European Aviation Safety Agency

EASA

TYPE-CERTIFICATE DATA SHEET

EV-97 VLA

Type Certificate Holder: EVEKTOR, spol. s r.o.

EVEKTOR, spol. s r.o. Kunovice-letiste 686 04 Kunovice CZECH REPUBLIC

Manufacturer: EVEKTOR-AEROTECHNIK a.s.

EVEKTOR-AEROTECHNIK a.s Letecka 1384 686 04 Kunovice CZECH REPUBLIC

For variants: EV-97 VLA

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(Reserved)

SECTION 1: EV-97 VLA

I. General

Data Sheet No.: EASA.A.029 Issue: 01 Date: February 4, 2005

1. a) Type: EV-97 VLA b) Variant: EV-97 VLA

2. Airworthiness Category: Very Light Aeroplane

3. Type Certificate Holder: EVEKTOR, spol. s r.o.

Kunovice-letiste 686 04 Kunovice CZECH REPUBLIC

4. Manufacturer: EVEKTOR-AEROTECHNIK a.s.

Letecka 1384 686 04 Kunovice CZECH REPUBLIC

5. Certification Application Date: April 6, 1999

6. CAA CZ Certification Date: January 13, 2003

7. The EASA Type Certificate replaces The CAA CZ Type Certificate No. 03-01

II. Certification Basis

1. Reference Date for determining

the applicable requirements: April 6, 1999

2. (Reserved)

3. (Reserved)

4. Certification Basis: As defined in CRI A-01, latest Issue

5. Airworthiness Requirements: JAR-VLA, issued April 26, 1990, including

amendments VLA/91/1, dated October 22, 1991 and VLA/92/1, dated January 1, 1992

6. Requirements elected to comply: None

7. EASA Special Conditions: None

8. EASA Exemptions: None

9. EASA Equivalent Safety Findings: None

10. EASA Environmental Standards: Noise: ICAO Annex 16, Volume I,

Chapter 10, FAR Part 36, Amdt.

36-22, Appendix G

Emission: N/A

III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Airplane EV-97 VLA G0 00-00 01

2. Description: The EV-97 VLA is a two-seater, all metal,

cantilever low-wing monoplane with two sideby-side seats, fixed tricycle landing gear,

tractor propeller.

3. Equipment: The minimum equipment is listed in approved

EV-97 VLA Flight Manual, No. EVLA 001.00-

AS.

4. Dimensions:

 Wing Span
 8.100 m

 Total Length
 5.980 m

 Maximum Height
 2.335 m

 Wing Area
 9.790 m²

5. Engine/s: No.

Model: Rotax 912 S Type Certificate: TW9 – ACG

issued by AUSTRO CONTROL (Austria)

5.1 Engine Limits: Maximum Take off Power

73.5 kW/5800 RPM (max. 5 min.)

Maximum Continuous Power 69 kW/5500 RPM

6. (Reserved)

7. Propeller/s: No.

Model V 230E Type Certificate CAA CZ 96-04

Number of blades
Diameter:
2
Diameter:
1625 mm
Sense of Rotation
clockwise

8. Fluids:

8.1 Fuel: EUROSUPER RON unleaded according to EN

228 or DIN 51607 or AVGAS 100 LL.

8.2 Oil: Automotive oil, classification SF, SG,

according to API.

8.3 Coolant: Combinational cooling: water and air

9. Fluid capacities:

9.1 Fuel: Total: 67 liters

Usable: 64.5 liters

9.2 Oil: Maximum: 3.0 liters

Minimum: 2.0 liters

10. Air Speeds:

Design Manoeuvring Speed V_A 200 km/h IAS

Maximum Flap Extended Speed VFE

flaps 15° V_{FE15} 145 km/h IAS flaps 40° V_{FE40} 135 km/h IAS

Maximum structural cruising speed V_{NO} 205 km/h IAS

Never exceed speed V_{NE} 270 km/h IAS

11. (Reserved)

12. Operational: VFR Day

Flight into expected or actual icing conditions

is prohibited.

13. Maximum Masses:

Take-off 575 kg Empty Weight 340 kg \pm 3%

14. Centre of Gravity Range: 13,8 - 32 % MAC

[MAC is 1230mm; 0% MAC is at 20mm aft of

the refference datum]

15. Datum: The reference datum is the leading edge of the

wing

16. (Reserved)

17. Levelling Means:

Longitudinal axis: leveling point No. 1L and 2L Lateral axis: leveling point No. 3L and 3L

18. Minimum Flight Crew: 1 (Pilot)

Maximum Passenger Seating Capacity:

20. (Reserved)

Baggage / Cargo Compartments
 Max. 15 kg aft of the seat backrest bulkhead.

22. Wheels and Tyres

Main wheel: G5 06-27 01 and G5 06-27 02

Tyres and tubes: Goodyear 15x6.00-6 6PR (P/N 156E61-3)

Nose wheel: G5 00-31 01

Tyre and tube: Goodyear 15x6.00-6 6PR (P/N 156E61-3)

IV. Operating and Service Instructions

Airplane Flight Manual (AFM) EVLA 001.00-AS

Airplane Maintenance Manual (AMM)

(incl. Airworthiness Limitations) EVLA 004.01-AS

Illustrated Parts Catalogue: EVLA 005.01-AS

Maintenance Manual for ROTAX Engine Type 912 Series: 899 372

Propeller Operation Manual V230E: LV-DZ-05/2001-01

V. Notes

None.

SECTION 2: Changes/Variants

(Reserved)