GEN.420(e) Flights over water		
LIFE JACKETS - HELICOPTERS		
The life jacket should be accessible from the seat or berth of the person for whose use it is provided, with a safety belt or harness fastened.		
GM OPS.GEN.420(a)-(e) Flights over water		
LIFE JACKETS - ALL AIRCRAFT		
Seat cushions are not considered to be flotation devices.		
GM OPS.GEN.420(a), (d) and (f) Flights over water		

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EQUIPMENT FOR MAKING DISTRESS SIGNALS - SAILPLANES, BALLOONS, AEROPLANES AND HELICOPTERS		
The equipment for making distress signals is described in ICAO Annex 2.		
AMC OPS.GEN.420(a), (d) and (g) Flights over water	 Not a practical requirement CAT operations shall be excluded. The equipment should be repeated here, it shouldn't be necessary to purchase ICAO documents too. The header of this AMC tailored to helicopters and seaplanes/sailplanes and there is therefore probably an editorial error. This should read AMC OPS GEN 420 (c) (instead of (d)) as paragraph 2 cannot be complied with for aeroplanes other than sailplanes and seaplanes. 	1. This AMC has been deleted for CAT.IDE but maintained for NCC.IDE.
RISK ASSESSMENT		
 When conducting the risk assessment, the pilot-in-command should base his/her decision, as far as is practicable, on the implementing rules and AMCs applicable to the operation of aircraft. 		

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2.		pilot-in-command should, for the determining the risk, take the wing operating environment and conditions into account:		
	a.	Sea state;		
	b.	Sea and air temperatures;		
	c.	The distance from land suitable for making an emergency landing;		
	d.	The availability of search and rescue facilities.		
АМС	OPS	G.GEN.420(f) Flights over water		
LIFE	-SAV	ING RAFTS - HELICOPTERS		
	1.	At least 50% of the life rafts carried should be deployable by remote control.		
	2.	Rafts which are not deployable by remote control and which have a mass of more than 40 kg should be equipped with some means of mechanically assisted deployment.		
АМС	OPS	GGEN.420(h) Flights over water		
LIFE	JACK	(ETS		
The	mean	s of electric illumination should be a survivor locator light.	1. No such requirements today for small GA-aircraft.	1. This is an ICAO requirement.

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GM OPS.GEN.425.H Ditching - Helicopters		
PERFORMANCE CLASS 2 TAKE-OFF AND LANDING		
Helicopters operated in Performance Class 2 and taking off or landing over water are exposed to a critical power unit failure.		
GM OPS.GEN.430 Emergency Locator Transmitter (ELT)		
DEFINITION		
An Emergency Locator Transmitter (ELT) is a generic term describing equipment which broadcasts distinctive signals on designated frequencies and, depending on application, may be activated by impact or may be manually activated.		
AMC1 OPS.GEN.430 Emergency Locator Transmitter (ELT)		
ELT BATTERIES – MOTOR-POWERED AIRCRAFT		
Batteries used in the ELTs should be replaced (or recharged, if the battery is rechargeable) when the equipment has been in use for more than 1 cumulative hour, and also when 50% of their useful life (or for rechargeable, 50% of their useful life of charge), as established by the equipment manufacturer has expired. The new expiry date for the replacement (or recharged) battery should be legibly marked on the		

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outside of the equipment. The battery useful life (or useful life of charge) requirements of this paragraph do not apply to batteries (such as water-activated batteries) that are essentially unaffected during probable storage intervals.		
AMC2 OPS.GEN.430 Emergency Locator Transmitter (ELT)		
GENERAL – TYPES OF ELT AND GENERAL TECHNICAL SPECIFICATIONS		
1. The ELT required by this provision should be one of the following:		
a. Automatic Fixed (ELT(AF)). An automatically activated ELT which is permanently attached to an aircraft and is designed to aid SAR teams in locating the crash site;		
b. Automatic Portable (ELT(AP)). An automatically activated ELT, which is rigidly attached to an aircraft before a crash, but is readily removable from the aircraft after a crash. It functions as an ELT during the crash sequence. If the ELT does not employ an integral antenna, the aircraft-mounted antenna may be disconnected and an auxiliary antenna (stored on the ELT case) attached to the ELT. The ELT can be tethered to a survivor or a life raft. This type of ELT is intended to aid SAR teams in locating the crash site or survivor(s);		
c. Automatic Deployable (ELT(AD)). An ELT which is rigidly attached to the aircraft before the crash and which is automatically ejected, deployed and activated by an impact, and, in some cases, also by hydrostatic sensors. Manual		

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		deployment is also provided. This type of ELT should float in water and is intended to aid SAR teams in locating the crash site; or		
	d.	Survival ELT (ELT(S)). An ELT which is removable from an aircraft, stowed so as to facilitate its ready use in an emergency, and manually activated by a survivor. An ELT(S) may be activated manually or automatically (e.g. by water activation). It should be designed to be tethered to a life raft or a survivor.		
2.	the a far a so a	ninimize the possibility of damage in the event of crash impact, automatic ELT should be rigidly fixed to the aircraft structure, as aft as is practicable, with its antenna and connections arranged as to maximize the probability of the signal being transmitted r a crash.		
3.	prov with	ELT carried should operate in accordance with the relevant visions of ICAO Annex 10, Volume III and should be registered the national agency responsible for initiating search and rescue ther nominated agency.		
4.	ELTs MHz	s should be able to transmit on 121.5 MegaHertz (MHz) and 406	1. MS: The ELT rule does not include the frequency requirement, asking for 121.5 and 406 MHz capability. This, however, is included in AMC OPS.GEN.430 No.2, which is not acceptable.	1. The frequencies have been moved to IR level.
АМС	OPS	GGEN.430.H(b)(2) Emergency Locator Transmitter (ELT)		

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ELT(S) - HELICOPTERS		
An ELT(AP) may be used to replace one required ELT(S) provided that it meets the ELT(S) requirements. A water-activated ELT(S) is not an ELT(AP).		
GM OPS.GEN.435 Survival equipment – Motor powered aircraft		
AREAS IN WHICH SEARCH AND RESCUE WOULD BE ESPECIALLY DIFFICULT'		
The expression 'areas in which search and rescue would be especially difficult' should be interpreted, in this context, as meaning:		
1. Areas so designated by the competent authority responsible fo managing search and rescue; or	r	
 Areas that are largely uninhabited and where: a. The competent authority responsible for managing search and rescue has not published any information to confirm whethe search and rescue would be or would not be especially difficult and 	er ;;	
 b. The competent authority referred to in 1. does not, as a matte of policy, designate areas as being especially difficult fo search and rescue. 		
AMC OPS.GEN.435(a)(3) Survival equipment- Motor powered aircraft		

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ADD	ITIONAL SURVIVAL EQUIPMENT	1. The carriage of the knife/ice saw by passenger/crew may conflict with security requirements. Clarify that the knife is part of the aircraft equipment.	1. The security aspects have been reviewed by the Review Group and the text aligned with CAT.IDE.
1.	The following additional survival equipment should be carried when required:		
	a. 500 ml of water for each 4, or fraction of 4, persons on board;		
	b. One knife;		
	c. First Aid Equipment;		
	d. One set of Air/Ground codes;		
2.	In addition, when polar conditions are expected, the following should be carried:	1. MS: Polar definition need to be provided.	1. Polar conditions should be determined by the responsible
	a. A means for melting snow;		authority of the area.
	b. 1 snow shovel and 1 ice saw;		
	c. Sleeping bags for use by 1/3 of all persons on board and space blankets for the remainder or space blankets for all passengers on board;		
	d. 1 Arctic/Polar suit for each crew member carried.		
3.	If any item of equipment contained in the above list is already carried on board the aircraft in accordance with another requirement, there is no need for this to be duplicated.		
AMO	COPS.GEN.440(a) High altitude flights - Oxygen		1. Deleted - not applicable to NCC.IDE

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BREATHING OXYGEN		
Breathing oxygen should be provided by a quick-donning mask (See GM OPS.CAT.440(b)(1)).		
AMC OPS.GEN.440(a)(1)(i) High altitude flights – Oxygen		
PASSENGER OXYGEN SUPPLY BETWEEN 10 000 FT AND 13 000 FT		
On routes where the oxygen is necessary to be carried for 10% of the passengers for the flight time between 10 000 ft and 13 000 ft, the oxygen may be provided by:		
1. a plug-in or drop-out oxygen system with sufficient outlets and dispensing units uniformly distributed throughout the cabin so as to provide oxygen to each passenger at his/her own discretion when seated on his/her assigned seat; or		
2. portable bottles, when a cabin crew member is carried on board such flight.		
AMC OPS.GEN.440.A(a)(2) High altitude flights – Oxygen		
MAXIMUM ALTITUDE WITHOUT A PASSENGER OXYGEN SYSTEM – AEROPLANES		
For complex motor-powered pressurised aeroplanes and for pressurised aeroplanes used in commercial operations, the maximum altitude up to		

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which an aeroplane can operate without a passenger oxygen system being installed and capable of providing oxygen to each cabin occupant, should be established using an emergency descent profile which takes into account the following conditions:			
1.	17 seconds time delay for pilot's recognition and reaction, including mask donning, for trouble shooting and configuring the aeroplane for the emergency descent (emergency descent data/charts established by the aeroplane manufacturer and published in the Aircraft Flight Manual (AFM), and/or the AFM should be used to ensure uniform application of the option);		
2.	Maximum Operational Speed (VMO) or the airspeed approved in the AFM for emergency descent, (emergency descent data/charts established by the aeroplane manufacturer and published in the AFM, and/or AFM should be used to ensure uniform application of the option), whichever is the less;		
3.	All engines operative;		
4.	The estimated mass of the aeroplane at the top of climb.		
GM	OPS.GEN.440 High altitude flights – Oxygen		
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	high altitude flights concept is dealt with in detail in the ICAO Manual ivil Aviation Medicine.	1. IND: GM to be deleted taking into account the different approaches of	1. Accepted.

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	the FAA and JAA rules, more "state of the art" that the out-dated ICAO standards on oxygen.	
GM OPS.GEN.440(a)(2)(i) and (a)(3) High altitude flights – Oxygen		
DETERMINATION OF THE PROPORTION OF THE PASSENGERS FOR OXYGEN SUPPLY		
1. The determination of the proportion of the passengers for which oxygen should be supplied depends on the circumstances (e.g. maximum altitude, duration of the flight) of the flight to be undertaken and the performance of the aircraft.		
2. For pressurised aircraft not involved in commercial air transport operations, the requirements in Table 1 of OPS.CAT.440 may be used as guidance material.		
GM OPS.GEN.440(b) High altitude flights – Oxygen		
DUTIES ESSENTIAL TO THE SAFE OPERATION OF AN AIRCRAFT		
Flight duties which are essential to the safe operation of an aircraft include at least the control of aircraft in the flight path and the reaction to any normal, abnormal or emergency conditions.		
AMC OPS.GEN.450 Marking of break-in points		

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COLOUR AND CORNERS' MARKING		
 The colour of the markings should be red or yellow and, if necessary, should be outlined in white to contrast with the background. 		
 If the corner markings are more than 2 m apart, intermediate lines 9 centimetres (cm) x 3 cm should be inserted so that there is no more than 2 m between adjacent markings. 		
AMC1 OPS.GEN.455 First-aid kits		
OTHER THAN COMPLEX MOTOR-POWERED AIRCRAFT AND BALLOONS		
First-Aid Kits (FAKs) according to DIN 13164 or DIN 13157 are considered to meet the objective of OPS.GEN.455.	 DIN is a German norm. What about other norms? Proposal: Introduce all other accepted norms in the AMC or international standard. Do not refer to documents not available to the reader. 	1. & 2. A first-aid kit that is DIN compliant is one possible means to fulfil the requirement. However the AMC is providing the list of equipment needed to be contained in the FAK and this will be acceptable even without being DIN compliant.
AMC2 OPS.GEN.455 First-aid kits		
COMPLEX MOTOR-POWERED AIRCRAFT	1. The contents of the FAK have been amended and there is no justification. Proposal: Remove the amendment	1. The first-aid kit content has been amended taking into account the items for the on

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	and leave the contents "as is".	board medical supply introduced in Amendment 33A to ICAO Annex 6 Part I that are applicable since 19 November 2009.
 First-Aid Kits (FAKs) should be equipped with appropriate and sufficient medications and instrumentation. However, these kits should be adapted by the operator according to the characteristics of the operation (scope of operation, flight duration, number and demographics of passengers etc.). 	 FAK for CMPA should not be allowed to be adapted to the characteristics of the operation. But no problem if the operator wants to add some items. Since, according to 1., " these kits should be adapted by the operator according to the characteristics of the operation", does it mean that the equipment and medications presented in 2. are recommended, and not required? IV cannulae and bronchodilator are prescriptive items in some countries, such as the USA. The requirement for all presented items may bring the necessity to make different FAKs available for the operators, depending on the country of operation. This will impose a severe burden on operators to have to control the configuration of the on board FAK on a flight-by-flight basis. 	 Text changed to allow the operator to complement the kit instead of adapting it, to ensure the minimum content. Obligation to carry the items listed in the AMC: The operator may choose to propose an alternative MC. This must include a risk assessment to provide proof that the alternative means of compliance provide the same safety level as the AMC published by EASA.
2. The following should be included in the FAKs:		

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a.	Equipment:i.Bandages (assorted sizes);ii.Burns dressings (unspecified);iii.Wound dressings (large and small);iv.Adhesive dressings (assorted sizes);v.Adhesive tape;vi.Adhesive tape;vi.Safety pins;vii.Scissors;ix.Antiseptic wound cleaner;x.Disposable resuscitation aid;xi.Disposable gloves;xii.IV cannulae (if IV fluids are carried in the FAK, a sufficient supply of IV cannulae should be stored there as well).	 In order to use IV cannulae other equipment is required (such as wipes, a tourniquet, an appropriate IV dressing), which are included within the EMK requirements. IV cannulae should be carried in the emergency medical kit only, on medical, security and safety grounds. It is inappropriate to house IV cannulae in a FAK. Proposed text: Delete 2.a.xii 	1. Accepted. IV cannulae should be in the EMK only. None of the medication listed for FAK is injectable.
b.	Medications: i. Simple analgesic (may include liquid form); ii. Antiemetic; iii. Nasal decongestant; iv. Gastrointestinal antacid; v. Anti-diarrhoeal medication (for aircraft carrying more	1. A bronchial dilator is a prescription- only medication and should not be included in the contents of a first aid kit. It is a new requirement and cannot be justified for inclusion in a first aid kit; such medication is appropriately included in the list for an extended medical kit.	1. Accepted.

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	vi.	than 9 passengers); Bronchial dilator spray.	Delete paragraph 2.d.vi	
c.	Oth(i. ii. iii. iv. v.	er: A list of contents in at least 2 languages (English and one other). This should include information on the effects and side effects of medications carried; First-aid handbook; Medical incident report form; Biohazard disposal bags; Ground/Air visual signal code for use by survivors.	1. A ground/air visual signal code for use by survivors is a historical item that has no relevance to safety in the modern era. Justification: the contents list for a first-aid kit should only include items that add value and are relevant to current airline operations. Delete paragraph 2.c.v. It has been removed from ICAO amendment to Annex 6 Part I, effective from November 2009.	1. Accepted.
d.		eye irrigator, whilst not required to be carried in the FAK, uld, where possible, be available for use on the ground.		
e.		security reasons, items such as scissors should be stored urely.		
AMC OPS	GEN	I.455(d) First-aid kits		
MAINTENA	ANCE	OF FIRST AID KITS		
To be mai	ntain	ed first aid kits should be:		

A: R	Rule	B: Summary of comments	C: Reason for change, remarks
1.	inspected periodically to confirm, to the extent possible, that contents are maintained in the condition necessary for their intended use; and		
2.	replenished at regular intervals, in accordance with instructions contained on their labels, or as circumstances warrant.		
	LOPS.GEN.460(a) and (b) Airborne Collision Avoidance System AS) II	 It is proposed to replace this GM by inserting as published the entire contents of Attachments A and B to ICAO PANS-OPS (Doc 8168), Volume I, Part III, Chapter 3, replacing the existing NPA texts. It should be noted that the replacement text is largely the same as that published in the NPA but includes revisions that reflect developments and changes introduced since the original JAA TGL 11 text was developed. This GM is not related to this regulation. It is not useful and should therefore be deleted. In-stead, this guidance material should be put in a separate booklet without a link to this rule. 	1. & 2. This GM has been maintained and moved to Subpart NCC.OP together with the associated IR. This GM provides explanatory information to the IR.
GEN	ERAL		

A: Rule	2	B: Summary of comments	C: Reason for change, remarks
e: O	ICAO PANS-OPS, Volume 1 ; ICAO PANS-ATM ; and	1. The list of Guidance Material shown in GM1 was developed and published by the JAA in JAR-OPS 1 ACJ OPS 1.398 many years before ICAO published instructions for the operation of ACAS and training guidelines for pilots in PANS-OPS (Doc 8168). The text of GM2 had its origins in the JAA Temporary Guidance Leaflet 11 (which itself had been based upon Attachment E of ICAO State Letter AN 7/1.3.7.2-97/77, since superseded) and does not now accurately reflect current ICAO guidance. Subsequently, ICAO has published comprehensive instructions for the operation of ACAS and training guidelines for pilots in PANS-OPS, Volume I, Part III, Chapter 3 and Attachments A and B thereto with the specific intention that all pilots of aeroplanes and helicopters that are equipped with ACAS shall be taught and operate this equipment in accordance with exactly the same instructions. Note should be taken of ICAO Annex 6 Part II, Chapter 9, paragraph 9.1.2 that was designed to include pilots of aeroplanes whose operation would not necessarily be	 The GM has been updated to better reflect the referenced document and to address the general aviation ICAO Annex 6 Part II, Chapter 9, paragraph 9.1.2 requirements. The GM has been updated to align with proposed changes.

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	governed by an 'operations manual' (e.g. General Aviation). To address this omission in the NPA, and to retain the requirement that operators should specify identical procedures and training, replacement text is proposed, making use of words published in the Annex 6 Part II Standard. References in the NPA GM2 to TCAS II Version 6.04A and to ACAS III are redundant. 2. This GM only refers to ICAO documents and is not useful. We suggest to delete it	
2. Additional guidance material on ACAS may be referred to, including information available from such sources as Eurocontrol.		
GM2 OPS.GEN.460(a) and (b) Airborne Collision Avoidance System (ACAS) II		
ACAS FLIGHT CREW TRAINING PROGRAMMES		
1. During the implementation of ACAS, several operational issues were identified which had been attributed to deficiencies in flight crew training programmes. As a result, the issue of flight crew training has been discussed within the ICAO, which has developed guidelines for operators to use when designing training programmes.		

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2.	This guidance material contains performance-based training objectives for ACAS II flight crew training. Information contained in this paper related to Traffic Advisories (TAs) is also applicable to ACAS I and ACAS III users. The training objectives cover five areas: theory of operation; pre-flight operations; general in-flight operations; response to TAs; and response to Resolution Advisories (RAs).		
3.	The information provided is valid for TCAS II version 6.04A as well as for version 7 (ACAS II). Where differences arise, these are identified.		
4.	The performance based training objectives are further divided into the areas of: academic training; manoeuvre training; initial evaluation and recurrent qualification. Under each of these four areas, the training material has been separated into those items which are considered essential training items and those which are considered desirable. In each area, objectives and acceptable performance criteria are defined.		
5.	 ACAS Academic Training: a. This training is typically conducted in a classroom environment. The knowledge demonstrations specified in this section may be completed through the successful completion of written tests or through providing correct responses to non-real-time Computer Based Training (CBT) questions. 		
	b. Essential items:		

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i. Theory of operation. The flight crew member should demonstrate an understanding of ACAS operation and the criteria used for issuing TAs and RAs. This training should address the following topics:		
ii. System operation. Objective: To demonstrate knowledge of how ACAS functions. Criteria: The flight crew member should demonstrate an understanding of the following functions:		
1. Surveillance:		
a. ACAS interrogates other transponder-equipped aircraft within a nominal range of 14 nautical miles (nm).		
b. ACAS surveillance range can be reduced in geographic areas with a large number of ground interrogators and/or ACAS II-equipped aircraft.		
If the operator's ACAS implementation provides for the use of the Mode S extended squitter, the normal surveillance range may be increased beyond the nominal 14 nm. However, this information is not used for collision avoidance purposes.		
2. Collision avoidance:		
 a. TAs can be issued against any transponder-equipped aircraft which responds to the ICAO Mode C interrogations, even if the aircraft does not have altitude reporting capability. 		
 RAs can be issued only against aircraft that are reporting altitude and in the vertical plane only. 		
c. RAs issued against an ACAS-equipped intruder are co-ordinated to ensure complementary RAs are issued.		

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d. Failure to respond to an RA deprives own aircraft of the collision protection provided by own ACAS.		
e. Additionally, in ACAS-ACAS encounters, it also restricts the choices available to the other aircraft's ACAS and thus renders the other aircraft's ACAS less effective than if own aircraft were not ACAS equipped.		
B. Advisory thresholds. Objective: To demonstrate knowledge of the criteria for issuing TAs and RAs. Criteria: The flight crew member should demonstrate an understanding of the methodology used by ACAS to issue TAs and RAs and the general criteria for the issuance of these advisories, including the following:		
1. ACAS advisories are based on time to Closest Point of Approach (CPA) rather than distance. The time should be short and vertical separation should be small, or projected to be small, before an advisory can be issued. The separation standards provided by air traffic services are different from the miss distances against which ACAS issues alerts;		
2. Thresholds for issuing a TA or an RA vary with altitude. The thresholds are larger at higher altitudes;		
3. A TA occurs from 15 to 48 seconds and an RA from 15 to 35 seconds before the projected CPA;		
4. RAs are chosen to provide the desired vertical miss distance at CPA. As a result, RAs can instruct a climb or descent through the intruder aircraft's altitude.		
C. ACAS limitations. Objective: To verify that the flight crew member is aware of the limitations of ACAS. Criteria: The flight crew member		

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should demonstrate knowledge and understanding of ACAS limitations, including the following:		
1. ACAS will neither track nor display non-transponder-equipped aircraft, nor aircraft not responding to ACAS Mode C interrogations;		
2. ACAS will automatically fail if the input from the aircraft's barometric altimeter, radio altimeter or transponder is lost.		
a. In some installations, the loss of information from other on board systems such as an Inertial Reference System (IRS) or Attitude Heading Reference System (AHRS) may result in an ACAS failure. Individual operators should ensure that their flight crews are aware of the types of failure which will result in an ACAS failure.		
b. ACAS may react in an improper manner when false altitude information is provide to own ACAS or transmitted by another aircraft. Individual operators should ensure that their flight crews are aware of the types of unsafe conditions which can arise. Flight crews should ensure that when they are advised, if their own aircraft is transmitting false altitude reports, an alternative altitude reporting source is selected, or altitude reporting is switched off;		
3. Some aeroplanes within 380 ft above ground level (agl) (nominal value) are deemed to be 'on ground' and will not be displayed. If ACAS is able to determine an aircraft below this altitude is airborne, it will be displayed;		
4. ACAS may not display all proximate transponder-equipped aircraft in areas of high density traffic;		

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5. The bearing displayed by ACAS is not sufficiently accurate to support the initiation of horizontal manoeuvres based solely on the traffic display;		
6. ACAS will neither track nor display intruders with a vertical speed in excess of 10 000 ft/min. In addition, the design implementation may result in some short-term errors in the tracked vertical speed of an intruder during periods of high vertical acceleration by the intruder;		
7. Ground Proximity Warning Systems/Ground Collision Avoidance Systems (GPWSs/GCASs) warnings and wind-shear warnings take precedence over ACAS advisories. When either a GPWS/GCAS or wind- shear warning is active, ACAS aural annunciations will be inhibited and ACAS will automatically switch to the 'TA only' mode of operation.		
D. ACAS inhibits. Objective: To verify that the flight crew member is aware of the conditions under which certain functions of ACAS are inhibited. Criteria: The flight crew member should demonstrate knowledge and understanding of the various ACAS inhibits, including the following:		
1. "Increase Descent" RAs are inhibited below 1 450 ft agl;		
2. "Descend" RAs are inhibited below 1 100 ft agl;		
3. All RAs are inhibited below 1 000 ft agl;		
4. All TA aural annunciations are inhibited below 500 ft agl (1 000ft version 6.04A);		

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5. Altitude and configuration under which "Climb" and "Increase Climb" RAs are inhibited. ACAS can still issue "Climb" and "Increase Climb" RAs when operating at the aeroplane's certified ceiling. (In some aircraft types, "Climb" or "Increase Climb" RAs are never inhibited.)		
ii. Operating procedures. The flight crew member should demonstrate the knowledge required to operate the ACAS avionics and interpret the information presented by ACAS. This training should address the following:		
A. Use of controls. Objective: To verify that the pilot can properly operate all ACAS and display controls.		
Criteria: Demonstrate the proper use of controls including:		
1. Aircraft configuration required to initiate a self-test;		
2. Steps required to initiate a self-test;		
3. Recognising when the self-test was successful and when it was unsuccessful. When the self-test is unsuccessful, recognising the reason for the failure and, if possible, correcting the problem;		
4. Recommended usage of range selection. Low ranges are used in the terminal area and the higher display ranges are used in the en-route environment and in the transition between the terminal and en-route environment;		

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5. Recognising that the configuration of the display does not affect the ACAS surveillance volume;		
6. Selection of lower ranges when an advisory is issued, to increase display resolution;		
7. Proper configuration to display the appropriate ACAS information without eliminating the display of other needed information;		
8. If available, recommended usage of the above/below mode selector. The above mode should be used during climb and the below mode should be used during descent;		
9. If available, proper selection of the display of absolute or relative altitude and the limitations of using this display if a barometric correction is not provided to ACAS.		
B. Display interpretation. Objective: To verify that the flight crew member understands the meaning of all information that can be displayed by ACAS. The wide variety of display implementations require the tailoring of some criteria. When the training programme is developed, these criteria should be expanded to cover details for an operator's specific display implementation. Criteria: The flight crew member should demonstrate the ability to properly interpret information displayed by ACAS, including the following:		
1. Other traffic, i.e. traffic within the selected display range that is not proximate traffic, or causing a TA or RA to be issued;		
2. Proximate traffic, i.e. traffic that is within 6 nm and ± 1 200 ft;		

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3. Non-altitude reporting traffic;		
4. No bearing TAs and RAs;		
 Off-scale TAs and RAs. The selected range should be changed to 6. TAs. The minimum available display range which allows the traffic to be displayed should be selected, to provide the maximum display resolution; 		
7. RAs (traffic display). The minimum available display range of the traffic display which allows the traffic to be displayed should be selected, to provide the maximum display resolution;		
8. RAs (RA display). Flight crew members should demonstrate knowledge of the meaning of the red and green areas or the meaning of pitch or flight path angle cues displayed on the RA display Flight crew members should also demonstrate an understanding of the RA display limitations, i.e. if a vertical speed tape is used and the range of the tape is less than 2 500 ft/min, an increase rate RA cannot be properly displayed;		
9. If appropriate, awareness that navigation displays oriented on "Track-Up" may require a flight crew member to make a mental adjustment for drift angle when assessing the bearing of proximate traffic.		
C. Use of the TA only mode. Objective: To verify that a flight crew member understands the appropriate times to select the TA only mode of operation and the limitations associated with using this mode. Criteria: The flight crew member should demonstrate the following:		

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1. Knowledge of the operator's guidance for the use of TA only;		
2. Reasons for using this mode. If TA only is not selected when an airport is conducting simultaneous operations from parallel runways separated by less than 1 200 ft, and to some intersecting runways, RAs can be expected. If for any reason TA only is not selected and an RA is received in these situations, the response should comply with the operator's approved procedures.	1. The detail of the ACAS training as shown in C.2 is outdated, even if may still be included in ICAO GM. OPS.GEN.460 clearly states – in compliance with the current ICAO PANS-OPS – that "ACAS shall be used in normal conditions during flight in a mode that enables Resolution Advisories (RAs)"	1. The conditions to use TA only mode have been updated.
	"Normal conditions" do include parallel runway operations, and these situations should no longer be promulgated as cases when to switch off RA. Switching to "TA ONLY" is restricted to "non-normal" or "abnormal" conditions in connection with technical malfunctions that might prevent correct compliance with RAs.	
3. All TA aural annunciations are inhibited below 500 ft agl (1 000 ft agl for version 6.04A). As a result, TAs issued below 500 ft agl may not be noticed unless the TA display is included in the routine instrument scan.		
D. Crew co-ordination. Objective: To verify that the flight crew member understands how ACAS advisories will be handled. Criteria: The flight crew member should demonstrate knowledge of the crew procedures that should be used when responding to TAs and RAs,		

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including the following:		
1. Task sharing between the pilot flying and the pilot not flying;		
2. Expected call-outs;		
3. Communications with Air Traffic Control (ATC).		
E. Phraseology requirements. Objective: To verify that the flight crew member is aware of the requirements for reporting RAs to the controller. Criteria: The flight crew member should demonstrate the following:		
1. The use of the phraseology contained in ICAO PANS-OPS;		
2. An understanding of the procedures contained in ICAO PANS-ATM and ICAO Annex 2;		
3. The understanding that verbal reports should be made promptly to the appropriate ATC unit:		
a. whenever any manoeuvre has caused the aeroplane to deviate from an air traffic clearance;		
b. when, subsequent to a manoeuvre that has caused the aeroplane to deviate from an air traffic clearance, the aeroplane has returned to a flight path that complies with the clearance; and/or		
c. when air traffic issue instructions that, if followed, would cause the crew to manoeuvre the aircraft contrary to an RA with which they are complying.		
F. Reporting requirements. Objective: To verify that the flight crew member is aware of the requirements for reporting RAs to the operator. Criteria: The flight crew member should demonstrate knowledge of where		

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information can be obtained regarding the need for making written reports to various states when an RA is issued. Various states have different reporting requirements and the material available to the flight crew member should be tailored to the operator's operating environment. For operators involved in commercial operations, this responsibility is satisfied by the flight crew member reporting to the operator according to the applicable reporting requirements.		
c. Non-essential items: Advisory thresholds. Objective: To demonstrate knowledge of the criteria for issuing TAs and RAs. Criteria: The flight crew member should demonstrate an understanding of the methodology used by ACAS to issue TAs and RAs and the general criteria for the issuance of these advisories, including the following:		
i. The minimum and maximum altitudes below/above which TAs will not be issued;		
ii. When the vertical separation at CPA is projected to be less than the ACAS-desired separation, a corrective RA which requires a change to the existing vertical speed will be issued. This separation varies from 300 ft at low altitude to a maximum of 700 ft at high altitude;		
iii. When the vertical separation at CPA is projected to be just outside the ACAS-desired separation, a preventive RA which does not require a change to the existing vertical speed will be issued. This separation varies from 600 to 800 ft;		
iv. RA fixed range thresholds vary between 0.2 and 1.1 nm.		

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6. ACAS manoeuvre training:		
a. Demonstration of the flight crew member's ability to use ACAS displayed information to properly respond to TAs and RAs, should be carried out in a flight simulator equipped with an ACAS display and controls similar in appearance and operation to those in the aircraft. If a flight simulator is utilised, CRM should be practised during this training.		
b. Alternatively, the required demonstrations can be carried out by means of an interactive CBT with an ACAS display and controls similar in appearance and operation to those in the aircraft. This interactive CBT should depict scenarios in which real-time responses should be made. The flight crew member should be informed whether or not the responses made were correct. If the response was incorrect or inappropriate, the CBT should show what the correct response should be.		
c. The scenarios included in the manoeuvre training should include: corrective RAs; initial preventive RAs; maintain rate RAs; altitude crossing RAs; increase rate RAs; RA reversals; weakening RAs; and multi-aircraft encounters. The consequences of failure to respond correctly should be demonstrated by reference to actual incidents such as those publicised in Eurocontrol ACAS II 'safety flash' Bulletins.		
i. TA responses. Objective: To verify that the pilot properly interprets and responds to TAs. Criteria: The pilot should demonstrate the following:		
A. Proper division of responsibilities between the pilot flying and the pilot not flying. The pilot flying should fly the aircraft using any type-specific procedures and be prepared to respond to any RA that might follow. For aircraft without an RA pitch display, the pilot flying should		

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consider the likely magnitude of an appropriate pitch change. The pilot not flying should provide updates on the traffic location shown on the ACAS display, using this information to help visually acquire the intruder;		
B. Proper interpretation of the displayed information. Flight crew members should confirm that the aircraft they have visually acquired is that which has caused the TA to be issued. Use should be made of all information shown on the display, note being taken of the bearing and range of the intruder (amber circle), whether it is above or below (data tag), and its vertical speed direction (trend arrow);		
C. Other available information should be used to assist in visual acquisition, including ATC "party-line" information, traffic flow in use, etc.;		
D. Because of the limitations described, the pilot flying should not manoeuvre the aircraft based solely on the information shown on the ACAS display. No attempt should be made to adjust the current flight path in anticipation of what an RA would advise, except that if own aircraft is approaching its cleared level at a high vertical rate with a TA present, vertical rate should be reduced to less than 1 500 ft/min;		
E. When visual acquisition is attained, and as long as no RA is received, normal right of way rules should be used to maintain or attain safe separation. No unnecessary manoeuvres should be initiated. The limitations of making manoeuvres based solely on visual acquisition, especially at high altitude or at night, or without a definite horizon should be demonstrated as being understood.		
ii. RA responses. Objective: To verify that the pilot properly interprets and responds to RAs. Criteria: The pilot should demonstrate the following:		

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A. Proper response to the RA, even if it is in conflict with an ATC instruction and even if the pilot believes that there is no threat present;		
B. Proper task sharing between the pilot flying and the pilot not flying. The pilot flying should respond to a corrective RA with appropriate control inputs. The pilot not flying should monitor the response to the RA and should provide updates on the traffic location by checking the traffic display. Proper Crew Resource Management (CRM) should be used;		
C. Proper interpretation of the displayed information. The pilot should recognise the intruder causing the RA to be issued (red square on display). The pilot should respond appropriately;		
D. For corrective RAs, the response should be initiated in the proper direction within five seconds of the RA being displayed. The change in vertical speed should be accomplished with an acceleration of approximately ¼ g (gravitational acceleration of 9.81 m/sec ²);		
E. Recognition of the initially displayed RA being modified. Response to the modified RA should be properly accomplished, as follows:		
1. For increase rate RAs, the vertical speed change should be started within two and a half seconds of the RA being displayed. The change in vertical speed should be accomplished with an acceleration of approximately ¹ / ₃ g;		
2. For RA reversals, the vertical speed reversal should be started within two and a half seconds of the RA being displayed. The change in vertical speed should be accomplished with an acceleration of approximately ¹ / ₃ g;		

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3. For RA weakenings, the vertical speed should be modified to initiate a return towards the original clearance;		
4. An acceleration of approximately ¼ g will be achieved if the change in pitch attitude corresponding to a change in vertical speed of 1 500 ft/min is accomplished in approximately five seconds, and of ⅓ g if the change is accomplished in approximately three seconds. The change in pitch attitude required to establish a rate of climb or descent of 1 500 ft/min from level flight will be approximately six degrees when the True Air Speed (TAS) is 150 kt, four degrees at 250 kt, and two degrees at 500 kt. (These angles are derived from the formula: 1000 divided by TAS.)		
F. Recognition of altitude crossing encounters and the proper response to these RAs;		
G. For preventive RAs, the vertical speed needle or pitch attitude indication, should remain outside the red area on the RA display;		
H. For maintain rate RAs, the vertical speed should not be reduced. Pilots should recognise that a maintain rate RA may result in crossing through the intruder's altitude;		
I. When the RA weakens, or when the green 'fly to' indicator changes position, the pilot should initiate a return towards the original clearance, and when "clear of conflict" is annunciated, the pilot should complete the return to the original clearance;		
J. The controller should be informed of the RA as soon as time and workload permit, using the standard phraseology;		

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K. When possible, an ATC clearance should be complied with while responding to an RA. For example, if the aircraft can level at the assigned altitude while responding to RA (an "adjust vertical speed" RA (version 7), "reduce climb" or "reduce descent" RA (version 6.04A), it should be done; the horizontal (turn) element of an ATC instruction should be followed;		
L. Knowledge of the ACAS multi-aircraft logic and its limitations, and that ACAS can optimise separations from two aircraft by climbing or descending towards one of them. For example, ACAS only considers intruders which it considers to be a threat when selecting an RA. As such, it is possible for ACAS to issue an RA against one intruder which results in a manoeuvre towards another intruder which is not classified as a threat. If the second intruder becomes a threat, the RA will be modified to provide separation from that intruder.		
7. ACAS initial evaluation:		
a. The flight crew member's understanding of the academic training items should be assessed by means of a written test or interactive CBT that records correct and incorrect responses to phrased questions.		
b. The flight crew member's understanding of the manoeuvre training items should be assessed in a flight simulator equipped with an ACAS display and controls similar in appearance and operation to those in the aircraft the flight crew member will fly, and the results assessed by a qualified instructor, inspector, or check airman. The range of scenarios should include: corrective RAs; initial preventive RAs; maintain rate RAs; altitude crossing RAs; increase rate RAs; RA reversals; weakening RAs; and multi-threat encounters. The scenarios should also include demonstrations of the consequences of not responding to RAs, slow or	1. ECA does not recognise the figure of check airman.	1. Reference to "check airman" has been deleted.

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late responses, and manoeuvring opposite to the direction called for by the displayed RA.		
c. Alternatively, exposure to these scenarios can be conducted by means of an interactive CBT with an ACAS display and controls similar in appearance and operation to those in the aircraft the pilot will fly. This interactive CBT should depict scenarios in which real-time responses should be made and a record made of whether or not each response was correct.		
8. ACAS recurrent training:		
a. ACAS recurrent training ensures that flight crew members maintain the appropriate ACAS knowledge and skills. ACAS recurrent training should be integrated into and/or conducted in conjunction with other established recurrent training programmes. An essential item of recurrent training is the discussion of any significant issues and operational concerns that have been identified by the operator. Recurrent training should also address changes to ACAS logic, parameters or procedures and to any unique ACAS characteristics which flight crew members should be made aware of.		
b. It is recommended that operator's recurrent training programmes using flight simulators include encounters with conflicting traffic when these simulators are equipped with ACAS. The full range of likely scenarios may be spread over a two-year period. If a flight simulator, as described above, is not available, use should be made of an interactive CBT that is capable of presenting scenarios to which pilot responses should be made in real-time.	1. Requirement to train the full range of scenarios over a 2 year period is very restrictive and does not comply with the spirit of the GM principle.	1. Reference to a 2 year period of time is adapted to a mere generic guidance.

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AMC OPS.GEN.465.A(c) Terrain Awareness Warning System (TAWS) - Aeroplanes		
TAWS WARNING		
The warning referred in OPS.GEN.465.A(c) should be provided by aural signals that may be supplemented by visual signals.		
GM1 OPS.GEN.465.A Terrain Awareness Warning System (TAWS) - Aeroplanes		
TAWS CLASS A AND TAWS CLASS B		
The minimum performance standards for TAWS Class A and TAWS Class B equipment are described in the Agency's ETSO-C151a .		
GM2 OPS.GEN.465.A Terrain Awareness Warning System (TAWS) - Aeroplanes		Transposed in NCC.OP and associated to NCC.OP.210. Text aligned with the CAT GM.
GUIDANCE MATERIAL FOR TAWS FLIGHT CREW TRAINING PROGRAMMES		
1. Introduction:		
a. This guidance material contains performance-based training objectives for TAWS flight crew training.		

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t	The training objectives cover five areas: theory of operation; pre-flight operations; general in-flight operations; response to TAWS cautions; and response to TAWS warnings.		
C	The term 'TAWS' in this guidance material means a Ground Proximity Warning System (GPWS) enhanced by a forward- looking terrain avoidance function. Alerts include both cautions and warnings.		
C	1. The content of this guidance material is intended to assist operators who are producing training programmes. The information it contains has not been tailored to any specific aircraft or TAWS equipment, but highlights features which are typically available where such systems are installed. It is the responsibility of the individual operator to determine the applicability of the content of this guidance material to each aircraft and TAWS equipment installed and their operation. Operators should refer to the AFM and/or Aircraft/Flight Crew Operating Manual A/FCOM, or similar documents, for information applicable to specific configurations. If there should be any conflict between the content of this guidance material and that published in the other documents described above, then information contained in the AFM or A/FCOM will take precedence.		
2. 9	Scope:		
ā	 The scope of this guidance material is designed to identify training objectives in the areas of: academic training; manoeuvre training; initial evaluation; and recurrent 		

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	qualification. Under each of these four areas, the training material has been separated into those items which are considered essential training items and those which are considered to be desirable. In each area, objectives and acceptable performance criteria are defined.		
	b. No attempt is made to define how the training programme should be implemented. Instead, objectives are established to define the knowledge that a pilot operating a TAWS is expected to possess and the performance expected from a pilot who has completed TAWS training. However, the guidelines do indicate those areas in which the pilot receiving the training should demonstrate his/her understanding, or performance, using a real-time, interactive training device, ie a flight simulator. Where appropriate notes are included within the performance criteria which amplify or clarify the material addressed by the training objective.		
3.	Performance-based training objectives:		
	a. TAWS academic training:		
i.	This training is typically conducted in a classroom environment. The knowledge demonstrations specified in this section may be completed through the successful completion of written tests or by providing correct responses to non-real-time CBT questions.		
ii.	Theory of operation. The pilot should demonstrate an understanding of TAWS operation and the criteria used for issuing cautions and warnings. This training should address system operation. Objective:		

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	To demonstrate knowledge of how a TAWS functions. Criteria: The pilot should demonstrate an understanding of the following functions:		
Α.	Surveillance:		
1.	The GPWS computer processes data supplied from an air data computer, a radio altimeter, an Instrument Landing System (ILS)/Microwave Landing System(MLS)/Multi-Mode (MM) receiver, a roll attitude sensor, and flap and gear selector position sensors.		
2.	The forward looking terrain avoidance function utilises an accurate source of known aircraft position, such as that which may be provided by a Flight Management System (FMS) or GPS, or an electronic terrain database. The source and scope of the terrain, obstacle and airport data, and features such as the terrain clearance floor, the runway picker, and geometric altitude (where provided), should all be described.		
3.	Displays required to deliver TAWS outputs include a loudspeaker for voice announcements, visual alerts (typically amber and red lights), and a terrain awareness display (that may be combined with other displays). In addition, means should be provided for indicating the status of the TAWS and any partial or total failures that may occur.		
В.	Terrain avoidance. Outputs from the TAWS computer provide visual and audio synthetic voice cautions and warnings to alert the flight crew about potential conflicts with terrain and obstacles.		

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С.	Alert thresholds. Objective: To demonstrate knowledge of the criteria for issuing cautions and warnings. Criteria: The pilot should be able to demonstrate an understanding of the methodology used by a TAWS to issue cautions and alerts and the general criteria for the issuance of these alerts, including:		
1.	basic GPWS alerting modes specified in the ICAO Standard:		
Мос	le 1: excessive sink rate;		
Мос	le 2: excessive terrain closure rate;		
Мос	le 3: descent after take-off or go-around;		
Мос	le 4: unsafe proximity to terrain;		
Мос	le 5: descent below ILS glide slope (caution only); and		
2.	an additional, optional alert mode:- Mode 6: radio altitude call-out (information only); TAWS cautions and warnings which alert the flight crew to obstacles and terrain ahead of the aircraft in line with or adjacent to its projected flight path (Forward-Looking Terrain Avoidance (FLTA) and Premature Descent Alert (PDA) functions).		
D.	TAWS limitations. Objective: To verify that the pilot is aware of the limitations of TAWS. Criteria: The pilot should demonstrate knowledge and an understanding of TAWS limitations identified by the manufacturer for the equipment model installed, such as:		

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1.	navigation should not be predicated on the use of the terrain display;		
2.	unless geometric altitude data is provided, use of predictive TAWS functions is prohibited when altimeter subscale settings display 'QFE';		
3.	nuisance alerts can be issued if the aerodrome of intended landing is not included in the TAWS airport database;		
4.	in cold weather operations, corrective procedures should be implemented by the pilot unless the TAWS has in-built compensation, such as geometric altitude data;		
5.	loss of input data to the TAWS computer could result in partial or total loss of functionality. Where means exist to inform the flight crew that functionality has been degraded, this should be known and the consequences understood;		
6.	radio signals not associated with the intended flight profile (eg ILS glide path transmissions from an adjacent runway) may cause false alerts;		
7.	inaccurate or low accuracy aircraft position data could lead to false or non-annunciation of terrain or obstacles ahead of the aircraft; and		
8.	Minimum Equipment List (MEL) restrictions should be applied in the event of the TAWS becoming partially or completely unserviceable. (It should be noted that basic GPWS has no forward-looking capability.)		

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E.	TAWS inhibits. Objective: To verify that the pilot is aware of the conditions under which certain functions of a TAWS are inhibited. Criteria: The pilot should demonstrate knowledge and an understanding of the various TAWS inhibits, including the following:		
1.	A means of silencing voice alerts;		
2.	A means of inhibiting ILS glide path signals (as may be required when executing an ILS back beam approach);		
3.	A means of inhibiting flap position sensors (as may be required when executing an approach with the flaps not in a normal position for landing);		
4.	A means of inhibiting the FLTA and PDA functions;		
5.	A means of selecting or deselecting the display of terrain information, together with appropriate annunciation of the status of each selection.		
b.	Operating procedures. The pilot should demonstrate the knowledge required to operate TAWS avionics and to interpret the information presented by a TAWS. This training should address the following topics:		
i.	Use of controls. Objective: To verify that the pilot can properly operate all TAWS controls and inhibits. Criteria: The pilot should demonstrate the proper use of controls, including the following:		
Α.	The means by which, before flight, any equipment self-test functions can be initiated;		

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В.	The means by which TAWS information can be selected for display;		
C.	The means by which all TAWS inhibits can be operated and what the consequent annunciations mean with regard to loss of functionality.		
ii.	Display interpretation. Objective: To verify that the pilot understands the meaning of all information that can be annunciated or displayed by a TAWS. Criteria: The pilot should demonstrate the ability to properly interpret information annunciated or displayed by a TAWS, including the following:		
Α.	Knowledge of all visual and aural indications that may be seen or heard;		
в.	Response required on receipt of a caution;		
C.	Response required on receipt of a warning;		
D.	Response required on receipt of a notification that partial or total failure of the TAWS has occurred (including annunciation that the present aircraft position is of low accuracy).		
iii.	Use of basic GPWS or use of the FLTA function only. Objective: To verify that the pilot understands what functionality will remain following loss of the GPWS or of the FLTA function. Criteria: The pilot should demonstrate knowledge of the following:		
Α.	How to recognise un-commanded loss of the GPWS function, or how to isolate this function and how to recognise the level of the remaining Controlled Flight Into Terrain (CFIT) protection (essentially, this is the FLTA function);		

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В.	How to recognise un-commanded loss of the FLTA function, or how to isolate this function and how to recognise the level of the remaining CFIT protection (essentially, this is the basic GPWS).		
iv.	Crew co-ordination. Objective: To verify that the pilot adequately briefs other flight crew members on how TAWS alerts will be handled. Criteria: The pilot should demonstrate that the pre-flight briefing addresses procedures that will be used in preparation for responding to TAWS cautions and warnings, including the following:		
А.	The action to be taken, and by whom, in the event that a TAWS caution and/or warning is issued;		
В.	How multi-function displays will be used to depict TAWS information at take-off, in the cruise and for the descent, approach, landing (and any go-around). This will be in accordance with procedures specified by the operator, who will recognise that it may be more desirable that other data is displayed at certain phases of flight and that the terrain display has an automatic 'pop-up' mode in the event that an alert is issued.		
v.	Reporting requirements. Objective: To verify that the pilot is aware of the requirements for reporting alerts to the controller and other authorities. Criteria: The pilot should demonstrate knowledge of the following:		
Α.	When, following recovery from a TAWS alert or caution, a transmission of information should be made to the appropriate air traffic control unit;		

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В.	The type of written report which is required, how it is to be compiled, and whether any cross reference should be made in the aircraft technical log and/or voyage report (in accordance with procedures specified by the operator), following a flight in which the aircraft flight path has been modified in response to a TAWS alert, or if any part of the equipment appears not to have functioned correctly.		
vi.	Alert thresholds. Objective: To demonstrate knowledge of the criteria for issuing cautions and warnings. Criteria: The pilot should be able to demonstrate an understanding of the methodology used by a TAWS to issue cautions and warnings and the general criteria for the issuance of these alerts, including:		
Α.	awareness of the modes associated with basic GPWS, including the input data associated with each; and		
В.	awareness of the visual and aural annunciations that can be issued by TAWS and how to identify which are cautions and which are warnings.		
с.	TAWS manoeuvre training. The pilot should demonstrate the knowledge required to respond correctly to TAWS cautions and warnings. This training should address the following topics:		
i.	Response to cautions:		
Α.	Objective: To verify that the pilot properly interprets and responds to cautions. Criteria: The pilot should demonstrate an understanding of the need, without delay:		

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1.	to initiate action required to correct the condition which has caused the TAWS to issue the caution and to be prepared to respond to a warning, if this should follow; and		
2.	if a warning does not follow the caution, to notify the controller of the new position, heading and/or altitude/flight level of the aircraft, and what the pilot-in-command intends to do next.		
В.	The correct response to a caution might require the pilot:		
1.	to reduce a rate of descent and/or to initiate a climb;		
2.	to regain an ILS glide path from below, or to inhibit a glide path signal if an ILS is not being flown;		
3.	to select more flap, or to inhibit a flap sensor if the landing is being conducted with the intent that the normal flap setting will not be used;		
4.	to select gear down; and/or		
5.	to initiate a turn away from the terrain or obstacle ahead and towards an area free of such obstructions if a forward-looking terrain display indicates that this would be a good solution and the entire manoeuvre can be carried out in clear visual conditions.		
ii.	Response to warnings. Objective: To verify that the pilot properly interprets and responds to warnings. Criteria: The pilot should demonstrate an understanding of the following:		
Α.	The need, without delay, to initiate a climb in the manner specified by the operator;		

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В.	The need, without delay, to maintain the climb until visual verification can be made that the aircraft will clear the terrain or obstacle ahead or until above the appropriate sector safe altitude (if certain about the location of the aircraft with respect to terrain) even if the TAWS warning stops. If, subsequently, the aircraft climbs up through the sector safe altitude, but the visibility does not allow the flight crew to confirm that the terrain hazard has ended, checks should be made to verify the location of the aircraft and to confirm that the altimeter subscale settings are correct;		
C.	When the workload permits, that the flight crew should notify the air traffic controller of the new position and altitude/flight level, and what the pilot-in-command intends to do next;		
D.	That the manner in which the climb is made should reflect the type of aircraft and the method specified by the aircraft manufacturer (which should be reflected in the operations manual) for performing the escape manoeuvre. Essential aspects will include the need for an increase in pitch attitude, selection of maximum thrust, confirmation that external sources of drag (e.g. spoilers/speed brakes) are retracted, and respect of the stick shaker or other indication of eroded stall margin;		
E.	That TAWS warnings should never be ignored. However, the pilot's response may be limited to that which is appropriate for a caution, only if:		
1.	the aircraft is being operated by day in clear, visual conditions; and		
ii.	it is immediately clear to the pilot that the aircraft is in no danger in respect of its configuration, proximity to terrain or current flight		

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	path.		
d.	TAWS initial evaluation:		
i.	The flight crew member's understanding of the academic training items should be assessed by means of a written test.		
ii.	The flight crew member's understanding of the manoeuvre training items should be assessed in a flight simulator equipped with TAWS visual and aural displays and inhibit selectors similar in appearance and operation to those in the aircraft which the pilot will fly. The results should be assessed by a synthetic flight instructor, synthetic flight examiner, type rating instructor or type rating examiner.		
iii.	The range of scenarios should be designed to give confidence that proper and timely responses to TAWS cautions and warnings will result in the aircraft avoiding a CFIT accident. To achieve this objective, the pilot should demonstrate taking the correct action to prevent a caution developing into a warning and, separately, the escape manoeuvre needed in response to a warning. These demonstrations should take place when the external visibility is zero, though there is much to be learnt if, initially, the training is given in 'mountainous' or 'hilly' terrain with clear visibility. This training should comprise a sequence of scenarios, rather than be included in Line Orientated Flying Training (LOFT).		
iv.	A record should be made, after the pilot has demonstrated competence, of the scenarios that were practised.		

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e.	TAWS recurrent training:		
i.	TAWS recurrent training ensures that pilots maintain the appropriate TAWS knowledge and skills. In particular, it reminds pilots of the need to act promptly in response to cautions and warnings, and of the unusual attitude associated with flying the escape manoeuvre.		
ii.	An essential item of recurrent training is the discussion of any significant issues and operational concerns that have been identified by the operator. Recurrent training should also address changes to TAWS logic, parameters or procedures and to any unique TAWS characteristics of which pilots should be aware.		
f.	Reporting procedures:		
i.	Verbal reports. Verbal reports should be made promptly to the appropriate air traffic control unit:		
А.	whenever any manoeuvre has caused the aircraft to deviate from an air traffic clearance;		
в.	when, following a manoeuvre which has caused the aircraft to deviate from an air traffic clearance, the aircraft has returned to a flight path which complies with the clearance; and/or		
C.	when an air traffic control unit issues instructions which, if followed, would cause the pilot to manoeuvre the aircraft towards terrain or obstacle or it would appear from the display that a potential CFIT occurrence is likely to result.		

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ii.	Written reports. Written reports should be submitted in accordance with the operator's occurrence reporting scheme and they also should be recorded in the aircraft technical log:		
Α.	whenever the aircraft flight path has been modified in response to a TAWS alert (false, nuisance or genuine);		
В.	whenever a TAWS alert has been issued and is believed to have been false; and/or		
C.	if it is believed that a TAWS alert should have been issued, but was not.		
iii.	Within this guidance material, and with regard to reports:		
Α.	the term 'false' means that the TAWS issued an alert which could not possibly be justified by the position of the aircraft in respect to terrain and it is probable that a fault or failure in the system (equipment and/or input data) was the cause;		
В.	the term 'nuisance' means that the TAWS issued an alert which was appropriate, but was not needed because the flight crew could determine by independent means that the flight path was, at that time, safe;		
C.	the term 'genuine' means that the TAWS issued an alert which was both appropriate and necessary; and		
D.	the report terms described in GM2 OPS.GEN.A.465 3.f.iii are only meant to be assessed after the occurrence is over, to facilitate		

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subsequent analysis, the adequacy of the equipment and the programmes it contains. The intention is not for the flight crew to attempt to classify an alert into any of these three categories when visual and/or aural cautions or warnings are annunciated.		
AMC OPS.GEN.485.A Crash axes and crowbars - Aeroplanes	1. MS: The content of this AMC was in the rule text of OPS 1.795 (b)	1. Accepted. Moved to CAT.IDE.
	2. MS: Move to CAT requirements OPS.CAT.485.A Crash axes and crowbars.	
POSITION OF CRASH AXES AND CROWBARS FOR AEROPLANES USED IN COMMERCIAL AIR TRANSPORT OPERATIONS		
For aeroplanes used in commercial air transport operations, crash axes and crowbars located in the passenger compartment should be stored in a position not visible to passengers.		
AMC1 OPS.GEN.490.A Flight data recorder - Aeroplanes		
LIST OF PARAMETERS TO BE RECORDED FOR AEROPLANES FIRST ISSUED WITH AN INDIVIDUAL CERTIFICATE OF AIRWORTHINESS ON OR AFTER 1 JANUARY 2010		
1. The Flight Data Recorder (FDR) should, with reference to a timescale, record:	The "competent Authority responsible for type certification or supplemental	The Agency is responsible for the approval of novel and unique
a. the parameters listed in Table 1 of AMC1 OPS.GEN.490.A;b. the additional parameters listed in Table 2 of AMC1OPS.GEN.490.A, when the information data source for the parameter is	type certification". This wording is different from EU-OPS. Moreover, the Authority responsible for type certification or supplemental	designs, and also for the approval of dedicated parameters. Note: The competent authority mentioned in NPA OPS.GEN.001

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A: Rule	B: Summary of comments	C: Reason for change, remarks
either used by aeroplane systems or is available on the instrument panel for use by the flight crew to operate the aeroplane; and c. any dedicated parameters related to novel or unique design or operational characteristics of the aeroplane, as determined by the competent authority responsible for the type certification or supplemental type certification.	type certification would be EASA whereas EU-OPS refers to "the competent Authority" (NAA).	of NPA-2009-2b relates to the oversight of operations and not to certification. Note: in Regulation (EC) No 2042/2003 ² , the competent authority is defined as such in M.1: "1. for the oversight of the continuing airworthiness of individual aircraft and the issue of airworthiness review certificates the authority designated by the Member State of registry. 2. for the oversight of a maintenance organisation as specified in M.A. Subpart F, (i) the authority designated by the Member State where that organisation's principle place of business is located. 4. for the approval of maintenance programmes, (i) the authority designated by the Member State of registry."

² Commission Regulation (EC) No 2042/2003on the continuing airworthiness of aircraft and aeronautical products, parts and appliances, and on the approval of organisations and personnel involved in these tasks. *OJ L 315, 28.11.2003, p. 1.*

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A: Ru	A: Rule		B: Summary of comments	C: Reason for change, remarks
minim	The parameters to be recorded should meet the period ications (designated ranges, sampling intervals, account resolution in read-out) as defined in the relevance an Organisation for Civil Aviation Equipment (EURC	uracy limits and nt tables of the		
Tab	le 1 of AMC1 OPS.GEN.490.A			
No.*	Parameter			
1a	Time; or			
1b	Relative time count			
1c	Global Navigation Satellite System (GNSS) time synchronisation			
2	Pressure altitude			
3a	Indicated air speed; or			
3b	Calibrated air speed			
4	Heading (primary flight crew reference) - when true or magnetic heading can be selected, the primary heading reference, a discrete indicating selection, should be recorded			
5	Normal acceleration			
6	Pitch attitude]		

A: Rul	le	 B: Summary of comments	C: Reason for change, remarks
7	Roll attitude		
8	Manual radio transmission keying and Cockpit Voice Recorder (CVR)/FDR synchronisation reference.		
9	Engine thrust/power		
9a 9b	Parameters required to determine propulsive thrust/power on each engine		
-	Cockpit thrust/power lever position for aeroplanes with non-mechanically linked cockpit - engine control		
14	Total or Outside Air Temperature (OAT)		
16	Longitudinal acceleration (body axis)		
17	Lateral acceleration		
18 18a 18b 18c	Primary flight control surface and primary flight control pilot input (for multiple or split surfaces, a suitable combination of inputs is acceptable in lieu of recording each surface separately. For aeroplanes which have a flight control break- away capability that allows either pilot to operate the controls independently, record both inputs):		
	pitch axis		

		Ture Nee			50 / lug 2011
A: Ru	A: Rule		B: Summary of c	comments	C: Reason for change, remarks
	roll axis yaw axis				
19	Pitch trim surface position				
23	Marker beacon passage				
24	Warnings - in addition to the master warning each "red" warning (including smoke warnings from other compartments) should be recorded when the warning condition cannot be determined from other parameters or from the CVR				
25	Each navigation receiver frequency selection				
27	Air - ground status and, if the sensor is installed, each landing gear				
38	Selected barometric setting - to be recorded for the aeroplane in which the parameter is displayed electronically				
44	Selected flight path (all pilot selectable modes of operation) - to be recorded for the aeroplane in which the parameter is displayed electronically				
45	Selected decision height - to be recorded for the a displayed electronically	aeroplane in which	the parameter is		
75	All cockpit flight control input forces (for fly-by-wi	re flight control sy	stems, where]	

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1 410 1100	0.01

A: Ru	le		B: Summary of c	omments	C: Reason for change, remarks
	control surface position is a function of the displace only, it is not necessary to record this parameter):		rol input device		
75a	Control wheel				
75b	Control column				
75c	Rudder pedal cockpit input forces				
* depict	The number in the left hand column reflects the sered in EUROCAE ED-112.	rial number			
AEROF PARAN ON TH	IE 2 OF AMC1 OPS.GEN.490.A PLANES FOR WHICH THE INFORMATION DATA SOUR IETER IS EITHER USED BY AEROPLANE SYSTEMS OF E INSTRUMENT PANEL FOR USE BY THE FLIGHT CRE EROPLANE.	R IS AVAILABLE			
N o. *	Parameter				
10	Flaps: Trailing edge flap position and cockpit control selection				
11	Slats: Leading edge flap (slat) position and cockpit control selection				
12	Thrust reverse status				
13	Ground spoiler and speed brake				
13a	Ground spoiler position				

			50 / lug 2011
A: Ru	le	B: Summary of comments	
13b	Ground spoiler selection		
13c	Speed brake position		
13d	Speed brake selection		
15	Autopilot/autothrottle/Automatic Flight Control System (AFCS) mode and engagement status		
20	Radio altitude. For autoland/Category 3 operations, each radio altimeter should be recorded. It is acceptable to arrange them so that at least one is recorded every second.		
21 21a	Vertical deviation - the approach aid in use should be recorded. For autoland/Category 3 operations, each system should be recorded. It is acceptable to arrange them so that at least one is recorded every second).		
21b	ILS glide path		
21c	MLS elevation		
	GNSS approach path/IRNAV vertical deviation		

		Part-NCC	CRST	30 Aug 2011
A: Ru	le		B: Summary of comments	C: Reason for change, remarks
22 22a 22b 22c	Horizontal deviation - the approach aid in use should be recorded. For auto land/Category 3 operations, each system should be recorded. It is acceptable to arrange them so that at least one is recorded every second). ILS localiser MLS azimuth GNSS approach path/IRNAV lateral deviation			
26	DME 1 and 2 distances		1. MS: Align parameters 21 and 22 with table II-A.1 of EUROCAE ED 112.	1. Regulatory justification:
28 28a 28b 28c	GPWS/TAWS/GCAS status: Selection of terrain display mode, including pop- up display status Terrain alerts, including cautions and warnings and advisories On/off switch position			Consistency with table A3.1-1 in ICAO Annex 6 Part II (see parameters 21 and 22).
29	Angle of attack			
30 30a 30b 31	Low pressure warning (each system): Hydraulic pressure Pneumatic pressure Ground speed			
32	Landing gear:			

		Part-NCC	CRST	30 Aug 2011
A: Ru	A: Rule		B: Summary of comments	C: Reason for change, remarks
32a	Landing gear			
32b 33 33a 33b 33c 33d 33e 33f 34	Gear selector position Navigation data: Drift angle Wind speed Wind direction Latitude Longitude GNSS augmentation in use Brakes:			
34a 34b	Left and right brake pressure Left and right brake pedal position			
35 35a 35b 35c 35d 35e 35f	Additional engine parameters (if not already recorded in parameter 9 of Table 1 of AMC1 OPS.GEN.490.A and if the aeroplane is equipped with a suitable data source): Engine Pressure Ratio (EPR) N_1 Indicated vibration level N_2 Exhaust Gas Temperature (EGT)			

A: Ru	le	 B: Summary of comments	C: Reason for change, remarks
35g	Fuel flow		
35h	Fuel cut-off lever position		
	N ₃		

A: Ru	le	B: Summary of comments	C: Reason for change, remarks
36 36a 36b 36c 36d 36e	Traffic Alert and Collision Avoidance System (TCAS)/ACAS - a suitable combination of discrete should be recorded to determine the status of system: Combined control Vertical control Up advisory Down advisory Sensitivity level	1. IND:) Item 26 should contain GLS distance to threshold.	1. Regulatory justification: Consistency with table A3.1-1 in ICAO Annex 6 Part II (see parameter 26).
37	Wind-shear warning		
38a	Pilot		
38b	First officer		
39	Selected altitude (All pilot selectable modes of operation) - to be recorded for the aeroplane where the parameter is displayed electronically		
40	Selected speed (All pilot selectable modes of operation) - to be recorded for the aeroplane where the parameter is displayed electronically		
41	Selected Mach (All pilot selectable modes of operation) - to be recorded for the aeroplane where the parameter is displayed electronically		
42	Selected vertical speed (All pilot selectable modes of operation) - to be recorded for the aeroplane where the parameter is displayed electronically		Page 422 of 483
43	Selected heading (All pilot selectable modes of		

		Part-NCC	CRST	30 Aug 2011
A: Rı	A: Rule		B: Summary of comments	C: Reason for change, remarks
47	Multi-function/engine/alerts display format			
48	AC electrical bus status - each bus			
49	DC electrical bus status - each bus			
50	Engine bleed valve position			
51	APU bleed valve position			
52	Computer failure - critical flight and engine control system			
53	Engine thrust command			
54	Engine thrust target			
55	Computed centre of gravity			
56	Fuel quantity or fuel quantity in CG trim tank			
1				
57	Head up display in use	1. IND: Parameter 38: it should be added selected barometric setting	1. Editorial:	
58	Para visual display on		The parameter definition that appears in table II-A.1 of ED 112 was only partially transcribed	

A: Ru	le	B: Summary of comments	C: Reason for change, remarks
59	Operational stall protection, stick shaker and pusher activation		here.
60	Primary navigation system reference:		
60a	GNSS		
60b	Inertial Navigational System (INS)		
60c	VOR/DME		
60d	MLS		
60e	Loran C		
60f	ILS		
61	Ice detection		
62	Engine warning - each engine vibration		
63	Engine warning - each engine over temperature		
64	Engine warning - each engine oil pressure low		
65	Engine warning - each engine over speed		
66	Yaw trim surface position		
67	Roll trim surface position		
68	Yaw or sideslip angle		
69	De-icing and/or anti-icing systems selection		

		Part-NCC CRS	Т	30 Aug 2011
A: Rı	A: Rule		Summary of comments	C: Reason for change, remarks
70	Hydraulic pressure - each system			
71	Loss of cabin pressure *			
72	Cockpit trim control input position pitch - when mechanical means for control inputs are not available, cockpit display trim positions or trim command should be recorded			
73	Cockpit trim control input position roll - when mechanical means for control inputs are not available, cockpit display trim positions or trim command should be recorded			
74	Cockpit trim control input position yaw - when mechanical means for control inputs are not available, cockpit display trim positions or trim command should be recorded			
76	Event marker			
77	Date			
78	ANP or EPE or EPU			
* depic	The number in the left hand column reflects the seria ted in EUROCAE ED-112.	al number		
Tał	ble 3 of AMC2 OPS.GEN.490.A			

		Part-NCC		1.1	30 Aug 2011
A: Ru	le		B: Summary of commo	ents	C: Reason for change, remarks
additio syster		rameters for Aeroplanes equipped with electronic display			
No.	No.*	Parameter			
33	6	Selected barometric setting (each pilot's station)			
34	7	Selected altitude			
35	8	Selected speed			
36	9	Selected mach			
37	10	Selected vertical speed			
38	11	Selected heading			
39	12	Selected flight path			
40	13	Selected decision height			
41	14	EFIS display format			
42	15	Multi-function/engine/alerts display format			
*		umber in the centre column reflects the serial numbers ed in table A1.5 of EUROCAE ED-55.			
АМС3	OPS.C	SEN.490.A Flight data recorder - Aeroplanes			

		Part-NCC	CRST	30 Aug 2011
A: Ru	ıle		B: Summary of comments	C: Reason for change, remarks
WITH	AN IN	AMETERS TO BE RECORDED FOR AEROPLANES FIRST ISSUED DIVIDUAL CERTIFICATE OF AIRWORTHINESS ON OR AFTER 1 JP TO AND INCLUDING 31 MARCH 1998		
1.		light data recorder should, with reference to a timescale, d the parameters listed in Table 1 of AMC3 OPS.GEN.490.A.		
2.	certifi recore mass of Tal	determined by the competent authority responsible for type ication or supplemental type certification, the flight data der of aeroplanes having a maximum certificated take-off of 27 000 kg does not need to record parameters 14 and 15b ble 1 of AMC3 OPS.GEN.490.A when any of the following tions are met:		
	a.	The sensor is not readily available;		
	b.	Sufficient capacity is not available in the FDR system;		
	c.	A change is required in the equipment that generates the data.		
3.	certifi Ageno paran	determined by the competent authority responsible for type ication or supplemental type certification and agreed by the cy, the flight data recorder does not need to record neters 15b 23, 24, 25, 26, 27, 28, 29, 30 and 31 of Table 1 of 8 OPS.GEN.490.A, when any of the following conditions are		
	a.	The sensor is not readily available;		
	b.	Sufficient capacity is not available in the FDR system;		
	C.	A change is required in the equipment that generates the data;		

		Part-NCC	CRST	30 Aug 2011
A: Ru	A: Rule		B: Summary of comments	C: Reason for change, remarks
	d.	For navigational data (NAV frequency selection, DME distance, latitude, longitude, ground speed and drift), the signals are not available in digital form;		
	e.	When the above conditions have been met and compliance with this AMC would imply significant modifications to the aeroplane with a severe re-certification effort.		
4.	certifi Ageno paran	determined by the competent authority responsible for type ication or supplemental type certification and agreed by the cy, the flight data recorder does not need to record individual neters that can be derived by calculation from the other ded parameters.		
Tab	le 1 of	f AMC3 OPS.GEN.490.A		
Aerop kg	lanes v	vith a maximum certificated take-off mass exceeding 27 000		
No.	Parar	neter		
1	Time	or relative time count		
2	Press	sure altitude		
3	Indic	ated air speed		
4	Head	ing		
5	Norm	nal acceleration		

			50 //ug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
6	Pitch attitude		
7	Roll attitude		
8	Manual radio transmission keying unless an alternate means to synchronise FDR and CVR recordings is provided		
9	Power on each engine		
10	Trailing edge flap or cockpit control selection		
11	Leading edge flap or cockpit control selection		
12	Thrust reverse position (for turbo-jet aeroplanes only)		
13	Ground spoiler position and/or speed brake selection		
14	Outside air temperature or total air temperature		
15a	Autopilot engagement status		
15b	Autopilot operating modes, autothrottle and AFCS systems engagement status and operating modes.		
16	Longitudinal acceleration		
17	Lateral acceleration		
18	Primary flight controls - control surface position and/or pilot input (pitch, roll and yaw)		
19	Pitch trim position		

		CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
20	Radio altitude		
21	Glide path deviation		
22	Localiser deviation		
23	Marker beacon passage		
24	Master warning		
25	NAV 1 and NAV 2 frequency selection		
26	DME 1 and DME 2 distance		
27	Landing gear squat switch status		
28	Ground proximity warning system		
29	Angle of attack		
30	Hydraulics, each system (low pressure)		
31	Navigation data		
32	Landing gear or gear selector position		
AMC4	OPS.GEN.490.A Flight data recorder - Aeroplanes		
	OF PARAMETERS TO BE RECORDED FOR AEROPLANES FIRST ISSUED AN INDIVIDUAL CERTIFICATE OF AIRWORTHINESS BEFORE 1 JUNE		

	Part-NCC	CRST	30 Aug 2011
A: R	le	B: Summary of comments	C: Reason for change, remarks
1990			
1.	The flight data recorder should, with reference to a timescale, record the parameters listed in Table 1 of AMC4 OPS.GEN.490.A.		
2.	When determined by the competent authority responsible for type certification or supplemental type certification and agreed by the Agency, the FDR of aeroplanes with a maximum certificated take-off mass exceeding 27 000 kg that are of a type which was first type certificated after 30 September 1969 does not need to record the parameters 13, 14 and 15b in Table 1 of AMC4 OPS.GEN.490.A, when any of the following conditions are met:		
	a. Sufficient capacity is available on a FDR system;		
	b. The sensor is readily available;		
	 A change is not required in the equipment that generates the data. 		
3.	When so determined by the competent authority responsible for type certification or supplemental type certification and agreed by the Agency, the FDR does not need to record individual parameters that can be derived by calculation from the other recorded parameters.		
Ta	ble 1 of AMC4 OPS.GEN.490.A		
Aeroj Kg	planes with a maximum certificated take-off mass exceeding 27 000		

Part-NCo		C	CRST	30 Aug 2011
A: Ru	A: Rule		B: Summary of comments	C: Reason for change, remarks
No.	Parameter			
1	Time or relative time count			
2	Pressure altitude			
3	Indicated air speed			
4	Heading			
5	Normal acceleration			
6	Pitch attitude			
7	Roll attitude			
8	Manual radio transmission keying unless an alternate means to synchronise the FDR and CVR recordings is provided			
9	Power on each engine			
10	Trailing edge flap or cockpit control selection			
11	Leading edge flap or cockpit control selection			
12	Thrust reverse position (for turbo-jet aeroplanes only)			
13	Ground spoiler position and/or speed brake selection			
14	Outside air temperature or total air temperature			

	Part-NC	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
15a 15b	Autopilot engagement status Autopilot operating modes, autothrottle and AFCS, systems engagement status and operating modes.		
16 17	Longitudinal acceleration Lateral acceleration		
18	Primary flight controls - control surface position and/or pilot input (pitch, roll and yaw)		
19	Pitch trim position		
20	Radio altitude		
21	Glide path deviation		
22	Localiser deviation		
23	Marker beacon passage		
24	Master warning		
25	NAV 1 and NAV 2 frequency selection		
26	DME 1 and DME 2 distance		
27	Landing gear squat switch status		
28	Ground proximity warning system		

	Tatt Nee		
A: Rı	ile	B: Summary of comments	C: Reason for change, remarks
29	Angle of attack		
30	Hydraulics, each system (low pressure)		
31	Navigation data (latitude, longitude, ground speed and drift angle)		
32	Landing gear or gear selector position		
	ndix 1 to AMC3 and AMC4 OPS.GEN.490.A Flight data recorder oplanes		
RECO	ORMANCES SPECIFICATIONS FOR THE PARAMETERS TO BE RDED FOR AEROPLANES FIRST ISSUED WITH AN INDIVIDUAL IFICATE OF AIRWORTHINESS BEFORE 1 APRIL 1998		
1.	The parameters to be recorded should meet the performance specifications (designated ranges, recording intervals and accuracy limits) defined in Table 1 of Appendix 1 to AMC3 and AMC4 OPS.GEN.490.A.		
2.	FDR systems for which the recorded parameters do not comply with the performance specifications of Table 1 of Appendix 1 to AMC3 and AMC4 OPS.GEN.490.A (i.e. range, sampling intervals, accuracy limits and recommended resolution readout) may be acceptable to the competent authority responsible for the type certification or supplemental type certification.		
3.	All aeroplanes should record the following additional parameters, when further recording capacity is available:		
	a. Remaining parameters below, as applicable:		

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1 41 6 11 6 6	

A: Ru	le				B: Summary of comments	C: Reason for change, remarks
		i.	syster Monit Alerti	ational information from electronic display ms, such as EFIS, Electronic Centralised Aircraft or (ECAM) and Engine Indications and Crew ng System (EICAS). The following order of ty should be used:		
			Α.	Parameters selected by the flight crew relating to the desired flight path, e.g. barometric pressure setting, selected altitude, selected air speed, decision height, autoflight system engagement and mode indications if not recorded from another source;		
			В.	Display system selection/status, e.g. SECTOR, PLAN, ROSE, NAV, WXR, COMPOSITE, COPY;		
			C.	Warnings and alerts;		
			D.	The identity of displayed pages for emergency procedures and checklists.		
		ii.	for us	dation information including brake application se in the investigation of landing over-runs and sed take-offs;		
		iii.	Additi flow,	ional engine parameters (EPR, N_1 EGT, fuel etc.);		
	b.			ed parameter relating to novel or unique design al characteristics of the aeroplane.		
4.	OPS.C	GEN.49	D.A and	the alleviations specified in AMC3 AMC4 OPS.GEN.490.A, they should be an adding recording of missing parameters to		

				Part-NCC	CRST			30 Aug 2011
A: Ru	le				B: Summary of	comments	C: Re	ason for change, remarks
	-	DR system wou Account should	•	major upgrade of the the following:				
	a. The ex	tent of the mo	dification req	uired;				
	b. The do	wn-time period	1;					
	c. Equipm	nent software d	levelopment.					
5.	data acquisition recording the	on unit and the required parar f accident inve	flight data re neters, or the	e space on both the flight ecorder not allocated for e parameters recorded for acceptable to the				
6. availa	A sensor is co ble or can be ea		•	when it is already				
Tab	le 1 of Appen	dix 1 to AMC3	and AMC4	OPS.GEN.490.A				
Paran	neters Performa	nce Specificati	ons					
No.	Parameter	Range	Sampling interval in seconds	Accuracy limits (sensor input compared to FDR readout)	Recommended resolution in readout	Remarks		
1	Time or relative time count	24 hours	4	±0•125% per hour	1 second	Co-ordinated Univers Time (UTC) preferred where available, othe elapsed time	ł	
2	Pressure	-1 000 ft to	1	±100 ft to ±700 ft	5 ft	For altitude record er	ror	

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A: R	A: Rule					of comments	C: Reason for change, remarks	
	altitude	maximum certificated altitude of aircraft +5 000 ft				see EASA ETSO-C124	a	
3	Indicated air speed	50 kt to max V _{SO} Max VSO to 1•2 V _d	1	±5% ±3%	1 kt	VSO stalling speed or minimum steady fligh speed in the landing configuration VdF des diving speed	t	
4	Heading	360°	1	±2°	0•50			
5	Normal acceleration	-3 g to +6 g	0•125 ±	0•125 ±1% of maximum range excluding a datum error of ±5%	0∙004 g			
6	Pitch attitude	±75°	1	±2°	0•5°			
7	Roll attitude	±180°	1	±2°	0•5°			
8	Manual radio transmission keying	Discrete	1	-	-	On-off (one discrete) FDR/CVR time synchronisation signa complying with 4.2.1 EUROCAE ED-55 is considered to be an acceptable alternative means of compliance	l of	

A: Ru	le				B: Summary of comments		C: Reason for change, remarks	
9	Power on each engine	Full range	Each engine each second	±2%	0•2% of full range	Sufficient parameters EPR/N, or Torque/N _P appropriate to the particular engine sho recorded to determin power	as uld be	
10	Trailing edge flap or cockpit control selection	Full range or each discrete position	2	±5% or as pilot's indicator	0•5% of full range			
11	Leading edge flap or cockpit control selection	Full range or each discrete position	2	-	0•5% of full range			
12	Thrust reverser position	Stowed, in transit and reverse	Each reverser each second	±2% unless higher accuracy uniquely required	-			
13	Ground spoiler and/or speed brake selection	Full range or each discrete position	1	±2°	0•2% of full range			

A: Ru	le				B: Summary of comments		C: Reason for change, remarks	
14	Outside air temperature s or total air temperature	Sensor range	2	-	0•30			
15a	Autopilot engagement status							
15b	Autopilot operating modes, auto-throttle and AFCS systems engagement status and operating modes	A suitable combination of discretes	1		-			
16	Longitudinal acceleration	± 1 g	0•25	±1•5% of maximum range excluding a datum error of ±5%	0∙004 g			
17	Lateral acceleration	±1 g	0•25	±1•5% of maximum range excluding a datum error of ±5%	0∙004 g			
18	Primary flight controls,	Full range	1	±2° unless higher accuracy uniquely required	0•2% of full range	For aeroplanes with conventional control systems 'or' applies		

A: Ru	lle				B: Summary of comments		C: Reason for change, remarks	
	control surface positions and/or pilot input (pitch, roll, yaw)					For aeroplanes with r mechanical control sy 'and' applies For aeroplanes with s surfaces a suitable combination of inputs acceptable in lieu of recording each surfac separately	stems plit s is	
19	Pitch trim position	Full range	1	±3% unless higher accuracy uniquely required	0•3% of full range			
20	Radio altitude	-20 ft to +2 500 ft	1	± 2 ft or $\pm 3\%$ whichever is greater below 500 ft and $\pm 5\%$ above 500 ft	1 ft below 500 ft, 1 ft +5% of full range above 500 ft	As installed. Accuracy limits are recommend		
21	Glide path deviation	Signal range	1	±3%	0•3% of full range	As installed. Accuracy limits are recommend		
22	Localiser deviation	Signal range	1	±3%	0•3% of full range	As installed. Accuracy limits are recommend		
23	Marker beacon passage	Discrete	1	-	-	A single discrete is acceptable for all ma	rkers	
24	Master	Discrete	1	-	-			

A: Rı	ıle				B: Summary of comments		C: Reason for change, remarks	
	warning							
25	NAV 1 and 2 frequency selection	Full range	4	As installed	-			
26	DME 1 and 2 distance	0-200 nm	4	As installed	-	Recording of latitude longitude from INS o other navigation syst a preferred alternativ	r em is	
27	Landing gear squat switch status	Discrete	1	-	-			
28	Ground proximity warning system (GPWS)	Discrete	1	-	-			
29	Angle of attack	Full range	0•5	As installed	0•3% of full range			
30	Hydraulics	Discrete(s)	2	-	-			
31	Navigation data	As installed	1	As installed	-			
32	Landing	Discrete	4	As installed	-			

A: Ru	A: Rule				B: Summary of	comments	C: Re	ason for change, remarks	
	gear or ge selector position	ar							
AMC	L OPS.GEN.	490.H Flight data	a recorder -	Helicopters					
ISSU		TERS TO BE RECOR INDIVIDUAL CERT (2010							
1.	The FDR sl	hould, with referer	ce to a time	scale, record:					
	a. the	parameters listed	in Table 1 o	FAMC1 OPS.GEN	I.490.H;				
	OPS par the	additional parame S.GEN.490.H, when ameter is used by instrument panel helicopter; and	n the information the information of the informatio	ation data source stems or is avai	e for the lable on				
	or o det	dedicated parame operational charact ermined by the cont tification or supple	eristics of th mpetent aut	e helicopter as nority responsible	_				
2.		hould meet the op cations of EUROCA Part III .		•					
3.	performan sampling i	ms for which the re ce specifications o ntervals, accuracy ould be acceptable	FEUROCAE E limits and re	D-112 (i.e. rang commended res	le,				

	Part-NCC	CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
	responsible for type certification or supplemental type certification.		
Tab	le 1 of AMC1 OPS.GEN.490.H		
No.*	Parameter		
1	Time or relative time count		
2	Pressure altitude		
3	Indicated air speed		
4	Heading		
5	Normal acceleration		
6	Pitch attitude		
7	Roll attitude		
8	Manual radio transmission keying CVR/FDR synchronisation reference		
9	Power on each engine		
9a	Free power turbine speed (N_F)		
9b	Engine torque		
9c	Engine gas generator speed (N_G)		

			50 Aug 2011
A: Ru	e	 B: Summary of comments	C: Reason for change, remarks
9d	Cockpit power control position		
9e	Other parameters to enable engine power to be determined		
10a	Main rotor speed		
10b	Rotor brake (if installed)		
11 11a	Primary flight controls – Pilot input and/or control output position (if applicable)		
11b	Collective pitch		
11c	Longitudinal cyclic pitch		
11d	Lateral cyclic pitch		
11e	Tail rotor pedal		
11f	Controllable stabilator (if applicable)		
	Hydraulic selection		
12	Hydraulics low pressure (each system should be recorded.)		
13	Outside air temperature		
18	Yaw rate or yaw acceleration		
20	Longitudinal acceleration (body axis)		
21	Lateral acceleration		
25	Marker beacon passage		
26	Warnings - a discrete should be recorded for the master		

	Part-NCC	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
	warning, gearbox low oil pressure and sas failure. other 'red' warnings should be recorded where the warning condition cannot be determined from other parameters or from the cockpit voice recorder.		
27	Each navigation receiver frequency selection		
37	Engine control modes		
*	The number in the left hand column reflects the serial numbers depicted in EUROCAE ED-112		
Tab	le 2 of AMC1 OPS.GEN.490.H		
either	pters for which the information data source for the parameter is used by helicopter systems or is available on the instrument panel e by the flight crew to operate the helicopter		
No.*	Parameter		
14	AFCS mode and engagement status		
15	Stability augmentation system engagement (each system should be recorded)		
16	Main gear box oil pressure		
17	Gear box oil temperature		
17a	Main gear box oil temperature		

			50 Aug 2011
A: Ru	e	B: Summary of comments	C: Reason for change, remarks
17b	Intermediate gear box oil temperature		
17c	Tail rotor gear box oil temperature		
19	Indicated sling load force (if signals readily available)		
22	Radio altitude		
23	Vertical deviation - the approach aid in use should be recorded.		
23a	ILS glide path		
23b	MLS elevation		
23c	GNSS approach path		
24	Horizontal deviation - the approach aid in use should be		
24a	recorded.		
24b	ILS localiser		
24c	MLS azimuth		
	GNSS approach path		
28	DME 1 & 2 distances		
29	Navigation data		
29a	Drift angle		
29b	Wind speed		
29c	Wind direction		
29d	Latitude		

	Part-NCo	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
29e	Longitude		
29f	Ground speed		
30	Landing gear or gear selector position		
31	Engine exhaust gas temperature (T_4)		
32	Turbine Inlet Temperature (TIT/ITT)		
33	Fuel contents		
34	Altitude rate (vertical speed) - only necessary when available from cockpit instruments		
35	Ice detection		
36 36a	Helicopter Health and Usage Monitor System (HUMS) - only when information from the HUMS is used by the crew or aircraft system		
36b	Engine data		
36c	Chip detector		
36d	Track timing		
36e	Exceedance discretes		
	Broadband average engine vibration		
38	Selected barometric setting - to be recorded for helicopters where the parameter is displayed electronically		
38a	Pilot		

	Part-NCC	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
38b	First officer		
39	Selected altitude (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
40	Selected speed (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
41	Not used (selected mach)		
42	Selected vertical speed (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
43	Selected heading (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
44	Selected flight path (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
45	Selected decision height (all pilot selectable modes of operation) - to be recorded for the helicopters where the parameter is displayed electronically		
46	EFIS display format		
47	Multi-function/engine/alerts display format		

		Ραπ-ΝΟΟ		30 AUG 2011
A: Ru	ıle		B: Summary of comments	C: Reason for change, remarks
48	ever	nt marker		
*		number in the left hand column reflects the serial numbers cted in EUROCAE ED-112		
AMC	2 OPS	.GEN.490.H Flight data recorder - Helicopters		
MCTO CERTI HELIO ISSUE	om exc Ificat Coptei Ed WI	RAMETERS TO BE RECORDED FOR HELICOPTERS HAVING A CEEDING 3 175 KG AND FIRST ISSUED WITH AN INDIVIDUAL TE OF AIRWORTHINESS AFTER 1 JANUARY 2005 AND RS HAVING A MCTOM EXCEEDING 7 000 KG AND FIRST TH AN INDIVIDUAL CERTIFICATE OF AIRWORTHINESS AFTER ER 1988		
1.	The	FDR should, with reference to a timescale, record:		
	a.	for helicopters with a maximum certificated take-off mass between 3 175 kg and 7 000 kg, the parameters listed in Table 1 of AMC2 OPS.GEN.490.H;		
	b.	for helicopters with a maximum certificated take-off mass of more than 7 000 kg, the parameters listed in Table 2 of AMC2 OPS.GEN.490.H;		
	C.	any dedicated parameters relating to novel or unique design or operational characteristics of the helicopter; and		
	d.	the parameters listed in Table 3 of AMC2 OPS.GEN.490.H, for helicopters equipped with electronic display system.		
2.	certi	n determined by the competent authority responsible for type fication or supplemental type certification and agreed by the ncy, the FDR of helicopters with a maximum certificated take-		

Part-NCC	CRST
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A: Ru	ıle	B: Summary of comments	C: Reason for change, remarks
	off mass of more than 7 000 kg does not need to record parameter 19 of Table 2 of AMC2 OPS.GEN.490.H, if any of the following conditions are met:		
	a. The sensor is not available;		
	b. A change is required in the equipment that generates the data.		
3.	Individual parameters that can be derived by calculation from the other recorded parameters, need not be recorded, if determined by the competent authority responsible for type certification or supplemental type certification and agreed by the Agency.		
4.	The parameters to be recorded should meet, as far as is practicable, the performance specifications (designated ranges, sampling intervals, accuracy limits and minimum resolution in read-out) defined in the relevant tables of EUROCAE ED-55. The remarks columns of those tables are considered to be acceptable means of compliance with the parameter specifications.		
5.	Table 1 of AMC2 OPS.GEN.490.H refers to table A1-4 of EUROCAE ED-55, Table 2 of AMC2 OPS.GEN.490.H refers to table A1-2 of EUROCAE ED-55 and Table 3 of AMC2 OPS.GEN.490.H refers to parameters 6 to 15 of table A1.5 of EUROCAE ED-55.		
6.	If recording capacity is available, as many of the additional parameters specified in table A1.5 of EUROCAE ED-55 as is possible, should be recorded.		
7.	For the purpose of this AMC, a sensor is considered to be 'readily available' when it is already available or can be easily incorporated.		
8.	The term 'where practicable' used in the remarks column of table A1.5 of EUROCAE ED-55 means that account should be taken of		

		Part-NCC	CRST	30 Aug 2011
A: Ru	le		B: Summary of comments	C: Reason for change, remarks
	the	following:		
	a.	Whether the sensor is already available or can be easily incorporated;		
	b.	Whether sufficient capacity is available in the flight recorder system;		
	c.	For navigational data (nav frequency selection, DME distance, latitude, longitude, groundspeed and drift), whether the signals are available in digital form;		
	d.	The extent of modification required;		
	e.	The down-time period required;		
	f.	Equipment software development.		
Tab	le 1 d	of AMC2 OPS.GEN.490.H		
Helico kg	pters	with a maximum certificated take-off mass exceeding 3 175		
No.	Para	ameter		
1	Tim	e or relative time count		
2	Pres	ssure altitude		
3	Indi	icated air speed		
4	Неа	iding		
5	Nor	mal acceleration		

	Part-NCC	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
6	Pitch attitude		
7	Roll attitude		
8	Manual radio transmission keying		
9	Power on each engine (free power turbine speed and engine torque)/cockpit power control position (if applicable)		
10a	Main rotor speed		
10b	Rotor brake (if installed)		
11 11a	Primary flight controls - pilot input and control output position (if applicable)		
11b	Collective pitch		
11c	Longitudinal cyclic pitch		
11d	Lateral cyclic pitch		
11e	Tail rotor pedal		
11f	Controllable stabilator		
	Hydraulic selection		
12	Warnings		
13	Outside air temperature		
14	Autopilot engagement status		
15	Stability augmentation system engagement		

_	Part-NCC	CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
Tab	le 2 of AMC2 OPS.GEN.490.H		
Helico kg	pters with a maximum certificated take-off mass exceeding 7 000		
No.	Parameter		
1	Time or relative time count		
2	Pressure altitude		
3	Indicated airspeed		
4	Heading		
5	Normal acceleration		
6	Pitch attitude		
7	Roll attitude		
8	Manual radio transmission keying		
9	Power on each engine (free power turbine speed and engine torque)/cockpit power control position (if applicable)		
10a	Main rotor speed		
10b	Rotor brake (if installed)		
11	Primary flight controls - pilot input and control output position		

		Part-NCC CRST	30 Aug 2011
A: Ru	le	B: Summary of comme	nts C: Reason for change, remarks
11a	(if applicable)		
11b	Collective pitch		
11c	Longitudinal cyclic pitch		
11d	Lateral cyclic pitch		
11e	Tail rotor pedal		
11f	Controllable stabilator		
	Hydraulic selection		
12	Hydraulics low pressure		
13	Outside air temperature		
14	AFCS mode and engagement status		
15	Stability augmentation system engagement		
16	Main gear box oil pressure		
17	Main gear box oil temperature		
18	Yaw rate or yaw acceleration		
19	Indicated sling load force (if installed)		
20	Longitudinal acceleration (body axis)		
21	Lateral acceleration		
22	Radio altitude		

	Part-NC	C CRST	30 Aug 2011
A: Ru	le	B: Summary of comments	C: Reason for change, remarks
23	Vertical beam deviation (ILS glide path or MLS elevation)		
24	Horizontal beam deviation (ILS localiser or MLS azimuth)		
25	Marker beacon passage		
26	Warnings		
27	Reserved (Nav receiver frequency selection is recommended)		
28	Reserved (DME distance is recommended)		
29	Reserved (navigation data is recommended)		
30	Landing gear or gear selector position		
Tab	le 3 of AMC2 OPS.GEN.490.H		
Helico	pters equipped with electronic display systems		
No.	Parameter		
6	Selected barometric setting (each pilot station)		
7	Selected altitude		
8	Selected speed		
9	Selected mach		

	Part-NCC		30 AUG 2011
A: R	ule	B: Summary of comments	C: Reason for change, remarks
10	Selected vertical speed		
11	Selected heading		
12	Selected flight path		
13	Selected decision height		
14	EFIS display format		
15	Multi function/engine/alerts display format		
АМС	OPS.GEN.495.A Cockpit voice recorder – Aeroplanes		
GEN	ERAL		
1.	The CVR should, with reference to a timescale, record:		
	a. flight crew members' two-way voice communications by radio, interphone system and public address system, if installed;		
	 b. the aural environment of the cockpit, including, where practicable, without interruption, the microphone audic signals; and 		
	c. voice or audio signals identifying navigation or approach aids introduced into a headset or speaker.		
2.	The operational performance requirements for CVRs should be those laid down in EUROCAE Documents ED56 or ED56A (Minimum Operational Performance Requirements For Cockpit Voice Recorder		

Part-NCC		Part-NCC	CRST	30 Aug 2011	
A: R	ule		B: Summary of comments	C: Reason for change, remarks	
	Syste	ms) dated February 1988 and December 1993 respectively.			
АМС	C OPS.C	GEN.495.H Cockpit voice recorder - Helicopters			
GEN	ERAL				
1.	The	CVR should, with reference to a timescale, record:			
	a.	flight crew members' two-way voice communications via radio, interphone system and public address system;			
	b.	the aural environment of the cockpit, including, where practicable, without interruption, the microphone audio signals;			
	c.	voice or audio signals identifying navigation or approach aids introduced into a headset or speaker; and			
	d.	for helicopters not equipped with a flight data recorder, the parameters necessary to determine main rotor speed.			
2. The operational performance requirements of EUROCAE ED-56/56A should be considered to be acceptable means of compliance.					
АМС	C OPS.C	GEN.495 (c) Cockpit voice recorder			
RECO	ORDING	3			
	-	on the availability of electrical power, the CVR should start to arly as possible during the cockpit checks, prior to the flight			

	Part-NCC	CRST	30 Aug 2011	
A: F	Rule	B: Summary of comments	C: Reason for change, remarks	
	l the cockpit checks immediately following engine shutdown at the end ne flight.			
	C OPS.GEN.490 and OPS.GEN.495 Flight data recorder and kpit voice recorder			
CON	IBINATION RECORDERS			
2.	 A combination recorder is a flight recorder that records: a. all voice communications and the aural environment required by the applicable CVR AMC; and b. all parameters and specifications required by the applicable FDR AMC. When two combination recorders are installed, one should be located near the cockpit, in order to minimise the risk of data loss due to a failure of the wiring that gathers data to the recorder. The other should be located at the rear section of the aeroplane, in order to minimise the risk of data loss of the case of a crash. 		Regulatory justification: There is no such requirement in ICAO Annex 6 so this requirement is removed for NCC.	
3.	 For aeroplanes, compliance with CVR and FDR requirements may be achieved by: a. one combination recorder, if the aeroplane should be equipped with either a CVR or an FDR; b. one combination recorder, if an aeroplane with a maximum certificated take-off mass of 5 700 kg or less should be equipped with both a CVR and an FDR; or 		Regulatory justification: This has been upgraded to IR level, for consistency with requirements for CAT.	

Part-NCC	CRST
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A: R	ule	B: Summary of comments	C: Reason for change, remarks
	c. two combination recorders, if an aeroplane with a maximum certificated take-off mass of more than 5 700 kg should be equipped with both a CVR and an FDR.		
	1 OPS.GEN.500 Data link recording - Aeroplanes and copters		
GEN	ERAL		
1.	Depending on the date of type certification, the aircraft shall be capable of recording the messages as specified in AMC2 OPS.GEN.500.		
2.	As a means of compliance with OPS.GEN.500 (a)(2), the operator should enable correlation by providing information which allows an accident investigator to understand what data was provided to the aircraft and by which provider.	1. The requirement related to the last part of this section (and by which provider) couldn't be complied with. The term "provider" is not specified concerning the expected information. Identifications of the ATS provider are not part of the recorded information, unless it is part of the message.	1. Regulatory justification: If the message received contains no information on the identity of the provider, this information has to be derived by other means, which is over-demanding.
3.	The timing information associated with the data link communications messages required to be recorded by OPS.GEN.500 (a)(3) should be capable of being determined from the airborne-based recordings. This timing information should include at least the following:		Regulatory justification: The timing information is essential to reconstruct a sequence of exchanged
	a. The time each message was generated;		information.
	b. The time any message was available to be displayed by the crew;		

	Fart-Nec		50 AUG 2011	
A: Rule		B: Summary of comments	C: Reason for change, remarks	
	c. The time each message was actually displayed or recalled from a queue;d. The time of each status change.			
4.	The message priority should be recorded when it is defined by the protocol of the data link communication message being recorded.			
5.	 The expression 'taking into account the system architecture', in OPS.GEN.500 (a)(3), means that the recording of the specified information may be omitted if the existing source systems involved would require a major upgrade. The following should be considered: a. The extent of the modification required; b. The down-time period; c. Equipment software development. 			
6.	The intention is that new designs of source systems should include this functionality and support the full recording of the required information.			
7.	The applications to be recorded should meet the performance specifications defined in the relevant tables of part IV CNS/ATM recorder systems of EUROCAE ED-112.	 The tables in part IV of ED-112 are immature and contain inconsistencies; another reference should be given. FAA established an AC 20-160 to define the approved data set that has to be recorded. Accepting these requirements as an alternative means of compliance would provide one standard. This also would improve the envisaged approach for 	 Regulatory justification: There is no requirement on the applications to be recorded in ICAO Annex 6 Part II Section 3 or appendix 3.1. Only Table IV-B.1 of ED-112 contains recording specifications, but they are redisplaying the content of ED-93 in a condensed 	

Part-NCC	CRST	30 Aug 2011
A: Rule	B: Summary of comments	C: Reason for change, remarks
	harmonisation of rules between Europe and USA.	and non-exhaustive way. In addition recording performance requirements would better fit in AMC2 NCC.IDE.A.170.
		FAA AC 20-160 points at relevant parts of EUROCAE Document ED- 93. ED-93 is the most exhaustive guidance document available on this subject.
		It is proposed here to strike out this paragraph and add a reference to ED-93 in AMC2 NCC.IDE.A.170.
8. Depending on the availability of electrical power, the flight recorder should start to record as early as possible during the cockpit checks prior to engine start at the beginning of the flight until the cockpit checks immediately following engine shutdown at the end of the flight.		This provision has been upgraded to IR level. It is consistent with the requirement on audio (CVR) recording to start during the cockpit checks prior to engine start and last until the cockpit checks immediately following engine shutdown.
AMC2 OPS.GEN.500 Data link recording - Aeroplanes and Helicopters		
LIST OF APPLICATIONS		

			Part-NCC	CRST	30 Aug 2011
A: Rul	A: Rule			B: Summary of comments	C: Reason for change, remarks
1. For aeroplanes and helicopters first issued with a type certificate after 31 December 2009, data link communications messages that support the applications in Table 1 of AMC2 OPS.GEN.500 should be recorded.			s messages that		1. Regulatory justification: NCC.IDE.A.170 will apply to aircraft from 01 January 2016; therefore there is no need to make a distinction based on the date of delivery of the individual certificate of airworthiness.
					This AMC can be simplified to keep only the provisions related to after the mandatory date for data link communication recording:
					Table 1 has been removed and only table 2 has been kept.
Tabl	e 1 of AMC2 OF	PS.GEN.500			
Item No.	Application Type	Application Description	Required Recording Content		
1	Data link initiation	This includes any application used to log on to, or initiate, a data link service. In Future Air Navigation System (FANS)-1/A and Air Traffic Navigation (ATN), these are ATS Facilities Notification (AFN) and Context	C		

			Part-NCC	CRST	30 Aug 2011
A: Rule	3			B: Summary of comments	C: Reason for change, remarks
		Management (CM), respectively.			
2	Controller/pilo t communicatio n	This includes any application used to exchange requests, clearances, instructions and reports between the flight crew and controllers on the ground. In FANS-1/A and ATN, this includes the Controller Pilot Data Link Communications (CPDLC) application. It also includes applications used for the exchange of Oceanic Clearances (OCL) and Departure Clearances (DCL), as well as data link delivery of taxi clearances.	C		
3	Addressed surveillance (2)	This includes any surveillance application in which the ground sets up contracts for delivery of surveillance data. In FANS-1/A and ATN, this includes the Automatic Dependent Surveillance- Contract (ADS-C) application, Controller Access Parameters (CAP) and System Access	С		

					567/109/2011	
A: Rule				B: Summary of comments	C: Reason for change, remarks	
		Parameters (SAP).				
4	Flight information	This includes any application used for delivery of flight information data to specific aeroplanes. This includes Data Link-Automatic Terminal Information Service (D-ATIS), Data Link-Operational Terminal Information Service (D-OTIS), text weather services, Data Link-Flight Information System (D-FIS) and Notice to Airmen (NOTAM) delivery.	C			
		Terminal Weather Information for Pilots (TWIP)	М			
5	Broadcast surveillance (2)	This includes elementary and enhanced surveillance systems, as well as Automatic Dependent Surveillance- Broadcast (ADS-B), Terminal Information Service-Broadcast (TIS-B) and Flight Information System-Broadcast (FIS-B).	М			
6	AOC data	This includes any application transmitting or receiving data used for AOC purposes (in accordance with the ICAO	M*			

Part-NCC			Part-NC	C CRST	30 Aug 2011	
A: Rule				B: Summary of comments	C: Reason for change, remarks	
		definition of AOC). Such systems may also process AAC messages, but there is no requirement to record AAC messages				
7	Graphics (1)	This includes any application receiving graphical data to be used for operational purposes (i.e. excluding applications that are receiving such things as updates to manuals).	M*			
before the app	1 January 2010, c	nd helicopters first issued with a typ data link communications messages 2 of AMC2 OPS.GEN.500 should be 5.GEN.500	s that support			
Item No.	Application Type	Application Description	Required Recording Content	1. IND: Remove the dependency on the date of delivery of the ICA and replace sentences/tables 1 and 2 by a single table 1 covering all new manufactured aircraft.	1. Regulatory justification: NCC.IDE.A.170 will apply to aeroplanes with an individual	
1	Data link initiation	This includes any application used to log on to, or initiate, a data link service. In FANS-1/A and ATN, these are AFN and CM, respectively	C		certificate of airworthiness issued on or after 01 January 2016. This AMC can be simplified to keep only the provisions related to after the mandatory date for	
2	Controller/pilo	This includes any application	С		data link communication recording:	

A: Rule				B: Summary of comments	C: Reason for change, remarks
	t communicatio n	used to exchange requests, clearances, instructions and reports between the flight crew and controllers on the ground. In FANS-1/A and ATN, this includes the CPDLC application. It also includes applications used for the exchange of OCL and DCL, as well as data link delivery of taxi clearances.			Table 1 has been removed and only table 2 has been kept.
3	Addressed surveillance (2)	This includes any surveillance application in which the ground sets up contracts for delivery of surveillance data. In FANS-1/A and ATN, this includes the ADS-C application.	C		
		САР	C*		
		SAP	C*		
4	Flight information	This includes any application used for delivery of flight information data to specific aeroplanes. This includes D- ATIS, D-OTIS, text weather services, D-FIS and NOTAM delivery.	C		

A: Ru	le			
		TWIP	М	
5	Broadcast surveillance (2)	This includes elementary and enhanced surveillance systems, as well as ADS-B, TIS-B and FIS-B.	M*	
6	AOC data	This includes any application transmitting or receiving data used for AOC purposes (in accordance with the ICAO definition of AOC). Such systems may also process AAC messages, but there is no requirement to record AAC messages.	M*	
7	Graphics (1)	This includes any application receiving graphical data to be used for operational purposes (i.e. excluding applications that are receiving such things as updates to manuals).	M*	
<mark>gm o</mark> i Genef		a link recording - Aeroplanes and	l Helicopters	
1. The letters and expressions in Tables 1 and 2 of AMC2 OPS.GEN.500 have the following meaning:				

Part-NCC	CRST
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A: Rul	e			B: Summary of comments	C: Reason for change, remarks
a	a. C: Complete contents recorded				
b	b. M: Information that enables correlation with any associated records stored separately from the aeroplane.				
с		plications that cable, given the	are to be recorded only as far as is architecture of the system.		
d	d. F1: Graphics applications may be considered as AOC messages when they are part of a data link communications application service run on an individual basis by the operator itself in the framework of the operational control.				
e	e. F2: Where parametric data sent by the aeroplane, such as Mode S, is reported within the message, it should be recorded unless data from the same source is recorded on the FDR.				
			plications type in Tables 1 and 2 of AMC2 ed in Table 1 of GM OPS.GEN.500.		
Tabl	e 1 of GM	OPS.GEN.50	D		
Item No.	Applicati on Type	Messages	Comments		
1	СМ		CM is an ATN service		
2	AFN		AFN is a FANS 1/A service		
3	CPDLC		All implemented up and downlink messages to be recorded		

A: Ru	A: Rule			B: Summary of comments	C: Reason for change, remarks
4	ADS-C	FLIPCY	All contract requests and reports recorded		
		Position reports	Only used within FANS 1/A. Mainly used in oceanic and remote areas.		
5	ADS-B	Surveillanc e data	Information that enables correlation with any associated records stored separately from the aeroplane.		
6	D-FIS		D-FIS is an ATN service. All implemented up and downlink messages to be recorded		
7	TWIP	TWIP messages	Terminal weather information for pilots		
8	D-ATIS	ATIS messages	EUROCAE ED-89A Data Link Application System Document (DLASD) for the "ATIS" Data Link Service		
9	OCL	OCL messages	EUROCAE ED-106A Data Link Application System Document (DLASD) for "Oceanic Clearance" (OCL) Datalink Service		
10	DCL	DCL messages	EUROCAE ED-85A Data Link Application System Document (DLASD) for "Departure Clearance"		

A: Ru	A: Rule			B: Summary of comments	C: Reason for change, remarks
			Data-Link Service		
11	Graphics	Weather maps & other graphics	Graphics exchanged in the framework of procedures within the operational control, as specified in Part-OR. Information that enables correlation with any associated records stored separately from the aeroplane.		
12	AOC	Aeronautica l operational control messages	Messages exchanged in the framework of procedures within the operational control, as specified in Part-OR. Information that enables correlation with any associated records stored separately from the aeroplane. Definition in ED-112.		
13	Surveilla nce	CAP, SAP	Use definition in ED-93.		
	ADS-B Auto ADS-C AFN Airc AOC Aero	omatic Depend Automatic traft Flight Not onautical Oper	inistrative Communications dent Surveillance - Broadcast Dependent Surveillance – Contract ification rational Control al Information Service		

Part-NCC	CRST	30 Aug 2011
A: Rule	B: Summary of comments	C: Reason for change, remarks
ATSC Air Traffic Service Communication		
CAP Controller Access Parameters		
CPDLC Controller Pilot Data Link Communications		
CM Configuration/Context Management		
D-ATIS Data link ATIS		
D-FIS Data link Flight Information Service		
DCL Departure Clearance		
FANS Future Air Navigation System		
FLIPCY Flight Plan Consistency		
OCL Oceanic Clearance		
SAP System Access Parameters		
TWIP Terminal Weather Information for Pilots		
AMC OPS.GEN.505(d) Preservation of FDR and CVR recordings - Aeroplanes and Helicopters		
OPERATIONAL CHECKS AND EVALUATIONS OF RECORDINGS OF REPRESENTATIVE FLIGHT		
Whenever a recorder is required to be carried, the operator should:		
1. save the recordings for the period of operating time as required by OPS.GEN.490, OPS.GEN.495 and OPS.GEN.500, except that, for the purpose of testing ad maintaining recorders, up to one hour of the oldest recorded material at the time of testing may be erased;		

Part-NCC	CRST
i une nice j	CRST

A: Rule		B: Summary of comments	C: Reason for change, remarks
2.	keep a document which presents the information necessary to retrieve and convert the stored data into engineering units; and		
3.	at all times, preserve a record of at least one representative flight made within the last 12 months which includes a take-off, climb, cruise, descent, approach to landing and landing, together with a means of identifying the record with the flight to which it relates.	1. MS: Indications on the necessity to conduct recorder regular inspection and FDR regular calibration checks are missing. Align with ICAO Annex 6 recommended practices.	Accepted. The text has been modified and an operational check after every 5 years or according to a periodicity determined by the sensor manufacturer has been added.
GM OPS.GEN.505(b) and (c) Preservation of FDR and CVR recordings - Aeroplanes and Helicopters		1. MS: Indications on the necessity to conduct recorder regular inspection and FDR regular calibration checks are missing. Align with ICAO Annex 6 recommended practices.	Accepted. The text has been modified and an operational check after every 5 years or according to a periodicity determined by the sensor manufacturer has been added.
REM	OVAL OF RECORDERS, INSPECTIONS AND MAINTENANCE		
1.	The need for removal of the FDR data from the aircraft will be determined by the investigating authority with due regard to the seriousness of an occurrence and the circumstances, including the impact on the operation.		
2.	Procedures for the inspections and maintenance practices of the FDR and CVR systems are given in Attachment A of ICAO Annex 6, Part II and in Annex I-B of EUROCAE ED-112.		
GM OPS.GEN.505(d) Preservation of FDR and CVR recordings - Aeroplanes and Helicopters			

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A: R	ule	B: Summary of comments	C: Reason for change, remarks
	RATIONAL CHECKS AND EVALUATIONS OF RECORDINGS OF RESENTATIVE FLIGHT		
The	representative flight may not be possible to be preserved if:		
1.	there are technical reasons as to why all the data cannot be preserved; and/or		
2.	the aircraft may have been dispatched with unserviceable recording equipment, as permitted by the operators' MEL.		
	COPS.GEN.515(b) and OPS.GEN.520(a) Microphones - oplanes and Helicopters and Flight Crew interphone system		
HEA	DSETS		
1.	A headset consists of a communication device which includes two earphones to receive and a microphone to transmit audio signals to the aircraft's communication system. To comply with the minimum performance requirements, the earphones and microphone should match the communication system's characteristics and the cockpit environment. The headset should be adequately adjustable in order to fit the pilot's head. Headset boom microphones should be of the noise cancelling type.	1. IND: The sentence "Headset boom microphones should be of noise cancelling type" is not clear. Usually there is a squelch doing that and located in the audio panel.	1. The wording is consistent with CAT.IDE.
2.	If the intention is to utilise noise cancelling earphones, the operator should ensure that the earphones do not attenuate any aural warnings or sounds necessary for alerting the flight crew on matters related to the safe operation of the aircraft.		

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GM OPS.GEN.515(b) and OPS.GEN.520(a) Microphones - Aeroplanes and Helicopters and Flight Crew interphone system		
HEADSETS		
The term 'headset' includes any aviation helmet incorporating headphones and microphone worn by a flight crew member.		
AMC OPS.GEN.520 Flight Crew interphone system		
GENERAL		
The flight crew interphone system should not be of a handheld type.		
GM OPS.GEN.525(b) Communication equipment		
AERONAUTICAL EMERGENCY FREQUENCY		
The aeronautical emergency frequency is 121.5 MHz.		
AMC OPS.GEN.530 Pressure-altitude-reporting transponder		
GENERAL		

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1.	The SSR transponder of aircraft being operated under European air traffic control should comply with any applicable Single European Sky legislation.		
2.	If the Single European Sky legislation is not applicable, the SSR transponder should operate in accordance with the relevant provisions of Volume IV of ICAO Annex 10.		
3.	The SSR transponder may have additional capabilities, if so required by the applicable airspace requirements.	1.IND: This has a double standard. The requirement is for the additional SSR facilities to be carried when required by the airspace. The statement then gives the option of carriage. Incorrect use of the word 'may' creates this problem. The wording should be "The airspace may require additional SSR transponder capabilities."	1. Provision upgraded to IR level and wording clarified.
АМС	COPS.GEN.535(a) Navigation equipment		
VISU	JAL REFERENCE TO LANDMARKS		
	igation for flight under visual flight rules may be accomplished by al reference to landmarks.		
GM	OPS.GEN.535(a)(2) Navigation equipment		

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A: Rule	B: Summary of comments	C: Reason for change, remarks
APPLICABLE AIRSPACE REQUIREMENTS		
For aircraft being operated under European air traffic control, the applicable airspace requirements include the Single European Sky legislation.		
GM OPS.GEN.535(b) Navigation equipment		
NUMBER OF NAVIGATION EQUIPMENTS		
The requirement in OPS.GEN.535(b) may be met by means other than the duplication of equipment.	1. The GM is vague and does not offer any guidance. It can be deleted.	1. Accepted. GM has been deleted.
AMC OPS.GEN.540.A(b) Electronic Navigation Data Management - complex motor-powered aeroplanes		
NAVIGATION DATA PRODUCTS NEEDED FOR OPERATIONS IN ACCORDANCE WITH OPS.SPA		
 When an operator of a complex motor-powered aeroplane uses a navigation database which supports an airborne navigation application as a primary means of navigation, the navigation database supplier should hold a Type 2 Letter of Acceptance (LoA), or equivalent. 	 IND: EASA need to send this paragraph back to experts. It is currently understood that such Type 2 LoA are required only for P-RNAV approval, not for B-RNAV approval. Can B-RNAV be the primary means of navigation ? IND: From EASA's CONDITIONS 	 It is correct that P-RNAV operations require a specific approval, in accordance with Part- SPA and a type 2 LoA (or equivalent) is needed for this purpose. B-RNAV instead does not require a specific approval and can be used as the primary means of

A: Rule	B: Summary of comments	C: Reason for change, remarks
A: Rule	B: Summary of comments FOR THE ISSUANCE OF LETTERS OF ACCEPTANCE FOR NAVIGATION DATABASE SUPPLIERS BY THE AGENCY, the definition of a Navigation Database is as follows: Navigation Database - Data (such as navigation information, flight planning waypoints, airways, navigation facilities, SID, STAR) that is stored electronically in a system that supports an airborne navigation application. Jeppesen's Position on Navigation Databases and Acceptable Means of Compliance Navigation Databases covered under a Type 2 Letter of Acceptance (LoA) should include more database types than those cited above, and should be applicable to all phases of flight (not just airborne) including databases utilized for navigation from gate-to- gate. By expanding the definition beyond airborne to all phases of flight, EASA will be more closely harmonised with other regulatory agencies like the FAA. Other benefits of the expanded navigation database definition will include additional data being available for flight crews including the offering of three	C: Reason for change, remarks navigation. 2. Noted. This will be addressed in rulemaking task 20.002 on EFBs.

A: R	ule	B: Summary of comments	C: Reason for change, remarks
		include vertical data). These additional navigational database types include terrain, obstacles, and airport moving map, and each of these data types offer additional information for flight crews to utilize for gate-to-gate navigation. Therefore, Jeppesen proposes EASA expand the acceptable means of compliance to recognise and include these other forms of navigational databases. The EASA "CONDITIONS FOR THE ISSUANCE OF LETTERS OF ACCEPTANCE FOR NAVIGATION DATABASE SUPPLIERS BY THE AGENCY" should be reviewed accordingly.	
2.	If this airborne navigation application is needed for an operation requiring a specific approval in accordance with OPS.SPA, the operator's procedures should be based upon the Type 2 LoA acceptance process.		
3.	A Type 2 LoA is issued by the Agency in accordance with the Agency's Opinion Nr. 01/2005 on The Acceptance of Navigation Database Suppliers (hereinafter referred to as the Agency's Opinion Nr. 01/2005). The definitions of navigation database, navigation database supplier, data application integrator, Type 1 LoA and Type 2 LoA can be found in the Agency's Opinion Nr. 01/2005.		
4.	Equivalent to a Type 2 LoA is the FAA Type 2 LoA, issued in accordance with the Federal Aviation Administration (FAA) Advisory		

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Circular AC 20-153 , and the Transport Canada Civil Aviation (TCCA) 'Acknowledgement Letter of an Aeronautical Data Process' which uses the same basis.		
5. EUROCAE ED-76/Radio Technical Commission for Aeronautics (RTCA) DO-200A Standards for Processing Aeronautical Data contains guidance relating to the processes which the supplier may follow.		
Section V - Manuals, Logs and Records		Section deleted and incorporated in NCC.GEN.
AMC OPS.GEN.600 Documents and information to be carried on all aircraft		Text transposed in AMC1- NCC.GEN.145.
1. In case of loss or theft of documents the operation may continue until the flight reaches the base or a place where a replacement document can be provided.		
2. The documents and information may be available in a form other than on printed paper. Accessibility, usability and reliability should be assured.		
3. For commercial air transport operations, if the competent authority has issued an English translation of the Noise Certificate, this should be carried as well.		
4. The procedures and the visual signals for use by intercepting and intercepted aircraft are those contained in ICAO Annex 2. For non-commercial operators with complex motor-powered aircraft and commercial operators, this may be part of the operations manual.		
5. Any other documents that may be pertinent to the flight or		

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required by the States concerned with the flight may include:		
a. the ground-air signal codes for search and rescue purposes;		
b. cargo and/or passenger manifests; and		
c. forms to comply with reporting requirements.		
AMC OPS.GEN.605 Documents and information to be carried on non-commercial flights with complex motor-powered aircraft and aircraft used in commercial operations		Transposed in the IR, NCC.GEN.145.
INFORMATION SEARCH AND RESCUE SERVICES AND OPERATIONS MANUAL		
1. The information pertinent to the intended flight concerning search and rescue services and should be easily accessible in the cockpit.		
2. Portions of the operations manual relevant to the duties of the crew should be easily accessible to the crew on board the aircraft.		
AMC OPS.GEN.605(a)(7) Documents and information to be carried on non-commercial flights with complex motor-powered aircraft and aircraft used in commercial operations		Text not transposed – beyond the scope of Part-NCC.
EQUIVALENT DOCUMENT		
For other than complex motor-powered aircraft, the Minimum Equipment List (MEL) may be in the form of an equivalent document as specified in accordance with Part-21.		

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АМС	OPS.GEN.610 Journey log book		Moved to ORO.MLR.
GEN	ERAL		
1.	The aircraft journey log book should contain the following items:		
a.	aircraft registration;		
b.	date;		
с.	crew member names and duty assignments;		
d.	departure and arrival points and times;		
e.	purpose/nature of the flight;		
f.	incidents, observations (if any); and		
g.	signature of the pilot-in-command.		
2. requi	The journey log may be combined with the aircraft log book as red in M.A.305.		
	The information or parts thereof may be retained in a form other on printed paper. In such cases, an acceptable level of accessibility, ility and reliability should be assured.		
GM (OPS.GEN.610 Journey log book		Moved to ORO.MLR.
SERI	ES OF FLIGHTS		

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The term 'series of flights' is used to facilitate a single set of documentation.		
Section VI – Security		Text not transposed - beyond the scope of this regulation.
GM OPS.GEN.700 Disruptive Passenger Behavior		
GENERAL		
1. Operators engaged in the transportation of passengers should take into account that their passengers could obstruct the safe operation of the aircraft. Passenger behaviour may be affected by a variety of factors, including:		
a. limitations on personal 'freedom', such as restrictions on smoking or on the use of mobile phones;		
b. physical effects, such as from consummation of alcohol, illness, or taking of medication, possibly increased from effects of higher altitude and less available oxygen;		
c. social or psychological effects, such as from fear of flying, claustrophobia, or reluctance to follow instructions.		

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2. The pilot-in-command should consider preventive measures when the possibility of disruptive passenger behaviour is anticipated. Such measures could include, but are not limited to:		
a. communication with the potentially disruptive passenger in an effort to reduce the likelihood of disruptive behaviour;		
b. reseating a potentially disruptive passenger to an area where there is less risk of passenger interference;		
c. deny boarding to the potentially disruptive passenger or cancel the flight.		